

# Test Readiness Review (TRR)

Cameron Brown, Max Feldman,  
Tony Lima, Caleb Lipscomb,  
Erick Chewakin, Nick Lindholm,  
Jon Sobol, Ryan Niedzinski

Solid Propellant Additive Manufacturing (SPAM)

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# Agenda

<b>Overview</b>	→	Cameron
<b>Testing:</b>		
• <b>Laser Sintering</b>	→	Tony
• <b>Powder Bed</b>	→	Nick
• <b>Material Properties</b>	→	Erick
<b>Summary</b>	→	Erick

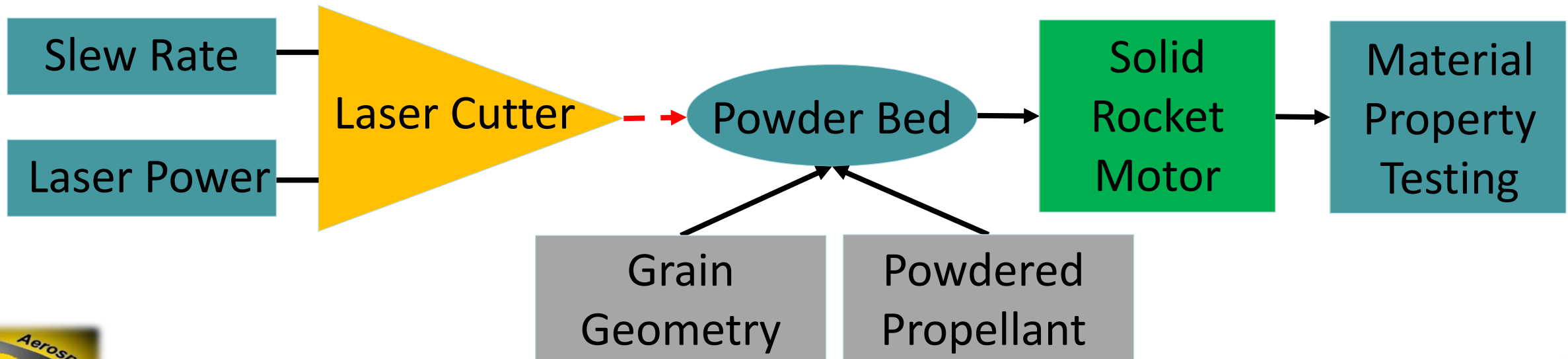
# Project Overview



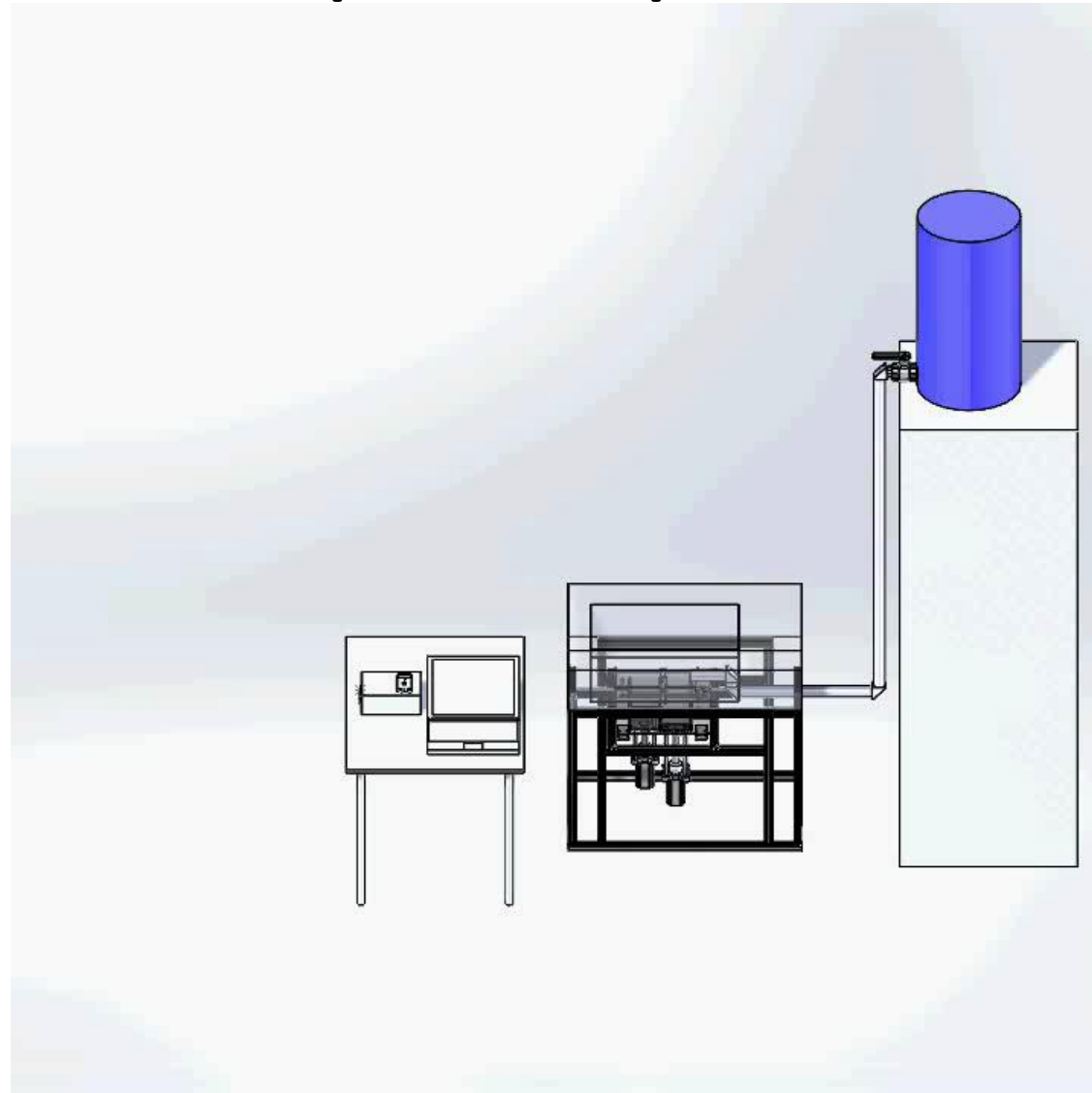
# Project Purpose

Additively Manufacture Solid Rocket Propellant

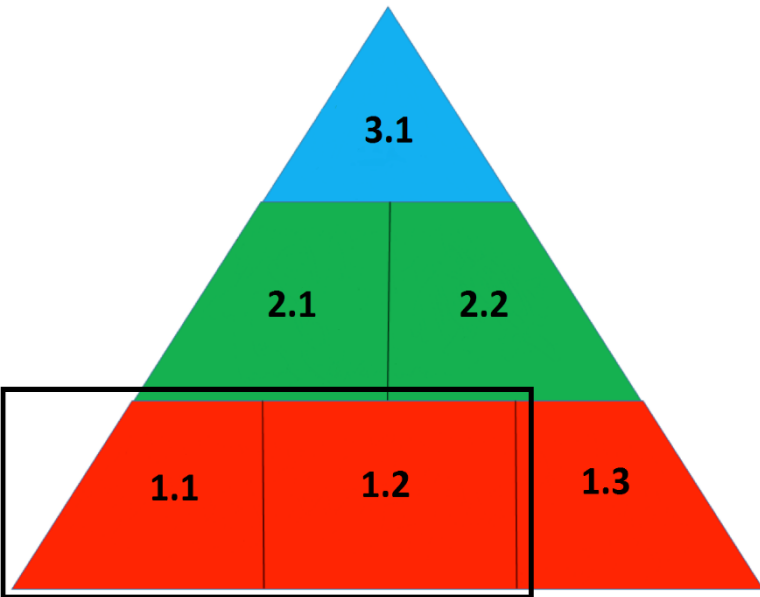
Sinter multiple  $\leq 1\text{mm}$  layers of Sucrose/ $\text{KNO}_3$  powder using a Laser Cutter



# Concept of Operations



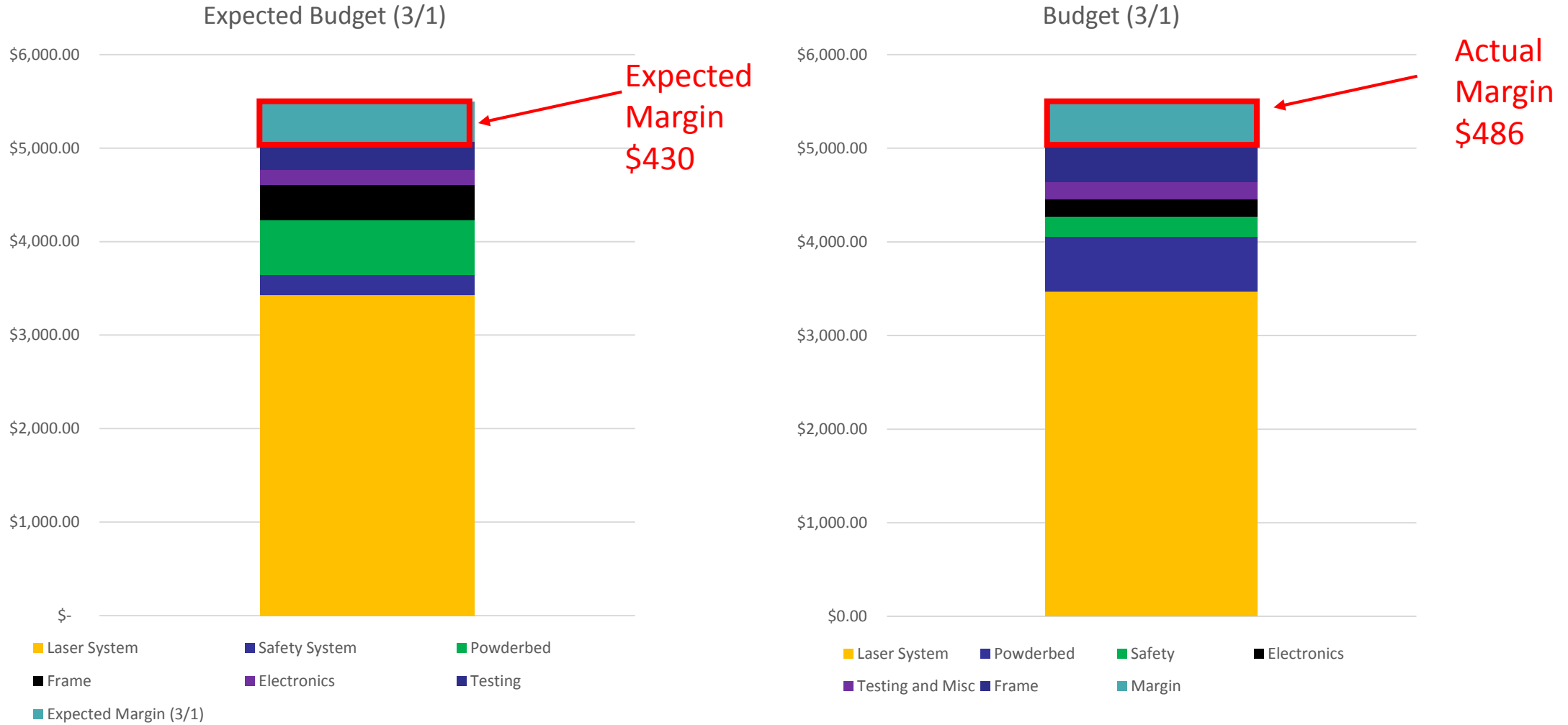
# Levels of Success




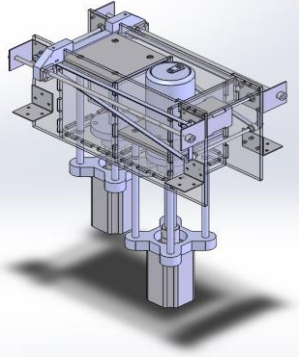
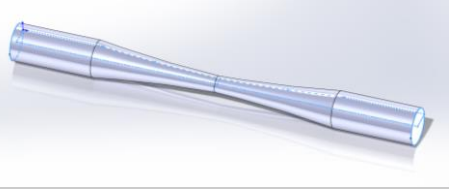
Level	Description	Achieved	Completion Date
1.1	Design 3D Printing System for Sucrose-KNO <sub>3</sub>	Yes	11/10/15
1.2	Characterize a Thermal Model for Propellant	Yes	1/30/16
1.3	Use Analogous Method to form Solid Propellant	No	3/10/16
2.1	Compare Material Properties (Casted vs Printed)	No	4/1/16
2.2	Print a Solid Rocket Motor Cylinder	No	3/23/16
3.1	Manufacture and Test 5 Different Grain Shapes	No	4/8/16



# Financial Status



# Critical Testing Elements

CPE	Description	Planned Completion Date
1) Laser Sintering Sucrose and KNO <sub>3</sub>	 <ul style="list-style-type: none"> <li>• Verify Thermal Model and Laser Energy Control</li> </ul>	<b>March 28<sup>th</sup></b>
2) Powder Bed Component Integration	 <ul style="list-style-type: none"> <li>• Component Integration and Tolerance Verification</li> <li>• Full Powder Bed Cycle Test</li> </ul>	<b>March 16<sup>th</sup></b>
3) Material Property Testing	 <ul style="list-style-type: none"> <li>• Validate Material Properties Between Casted and Printed Motors</li> </ul>	<b>April 7<sup>th</sup></b>



# Design Changes



Overview

Laser  
Testing

Powder Bed  
Testing

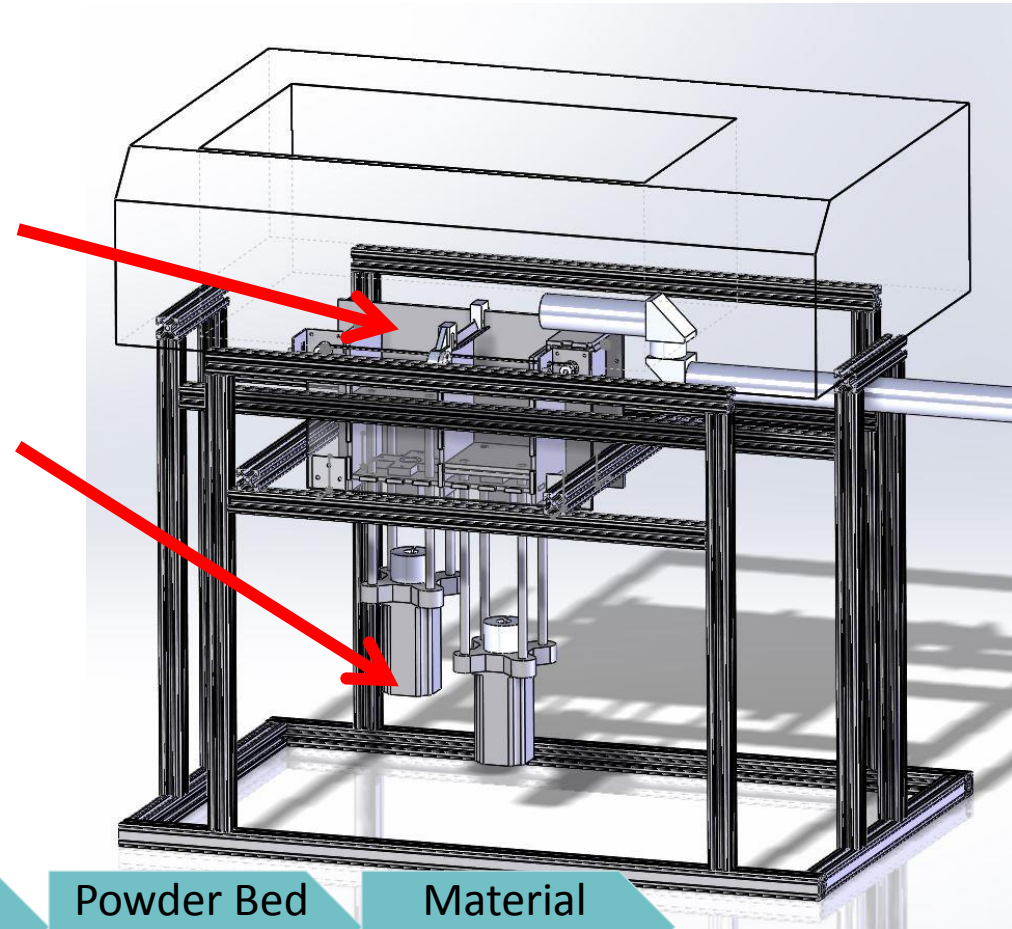
Material  
Testing

# Design Changes: Ventilation

**Problem**: Laser Cutter Filled with Fine Particulate During Testing

**Safe Engineering Solution**:

- Powdered propellant mixture atomized into enclosure here
- Side wall and additional fans for ventilation installed
- Safely remove particulate faster than build up



# Design Changes: Driver Heat

Problem: Motor Drivers Overheat Due to High Wattage Input

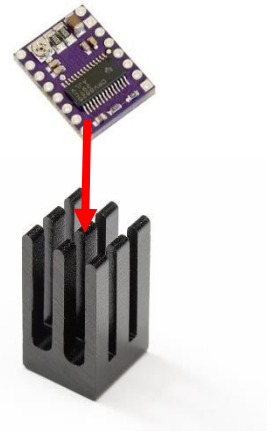
Impact: Behind Schedule on Electrical Component Testing

Learned: Need higher amp rated drivers (2000% more expensive)

Or modify existing drivers for better heat dissipation

Solution:

- Calculated wattage and required heat removal
- Purchased applicable heatsinks
- Solution verified by 3/4/16 (One Week Behind Schedule)



# Testing

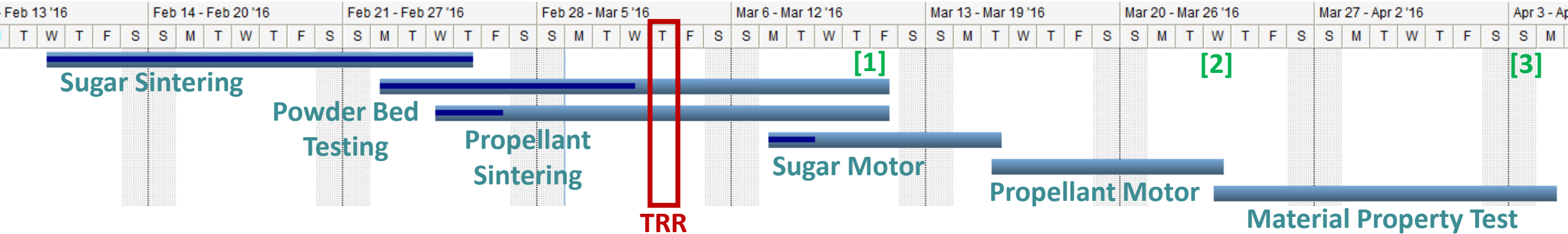
Overview

Laser  
Testing

Powder Bed  
Testing

Material  
Testing

# Testing Schedule



## Summary:

- **Behind schedule by 1 Week** due to unexpected obstacles
- **[1] Level 1 Success**: scheduled for 3/10/16
- **[2] Level 2 Success**: scheduled for 3/23/16
- **[3] Level 3 Success**: scheduled for 4/4/16



# Sugar Sintering Test: Overview

**Test Purpose:** Validate Sugar Thermal Model

Requirement	Description
1.3	Laser raises propellant temperature to 185°C

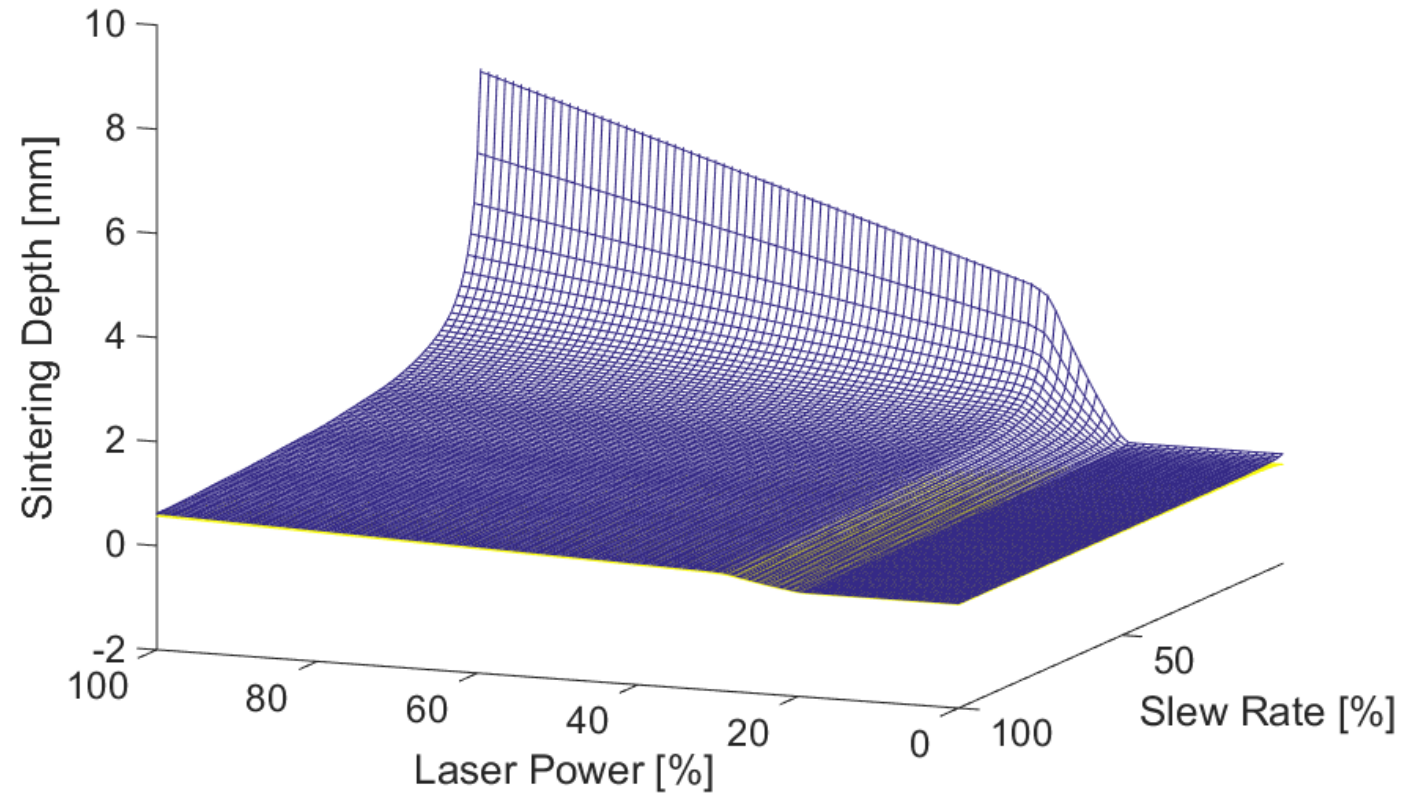
**Test Procedure:**

- Sinter small samples of sucrose at varying power and slew rates
- Record sintering depths with calipers
- Compare to model predictions

**Expected Results:**

- Exponential increase between 10% and 1% slew rate
- Not sensitive to Power

**Predicted Sintering Depths [mm] by Control Parameters**





# Sugar Sintering Test: Results

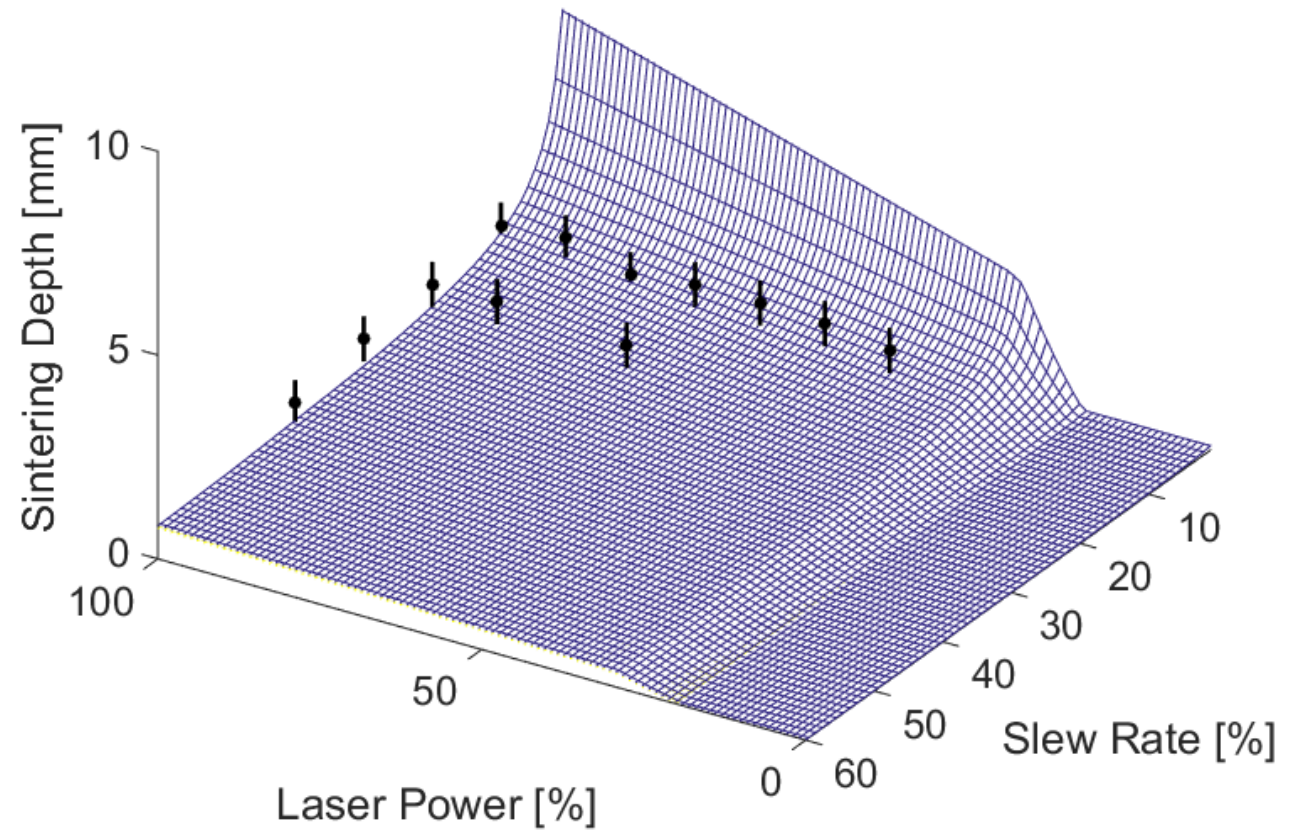
## Results:

- Most samples thicker than predicted
  - Likely caused by size of sugar granules
- Minimum Sintering Depth
- Inaccurate (>1 Std) at Power < 5%
  - Caused by heat conduction

## Future Analysis:

- Test goodness of fit ( $X^2$ )
- Update model with minimum thickness
- Re-test goodness of fit

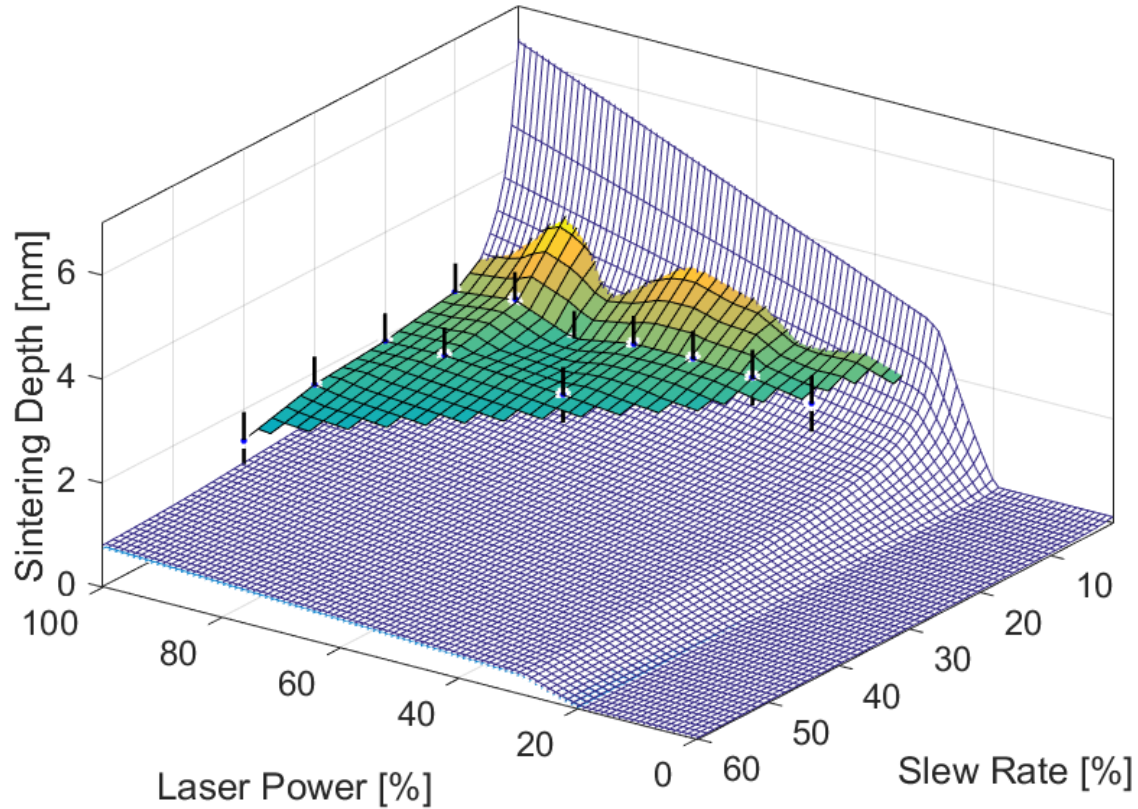
Predicted Vs. Measured Sintering Depths [mm]



# Sugar Sintering Test: Results

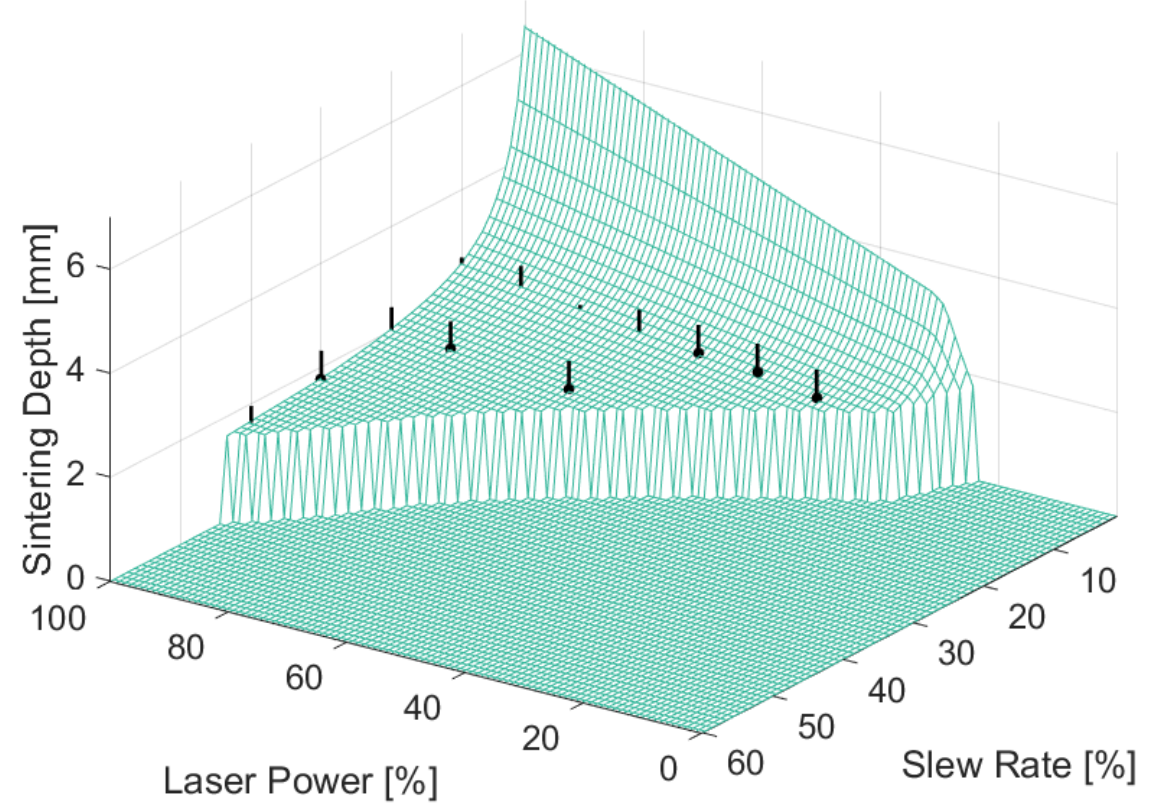
- First Model:  $X^2_{red} = 9.05$

Predicted vs Measured Sintering Depths (Old)



- Adjusted Model:  $X^2_{red} = 1.00$

Predicted vs Measured Sintering Depths (Updated)





# Propellant Sintering Test: Overview

**Test Purpose:** Validate Propellant Thermal Model

Requirement	Description
1.3	Laser raises propellant temperature to 185°C

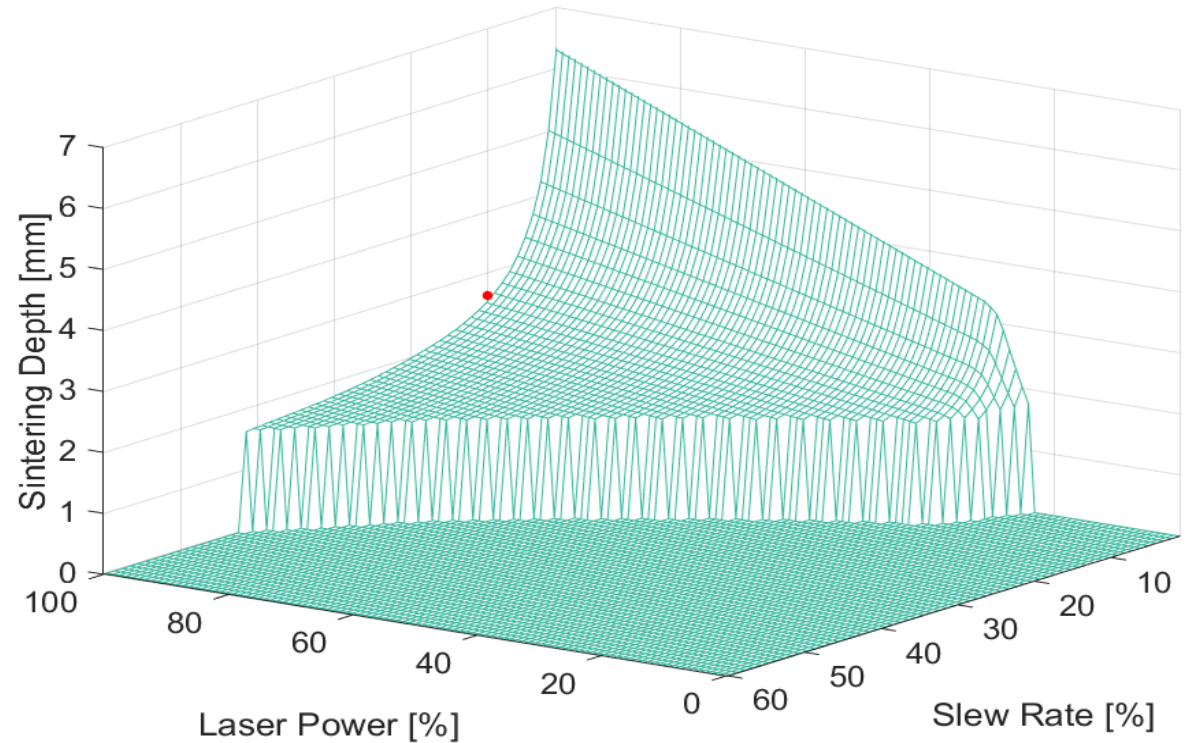
**Test Procedure: (Same as Previous Test)**

- Sinter small samples of **propellant** at varying power and slew rates
- Record sintering depths with calipers
- Compare to model predictions

**Expected Results:**

- Sintering depths should not change by more than 1 mm
- Average of 2 mm sintering depths at 10% slew rate

**Predicted Propellant Sintering Depths [mm]**



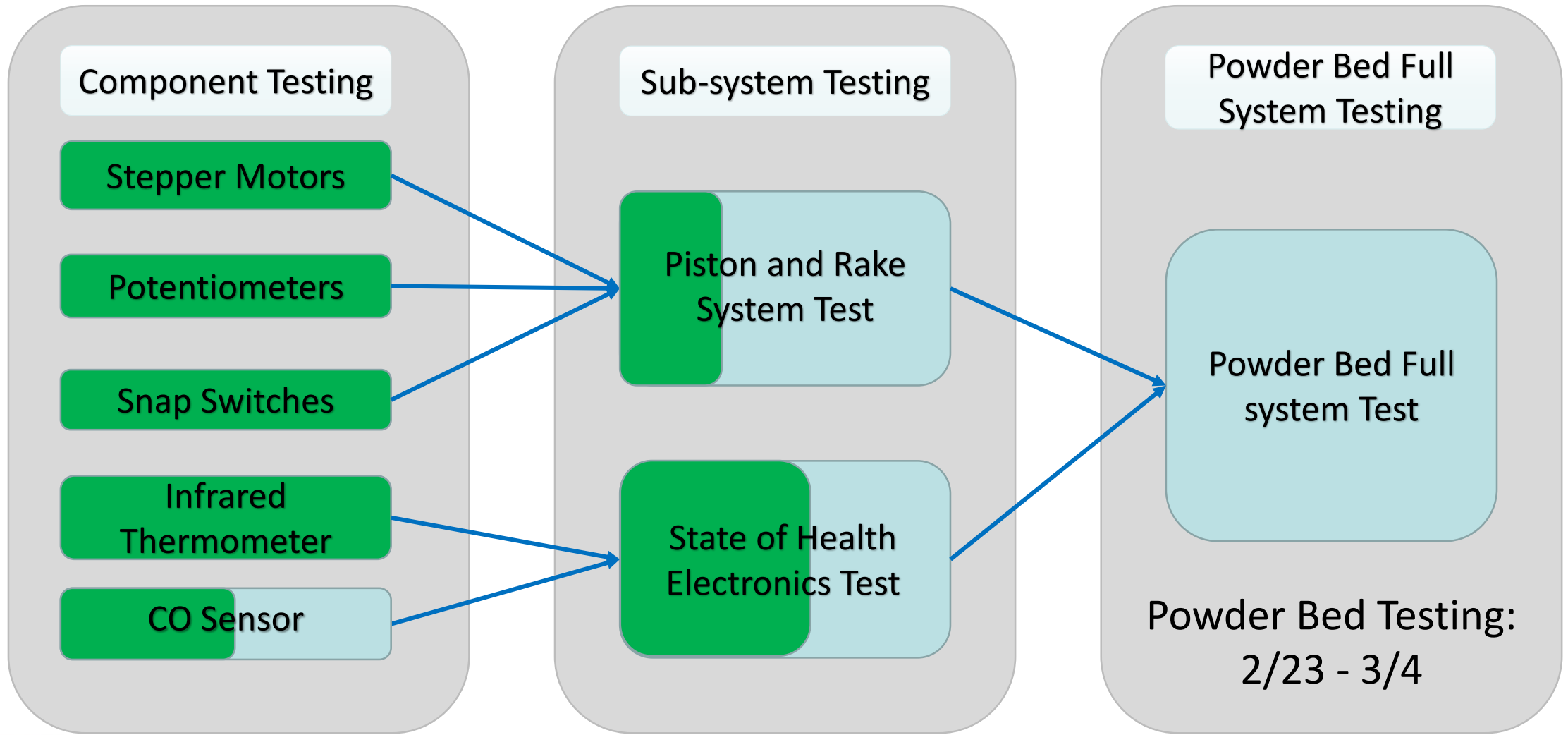
**This Test Will Accomplish Level 1 Success  
(3/10/16)**



# Laser Sintering Test Summary

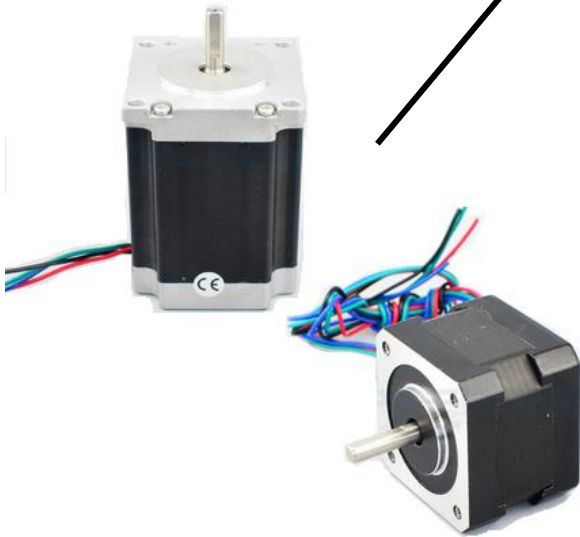
Test	Status (% Complete)	Total Required Testing Hours	Expected Completion Date
Sugar Sintering Test	100 %	10	2/16
Propellant Sintering	10 %	10	3/10
Sucrose Motor Print	20 %	5	3/15
Full Motor Printing	0 %	15	3/28

# Powder Bed Testing Flow

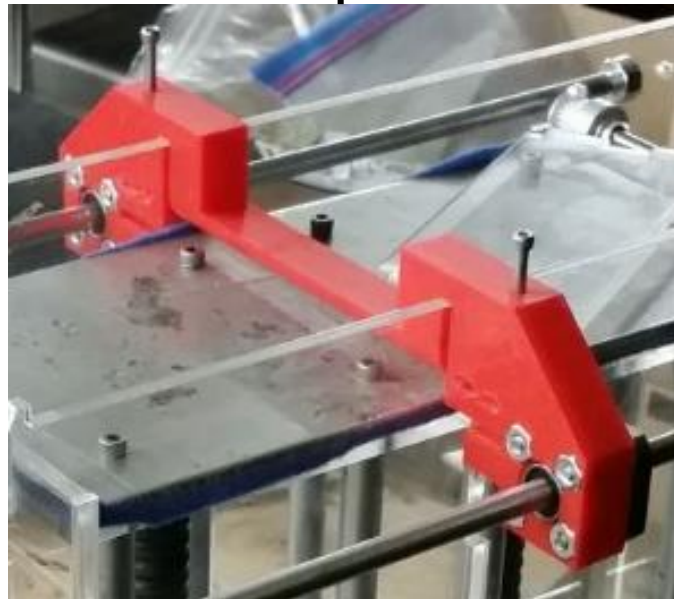


# Powder Bed Integration Tests

Full System Integration Tests



Piston Motor/Driver Tests



Rake Pull/Binding Tests



Piston Tests

Overview

Laser  
Testing

Powder Bed  
Testing

Material  
Testing



# Powder Bed Tests: Motor/Driver

## Test Purpose:

- Confirm motors meet Required Tolerances

## Test Procedure:

- Command known number of steps.
- Measure resultant motion and torque

## Expected Results:

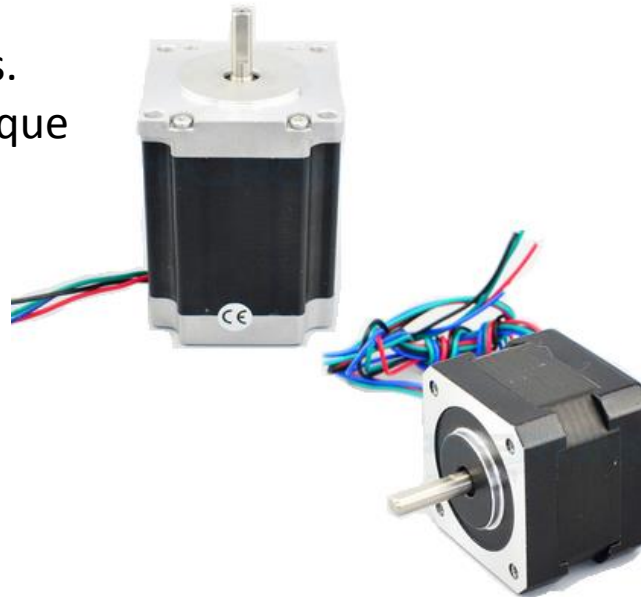
- Motors operate within specs

## Problem:

- Drivers overheat during operation

## Solution:

- Attach heat sinks to drivers



Motor	Expected Torque (N*m)	Expected Accuracy (deg/step)
NEMA-23 (Piston Motor 1)	2.4	1.8 (deg/step)
NEMA-23 (Piston Motor 2)	2.4	1.8 (deg/step)
NEMA-17 (Rake Motor)	0.43	1.8 (deg/step)

Table: Expected Motor Specifications

# Rake Neutral Friction Test

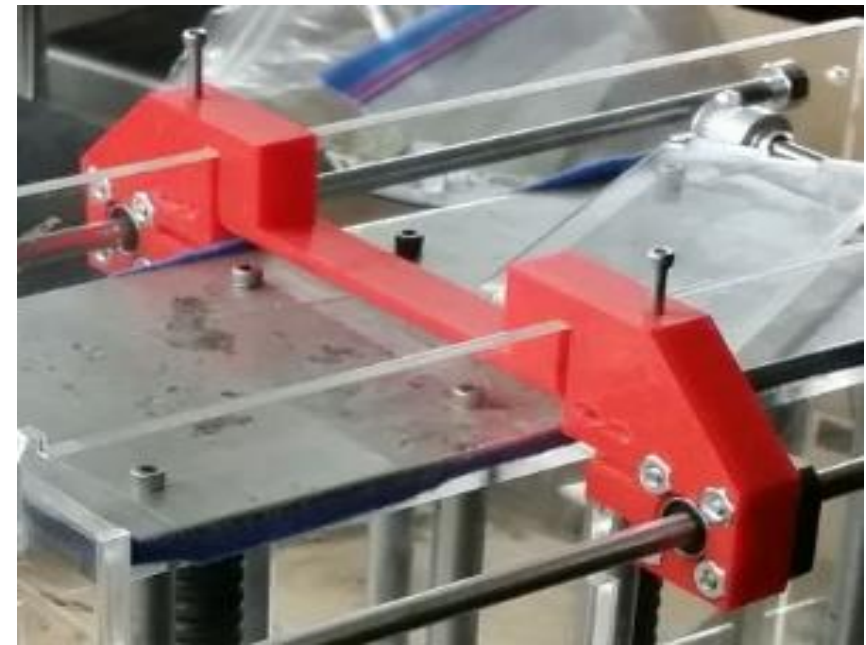
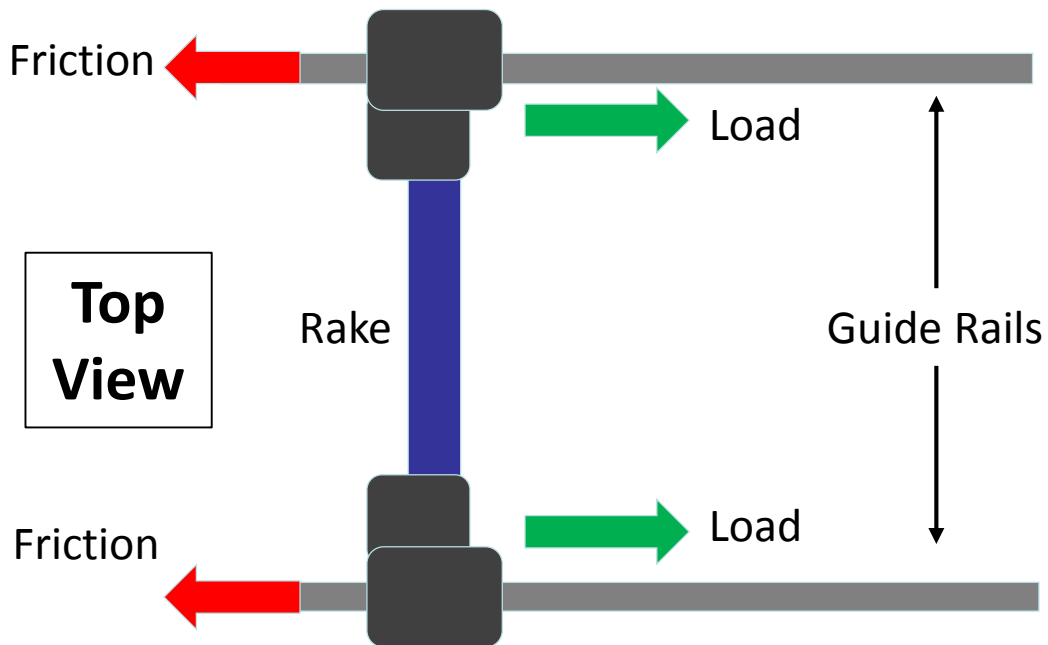
## Test Purpose:

- Can the motor move the rake and a layer of powder?
- What is the friction force in the assembly?

## Expected Results:

- The rake increases the expected load on the motor by less than a factor of 2

Requirement	Description
2.4.3	Rake shall deliver propellant to the print area



Manufactured Rake (Red)



# Powder Bed Integration: Lifts

## Test Purpose:

Confirm pistons are controlled and move as expected

Requirement Verified	Description
2.1.2	The Powder Bed shall move 1.0+0.3mm.
2.4.2.2	Pistons displace 2.5kg of propellant
2.4.2.3	Pistons move 2.85" vertically

## Expected Results:

1.8 degrees per step => 0.02 mm vertical movement

## Dial Micrometer:

Used to verify predicted lift tolerance.

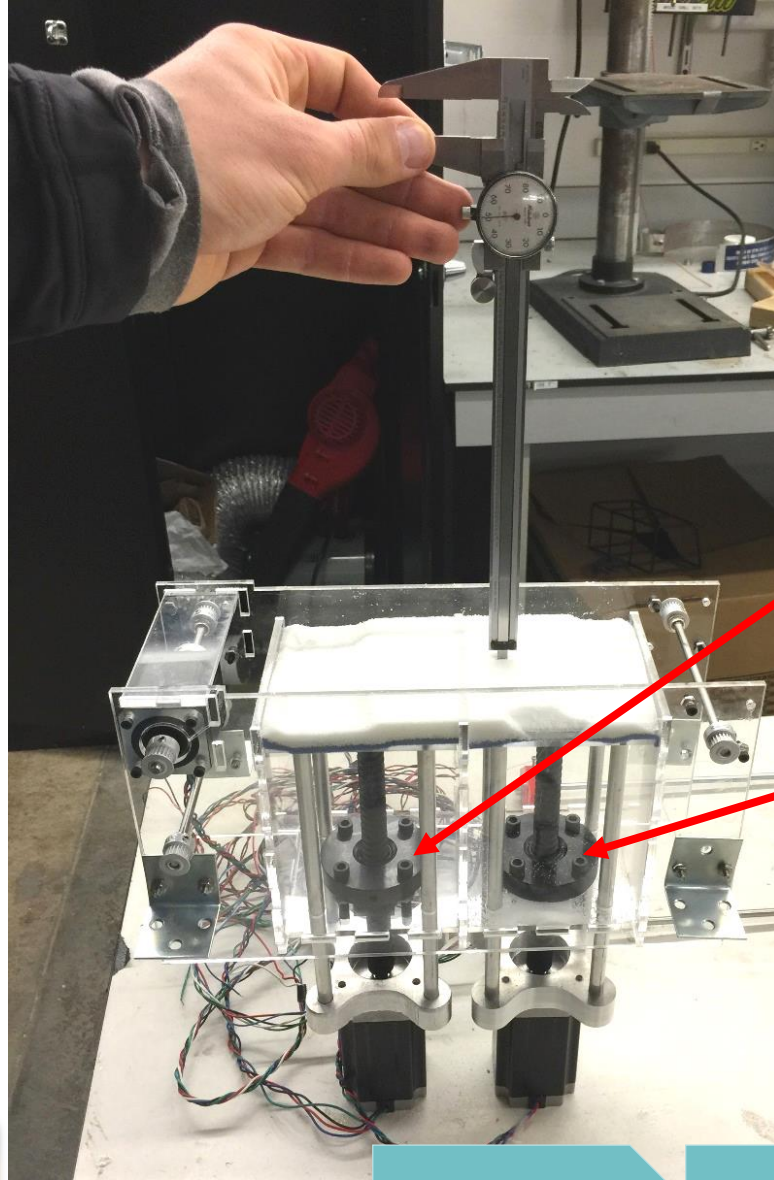
Accuracy: 0.001" = 0.0254mm



## Piston Lift Assembly:

Measure height increase for each motor step





# Layer Tolerance Test

- **2.4.1.1**: Reservoir vertical movement to an accuracy of  $2 \text{ mm} \pm 1 \text{ mm}$
- **2.4.2.1**: Sintering area vertical movement to an accuracy of  $2 \text{ mm} \pm 1 \text{ mm}$

Overview

Laser  
Testing

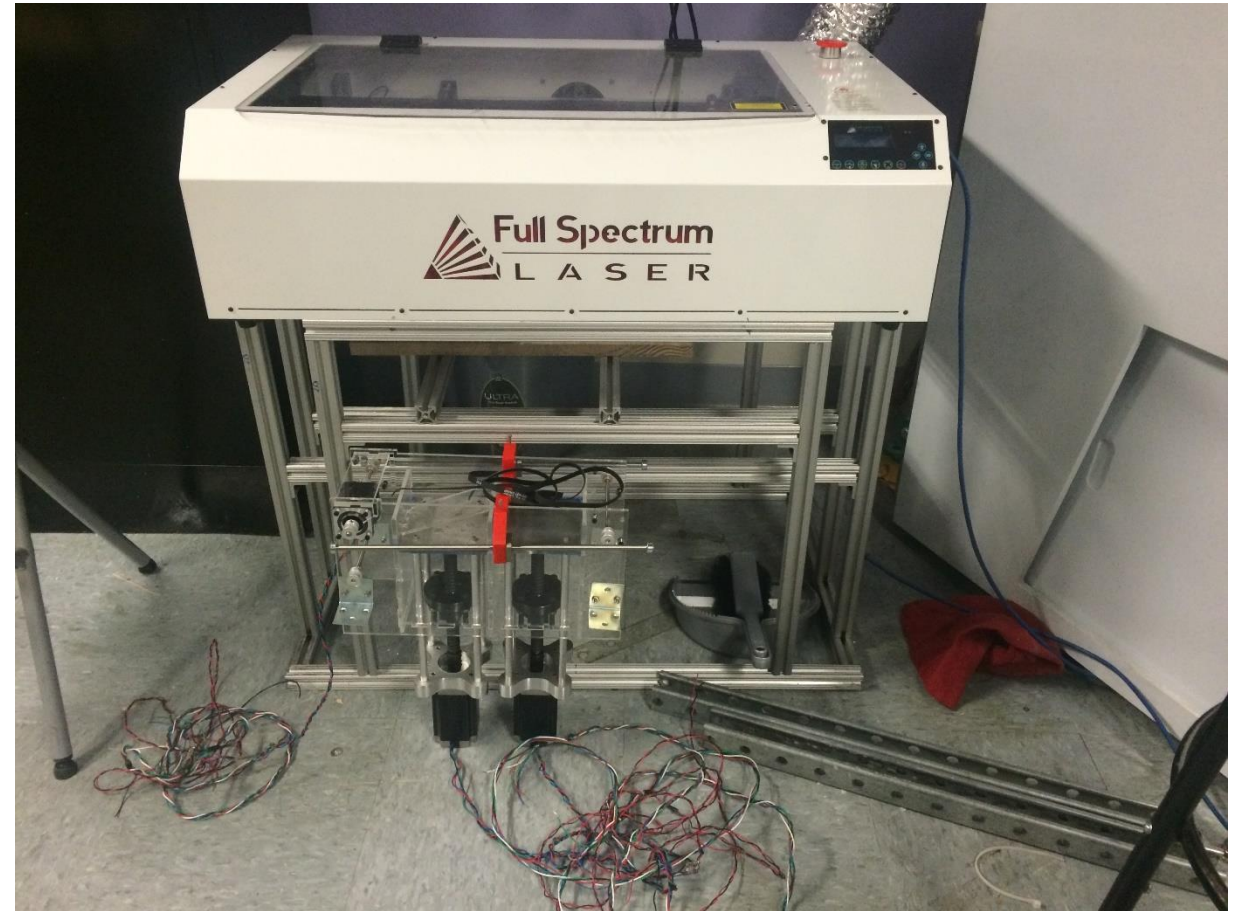
Powder Bed  
Testing

Material  
Testing

# Powder Bed System Integration Test

**Test Purpose:** Verify Powder Bed Requirements

Requirement Verified	Description
1.4	Layers of propellant 2.0+1.0 mm
2.4	Autonomously deliver unsintered material to manufacturing area.
2.4.1/2.4.2	Piston controlled chambers shall hold unsintered powder and the completed motor.
2.4.3	A Rake shall autonomously deliver unsintered powder.
2.4.5	Each cycle shall take no more than five minutes.





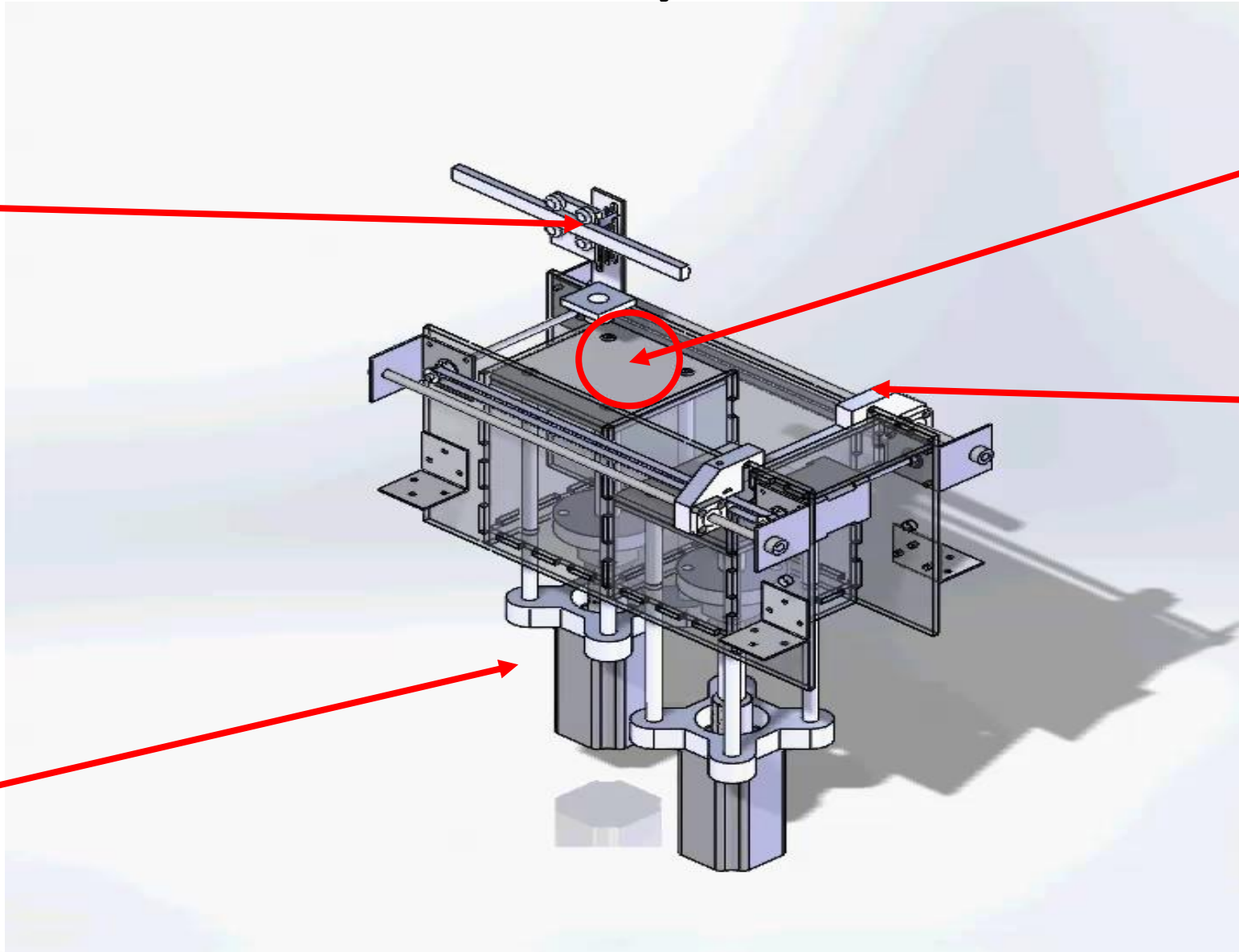
# Powder Bed Full Cycle Test: Overview

Dial  
Micrometer  
measures  
sintered  
depth

Sintering  
region

Pistons move  
1mm+/- .3mm  
of propellant to  
sintering region

Rake sweeps  
propellant to  
sintering  
region





# Powder Bed Testing Summary

Test	Status (% Complete)	Total Required Testing Hours	Expected Completion Date
Functional Component Testing	90%	10	3/4
Frictional Testing	10%	5	3/4
Lift Tolerance	25%	5	3/8
Rake Tolerance	0%	5	3/8
Full Cycle Test	0%	15	3/10

# Material Properties Testing

## Test Purpose:

- How does the printed propellant differ from cast propellant?
- Any differences that would indicate that printing is not a viable SRM manufacturing method?

Requirement Verified	Description
DR 3.1	Compare selected material properties of printed and cast propellants
DR 1.2	Printed layers bond

## Subtests:

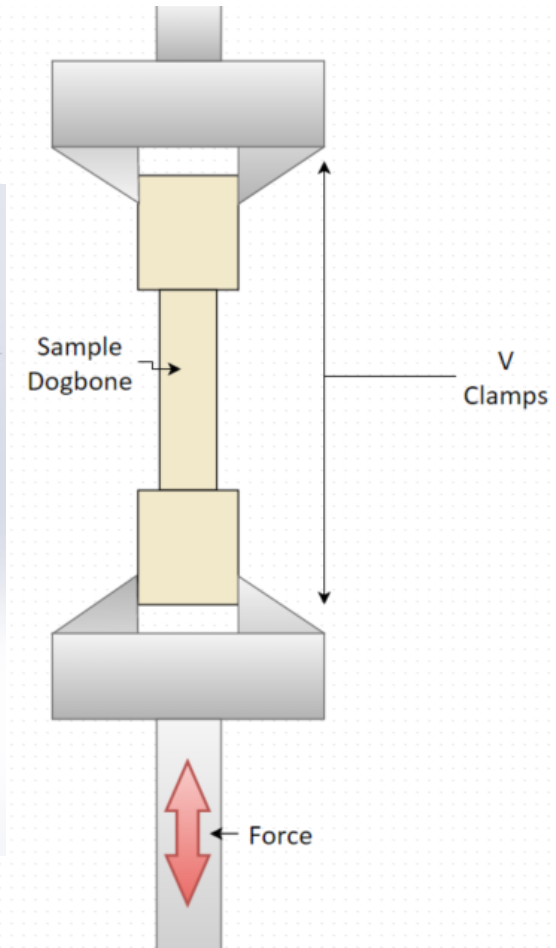
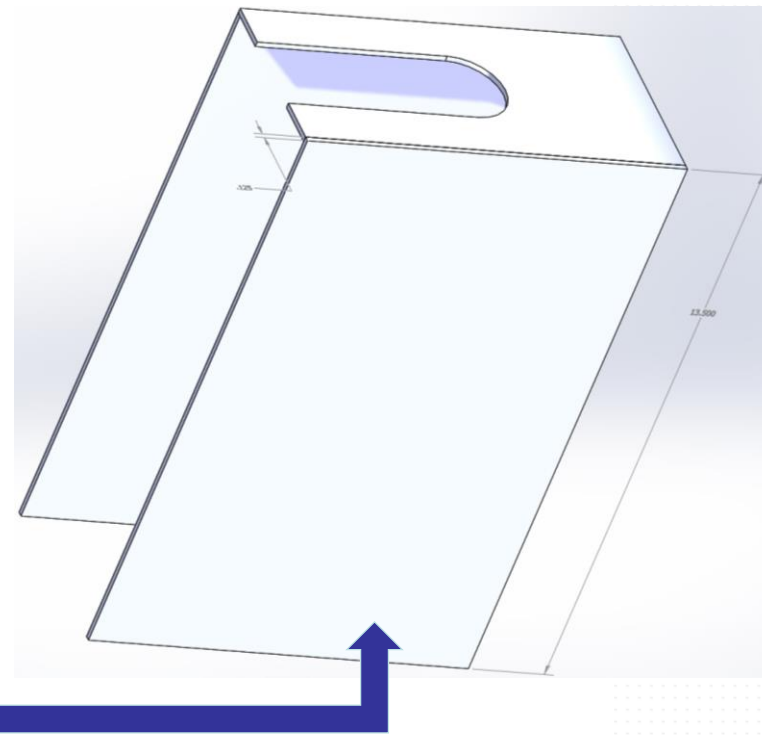
- Density
- Tensile Strength
- Crush Strength

## Expected Results:

- Printed propellant will be weaker, and less dense

# Material Properties: Strength

- Tensile strength, crush strength
- Differences between printed and cast propellant? Is printing viable?
- Deformation measured via camera
- Max stress of 24 Mpa
  - Ignition from thermoelastic effects at 72 MPa
  - 271 lbf for our sample with Safety factor of 2.4
- Debris fully contained for EH & S disposal



# Material Testing Summary

Test	Status (% Complete)	Total Required Testing Hours	Expected Completion Date
Material Properties	0%	25	3/30
Instron	0%	25	4/07

## Material Testing Plan:

- Material testing cannot begin until propellant has been successfully sintered and shaped
- Scheduled to test from 3/23/16 to 4/7/16

# SPAM Status Summary

- **Behind on Testing Schedule**
  - Still Within Margin

Level of Success	Description	% Complete	Deadline
1	Test Propellant Sintering	90%	3/10/16
2	Printed Material Testing	20%	3/23/16
3	Compare Multiple Grain Patterns	0%	4/7/16

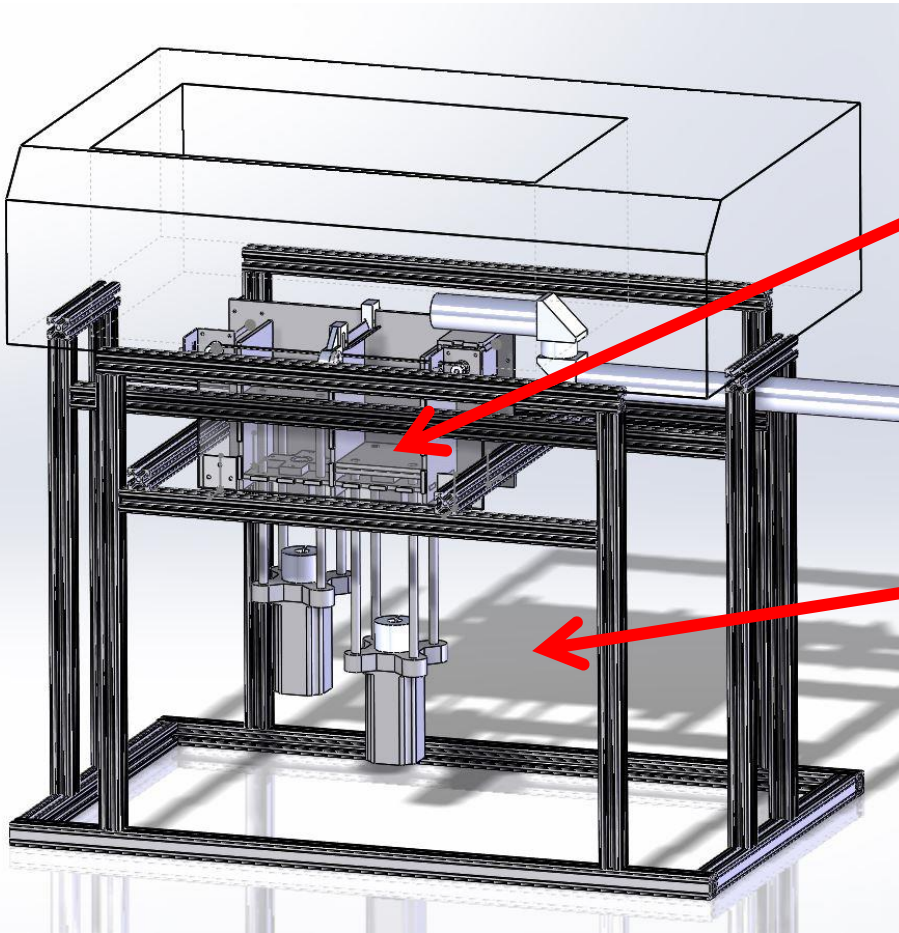


# Questions?

# Backup Slides

# Ventilation Changes (Back-up)

# Design Changes: Ventilation



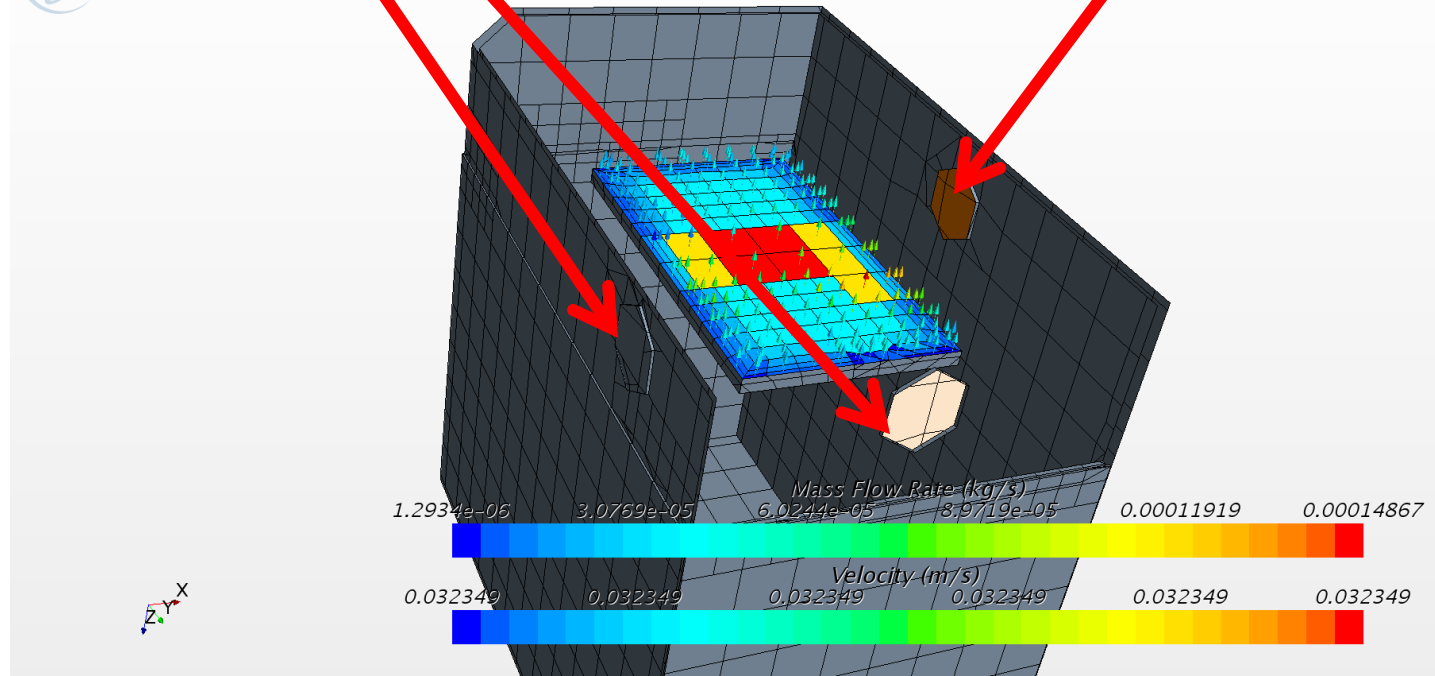
Fumes from Sucrose-KNO<sub>3</sub> mix are hazardous in large quantities

Side wall and additional fans for ventilation must be installed



# Design Changes: Ventilation

Component	Condition
Fan 1	$\dot{m} = .090 \text{ kg/s}$ (600 CFM)
Fan 2	$\dot{m} = .057 \text{ kg/s}$ (400 CFM)
Outlet Duct	Zero Pressure Outlet



# Powder Bed Additional Tests (Back-up)

# Powder Bed Component Tests:

## Acrylic 'Snap' Test



### Test Purpose:

Characterize acrylic fit and bonding options for stability, strength, and ease of manufacturing.

### Test Procedure:

- Laser cut patterns for matching peg-hole and peg-small hole.
- Assemble with glue/epoxy and write observations.

### Results:

- Matching peg-hole + epoxy hold but require effort to hold in place while curing.
- Peg-small hole snaps together without need for epoxy.

### Facilities/Tools:

- Laser cutter
- Epoxy

### Problem:

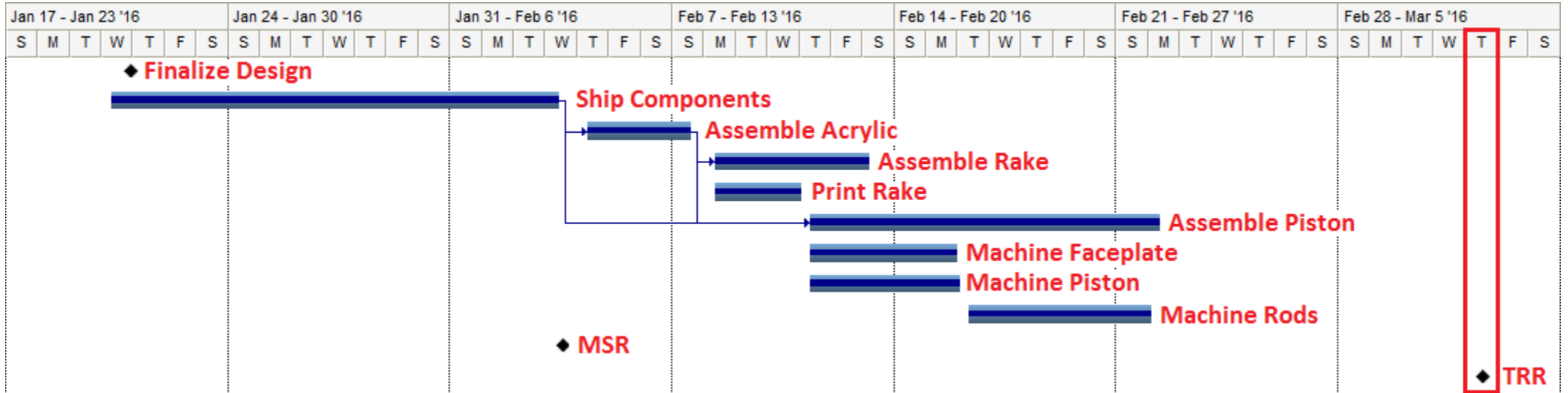
- Peg-small hole method used for Acrylic Body but results could not be replicated (pegs would not fit into holes)

### Solution:

- Backup acrylic used to cut matching peg-hole body.

Requirement Verified	Description
None	The acrylic body is not part of the functional requirements.

# Powder Bed Manufacturing Schedule



**Problem:** Completed 1 week behind schedule

**Impact:** Integration testing is on a tighter schedule

Level 1 success still expected to occur on March 10<sup>th</sup>





# All Tests Schedule (Back-up)



# REMAKE THIS CHEWBACCA Testing Schedule

