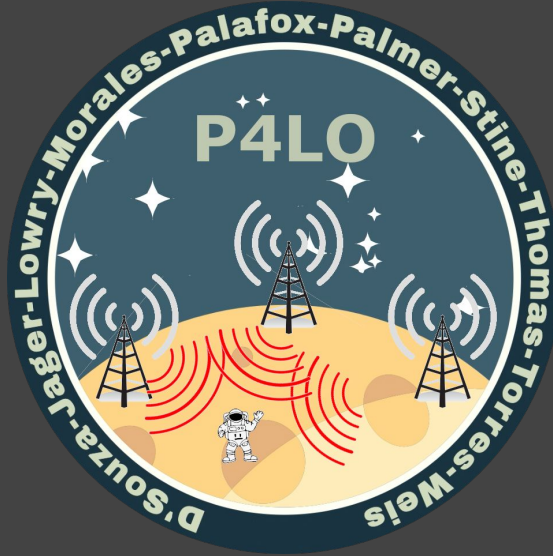


P4LO



Positioning For Lunar Operations

Team Advisor: Dr. Jade Morton



Overview

Test Readiness

Schedule

Budget



Agenda

1. Project Overview/Objectives
2. Test Readiness
3. Test Schedule
4. Budget



Project Overview



Overview

Test Readiness

Schedule

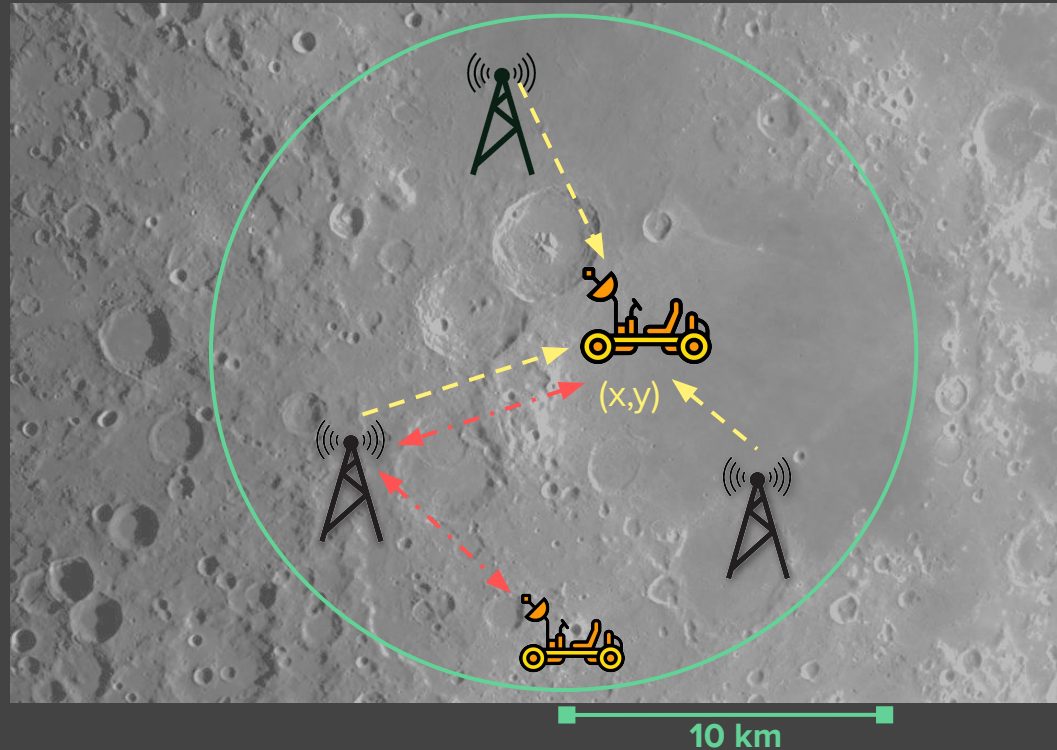
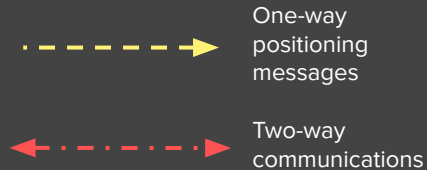
Budget

Background

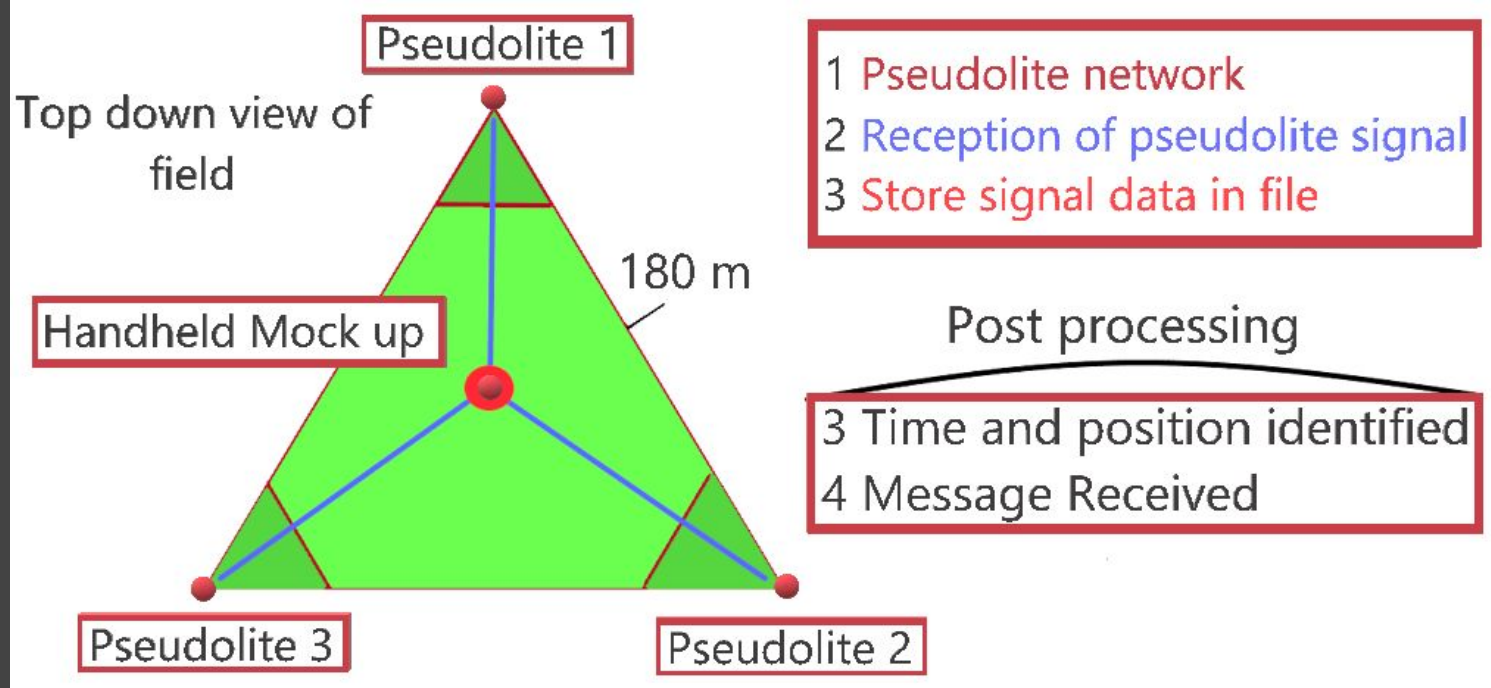


Lunar CONOPS

- Provide ground-based positioning and communication to the South Pole
- 10 km radius region

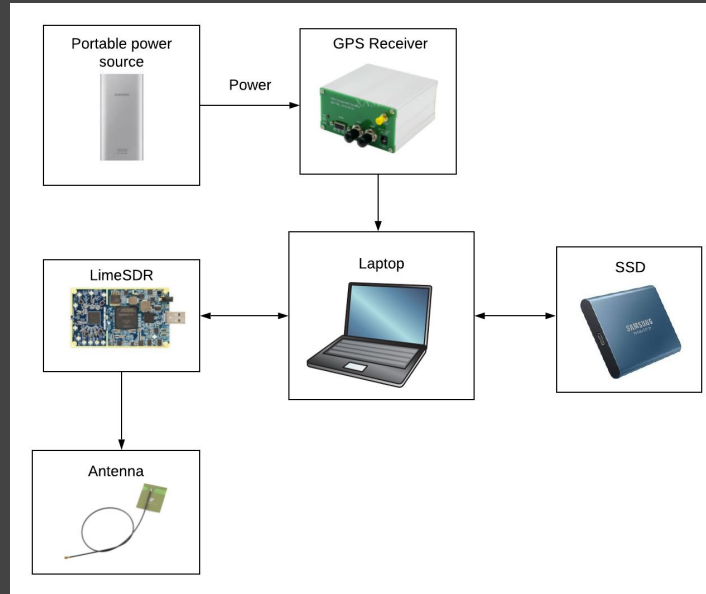


P4LO CONOPS



What is a Pseudolite?

- A portable ground station
 - Transmits positioning messages
 - Receives user communication messages



Pseudolite Hardware Block Diagram





Overview

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Mission Statement

P4LO (Positioning For Lunar Operations) is a prototype network of software defined radios that demonstrates an architecture for the lunar communication and positioning system LunaNet.

How will positioning work?

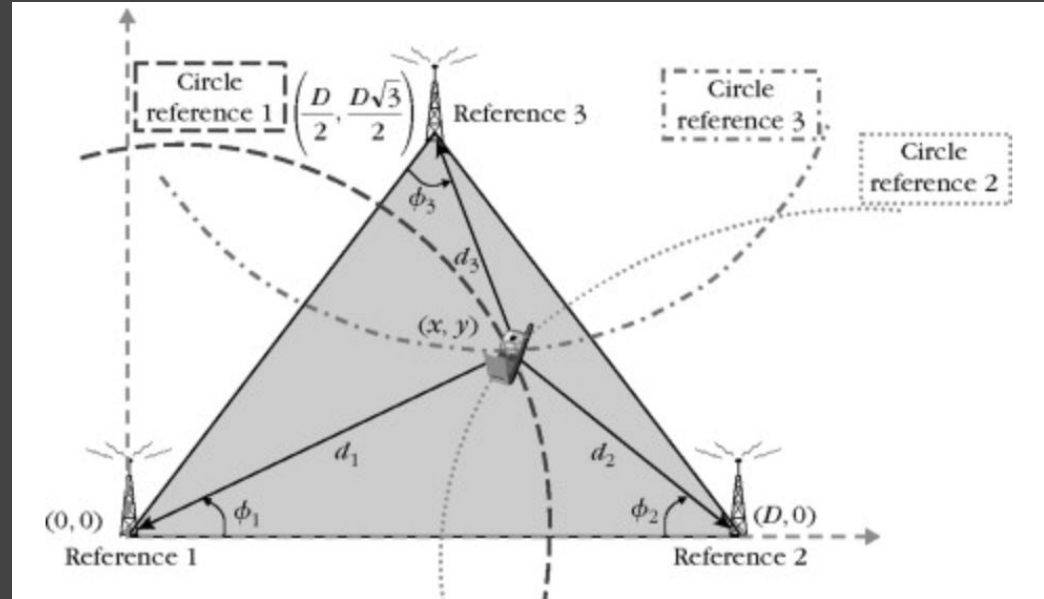
- Known pseudolite positions
- Time of arrival (TOA) -> **range**
- System of equations with 3 unknowns (x, y, clock bias)
- Least-squares solution to account for noise and measurement error

$$d_1 = c(t_1 - t_0) = \sqrt{x^2 + y^2}$$

$$d_2 = c(t_2 - t_0) = \sqrt{(D - x)^2 + y^2}$$

$$d_3 = c(t_3 - t_0) = \frac{1}{2} \sqrt{(D - 2x)^2 + (D\sqrt{3} - 2y)^2}$$

t_i : TOA at reference i ; c : speed of light





Overview

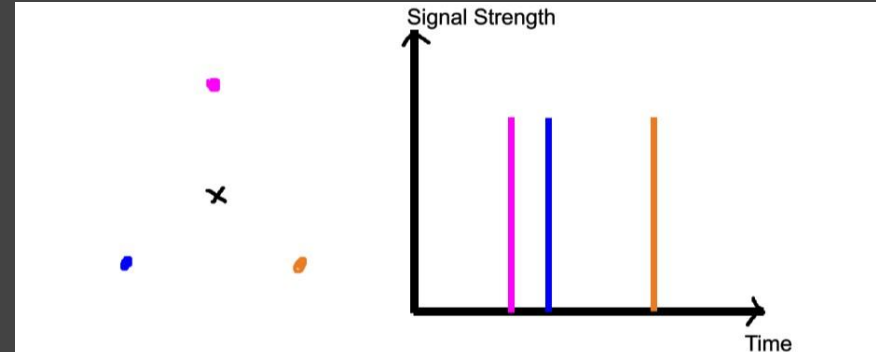
Test Readiness

Schedule

Budget

Range Determination - Bias Estimation

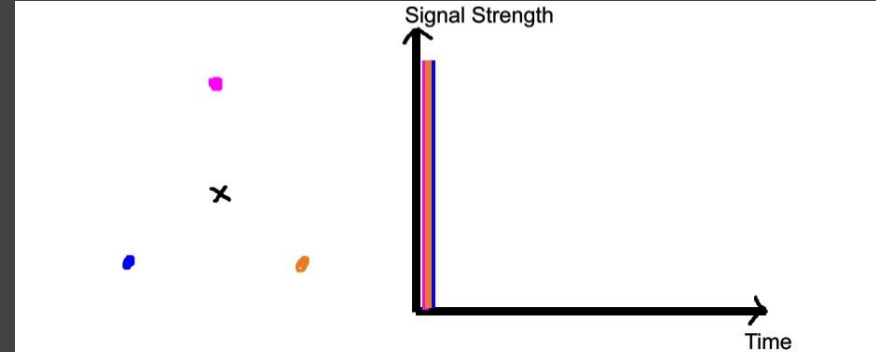
- **Measure synchronization bias** at a known location (center)
- **Subtract out bias** on subsequent measurements
- Does not require pseudolite synchronization
- Requires **high-sample rates** and a **stable oscillator** (GPSDO)





Range Determination - Bias Estimation

- **Measure synchronization bias** at a known location (center)
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Overview

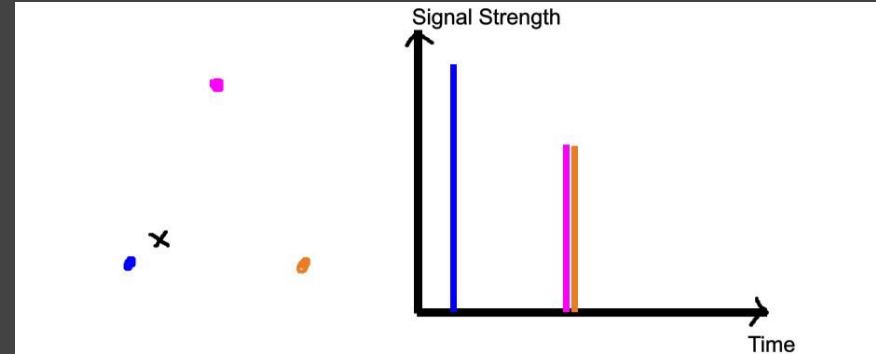
Test Readiness

Schedule

Budget

Range Determination - Bias Estimation

- **Measure synchronization bias** at a known location (center)
- **Subtract out bias** on subsequent measurements
- Does not require pseudolite synchronization
- Requires **high-sample rates** and a **stable oscillator** (GPSDO)





- Key Functional Requirements
 - Scalable Positioning System
 - Sub 10 (m) positioning within specified area
 - 30 ns 1-sigma transfer time
 - Communication Network
 - Communication operating between 2.4 - 2.48 GHz
 - 1 MHz Signal Bandwidth
 - Framework allowing for up to 170 users communicating within region
 - Data rate of at least 1 kilobit per second



Overview

Test Readiness

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Budget

Critical Elements

Signal Generation and Transmission

- LimeSDR - GNU Radio
- Antenna
- GPS Reference Oscillator (Bias Correction)
- Laptop and SSD

Signal Reception

- LimeSDR - GNU Radio
- Antenna
- GPS Reference Oscillator (Used as stable oscillator)
- Laptop and SSD

Signal Processing

- Acquisition and tracking
- Data extraction
- Positioning Solution



Test Readiness



Overview

Test Readiness

Schedule

Budget

Testing Overview

- SDR (Software Defined Radio) Communication
- Coarse Range Measurement
- Refined Range Measurement
- Clock Override
- Synchronization Bias Test
- Positioning Test
- Positioning + Communication Test

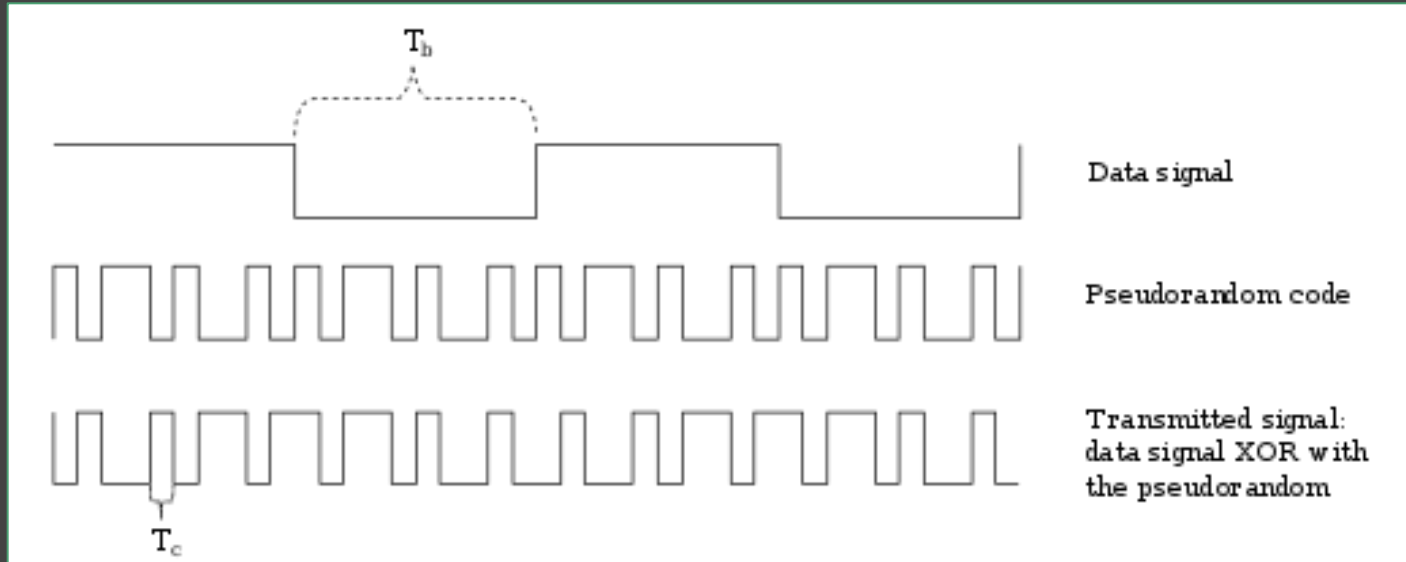


- Purpose:
 - LimeSDR transmission and reception of a test signal with information
- Design:
 - Hardware: LimeSDR, one computer, two antennas
 - Software: GNURadio FM transceiver script, sample audio file
- Test:
 - Transmit one FM audio file from SDR transmit port and receive it on the SDR receive port, play results
- Status: Done



Coarse Range Measurement (Acquisition)

- Purpose:
 - **Transmit pure code-division multiple access (CDMA) codes** to see if a receiver can determine code phase and therefore **determine range**





Coarse Range Measurement (Acquisition)

- Design:
 - Hardware: Two LimeSDRs, two laptops, two antennas
 - Software: Generated CDMA codes, GNURadio modulation and transmission, GNURadio IF signal reception, IF signal correlators
- Test:
 - Level 1 - Lab testing
 - Level 2 - Field testing
 - Level 3 - Multiple Transmitters
- Status:
 - CDMA code file generated, acquisition MATLAB code developed
 - Just need to go and actually send and receive files



Refined Range Measurement (Tracking)

- Purpose:
 - Integrate a **control loop** (tracking) in order to better determine code delay (range) and signal frequency (carrier)
 - Allows **data bit decoding**
 - **Better range accuracy**
- Design:
 - Hardware: Two LimeSDRs, two laptops, two antennas
 - Software: Matlab tracking loop code (Post signal acquisition and reception)
- Test:
 - Process received signal with tracking loop and attempt to decode data bits.
- Status:
 - Matlab tracking loop ready
 - Must finish range measurement tests first (acquisition)

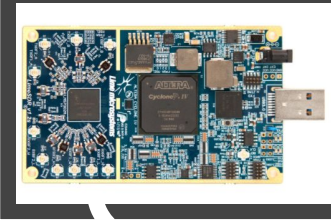


Clock Override Test

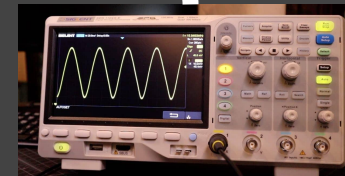
- Purpose:
 - **Verify frequency of overridden clock** in LimeSDR.
 - Should be at 10 MHz
- Design:
 - Hardware: Oscilloscope, GPSDO, LimeSDR
 - Software: GNU Radio
- Test:
 - Check oscilloscope FFT and verify peak at 10 MHz
- Status:
 - Currently working GPS-DO integration
 - After this is done, oscilloscope testing should only take a day or two.



GPSDO



LimeSDR

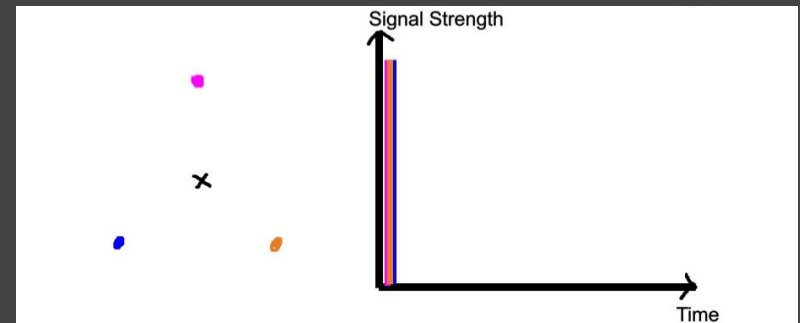
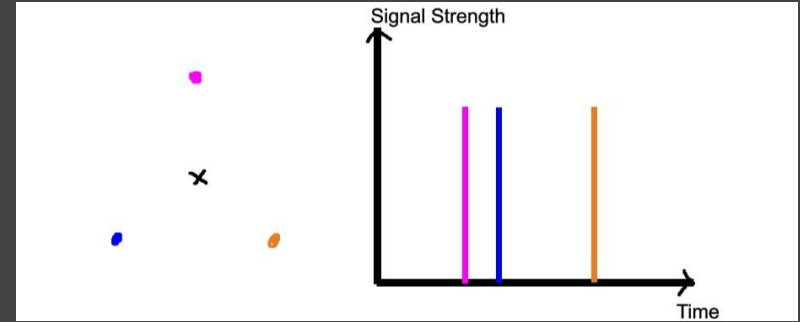


Oscilloscope



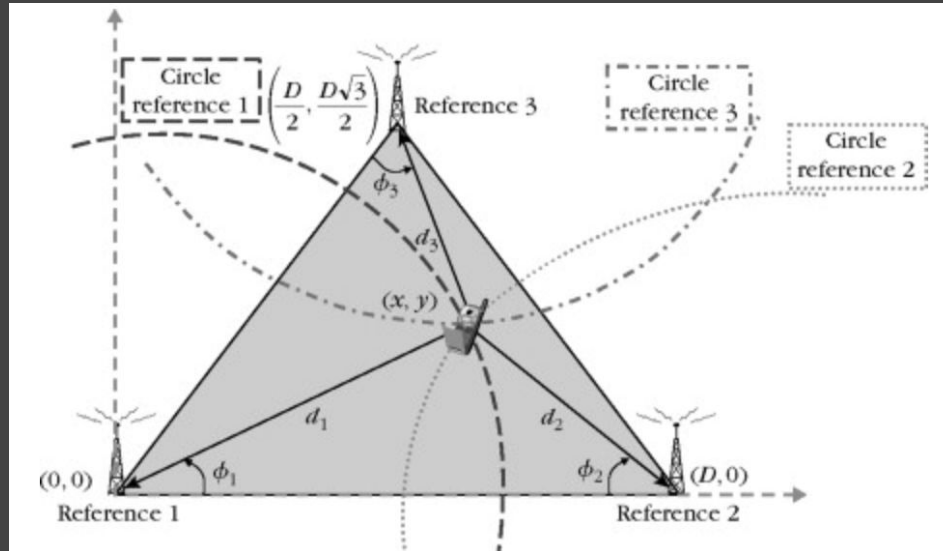
Synchronization Bias Test

- Purpose:
 - **Establish pattern of deviation for biases** by computing synchronization bias for each pseudolite
 - Peak structure should repeat every 1ms
- Design:
 - Hardware: Positioning pseudolite configuration (triangle) + receiver in the middle
 - Software: GNUradio
- Test:
 - Compute relative biases and establish pattern of variation for biases.
- Status:
 - Still setting up pseudolites



Positioning Test

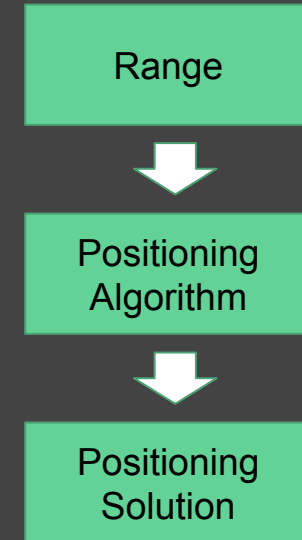
- Purpose:
 - Demonstrate pseudolite system **positioning capability** at testing location





Positioning Test

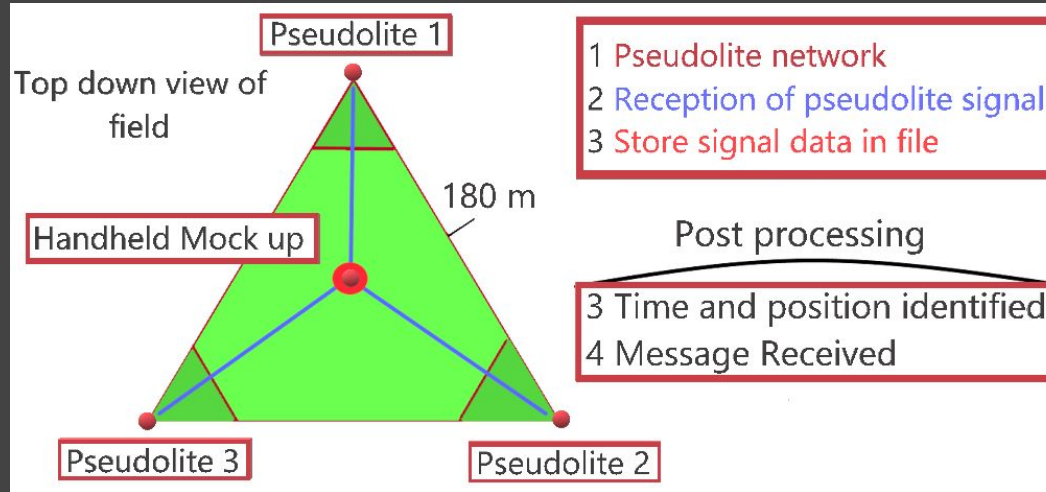
- Design:
 - Hardware: Pseudolites, User system
 - Software: Positioning Algorithm
- Test:
 - **Feed range measurements into positioning algorithm.** Determine the position of the user to sub 10m accuracy at testing location
- Status:
 - Positioning algorithm ready to go
 - Need previous tests to be completed first
 - Pseudolites need to be setup





Positioning + Communication Test

- Purpose:
 - Demonstrate **positioning and communication** capability

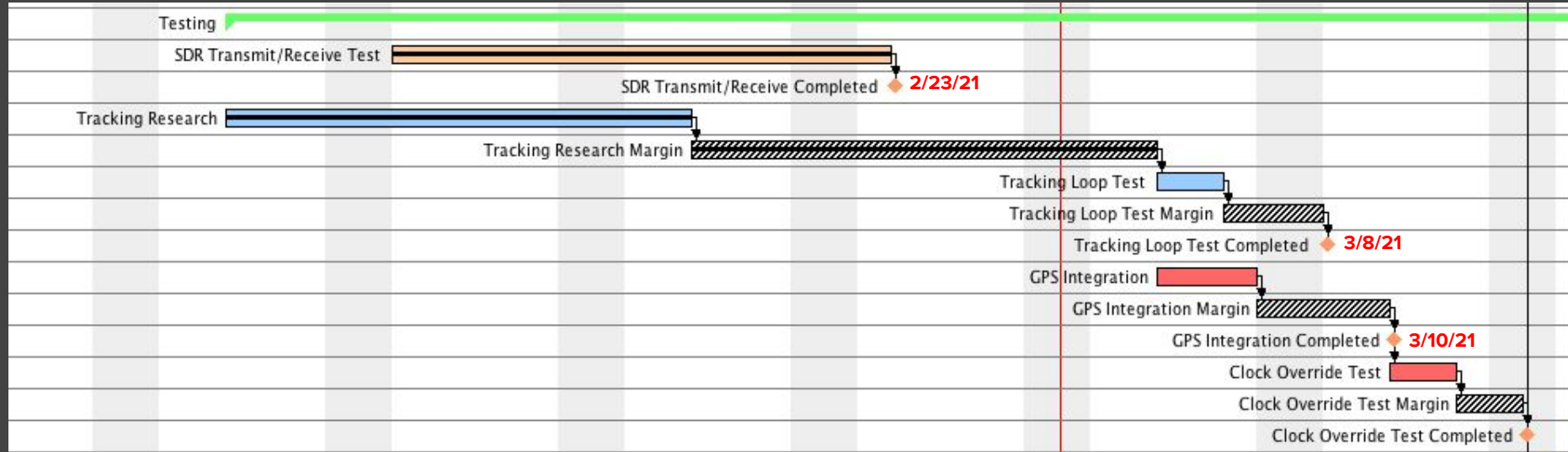




Schedule

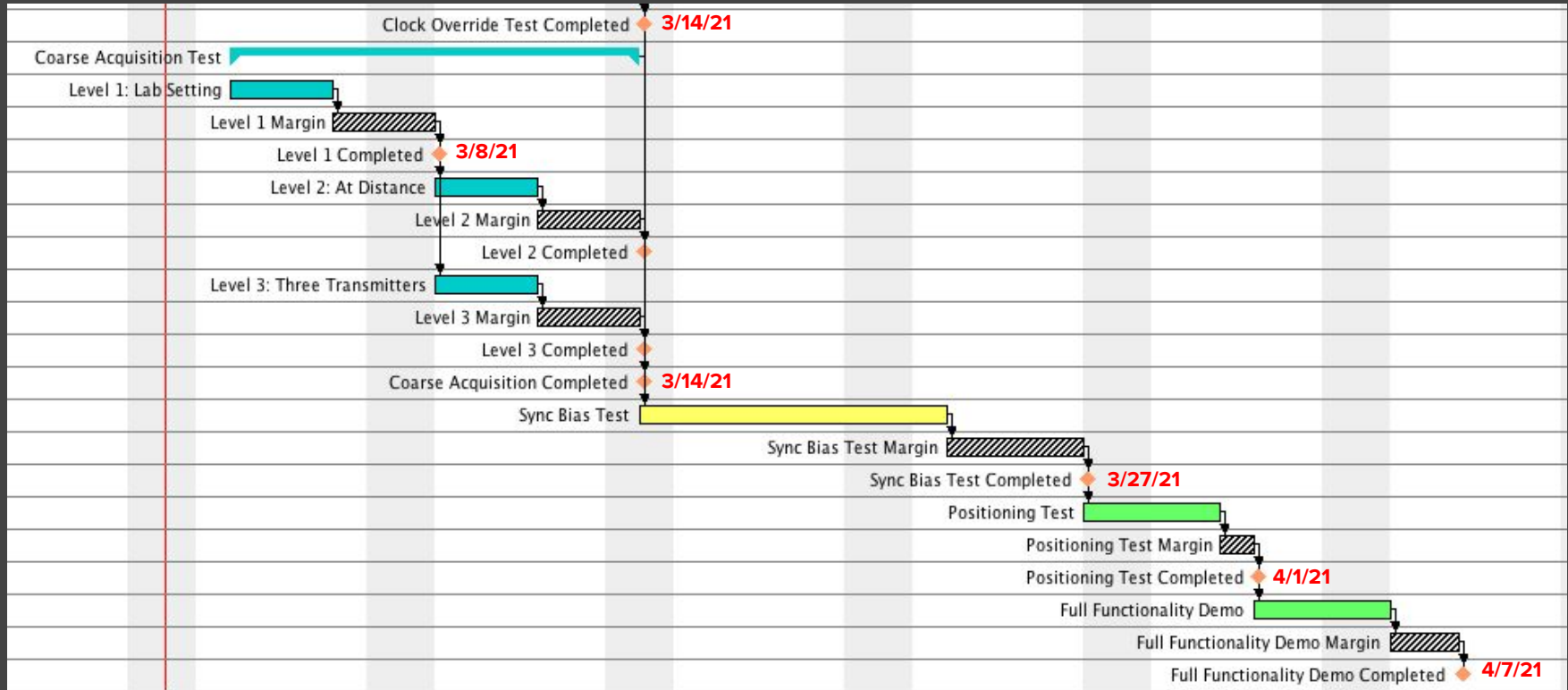


Schedule





Schedule





Budget



Overview

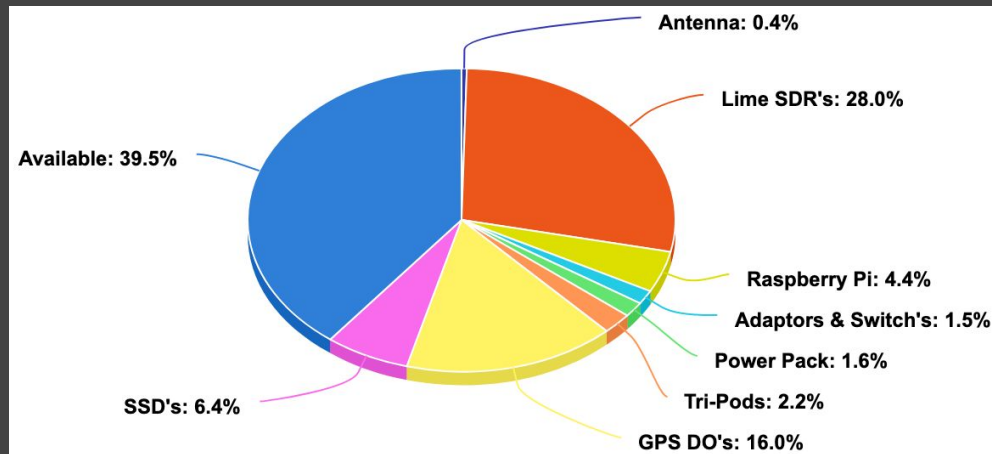
Test Readiness

Schedule

Budget

Cost Plan & Margin

Group	Cost (USD)
Dual TE Connectivity Antennas (4)	\$19.80
Lime SDR's (4)	\$1,399.80
Raspberry Pi (4)	\$220.00
Adapters & Connectors (4 Sets)	\$74.65
Miadi Portable Charger (Power Pack) (4)	\$79.96
5 Feet Tripods (4)	\$111.96
GPS Disciplined Oscillator (4)	\$769.53
Samsung External SSD 500Gb(4)	\$319.96
Budget Status	
Projected Purchases Total	\$3026.12
Current Up to date Purchase Amount Total	\$2862.19
Available Amount / Percentage	\$1973.68 ~ 39.48%





Overview

Test Readiness

Schedule

Budget



Major Item List & Status

Item	Status
Dual Band TE Connectivity Antenna (4)	In Possession (4)
Lime SDR's (4)	In Possession (4)
Laptops (4)	In Possession (4) Team Member Laptops
GPS Disciplined Oscillator (4)	Shipped From China (4) Max ETA (3/1)
Samsung External SSD 500Gb(4)	In Possession (4)

Obstacles: P-card access, Shipping Logistics & Chinese New Years Celebration



Acknowledgements

- Nicholas Rainville
- Jade Morton
- Dennis Akos
- Trudy Schwartz
- Kathryn Wingate
- John Mah
- Steve Taylor
- Brian Breitsch



Questions?



Backup Slides



References

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“Radio Waves Clip Png.” *Pngio*, pngio.com/PNG/a104378-radio-waves-clip-png.html.

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Muñoz David. *Position Location Techniques and Applications*. Academic Press, 2009.

Misra, Pratap, and Per Enge. Global Positioning System: Signals, Measurements, and Performance. Ganga-Jamuna Press, 2012.



Intro/Overview



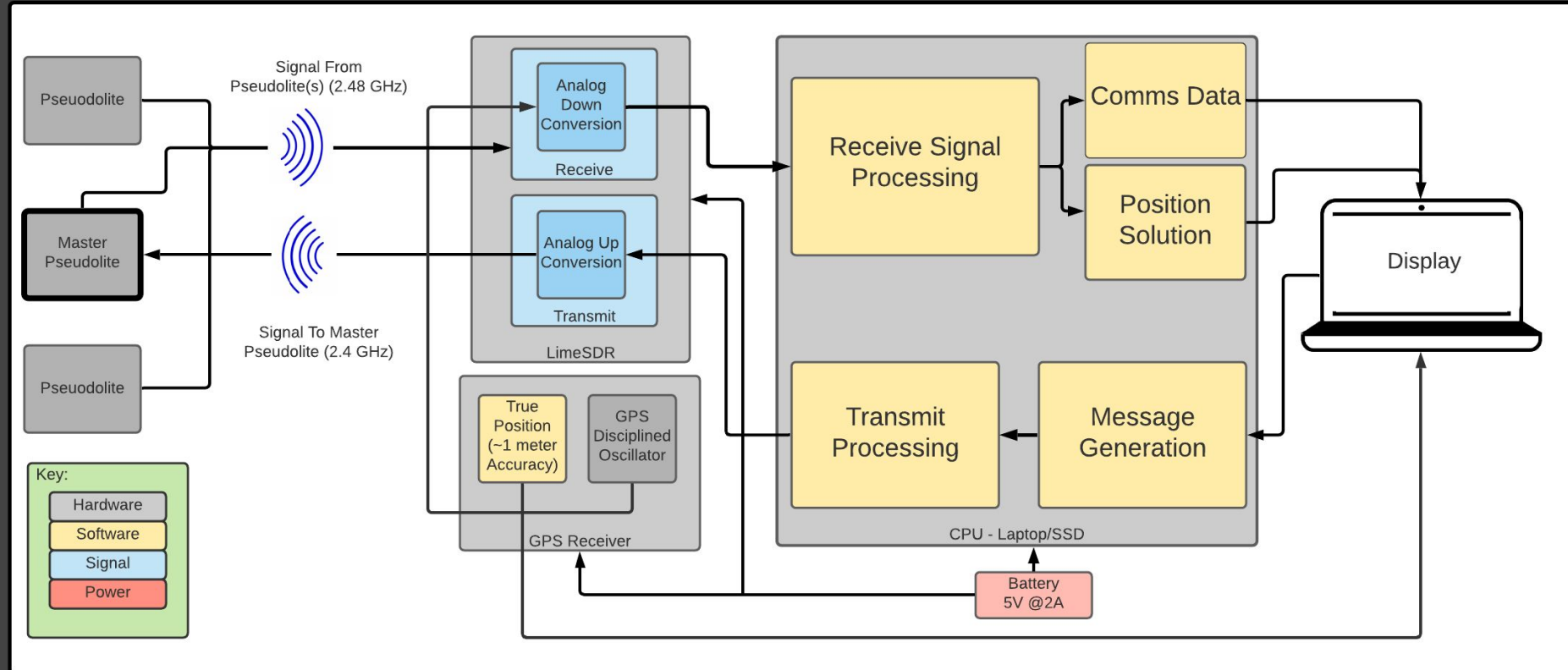
Overview

Schedule

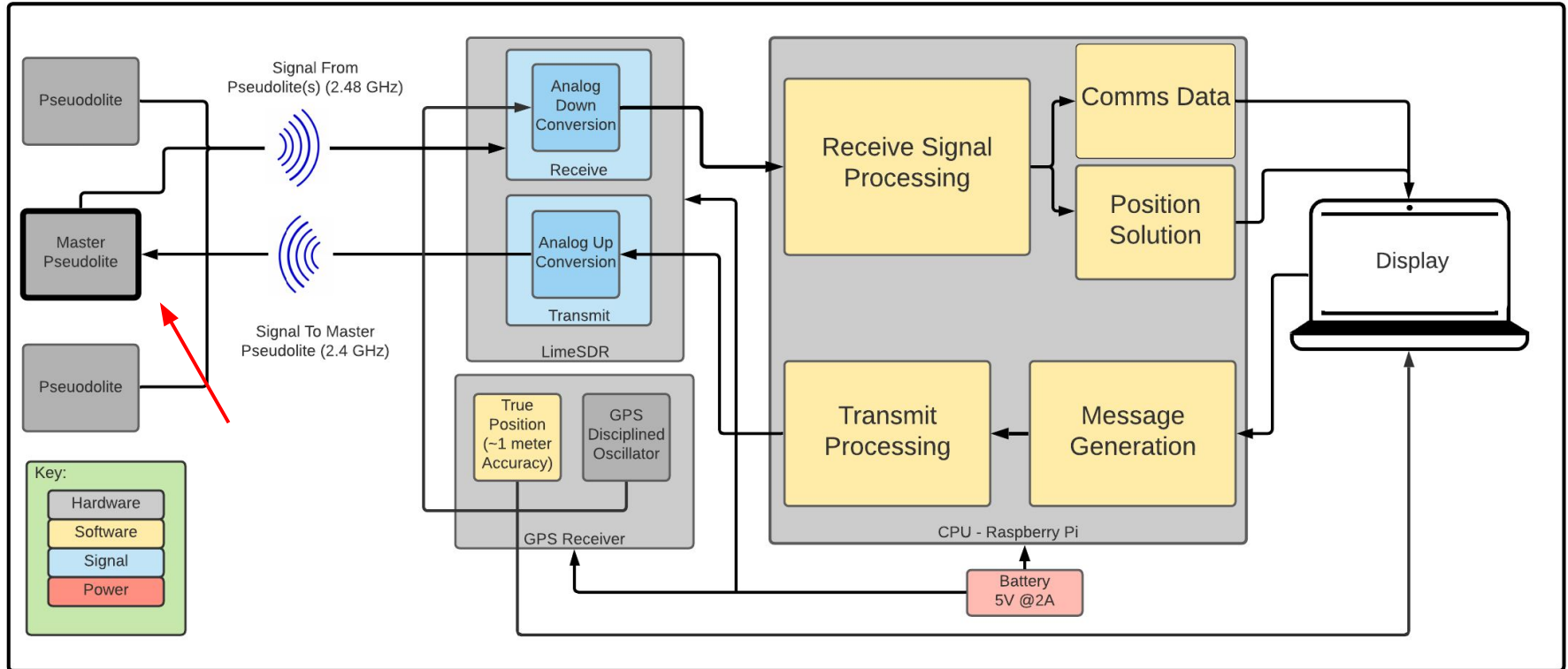
Manufacturing

Budget

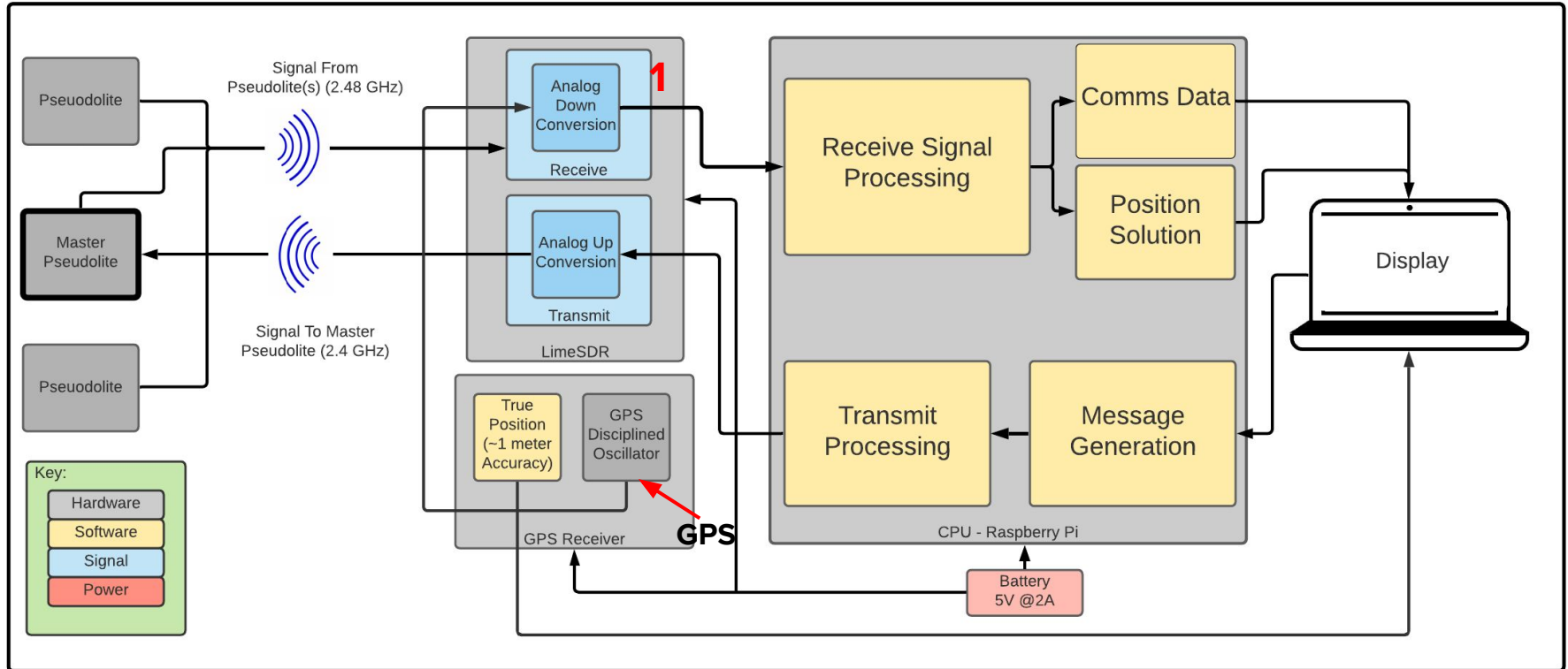
Overall FBD



Overall FBD

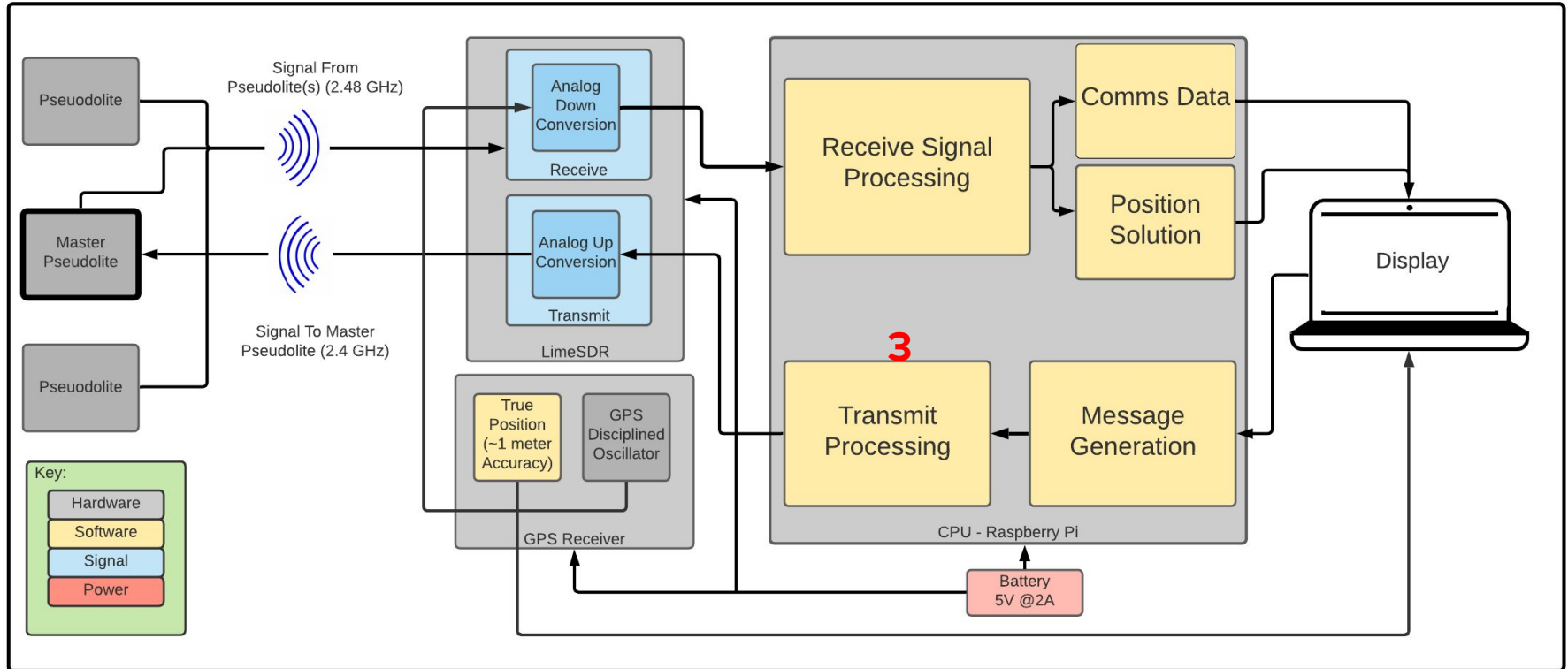


Overall FBD





Overall FBD





Overview

Schedule

Manufacturing

Budget

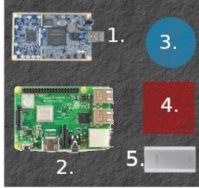
Scaled Down Testing (Placeholder Title)

- Scale from 2.4 km to 0.18 km dist. between transmitters
- Assume 2D (only x and y, no height)
- Transmitting on Earth is at least as complicated as on the Moon
 - Atmospheric losses
 - RF Interference
 - Man-made structures/materials interfering
- Not testing environmental constraints

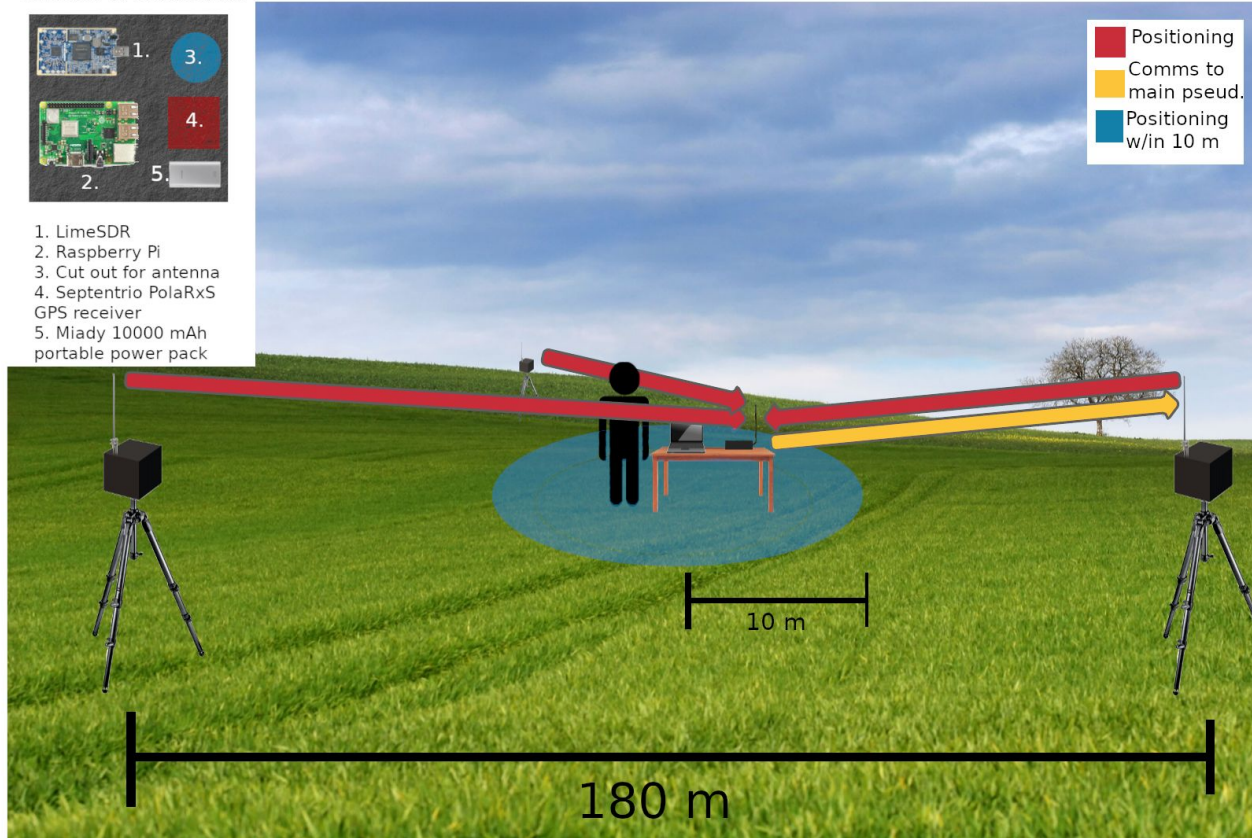
A scaled-down test in Boulder, CO is still able to validate the design for use on the lunar surface

Design Solution Set Up

Contents of Black Boxes



1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack





Overview

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Software Overview

GNU Radio

- Reference oscillator integration
- Signal reception and transmission
 - Modulation
 - Multiplexing
 - Up/Down Conversion
 - Acquisition
 - Tracking

Matlab

- Packet Creation
- Positioning algorithm
- For simulation
 - Modulation
 - Multiplexing
 - Up/Down Conversion
 - Acquisition
 - Tracking

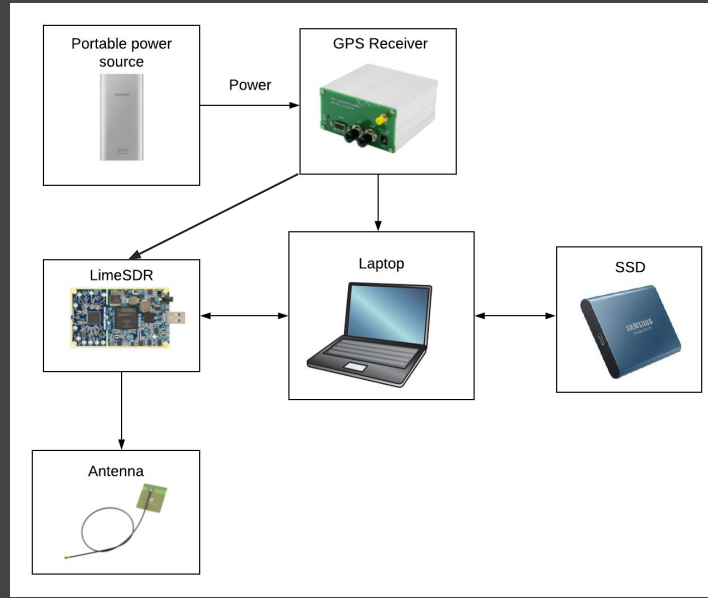
LimeSDR

- Overwrite clock with reference oscillator
- Integrate antenna

Team Created
Open Source Edited
Outside Leveraged

What is a Pseudolite?

- A portable ground station
 - Transmits positioning messages
 - Receives user communication messages



Pseudolite Hardware Block Diagram





Positioning Test

- Design:
 - Hardware: Pseudolites, User system
 - Software: Positioning Algorithm
- Test:
 - **Feed range measurements into positioning algorithm.** Determine the position of the user to sub 10m accuracy at testing location
- Status:
 - Positioning algorithm ready to go
 - Need previous tests to be completed first
 - Pseudolites need to be setup

$$\begin{bmatrix} \delta \hat{\mathbf{x}} \\ \delta \hat{b} \end{bmatrix} = (\mathbf{G}^T \mathbf{G})^{-1} \mathbf{G}^T \delta \boldsymbol{\rho}$$

$$\mathbf{G} = \mathbf{A} = \begin{bmatrix} -\frac{x^{(1)} - x_0}{R_0^{(1)}} & -\frac{y^{(1)} - y_0}{R_0^{(1)}} & -\frac{z^{(1)} - z_0}{R_0^{(1)}} & 1 \\ -\frac{x^{(2)} - x_0}{R_0^{(2)}} & -\frac{y^{(2)} - y_0}{R_0^{(2)}} & -\frac{z^{(2)} - z_0}{R_0^{(2)}} & 1 \\ -\frac{x^{(3)} - x_0}{R_0^{(3)}} & -\frac{y^{(3)} - y_0}{R_0^{(3)}} & -\frac{z^{(3)} - z_0}{R_0^{(3)}} & 1 \\ \vdots & \vdots & \vdots & \vdots \\ -\frac{x^{(m)} - x_0}{R_0^{(m)}} & -\frac{y^{(m)} - y_0}{R_0^{(m)}} & -\frac{z^{(m)} - z_0}{R_0^{(m)}} & 1 \end{bmatrix}$$

Geometry Matrix

$$\delta \boldsymbol{\rho} = \mathbf{G} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix}$$



Test Site Images











Manufacturing Slides



- **Signal Reception: SSD + Laptops instead of Raspberry Pi**
 - Due to computation restrictions, boot Linux off external SSDs connected to laptops for additional speed and memory
- **Signal Transmission: Pseudolites equipped with a reference oscillator**
 - GPS disciplined oscillator (GPSDO)
 - Parts have been selected and ordered.



Customer Value Proposition

- Develop testing scenario
 - Testing out a **scaled down version** of the system
 - What it takes to test the system
 - What works, what doesn't
- Develop/Integrate hardware & software
 - LimeSDR (Radio)
 - Performance metrics
 - External clock integration
 - GNU Radio v3.8 (Software)
 - Performance
 - Interfacing with LimeSDR



Overview

Schedule

Manufacturing

Budget

Manufacturing Overview

Purchase

- LimeSDR
- Reference Oscillator
- Antenna
- SSD
- Tripod

Integrate

- Pseudolite electronics
 - Antenna
 - GNU Radio
 - Laptop
- Pseudolite clock
 - Override internal clock
- Calibrate system

Develop / Manufacture

- GNU Radio script
- Communications Protocol
- Positioning Algorithm
- Pseudolite Housing



Overview

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Manufacturing Overview

Purchase

- LimeSDR
- Reference Oscillator
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Integrate

- Pseudolite electronics
 - Antenna
 - GNU Radio
 - Laptop
- **Pseudolite clock**
 - **Override internal clock**
- **Calibrate system**

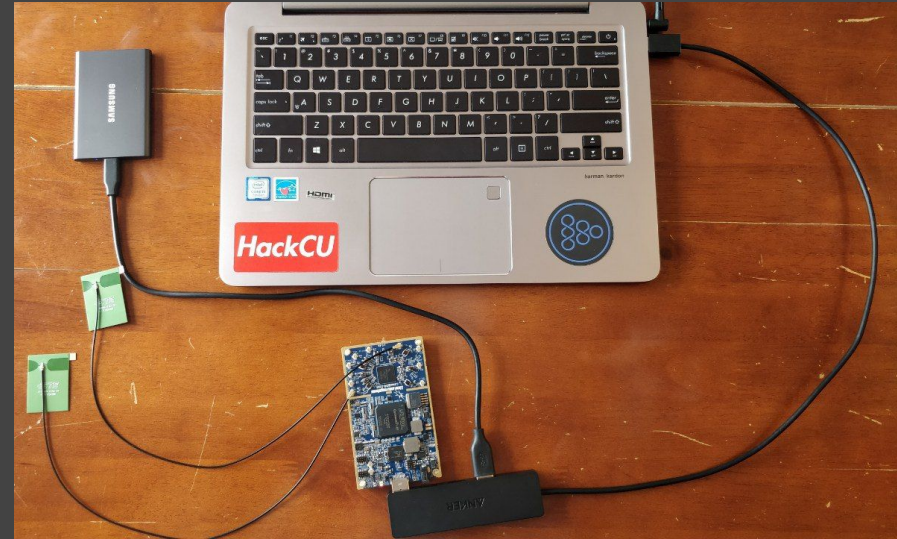
Develop / Manufacture

- GNU Radio script
- **Communications Protocol**
- **Positioning Algorithm**
- Pseudolite Housing

Integrate: Pseudolite Electronics

Task: Send and receive test signals using pseudolite electronics

- GNU Radio 3.8 (Software)
- Ubuntu Linux booted on a Samsung T7 500 GB External SSD
 - Running GNURadio
 - 1 GBps read/write
- Dual Band TE Connectivity Patch Antennas
- LimeSDR USB 3.0
- ASUS Zenbook (Core i5 @ 3.5 GHz)
- Write samples to file



Status: Done



Task: Override internal LimeSDR clock in each pseudolite

-

+



60



Overview

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Integrate: Calibrate System

Task: Measure and calibrate for total system bias (Pseudolite + GPSDO) when transmitting

Status: **Waiting** for overall system integration





Overview

Schedule

Manufacturing

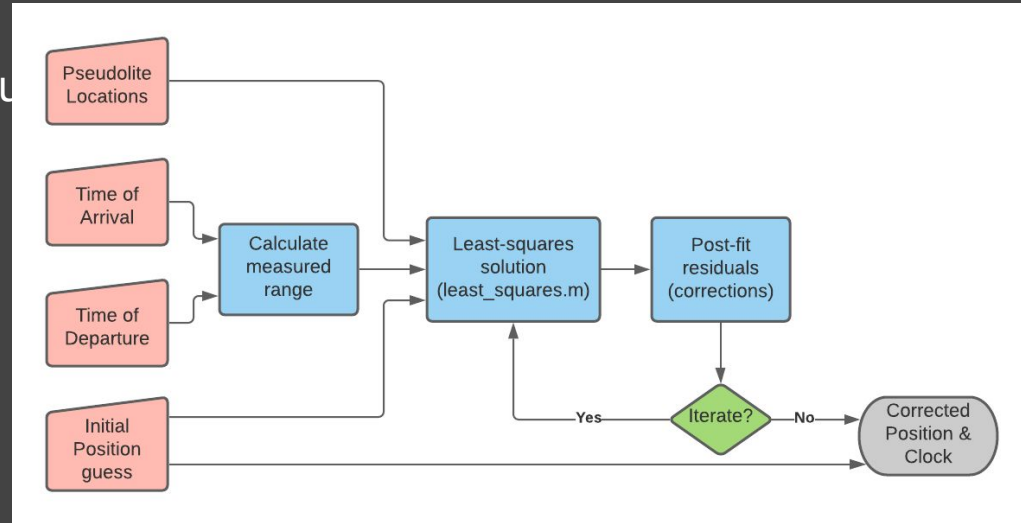
Budget

Develop: Positioning Algorithm

Task: Develop code to process received samples and then compute position.

Status:

- Signal processing script **in progress**
- Positioning script **done**



Positioning Script Block Diagram



Overview

Schedule

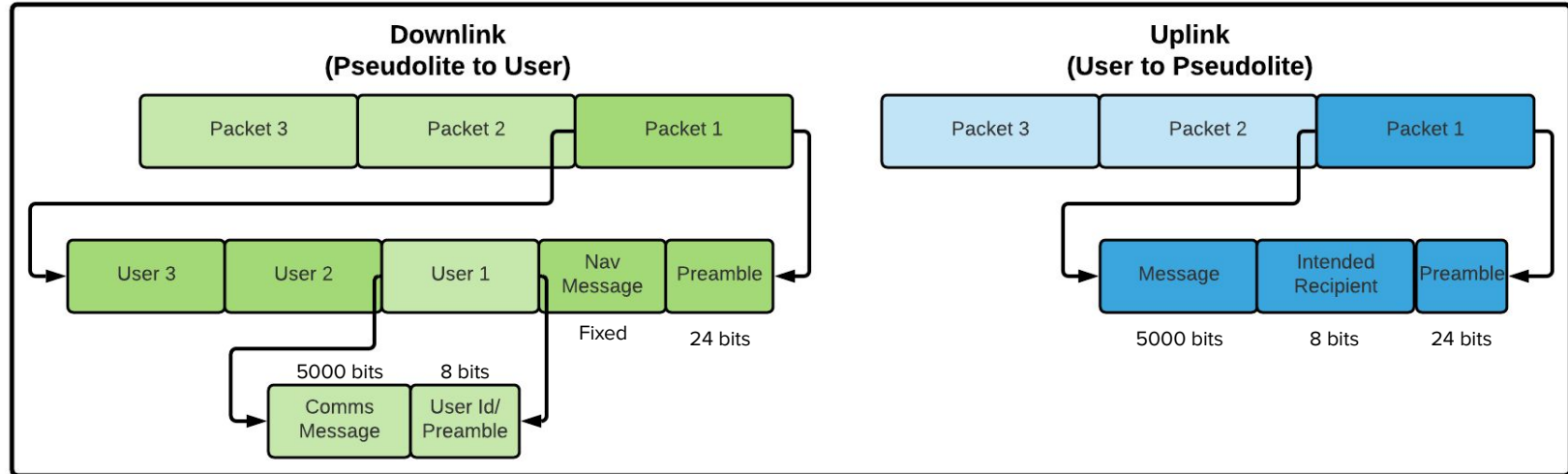
Manufacturing

Budget

Develop: Communications Protocol

Task: Format communication data into packets

Status: Done



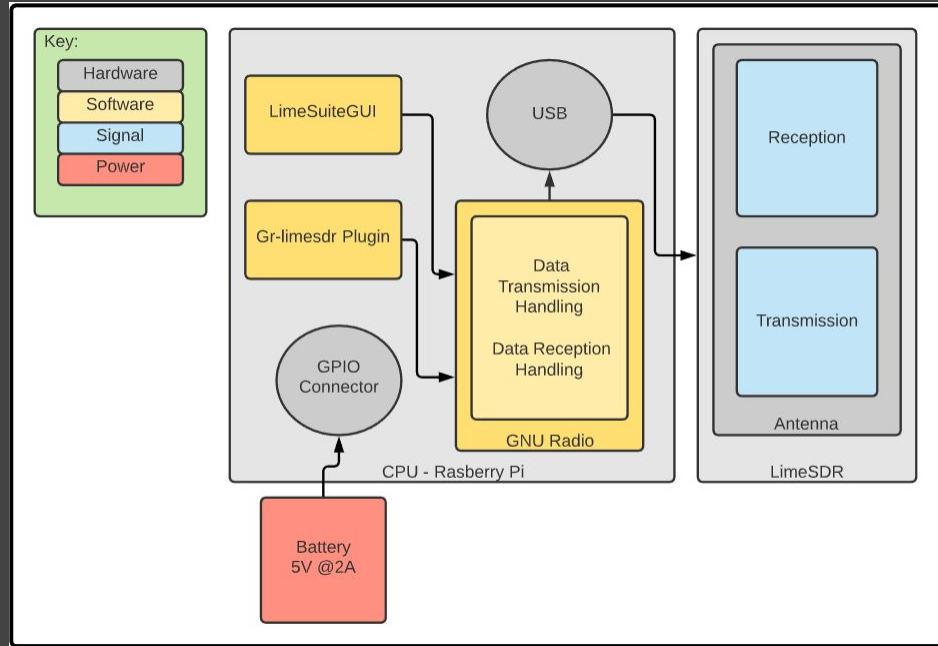


- GNU Radio system setup
- Signal detection between two transmitters
- Measured signal interference at testing location
- Preliminary CAD Housing Model Done
- Selected Referenced Oscillator



Hardware

Pseudolite Electronic Configuration

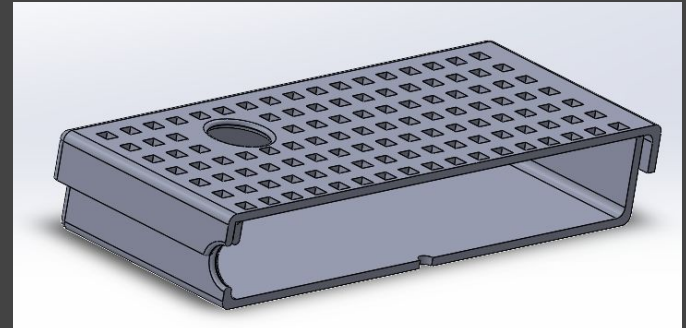
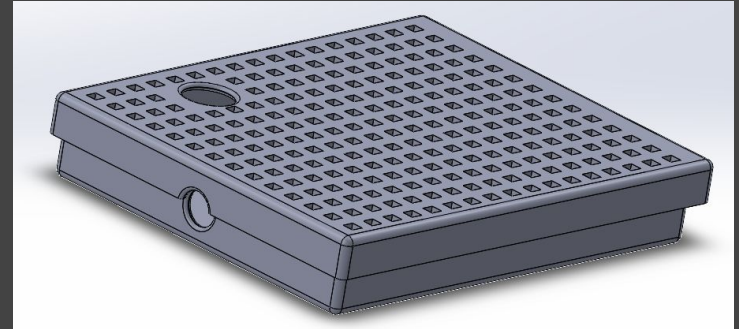


- Electronic/Hardware functional block diagram.
- DR 2.1.3: The system will operate using the LimeSDR electronics device
- The LimeSDR combined with the GNU Radio software package.
- FR1: The system must operate under a scalable LPS model
- How is this scalable?

DR 2.1.3: LimeSDR is in use

Mechanical Hardware Integration - CAD

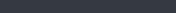
- CAD model of case for holding the limeSDR, GPS, power supply, and Raspberry Pi
- Holes for cable, antenna, and ventilation.
- To do: get a tripod so that the model can be made to attach





SDR - Requirements

- FR1: The system must operate under a scalable LPS model
 - DR 2.1.3: The system will operate using the LimeSDR electronics device
- FR2: Demonstrate SMS-like communication
 - DR 2.2.1: Device must demonstrate wireless transmission and reception of data
- FR 4: System will transmit and receive data between (2.4 - 2.48 GHz)
 - DR 2.4.1: Demonstrate uplink transmission at 2.4-2.48 GHz
 - DR 2.5.1: Demonstrate downlink transmission at 2.4-2.48 GHz

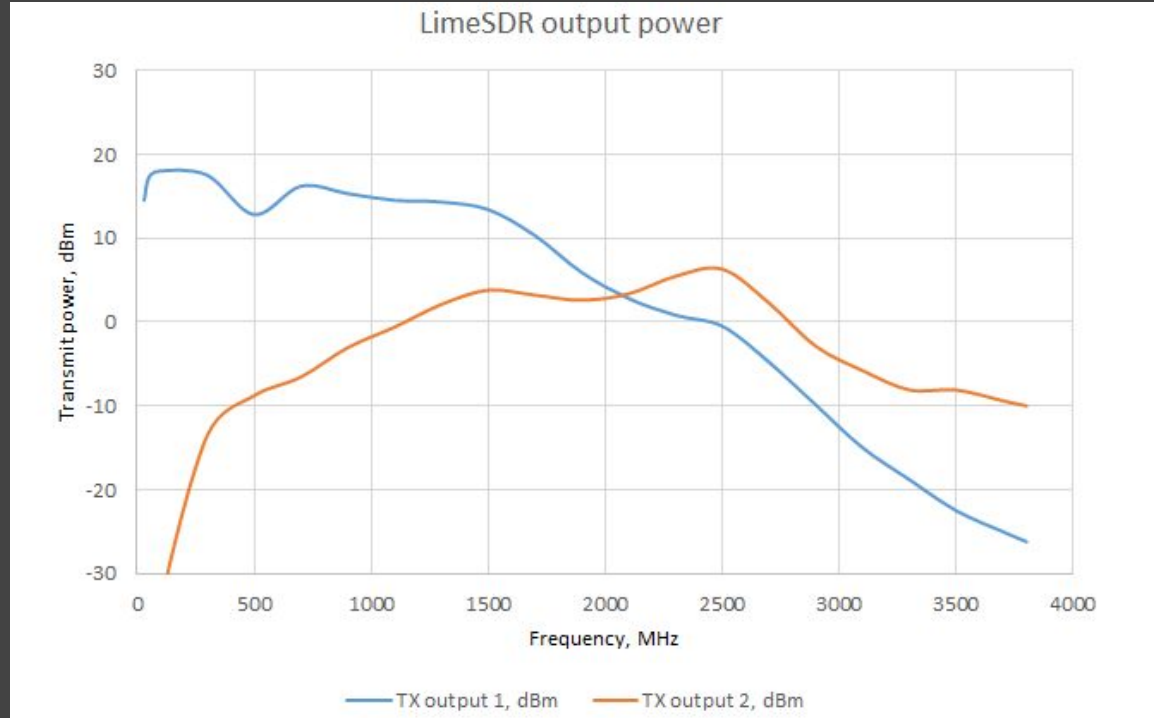


GNU Radio
THE FREE & OPEN SOFTWARE RADIO ECOSYSTEM

- 69



LimeSDR Output Power





Antenna - Requirements

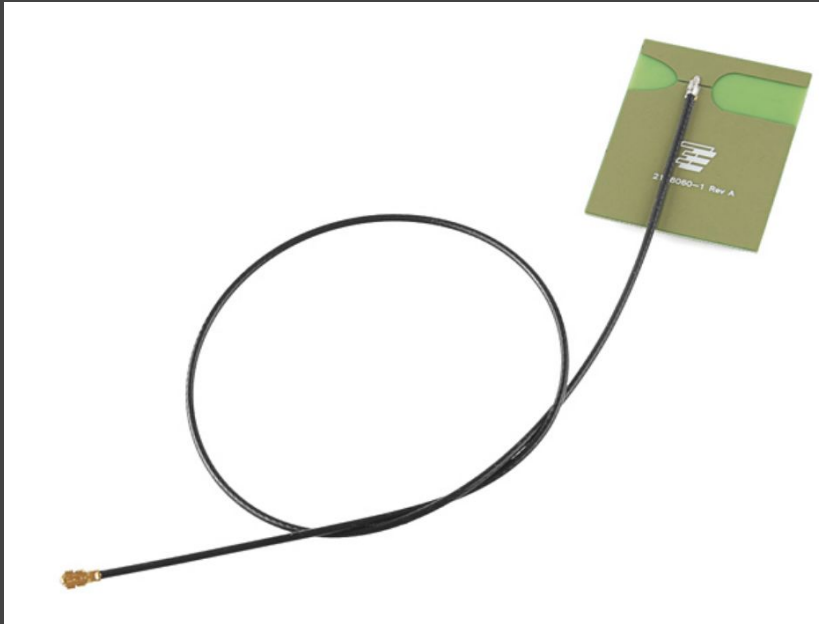
FR4: The system will transmit data on the S-Band frequency.

- DR 2.4.1 : Demonstrate uplink transmission at a frequency range between 2.4-2.48 GHz

FR5: The system will receive data on the S-Band frequency.

- DR 2.5.1 : Demonstrate downlink reception at a frequency range between 2.4-2.48 GHz

Dual Band-TE Connectivity Antenna



Dual Band-TE Connectivity

- Peak Gain 2dBi
- Embedded Antenna Style
- Freq Range: 2.4-3.8GHz
- Low Weight (<3.3gram)
- Low Cost (5 USD)
- Voltage Standing Wave Ratio (VSWR) - MAX 3:1



Antenna - Selection Analysis

Customer Requirement: S-Band (due to radio silent far side of the moon)

Frequency Range	PROS	CONS
S- Band: (2-4 GHz)	<ul style="list-style-type: none">• Previous communication satellites (NASA)• Unlicensed bandwidth (2.4-2.483 GHz) good for testing• Low Cost system	<ul style="list-style-type: none">• Large amount of interference (Many devices at this bandwidth)• Mainly used for large antennas (transmission)



Project Overview

Baseline Design

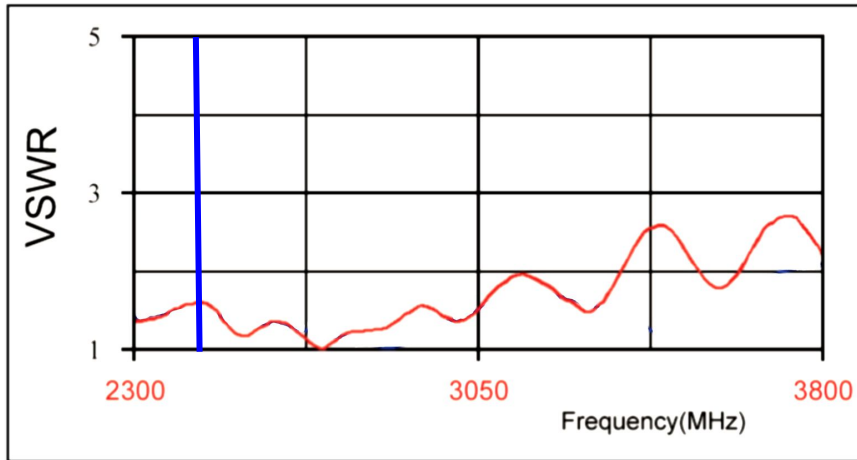
Feasibility

Summary



Antenna - Evidence of Feasibility

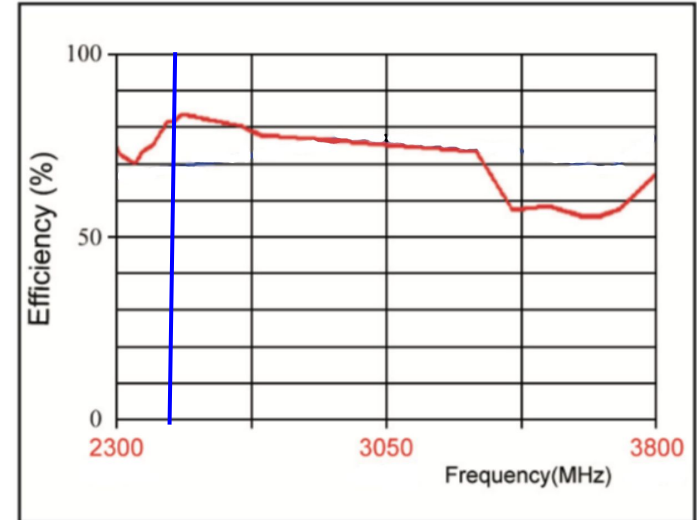
VSWR



DR 2.4.1, 2.5.1

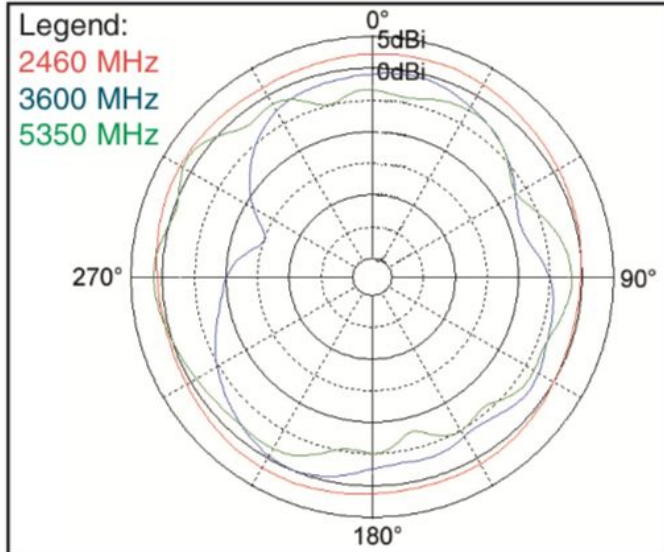
FEASIBLE

Efficiency

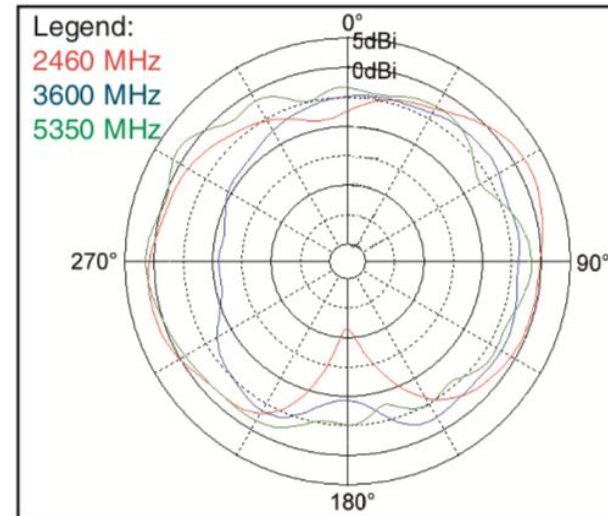


Antenna - Evidence of Feasibility

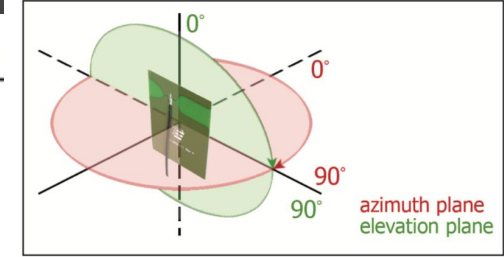
Azimuth



Elevation



Test Orientation in Free Space

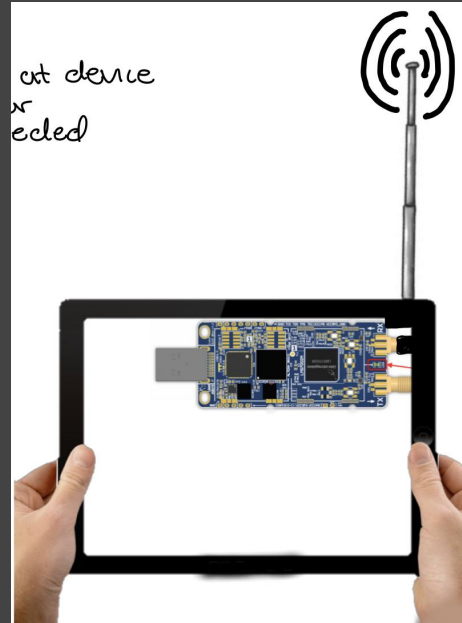
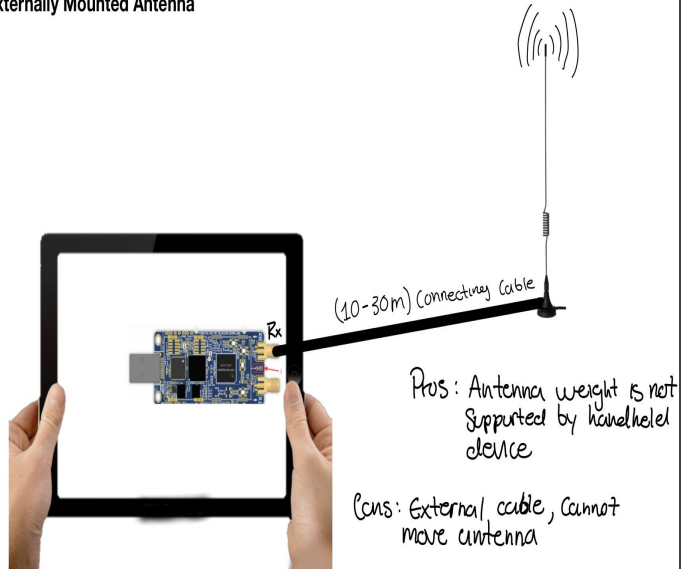




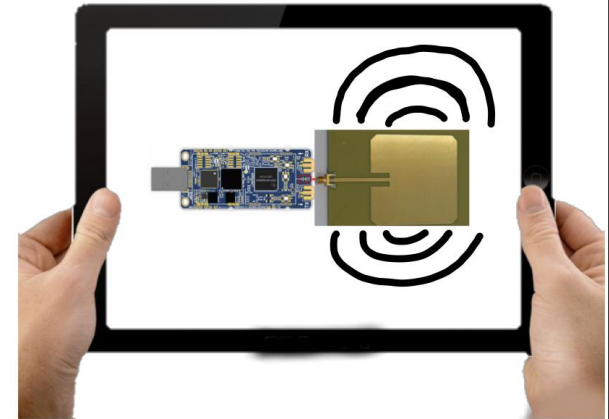
76

Antenna

Externally Mounted Antenna



INTERNALLY EMBEDDED ANTENNA





Antenna Tradestudy BU-Slides (Table 1)

Table1.Antenna Options with performance characteristics

Antenna Model	Antenna Type	Antenna Design Configuration	Gain (dBi)	Bandwidth(GHz)
TE Connectivity Antenna	Omnidirectional-DualBand	Embedded	2	2.4-3.8 GHz, 5.150-5.870 GHz
Laird-MAF94051	External Dual-Band Omnidirectional	Attached	2	2.4-2.5
Laird OC24006H	Omnidirectional/ horizontally polarized	External	6	2.4-2.5
Argain-N2420M	Single Band embedded	Embedded	2.81	2.4-2.49



Antenna Tradestudy BU-Slides (Table 2)

Table2.Antenna and other characteristics

Antenna Model	Weight (grams)	Unit-Cost (USD Currency)	Connector Type	Effective Area (Ae) or Effective Antenna Aperture (m^2)
TE Connectivity	3.3	4.79	Cable-Side Entry	0.001828
Laird-MAF94051	113.4	7.64	RP SMA Connector with 90 degree elbow	0.001814
Laird OC24006H	260	50.20	N-Female	0.004556
Argain-N2420M	0.5	1.80	IPEX/MHF/U FL	0.002186



Antenna Tradestudy BU (Effective Area Equations)

$$A_e = \frac{\lambda^2}{4\pi} G = \frac{c^2}{f^2} \times \frac{G}{4\pi}$$

Eq#1 is assuming Linear Gain and Eq#2 Assuming dB Gain

$$A_e = \frac{c^2}{f^2} \times \frac{10^{\frac{G (dB)}{10}}}{4\pi}$$



Table3. Point Matrix Criteria

Metric	High Score (3)	Medium Score(2)	Low Score(1)
Cost Effectiveness	low cost for a high performance and capabilities (price<5)	The antenna has an average cost for its capabilities(5-30)	The antenna has a high cost for its capabilities (Over 30)
Weight	The antenna is very light (Under 50 grams)	The antenna has a medium weight (from 50-200grams)	The antennas has a high weight value (over 200 grams)
Compatibility	The antenna can easily connect to the SDR, The user has no issues when using the antenna	The antenna is compatible with the sdr without any extra hardware but needs intervention for it to start working	The antenna is not compatible with the sdr without any extra hardware and needs intervention for it to start functioning
Performance/Specs	Omnidirectional, Operates in designated frequency, high bandwidth, good materials, Large Effective Area,etc.	The antenna has all the required specs but does not have good materials or other hardware constraints	The antenna doesn't satisfy all the needed requirements



Antenna TradeStudy BU-Slides (Table 4)

Metric	Weighting	Antenna #1 TE	Antenna #2 Laird-1	Antenna #3 Laird-2	Antenna #4 Argain
Cost	0.15	3	2	1	3
Weight	0.30	3	2	1	3
Compatibility	0.15	3	3	2	3
Performance/Specs	0.40	3	3	3	2
Total Score	100%	100%	80%	65%	90%



Component	Max Operating Power (Transmit)	Max Operating Power (Receive)
LimeSDR	4.5W (More like 2mW)	NA
Antenna	NA	3.0W
Raspberry Pi	NA	3.5W
Total	4.5W	6.5W
	or approx 2mW	

Battery Pack: Miady
- 5V 2A spread
evenly from two
outputs



Power Pack (Selected Battery)

Miady 10000mAh Dual USB Portable Charger





Scaling and Power

- Scaling
 - In the real-life application of this project we will use 2.4km side-lengths for the pseudolite geometry
 - Calculated using the horizon of the Moon.
 - Would also need to take into account any terrain on the Moon.
 - **Scaled down geometry** was created for **practical** reasons.
- Power
 - Current testing power is **2 mW** which is **more than enough** (even for a pseudolite configuration with 2.4km side-lengths)
 - As part of our testing we will **vary transmit power** and plot ranging accuracy vs. SNR in order to determine the **optimal transmit power**.



- Time: 1:07 pm



Navigation



Overview

Solution

CPE

DR

Risk & M

V&V

Planning



Positioning System Requirements

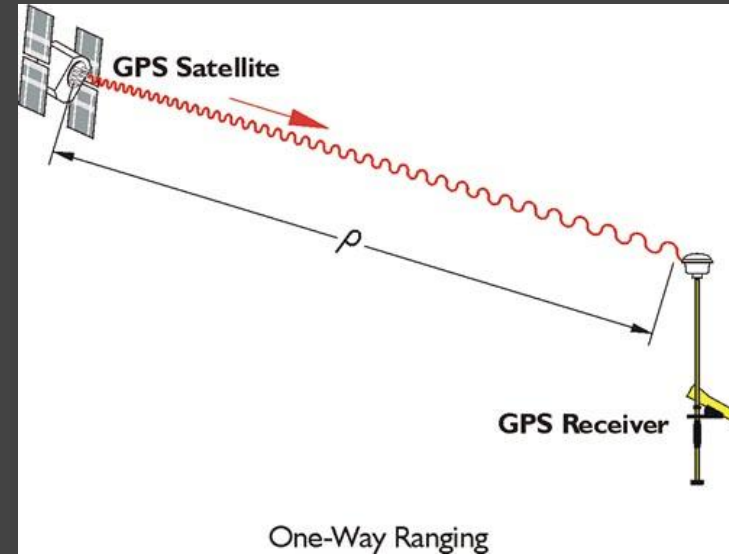
FR3: The system will be able to provide an architecture for a navigation solution with a 10 meter position accuracy and a 30 nanosecond 1- σ transfer time

- **DR 2.3.2** : Provide LPS coverage to the Moon's South Pole

Solution: Pseudolite geometry composed of a 0.18km equilateral triangle + GPS-disciplined clocks

One way ranging

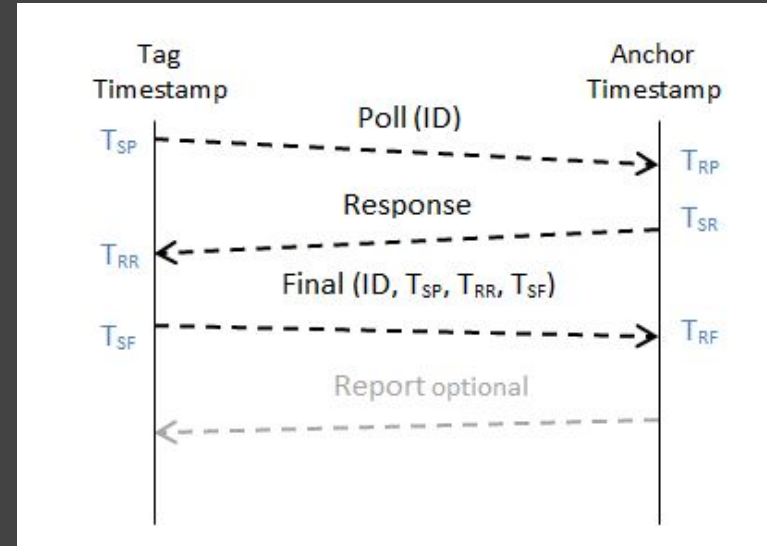
- Requires accurate clocks on both ends to determine the distance based on time differences.
- Lots of research/documentation available supporting the algorithms and systems.
- Only satellite transmits.
- The energy and time required is low due to only one transmission of data.





Two way ranging

- Requires clocks on both ends but does need them to be synced up with each other.
- Receiver must transmit data BACK to the transmitter - multiple times.
- **Both the satellite and the receiver know the location of the satellite.**
- The energy and processing requirements are high due to the fast data transmission requirements.





One way vs Two way ranging

	One way	Two way
Power required	Low	High
Position information	Only receiver	Both receive info
Clock synchronization	Required	Not required

Positioning Algorithm: Least-Squares Solution

- Reason for using least-squares method
- Position Guess and corrections

$$\begin{aligned}
 \mathbf{G} = \mathbf{A} = & \begin{bmatrix} -\frac{x^{(1)} - x_0}{R_0^{(1)}} & -\frac{y^{(1)} - y_0}{R_0^{(1)}} & -\frac{z^{(1)} - z_0}{R_0^{(1)}} & 1 \\ -\frac{x^{(2)} - x_0}{R_0^{(2)}} & -\frac{y^{(2)} - y_0}{R_0^{(2)}} & -\frac{z^{(2)} - z_0}{R_0^{(2)}} & 1 \\ -\frac{x^{(3)} - x_0}{R_0^{(3)}} & -\frac{y^{(3)} - y_0}{R_0^{(3)}} & -\frac{z^{(3)} - z_0}{R_0^{(3)}} & 1 \\ \vdots & \vdots & \vdots & \vdots \\ -\frac{x^{(m)} - x_0}{R_0^{(m)}} & -\frac{y^{(m)} - y_0}{R_0^{(m)}} & -\frac{z^{(m)} - z_0}{R_0^{(m)}} & 1 \end{bmatrix} \\
 \text{Geometry Matrix} & \\
 \delta \mathbf{p} = \mathbf{G} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix} &
 \end{aligned}$$

$$\begin{bmatrix} \delta \hat{\mathbf{x}} \\ \delta \hat{b} \end{bmatrix} = (\mathbf{G}^T \mathbf{G})^{-1} \mathbf{G}^T \delta \mathbf{p}$$

Overview

Solution

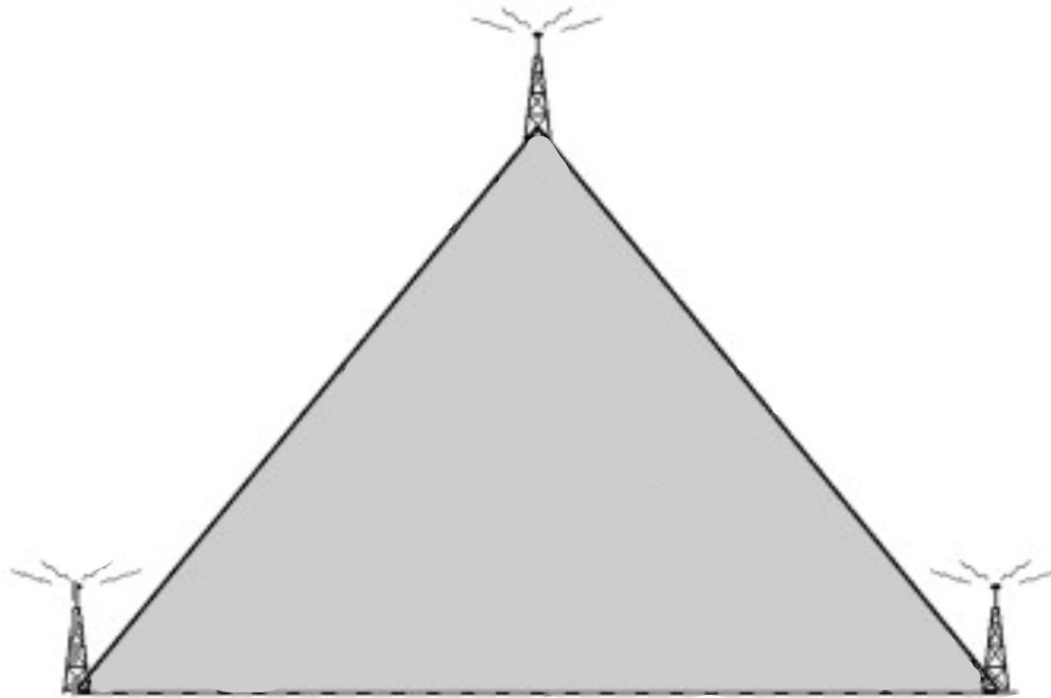
CPE

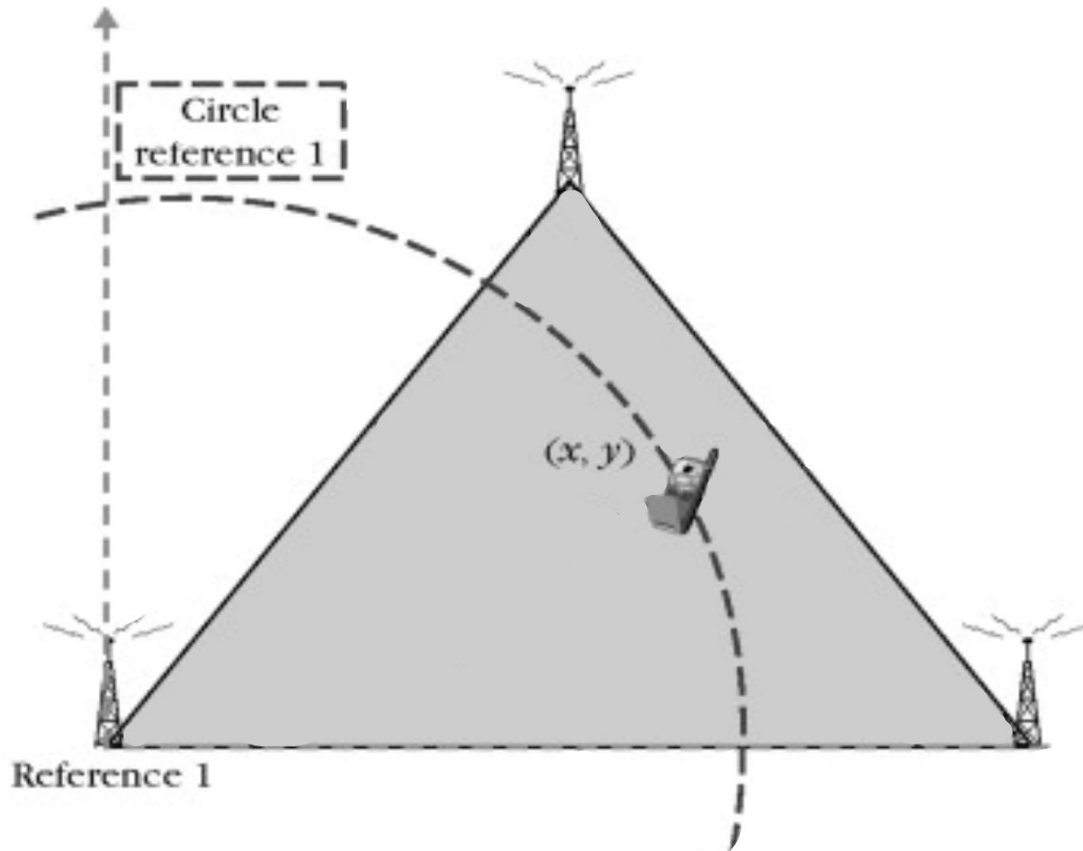
DR

Risk & M

V&V

Planning





Overview

Solution

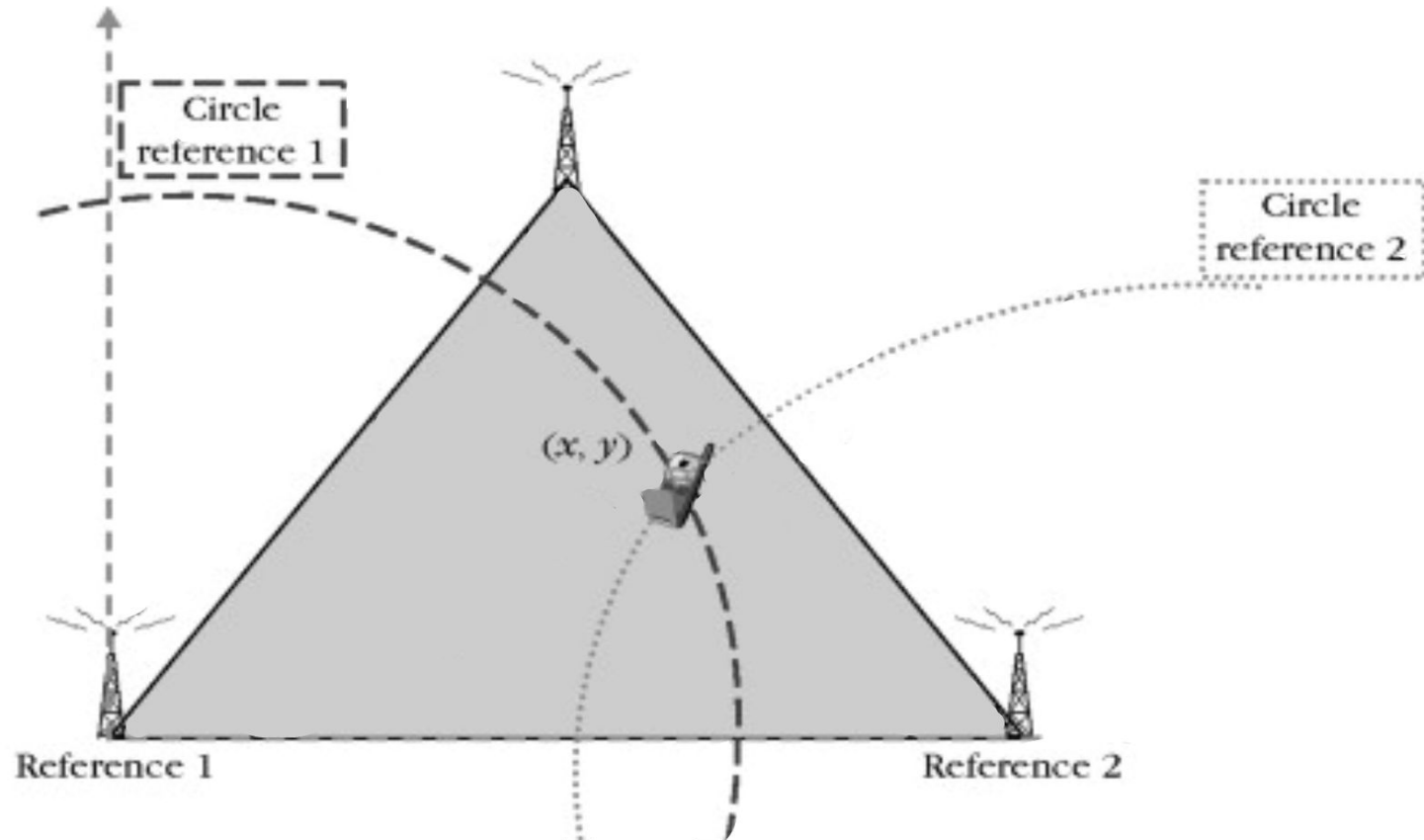
CPE

DR

Risk & M

V&V

Planning



Overview

Solution

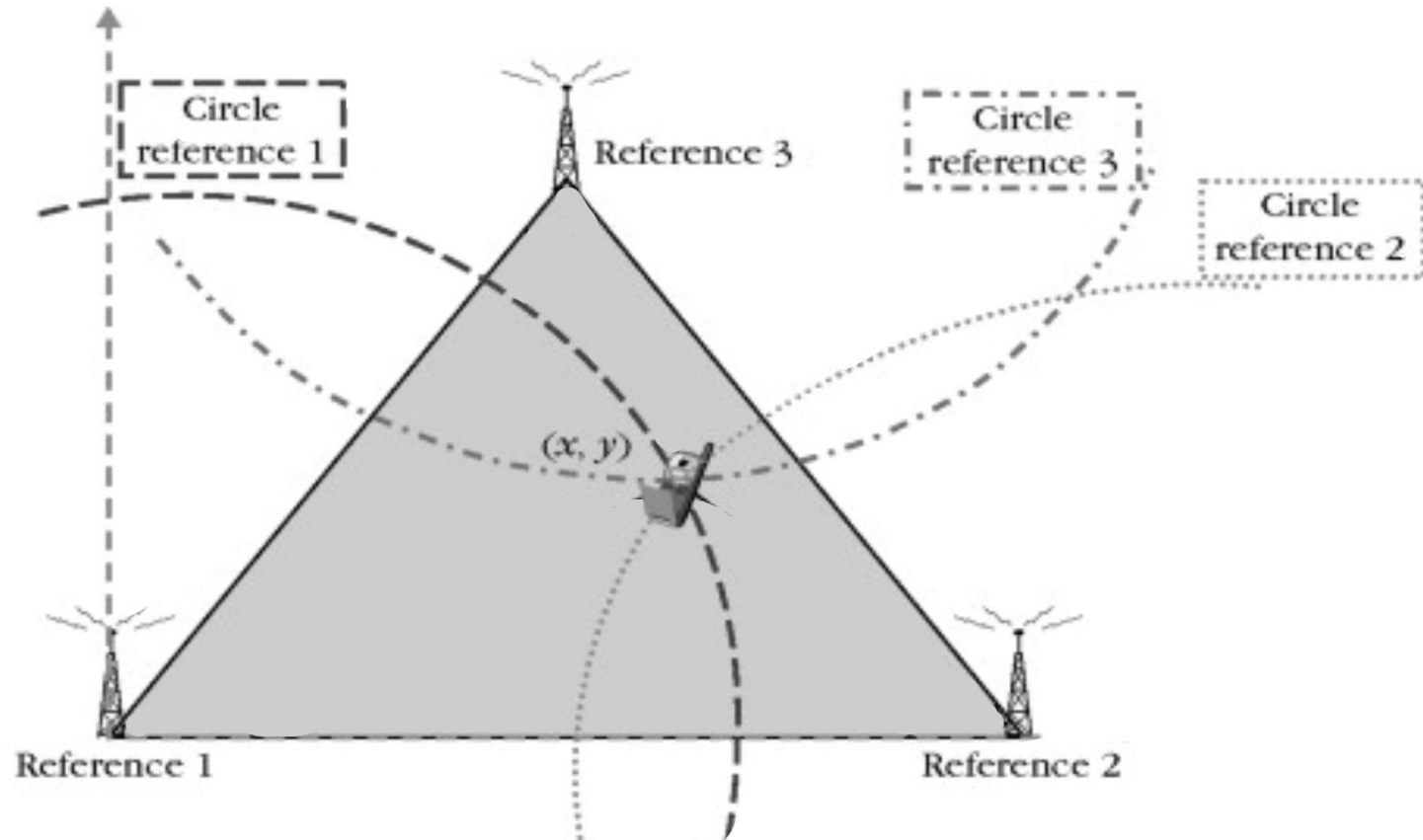
CPE

DR

Risk & M

V&V

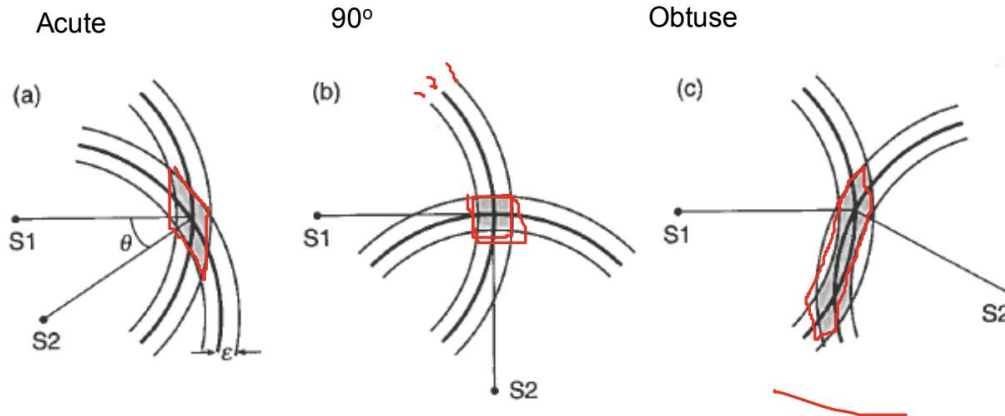
Planning



Positioning System - HDOP

Horizontal Dilution Of Precision

Definition: “Used to specify error propagation as a mathematical effect of navigation pseudolite geometry on positional measurement precision”.



$$\text{RMS horizontal error} = \sqrt{\sigma_E^2 + \sigma_N^2} = \sigma \cdot \text{HDOP} \quad (6.2)$$

$$\text{RMS vertical error} = \sigma_U = \sigma \cdot \text{VDOP} \quad (6.2)$$

$$\text{RMS 3-D error} = \sqrt{\sigma_E^2 + \sigma_N^2 + \sigma_U^2} = \sqrt{\sigma_x^2 + \sigma_y^2 + \sigma_z^2} = \sigma \cdot \text{PDOP} \quad (6.2)$$



Positioning System - HDOP

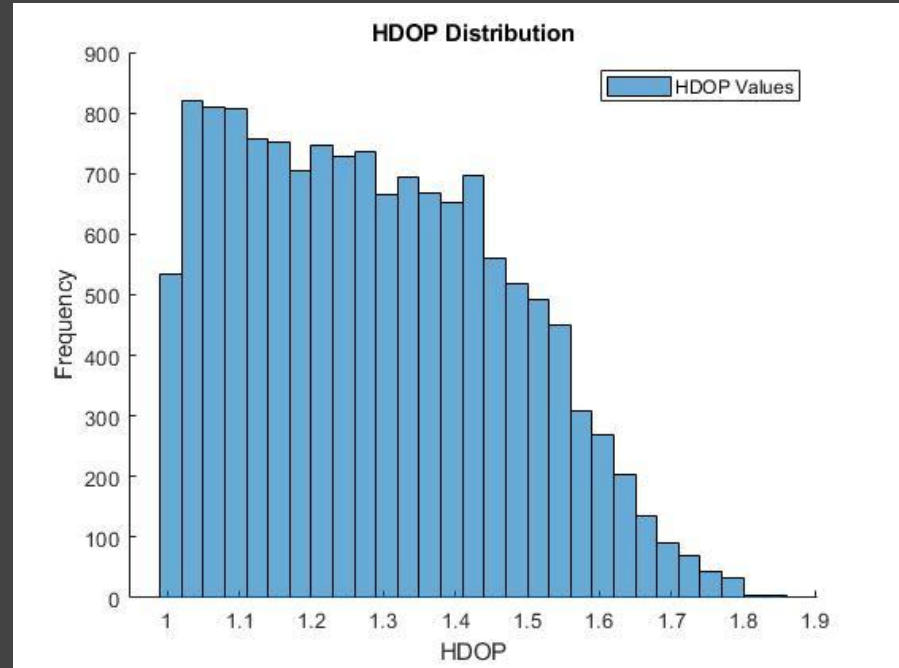
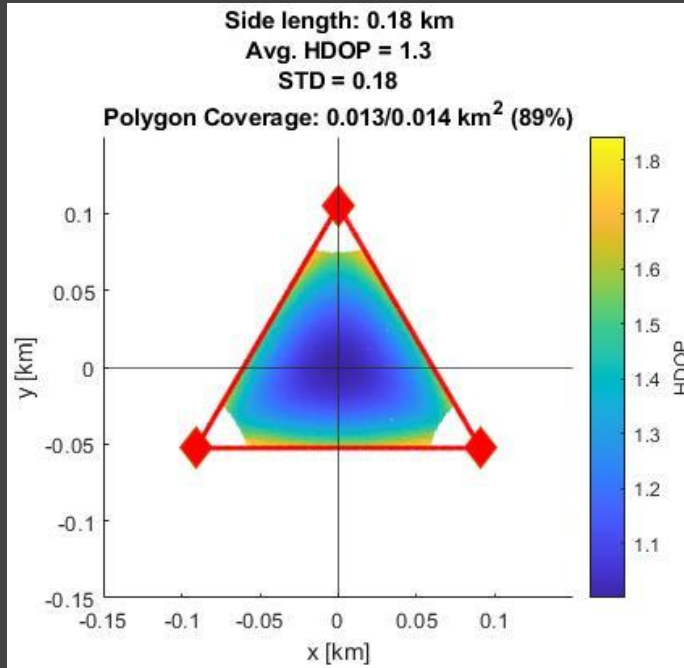
The DOPs provide a simple characterization of the user-satellite geometry. The more favorable the geometry, the lower the DOP. The lower the DOP and σ , the better the quality of the position estimate, in general. If a receiver is limited in the number of satellites it can track simultaneously, the user pays a price with larger rms position error due to higher DOP values.

$$\text{RMS horizontal error} = \sqrt{\sigma_E^2 + \sigma_N^2} = \sigma \cdot \text{HDOP} \quad (6.27a)$$

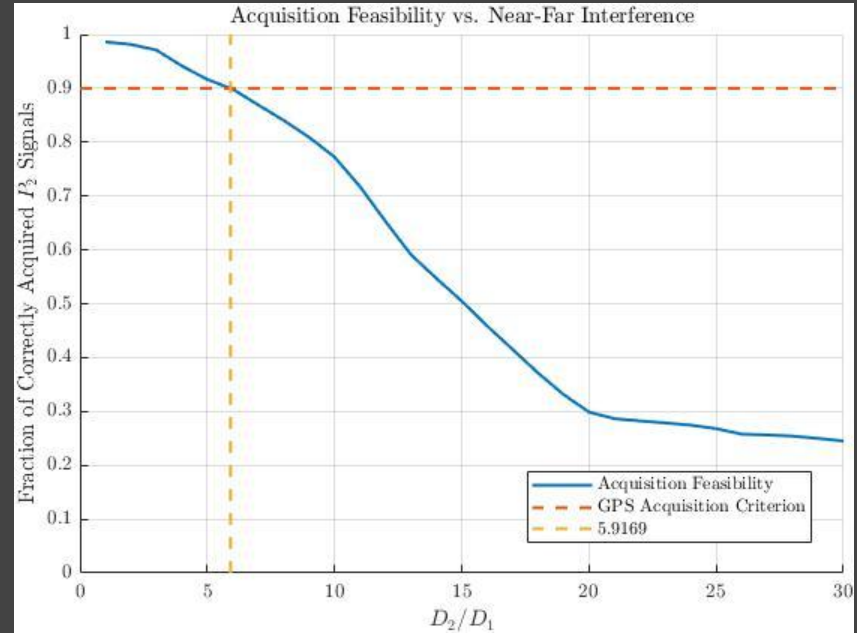
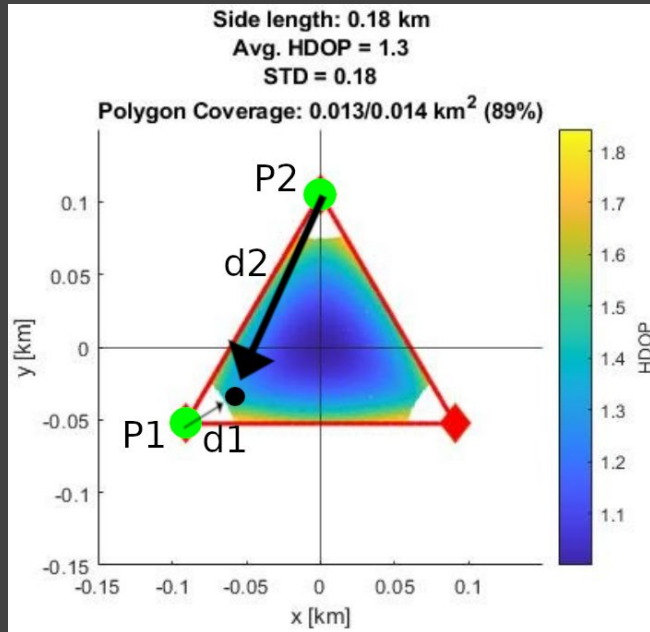
$$\text{RMS vertical error} = \sigma_U = \sigma \cdot \text{VDOP} \quad (6.27b)$$

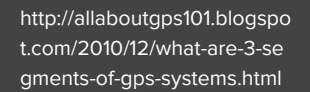
$$\text{RMS 3-D error} = \sqrt{\sigma_E^2 + \sigma_N^2 + \sigma_U^2} = \sqrt{\sigma_x^2 + \sigma_y^2 + \sigma_z^2} = \sigma \cdot \text{PDOP} \quad (6.27c)$$

Pseudolite Geometry

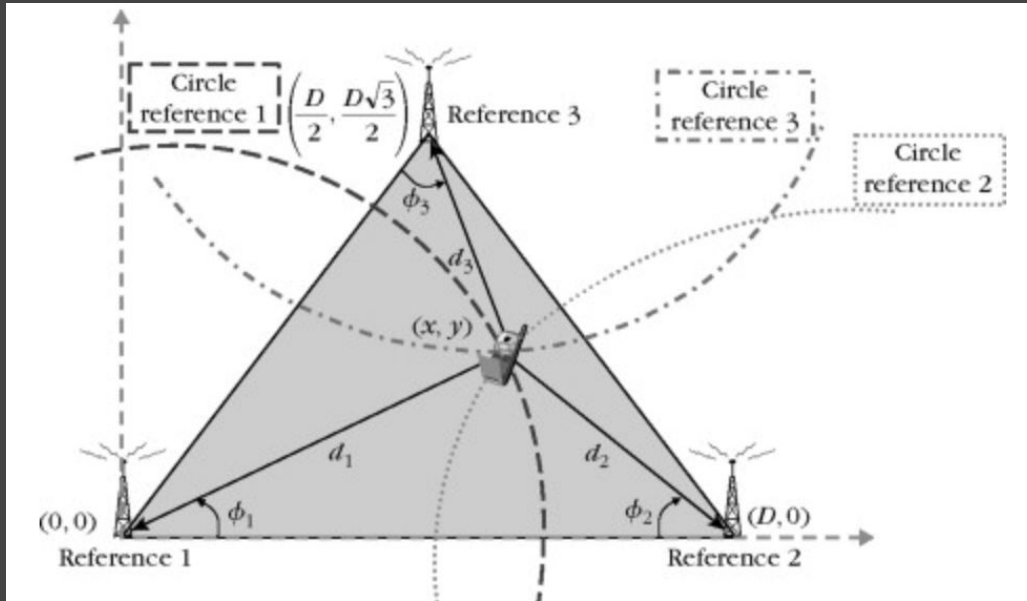


Near-Far Problem





Time of Arrival (TOA)



$$d_1 = c(t_1 - t_0) = \sqrt{x^2 + y^2}$$

$$d_2 = c(t_2 - t_0) = \sqrt{(D-x)^2 + y^2}$$

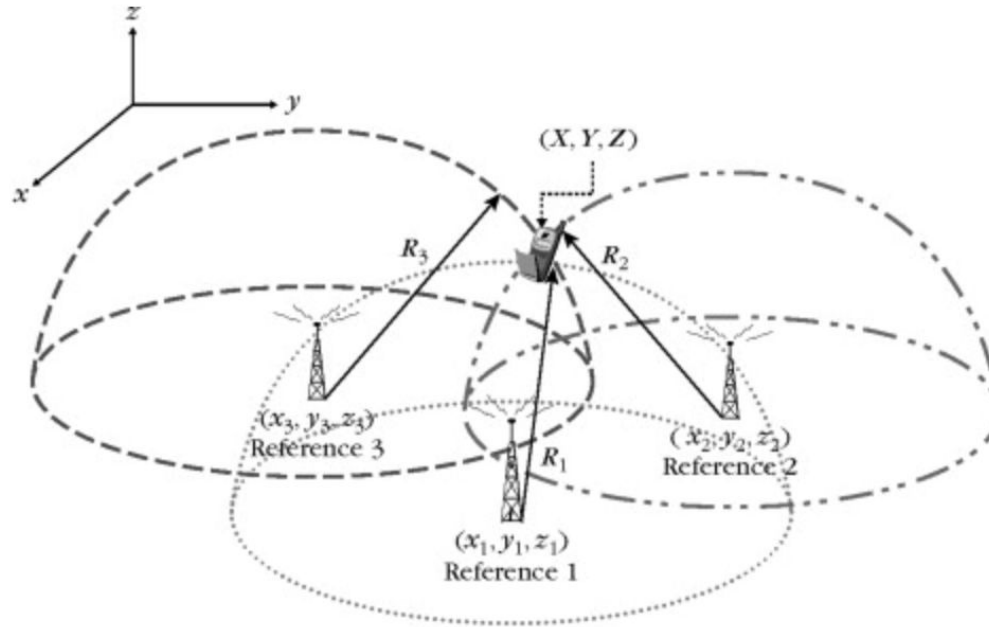
$$d_3 = c(t_3 - t_0) = \frac{1}{2}\sqrt{(D-2x)^2 + (D\sqrt{3}-2y)^2}$$

t_i : TOA at reference i ; c : speed of light

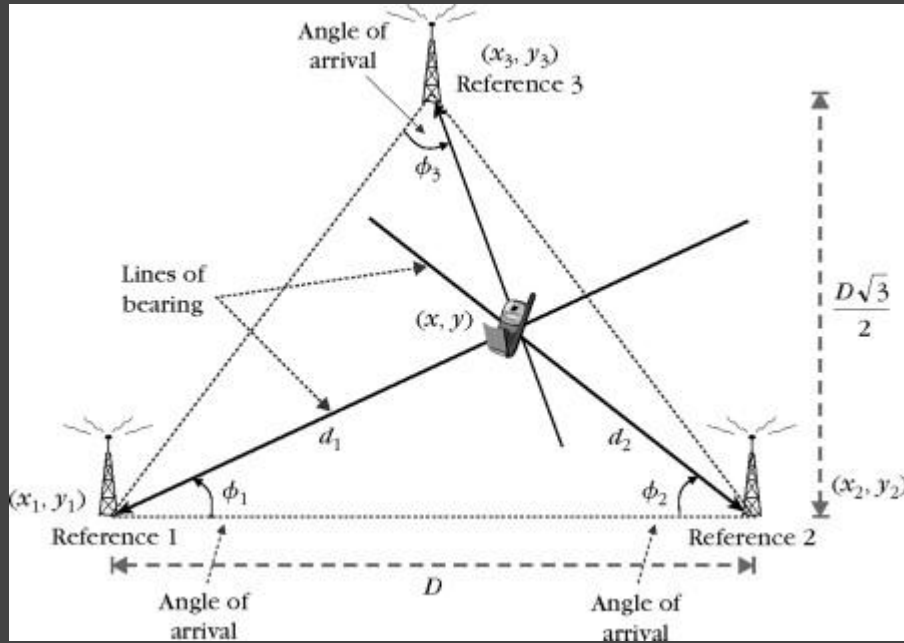


Time of Arrival (TOA)

$$R_i^2 = (X - x_i)^2 + (Y - y_i)^2 + (Z - z_i)^2, i = 1, 2, 3.$$



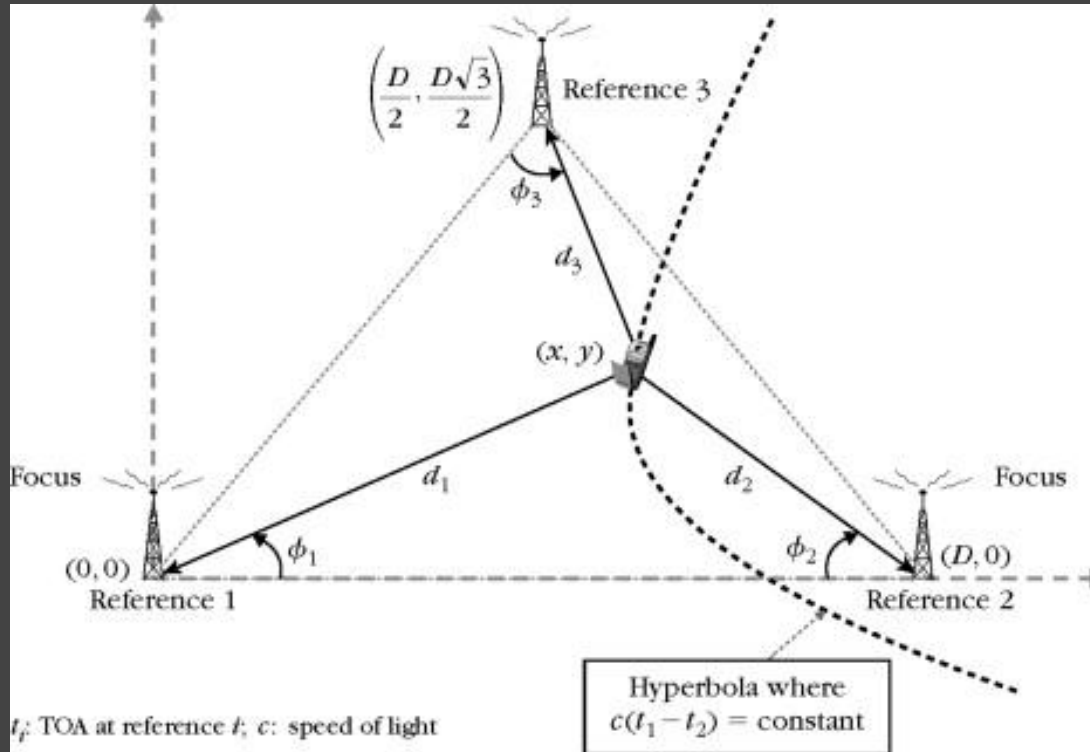
Angle of Arrival (AOA)



$$x = d_i \cos(\phi_i) + x_i,$$

$$y = d_i \sin(\phi_i) + y_i, i = 1, 2, 3.$$

Time Difference of Arrival (TDOA)





Time Difference of Arrival (TDOA)

$$d_{ij} = d_i - d_j = c(t_i - t_o) - c(t_j - t_o) = c(t_i - t_j), i = 1, 2, 3, j = 1, 2, 3, i \neq j.$$

$$d_i = d_{ij} + d_j;$$

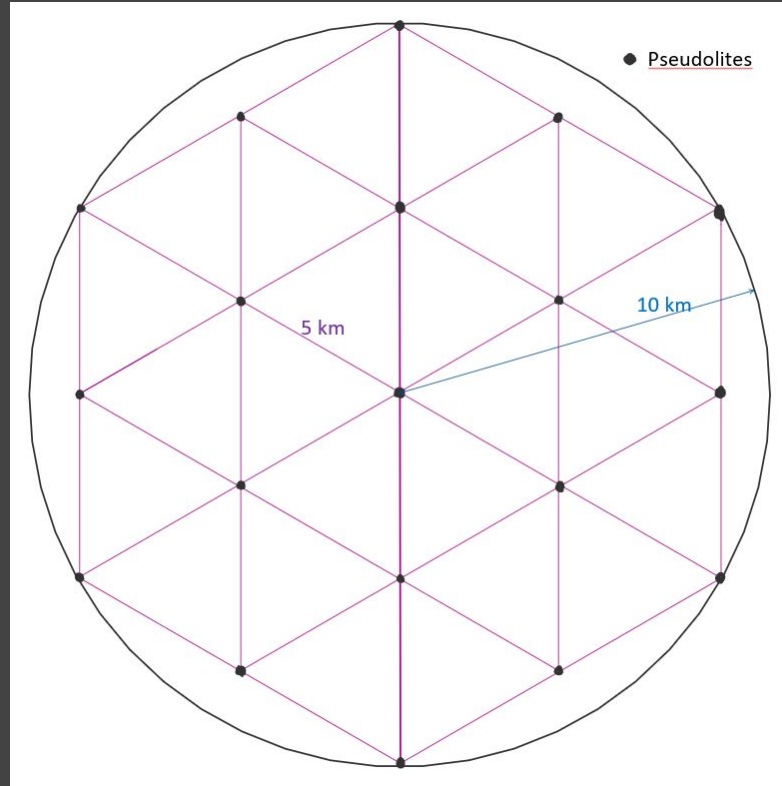
$$\begin{aligned} d_2^2 &= (d_{21} + d_1)^2 \\ &= (x_2 - x)^2 + (y_2 - y)^2 \\ &= x_2^2 - 2x_2x + x^2 + y_2^2 - 2y_2y + y^2 \\ &= x_2^2 - 2x_2x + y_2^2 - 2y_2y + d_1^2, \end{aligned}$$

$$d_1^2 = x^2 + y^2.$$

$$(d_{21}^2 - x_2^2 - y_2^2) + 2d_{21}d_1 = -2x_2x - 2y_2y,$$

$$(d_{31}^2 - x_3^2 - y_3^2) + 2d_{31}d_1 = -2x_3x - 2y_3y.$$

Lunar Pseudolite Geometry



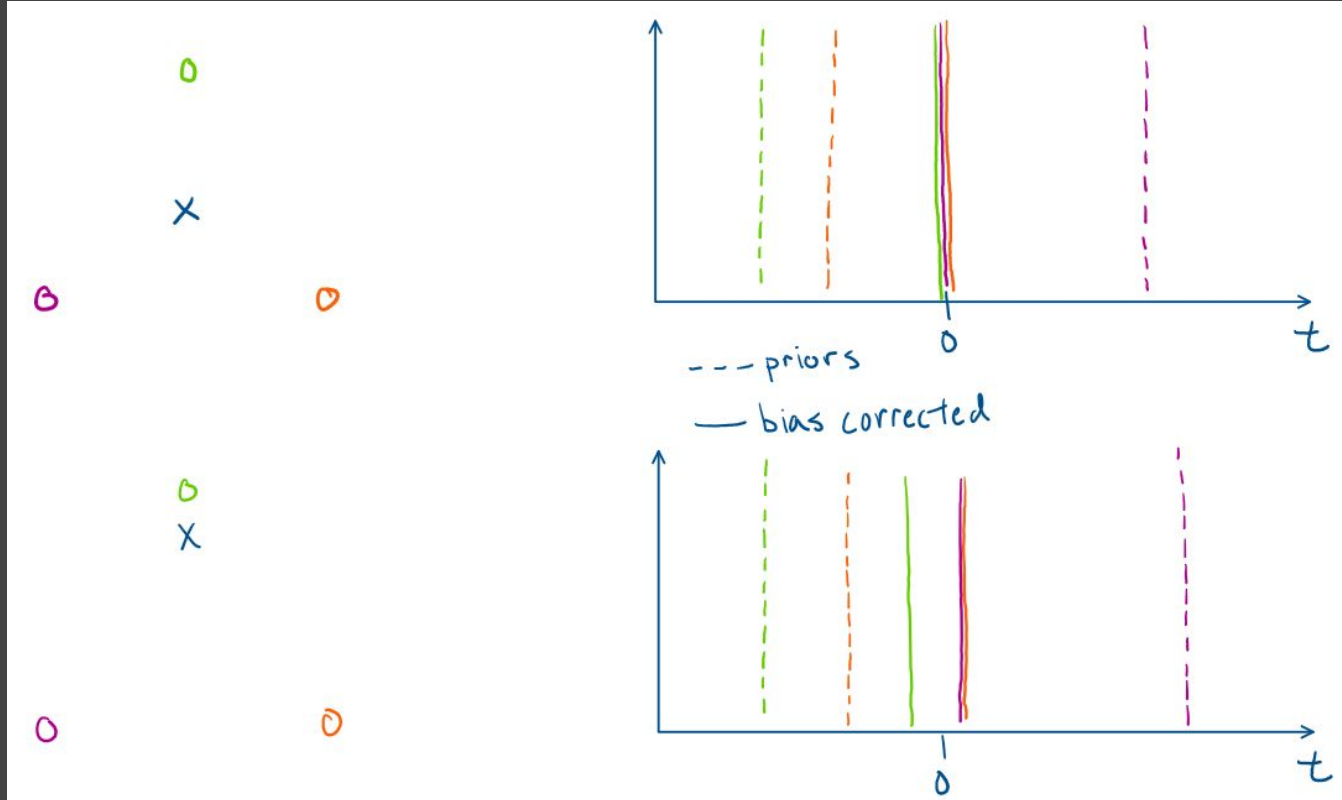


Tentative Trade/Analysis Aspects

- 111

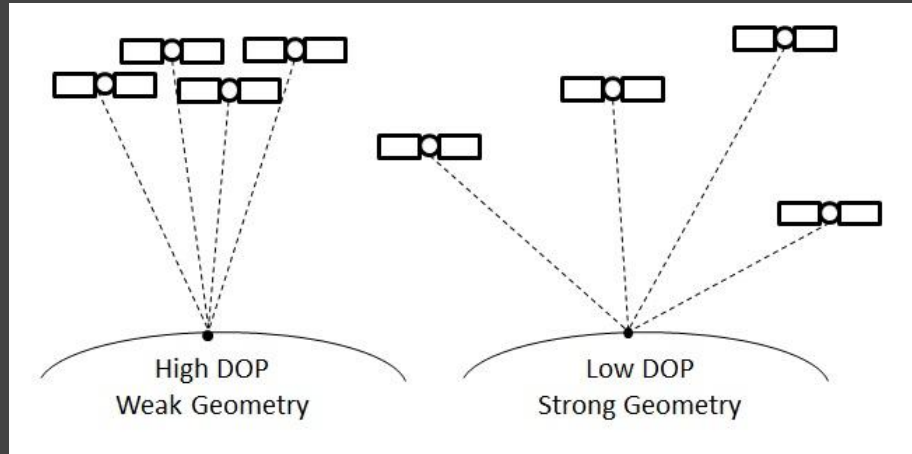


P4LO's Bootleg Positioning System



GNSS: Error sources

- Error sources can be divided into two categories:
 - Ranging error: signal quality, errors in transmitter location, environmental effects, receiver design, etc....
 - Geometry: Geometric Dilution of Precision (GDOP). Measures the quality of the geometric distribution of the satellites visible to the receiver.



<https://www.polyu.edu.hk/proj/gef/index.php/glossary/dilution-of-precision/>

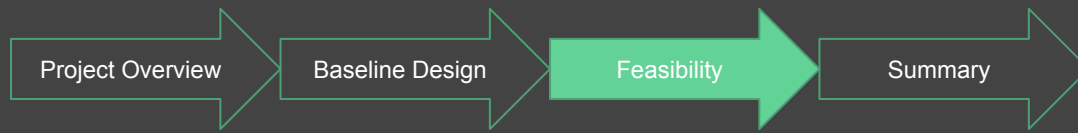


GNSS: Architecture

- 3 segments:
 - Ground control segment
 - Provides satellites with ephemeris data and almanac
 - Provides clock-correction factors and data on atmospheric effects
 - Keeps satellites “in check”
 - Space segment
 - Satellites which send out coded ranging signals, ephemerides, correction parameters and almanac.
 - User segment
 - Receiver (which in the case of this project, can also transmit SMS messages)
 - Acquires and track satellite signals.
 - Computes position solution.
 - For this project, THIS is the segment we’ll be working with. All other segments will be assumed as working.



Comms



Communications - Requirements

FR 2: The prototype will provide two-way SMS-like messaging

- **DR 2.1:** Device must have wireless transmission and reception of data
- **DR 2.2:** Communication data rate must be at least 200 bits/s

FR 3: The prototype will provide path to navigation solution with 10 meter positioning accuracy and 30 nanosecond 1-sigma transfer time

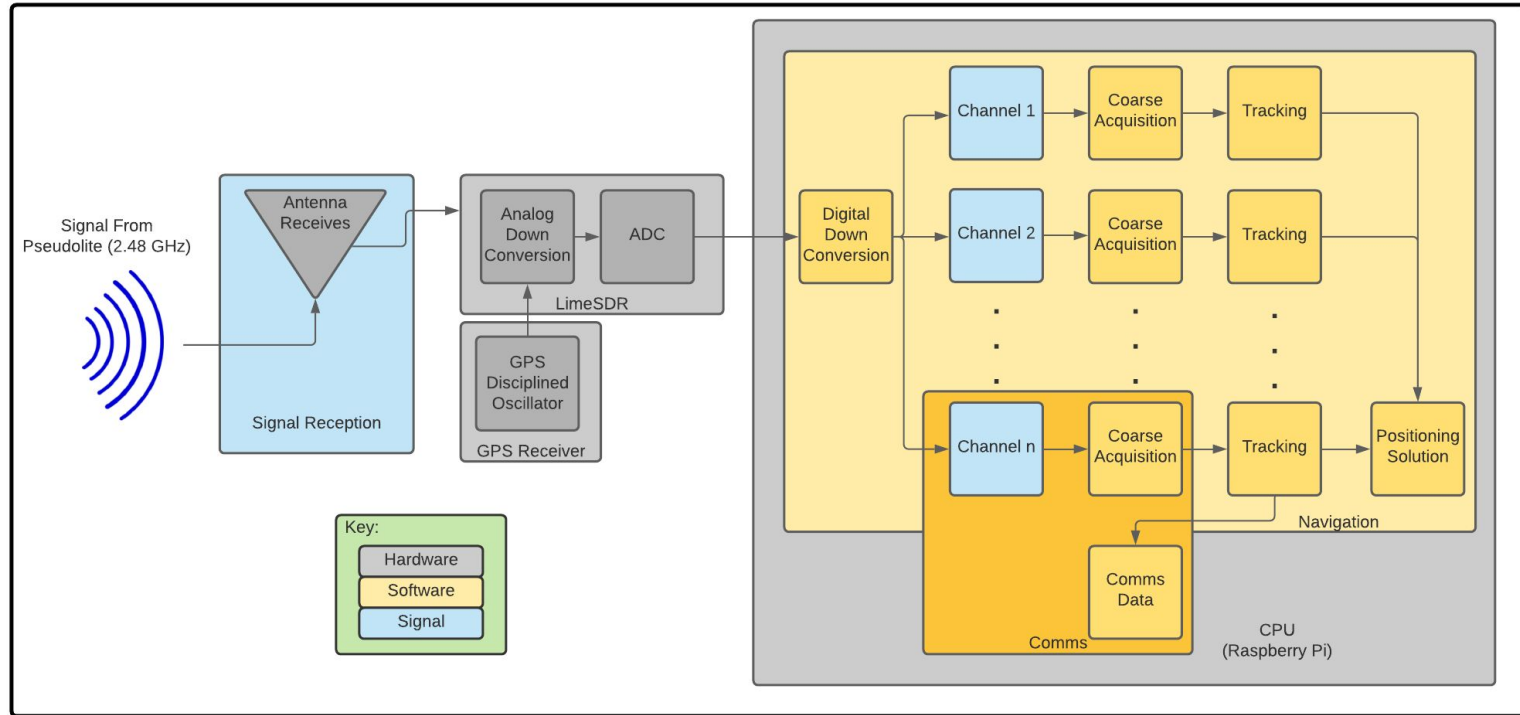
- **DR 3.1:** Receiver Signal to Noise Ratio must be at least 20 dB

FR 4/5: The communication link must operate at 2.4-2.48 GHz

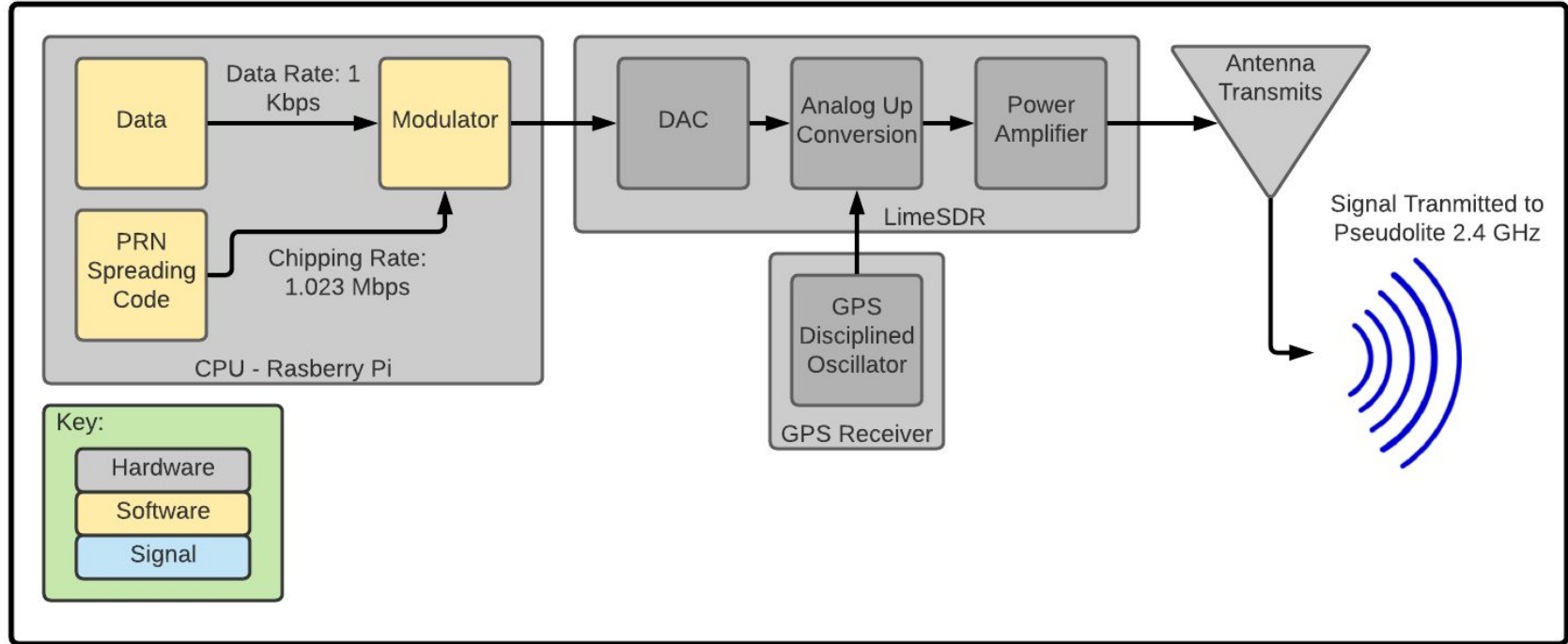
FR 6: The architecture must be extendable to 170 simultaneous users

FR 7: The communication link must have 3 dB channel bandwidth of 1 MHz

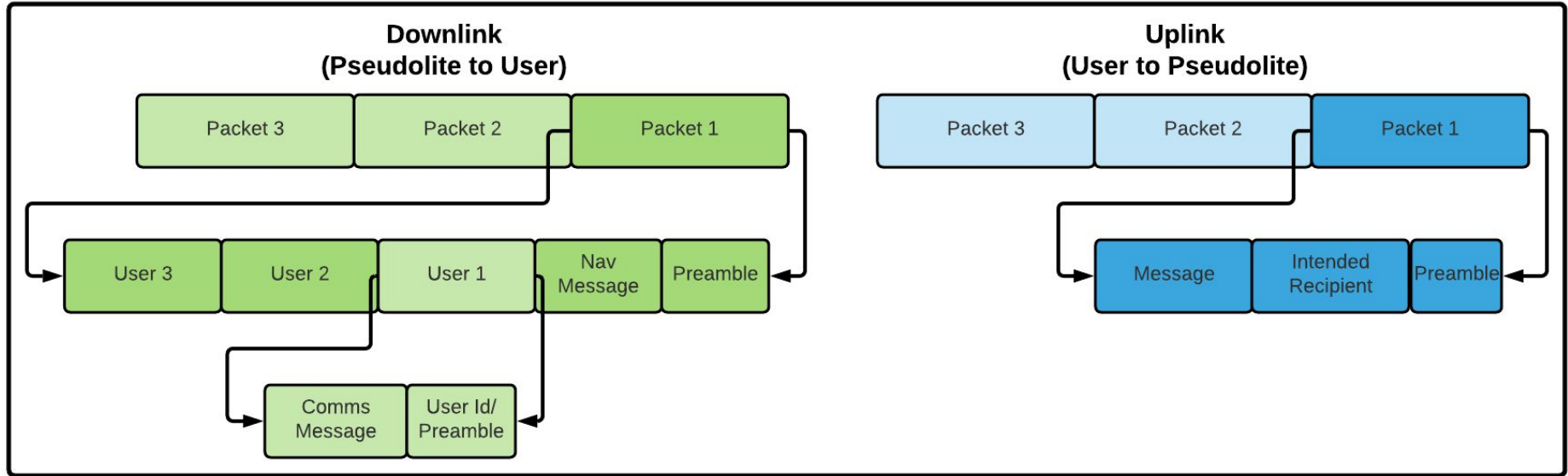
Receive FBD



Transmit FBD



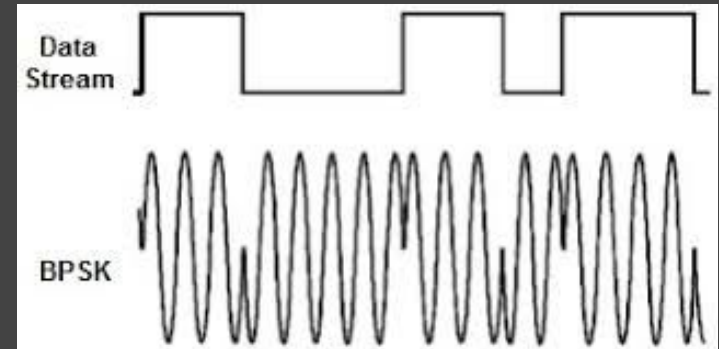
Signal FBD





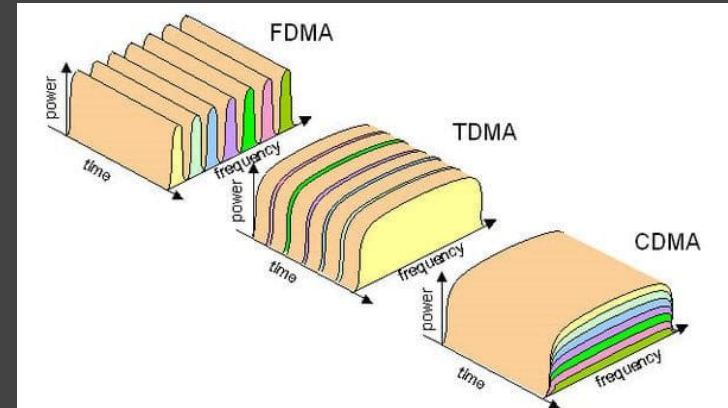
Link Design: Modulation

- Design Choice: Binary Phase Shift Keying (BPSK)
- Reasoning:
 - Low bit error rate (BER)
 - Low complexity
 - Able to use carrier phase to get even more precise signal time of arrival
 - Currently used in GPS implementations



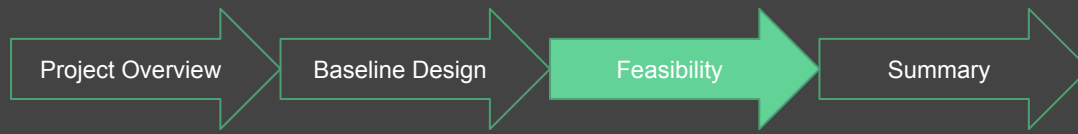
Link Design: Multiplexing

- Design: Code Division Multiple Access
- Reasoning:
 - Able to use phase of spreading code to calculate time of arrival (TOA)
 - Single Frequency
 - Asynchronous CDMA
 - Low cross-interference
- Specification (Design Solution):
 - 1023 bit Gold Code PRN chipping code
 - 1.023 MHz chipping rate
 - 1 ms code period (same as data rate)





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-
- Pulse Error
- Signal + Noise
- True Signal

$$\delta R = \frac{c_0}{2B\sqrt{2SNR}}$$

- FR 7: Link must operate at 1 MHz bandwidth
- Link Budget operates on a 48 dB-Hz minimum C/N_0

DR 3.1, FR7

FEASIBLE



Project Overview

Baseline Design

Feasibility

Summary

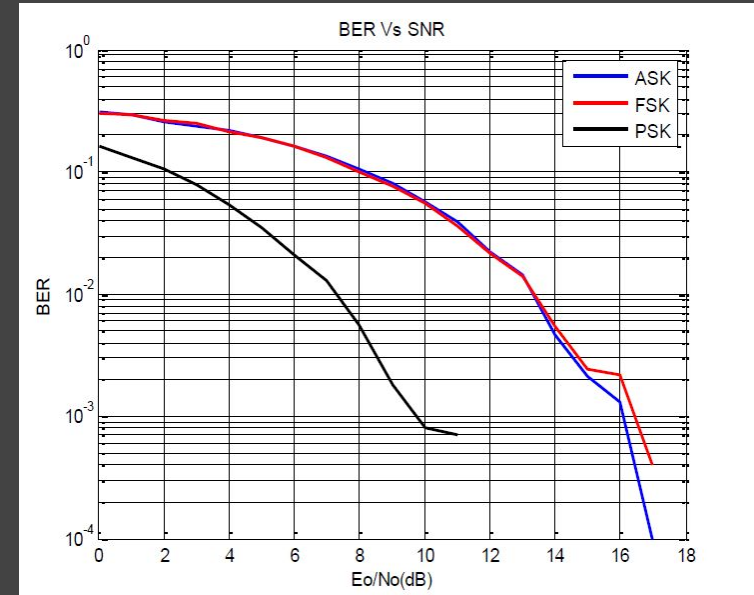


Modulation: Phase Shift Keying

- Modulation Scheme must be compatible with SDR
 - Compatible with LimeSDR
 - Compatible with GNU Radio
- Low Bit Error Rate
- Single Frequency
- Use carrier phase for additional position accuracy
- GPS uses Phase Shift Keying

DR 2.1, 2.2, 3.1, FR4, 5, 7
SDR Compatibility

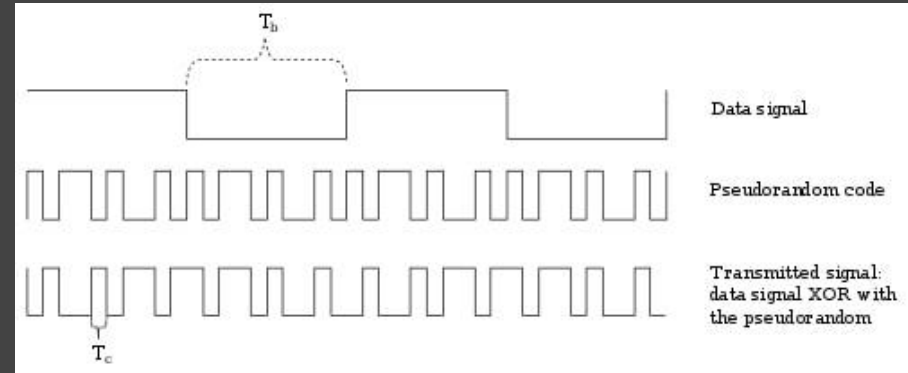
FEASIBLE





Multiplexing: CDMA

- CDMA used in GPS
- CDMA code rate relates to first level of position accuracy
- CDMA can operate on single frequency
- Receiver can receive multiple signals simultaneously
- CDMA with 1023 bit chip code:
 - One way ranging: more end users overall, more precise timing requirements
 - Two way ranging: each user contributes to multiple access interference



One Way Ranging: FR 6

MORE ANALYSIS

Two Way Ranging: FR 6

MORE ANALYSIS

DR 2.1, 3.1, FR 4, 5, 7

FEASIBLE



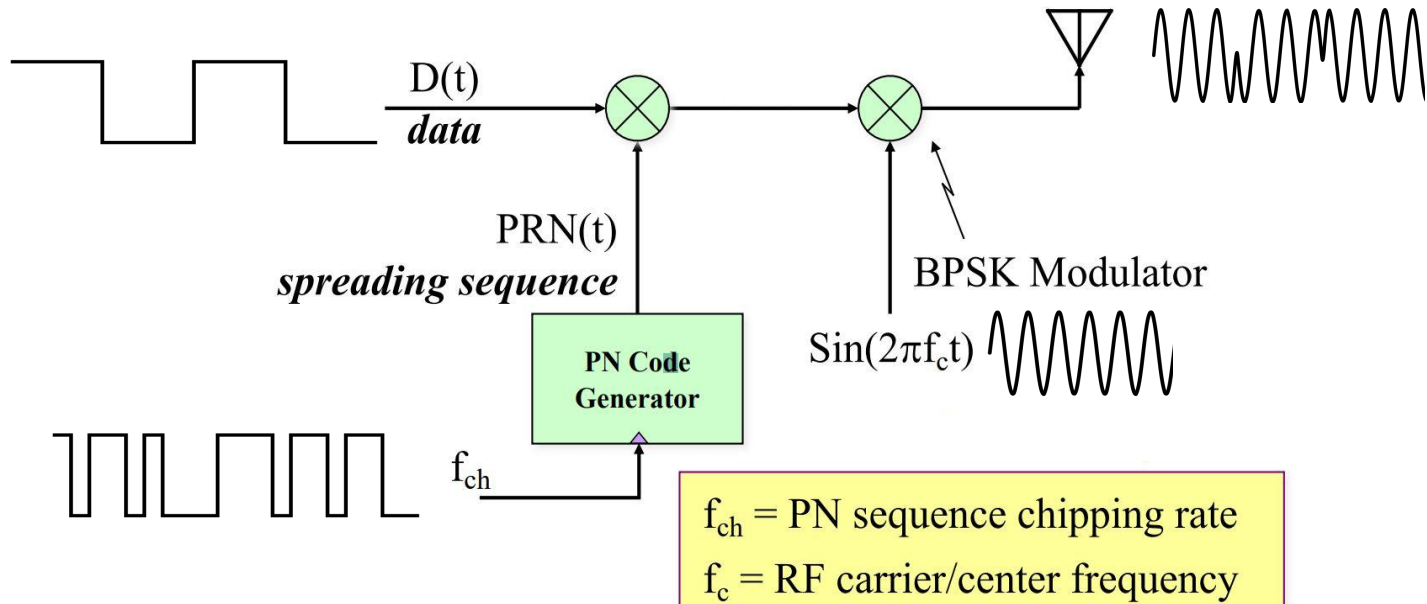
Overview

Schedule

Manufacturing

Budget

Signal Generation



Courtesy Logan Scott



Overview

Schedule

Manufacturing

Budget

Downlink Packet Breakdown

- Preamble
 - Unique 24 bit code sequence to allow for signal correlation
- Navigation Message
 - Element of set bit length providing timing for positioning algorithm
- User ID
 - 8 bit element containing a unique user identification sequence for each user to identify intended recipient
 - Designed to accommodate 30 user requirement
- Communications Message
 - 5000 bit text communication element broken into 8 bit characters
 - Message will be appended with ' ' (space) buffers to maintain constant size



Overview

Schedule

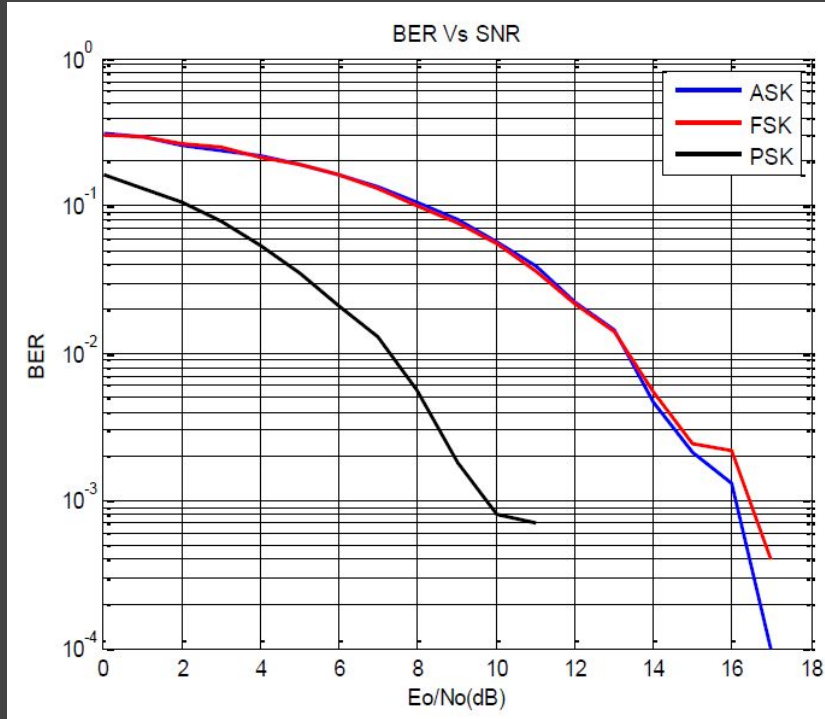
Manufacturing

Budget

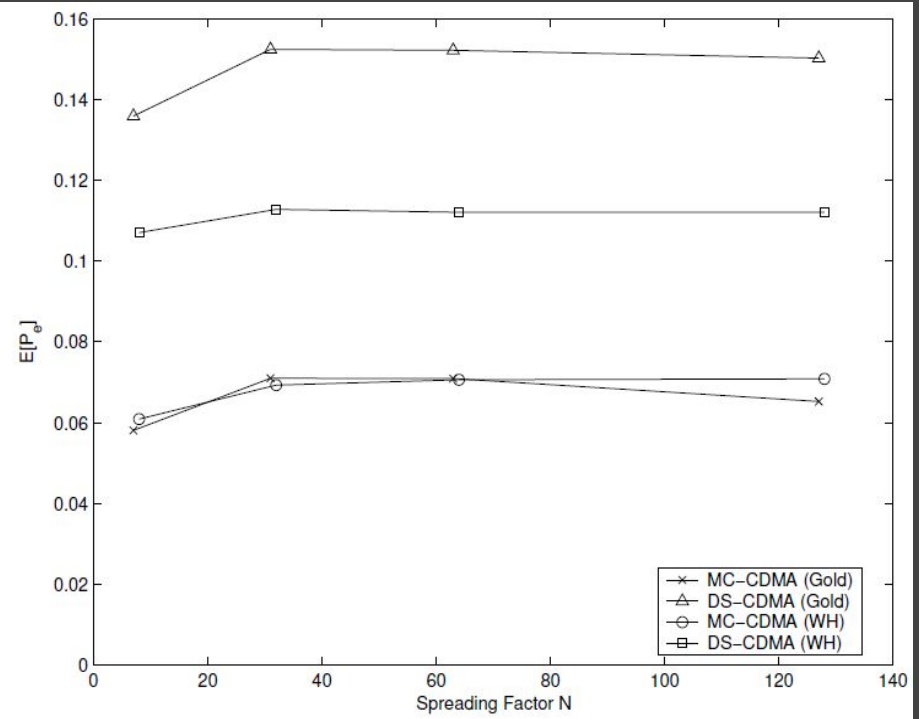
Uplink Breakdown

- Preamble
 - Unique 24 bit code sequence to allow for signal correlation
- User Identification
 - 8 bit element containing a unique user identification sequence for each user to identify transmitting user
 - Designed to accommodate 30 user requirement
- Communications Message
 - 5000 bit text communication element broken into 8 bit characters
 - Message will be appended with ' ' (space) buffers to maintain constant size

Bit Error Rates of Modulation Schemes



Bharati et. al.



Carey et. al.



Pseudolite Link Budget 1

PARAMETER	UPLINK (Receiver to Satellite)	DOWNLINK (Satellite to Receiver)	UNITS	Symbol	Reference
Speed of Light	3.0E+08	3.0E+08	m/s	$C = \lambda * f$	constant
Frequency	2.4	2.4	GHz	f	Input: system choice, X-band mil.com.sat
Wavelength	0.125	0.125	m	λ	
Range	10	10	km	R	Input: Geostationary Satellite [km]
Boltzman's Constant	1.380E-23	1.380E-23	W/(Hz-K)	k	constant
Data Parameters					
	Uplink	Downlink	Units	Symbol	Reference
Bit Error Rate / Probability of Bit Error	10-8	10-8	[-]	BER	Input: design requirement
Data Coding Scheme	QPSK	QPSK			Input: chosen modulation (SMAD Tab.13-10)
Required Bit Energy to Noise Ratio	12	12.0	dB	E_b/N_0	Lecture Pt 2, Slide 15
Data Rate	1000	1,000	bps (Hz)	R	Input: based on mission / objective
Carrier to Noise Ratio Density	42.00	42.00	dB-Hz	P_r/N_0	Lecture Pt 2, Slide 15
Required Design Margin	6.00	6.00	dB		Input: design rule (Hoffmann chap. 9.4.4)
Minimum Pr/No	48.00	48.00	dB-Hz		
Noise (applies to receiving elements)					
	Uplink	Downlink	Units	Symbol	Reference
Receiving Antenna Noise Temperature	400	400	K	T_a	Lecture Pt 2, Slide 13
Receiver Cable Loss	0.9	0.9	dB	L_c	SMAD Table 13-10
Receiver Noise Figure (based on receiver)	3.0	3.0	dB	NF	SMAD Table 13-10
Receiver Noise Factor	2.0	2.0	[-]	F	Lecture Pt 2, Slide 10
Receiver Noise Temperature	288.6	288.6	K	T_r	Lecture Pt 2, Slide 10
Reference Temperature	290	290	K	T_o	SMAD Eqn13-24
Receiver System Noise Temperature	580.32	580.32	K	T_s	Lecture Pt 2, Slide 13
Receiver System Noise Power	-200.96	-200.96	dBW-Hz	No	Lecture Pt 2, Slide 9
Receiver Parameters:					
	Uplink	Downlink	Units	Symbol	Reference
Receive Antenna Diameter	NA	NA	m	D	Input: given geometry from spacecraft
Receive Antenna Area	NA	NA	m ²	A	Geometry
Receive Antenna Efficiency	NA	NA	[-]	h	Input: typical value
Receive Antenna Effective Area	NA	NA	m ²	A_e	efficiency * area
Receive Antenna Gain	2.00	2.00	Power Ratio	G_r	Lecture Pt 1, Slide 14
Receive Antenna Beamwidth	180.0	180.0	Degrees	θ_r	Lecture Pt 1, Slide 16
Receive Antenna Pointing Accuracy	0.0	0	degrees	ϵ_r	Input: pointing error e for chosen system
Receive Antenna Pointing Loss	0.00	0.00	dB	L_{pr}	SMAD 13-21
Receiver Cable Loss (see noise)	-0.5	-0.5	dB	L_c	Input: typical value
Receiver Figure of Merit	0.00	0.00	dB/K	FOM	
Propagation Parameters:					
	Uplink	Downlink	Units	Symbol	Reference
Space Loss	-120.05	-120.05	dB	L_s	Lecture Pt 2, Slide 3
Atmospheric Attenuation (clear air)	0	0	dB	L_a	Lecture Pt 2, Slide 7
Polarization Loss	0	0	dB	L_p	Input: typical value



Satellite Link Budget 1

PARAMETER	UPLINK (Receiver to Satellite)	DOWNLINK (Satellite to Receiver)	UNITS	Symbol	Reference
Speed of Light	3.0E+08	3.0E+08	m/s	$C = \lambda * f$	constant
Frequency	2.5	2.5	GHz	f	Input: system choice, X-band mil.com.sat
Wavelength	0.120	0.120	m	λ	
Range	10000	10000	km	R	Input: Geostationary Satellite [km]
Boltzman's Constant	1.380E-23	1.380E-23	W/(Hz-K)	k	constant
Data Parameters	Uplink	Downlink	Units	Symbol	Reference
Bit Error Rate / Probability of Bit Error	10-6	10-6	[-]	BER	Input: design requirement
Data Coding Scheme	QPSK	QPSK			Input: chosen modulation (SMAD Tab.13-10)
Required Bit Energy to Noise Ratio	11	12.0	dB	E_b/N_0	Lecture Pt 2, Slide 15
Data Rate	1000	1,000	bps (Hz)	R	Input: based on mission / objective
Carrier to Noise Ratio Density	41.00	42.00	dB-Hz	P_r/N_0	Lecture Pt 2, Slide 15
Required Design Margin	3.00	6.00	dB		Input: design rule (Hoffmann chap. 9.4.4)
Minimum P_r/N_0	44.00	48.00	dB-Hz		
Noise (applies to receiving elements)	Uplink	Downlink	Units	Symbol	Reference
Receiving Antenna Noise Temperature	400	25	K	T_a	Lecture Pt 2, Slide 13
Receiver Cable Loss	0.9	0.9	dB	L_c	SMAD Table 13-10
Receiver Noise Figure (based on receiver)	3.0	1.0	dB	NF	SMAD Table 13-10
Receiver Noise Factor	2.0	1.3	[-]	F	Lecture Pt 2, Slide 10
Receiver Noise Temperature	288.6	75.1	K	T_r	Lecture Pt 2, Slide 10
Reference Temperature	290	290	K	T_o	SMAD Eqn13-24
Receiver System Noise Temperature	580.32	31.75	K	T_s	Lecture Pt 2, Slide 13
Receiver System Noise Power	-200.96	-213.58	dBW-Hz	N_0	Lecture Pt 2, Slide 9
Receiver Parameters:	Uplink	Downlink	Units	Symbol	Reference
Receive Antenna Diameter	NA	NA	m	D	Input: given geometry from spacecraft
Receive Antenna Area	NA	NA	m ²	A	Geometry
Receive Antenna Efficiency	NA	NA	[-]	h	Input: typical value
Receive Antenna Effective Area	NA	0.002	m ²	A_e	efficiency * area
Receive Antenna Gain	20.00	2.38	power Ratio	G_r	Lecture Pt 1, Slide 14
Receive Antenna Beamwidth	20.0	180.0	Degrees	θ_r	Lecture Pt 1, Slide 16
Receive Antenna Pointing Accuracy	0.2	0	degrees	ϵ_r	Input: pointing error ϵ for chosen system
Receive Antenna Pointing Loss	0.00	0.00	dB	L_{pr}	SMAD 13-21
Receiver Cable Loss (see noise)	-0.5	-0.5	dB	L_c	Input: typical value
Receiver Figure of Merit	0.03	0.08	dB/K	FOM	
Propagation Parameters:	Uplink	Downlink	Units	Symbol	Reference
Space Loss	-180.37	-180.37	dB	L_s	Lecture Pt 2, Slide 3
Atmospheric Attenuation (clear air)	0	0	dB	L_a	Lecture Pt 2, Slide 7
Polarization Loss	0	0	dB	L_p	Input: typical value



Pseudolite Link Budget 2

Transmitter Parameters:	Uplink	Downlink	Units	Symbol	Reference
Transmit Antenna Diameter	NA	NA	m	D	Switch Receive
Transmit Antenna Area	NA	NA	m ²	A	Switch Receive
Transmit Antenna Efficiency	NA	NA	[-]	h	Switch Receive
Transmit Antenna Effective Area	NA	NA	m ²	Ae	Switch Receive
Transmit Antenna Gain	2.00	2.00	dBi	Gt	Switch Receive
Transmit Antenna Beamwidth	180.00	180.00	degrees	qt	Switch Receive
Transmit Antenna Pointing Accuracy	0.00	0.15	degrees	et	Switch Receive
Transmit Antenna Pointing Loss	0.00	0.00	dB	Lpt	Switch Receive
Transmit Line Loss	-0.5	-0.5	dB	Lt	Input: based on chosen cable/geometry
Transmit Power	-27.0	-27.0	dBW	Pt	10*LOG10(Transmit power)
Transmit Power, Linear	0.002	0.002	W		Input: chosen transmitter
Effective Isotropic Radiated Power	-25.49	-25.49	dBW	EIRP	Sum of Power, Gain, and Losses in dB
Link Budget:	Uplink	Downlink	Units	Symbol	Reference
Effective Isotropic Radiated Power	-25.49	-25.49	dBW	EIRP	From Above
Propagation Losses	-120.05	-120.05	dB	L	Sum of Losses
Receive System Gain	1.50	1.50	dB	Gr	Sum of antenna gain and system losses
Received Power	-144.04	-144.04	dBW	Pr	Sum of Power sent out minus losses
System Noise Power	-200.96	-200.96	dBW-Hz	No	
Carrier to Noise Ratio Density	56.93	56.93	dB-Hz	Pr/No	
Minimum Pr/No	48.00	48.00	dB-Hz		
Link Margin	8.93	8.93	dB		



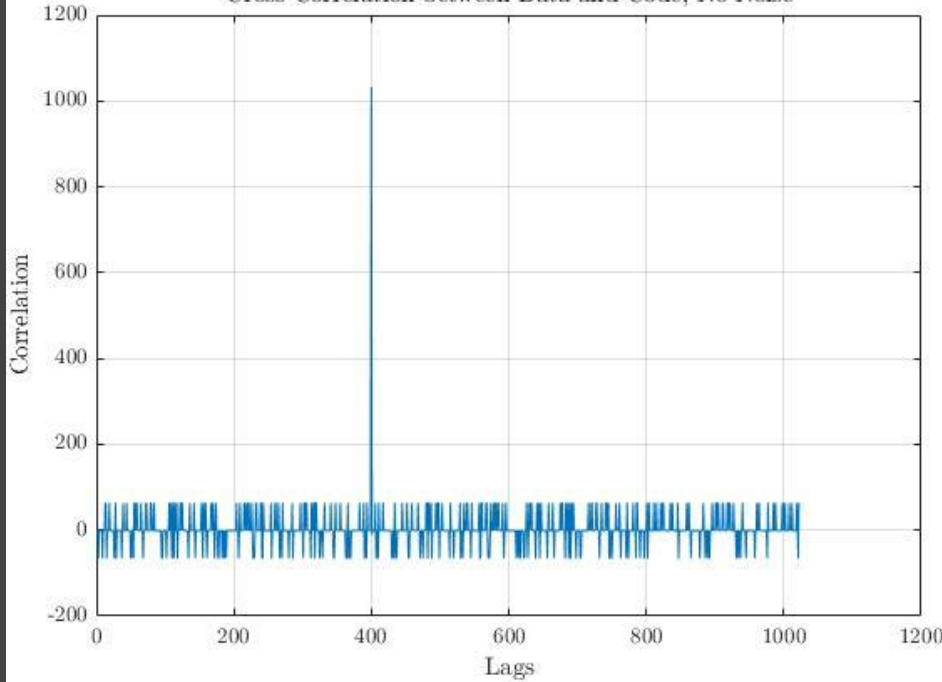
Satellite Link Budget 2

Transmitter Parameters:	Uplink	Downlink	Units	Symbol	Reference
Transmit Antenna Diameter	NA	NA	m	D	Switch Receive
Transmit Antenna Area	NA	NA	m ²	A	Switch Receive
Transmit Antenna Efficiency	NA	NA	[-]	h	Switch Receive
Transmit Antenna Effective Area	0.002	NA	m ²	Ae	Switch Receive
Transmit Antenna Gain	2.38	20.00	dBi	Gt	Switch Receive
Transmit Antenna Beamwidth	180.00	20.00	degrees	qt	Switch Receive
Transmit Antenna Pointing Accuracy	0.00	0.15	degrees	et	Switch Receive
Transmit Antenna Pointing Loss	0.00	0.00	dB	Lpt	Switch Receive
Transmit Line Loss	-0.5	-0.5	dB	Lt	Input: based on chosen cable/geometry
Transmit Power	3.0	14.1	dBW	Pt	10*LOG10(Transmit power)
Transmit Power, Linear	2	25.6	W		Input: chosen transmitter
Effective Isotropic Radiated Power	4.89	33.58	dBW	EIRP	Sum of Power, Gain, and Losses in dB
Link Budget:	Uplink	Downlink	Units	Symbol	Reference
Effective Isotropic Radiated Power	4.89	33.58	dBW	EIRP	From Above
Propagation Losses	-180.37	-180.37	dB	L	Sum of Losses
Receive System Gain	19.50	1.88	dB	Gr	Sum of antenna gain and system losses
Received Power	-155.97	-144.90	dBW	Pr	Sum of Power sent out minus losses
System Noise Power	-200.96	-213.58	dBW-Hz	No	
Carrier to Noise Ratio Density	44.99	68.68	dB-Hz	Pr/No	
Minimum Pr/No	44.00	48.00	dB-Hz		
Link Margin	0.99	20.68	dB		

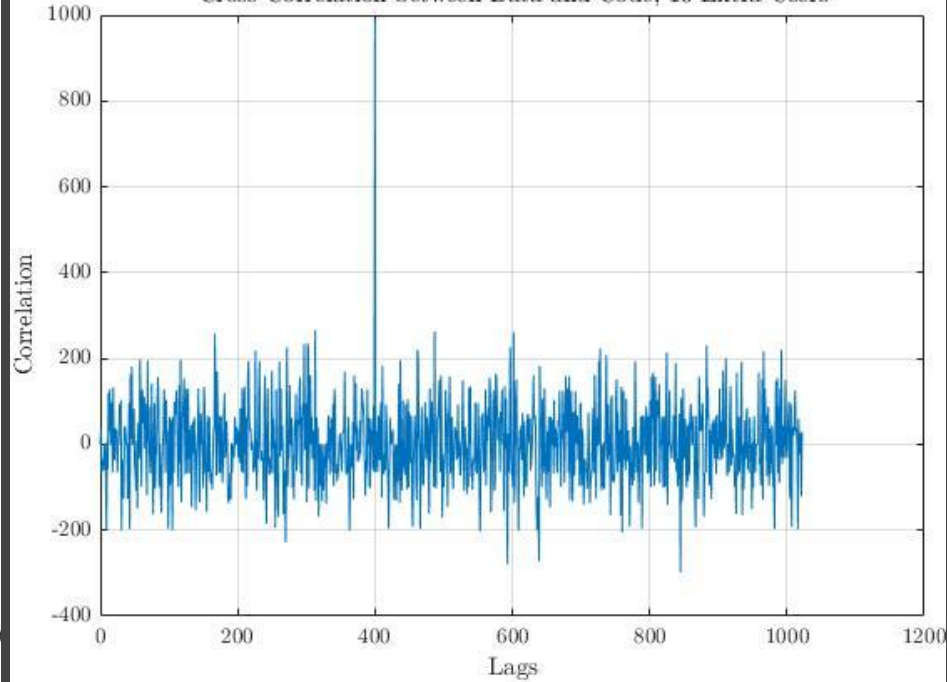


CDMA Signal Acquisition

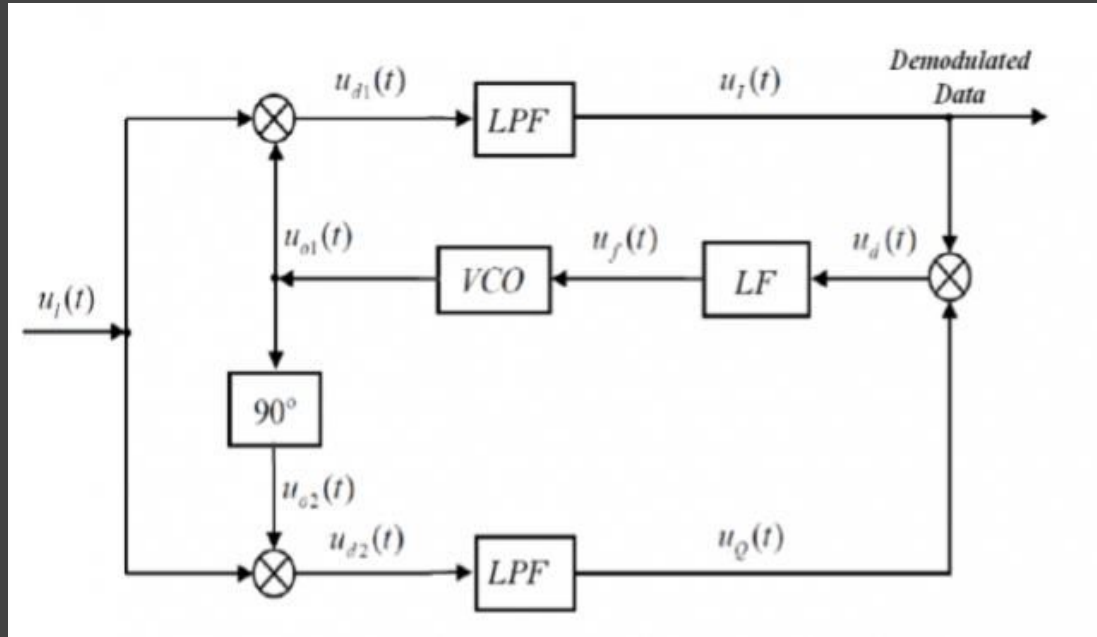
Cross Correlation between Data and Code, No Noise



Cross Correlation between Data and Code, 10 Extra Users



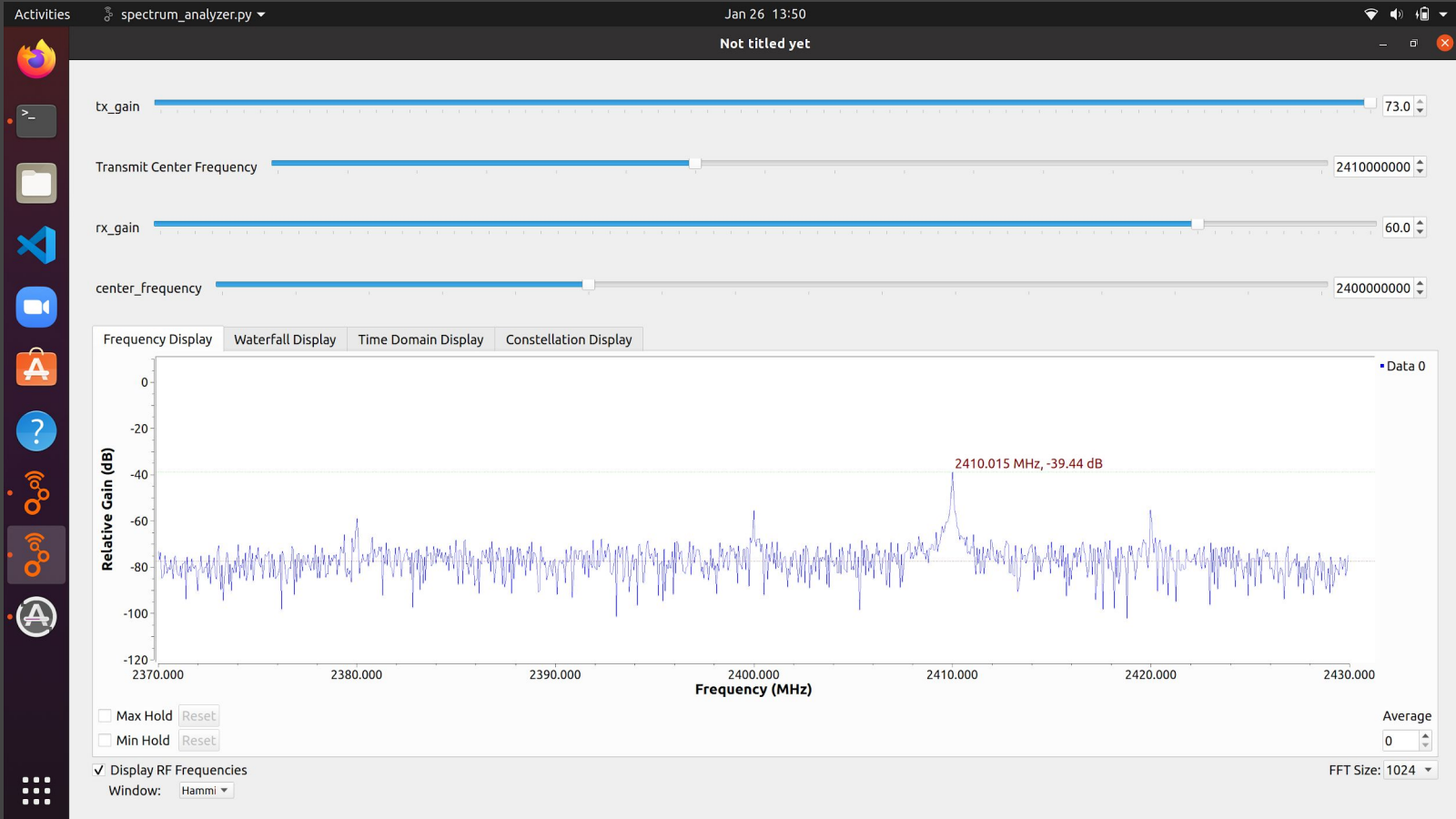
Simple Tracking Loop (Costas Loop)







LimeSDR Hardware/Software Validation





Overview

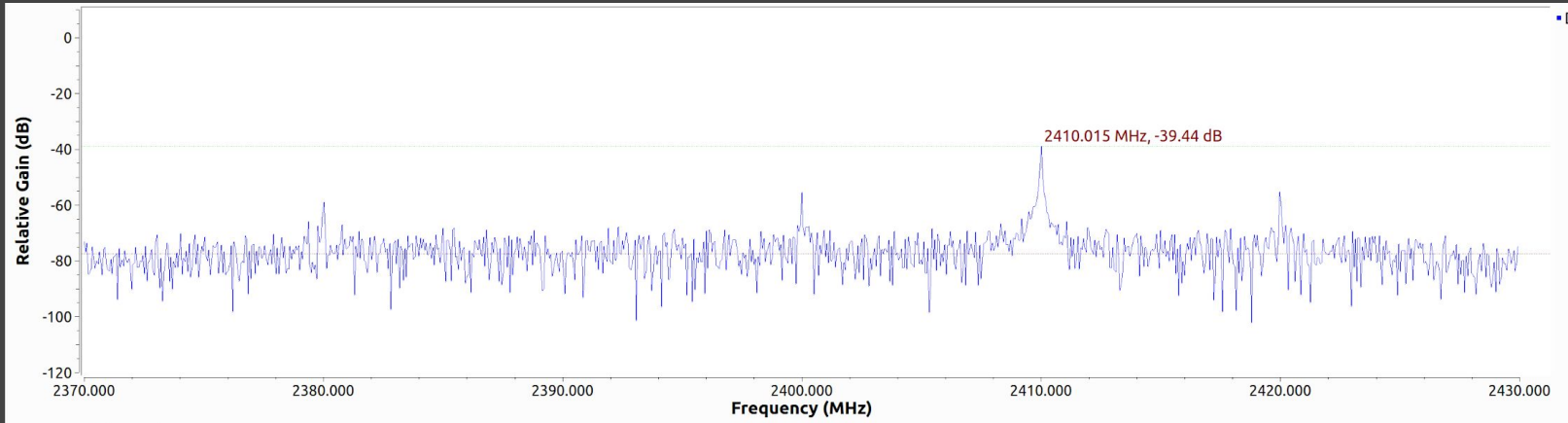
Schedule

Manufacturing

Budget

Signal Interference (Simple Transceiver)

- Preliminary interference and verification test resulted in positive results.
- Single tone SNR:
 - At 1 foot separation 40dB
 - At 590 feet (180 Meters) -15dB (Extrapolation)





Overview

Schedule

Manufacturing

Budget

Downlink Packet Breakdown

- Preamble
 - Unique 24 bit code sequence to allow for signal correlation
- Navigation Message
 - Element of set bit length providing timing for positioning algorithm
- User ID
 - 8 bit element containing a unique user identification sequence for each user to identify intended recipient
 - Designed to accommodate 30 user requirement
- Communications Message
 - 5000 bit text communication element broken into 8 bit characters
 - Message will be appended with ' ' (space) buffers to maintain constant size



Overview

Schedule

Manufacturing

Budget

Uplink Breakdown

- Preamble
 - Unique 24 bit code sequence to allow for signal correlation
- User Identification
 - 8 bit element containing a unique user identification sequence for each user to identify transmitting user
 - Designed to accommodate 30 user requirement
- Communications Message
 - 5000 bit text communication element broken into 8 bit characters
 - Message will be appended with ' ' (space) buffers to maintain constant size



Overview

Solution

CPE

DR

Risk & M

V&V

Planning

Gr-LimeSDR Installation Process

The goal is to create a universal **installation script**, which a person could run on a Linux terminal and have access to a LimeSDR Development Environment.

LimeSuiteGUI





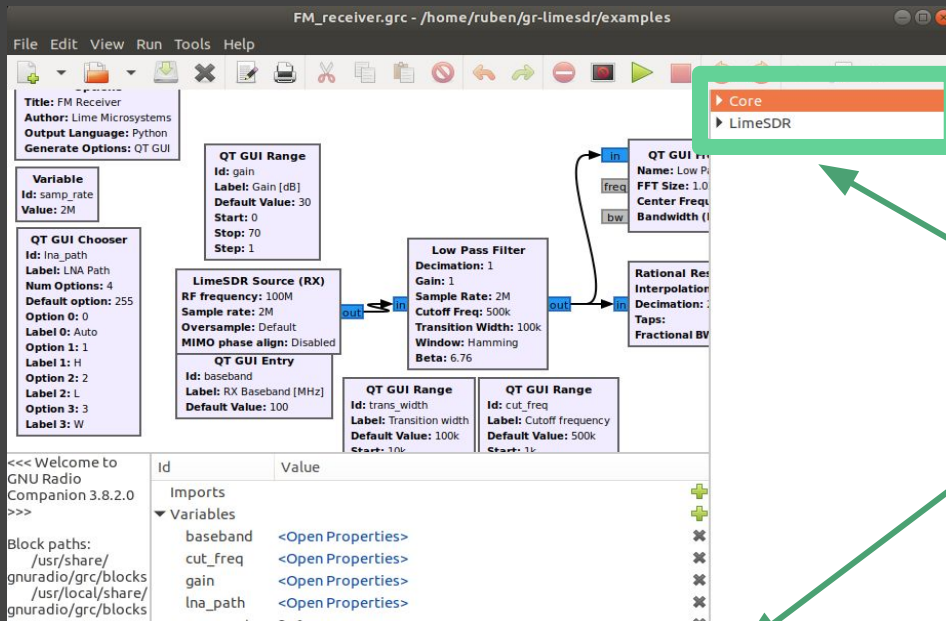
Gr-LimeSDR Installation Process

Integrating gr-limesdr plugin package.

Example script:

```
[3] - Gr-limeSDR Plugin Installation: (Source Code Installation)
[3.1] - Installing Dependencies:
-> sudo apt-get install libboost-all-dev swig
Just in case:
-> sudo apt-get install gnuradio-dev
[3.2] - Building gr-limesdr from source: (Only for GNU Radio 3.8!)
-> git clone https://github.com/myriadrf/gr-limesdr
-> cd gr-limesdr
-> git checkout gr-3.8
-> mkdir build
-> cd build
-> cmake ..
-> make
-> sudo make install
-> sudo ldconfig
```

Gr-LimeSDR Development Environment



- We have a head start in the environment setup!
- We can also verify that the GNU Radio communicates with the Gr-LimeSDR plugin
- We can already verify the LimeSDR communicates with the given software via the LimeSuiteGUI

```

$ LimeUtil --find
* [LimeSDR-USB, media=USB 3.0, module=FX3, addr=1d50:6108, serial=0009072C0289300F]
    
```



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Overview

Schedule

Manufacturing

Budget

Integration: LimeSDR + GNURadio

- gr-limesdr plugin for GNURadio
 - Provides LimeSDR Transmit and Receive blocks
 - Can specify Rx/Tx channels for multiple antennas
- Automatically finds the LimeSDR
- Configurable via .ini file
- These blocks feed samples into the well documented GNURadio sample stream

LimeSDR Sink (TX)

RF frequency: 100M

Sample rate: 2M

Oversample: Default

in

LimeSDR Source (RX)

RF frequency: 100M

Sample rate: 2M

Oversample: Default

MIMO phase align: Disabled

out



Project Risks and Mitigation



Overview

Solution

CPE

DR

Risk & M

V&V

Planning

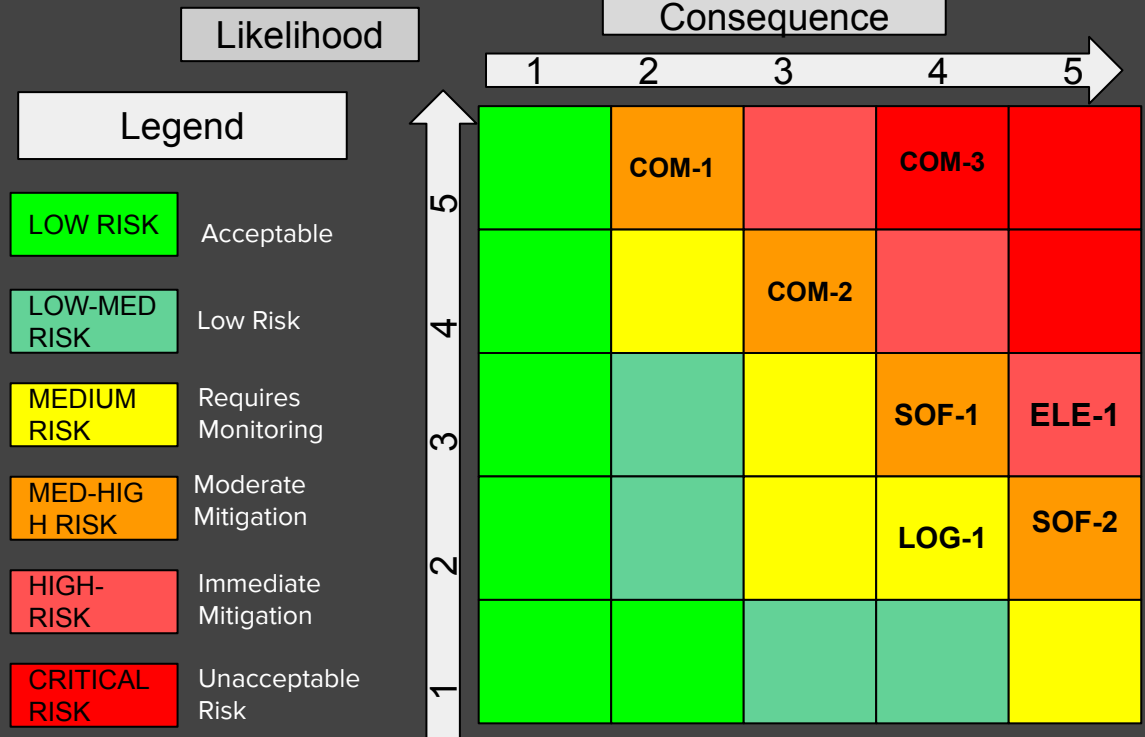
Risk Identification

Risk ID	Description	Likelihood	Consequence
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources	4	5
ELE-1	Time synchronization between pseudolite and GPS receiver	3	5
SOF-1	Team inexperience handling new software. For example, GNU radio	3	4
COM-2	Signal Interference in testing environment	4	3
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly	2	5
COM-1	Multiple User Communicating	5	2
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project	2	4



Risk Matrix

Risk ID	Description
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources
ELE-1	Time synchronization between pseudolite and GPS receiver
SOF-1	Team inexperience handling new software. For example, GNU radio
COM-2	Signal Interference in testing environment
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly
COM-1	Multiple User Communicating
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project

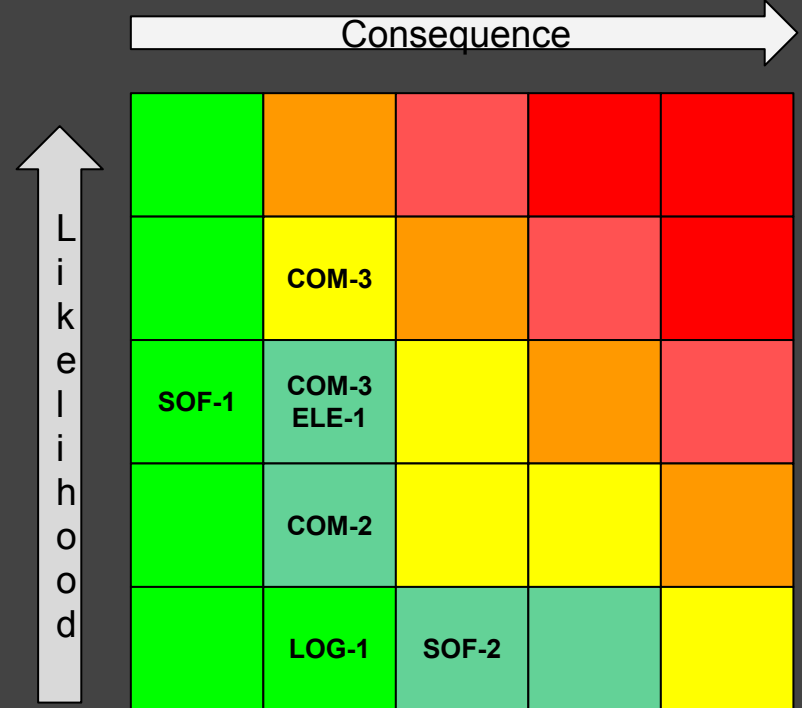






Mitigation Effects

Risk ID	Description
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources
ELE-1	Time synchronization between pseudolite and GPS receiver
SOF-1	Team inexperience handling new software. For example, GNU radio
COM-2	Signal Interference in testing environment
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly
COM-1	Multiple User Communicating
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project

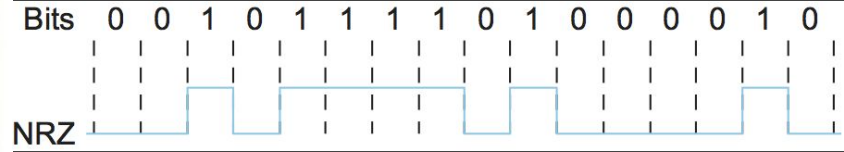
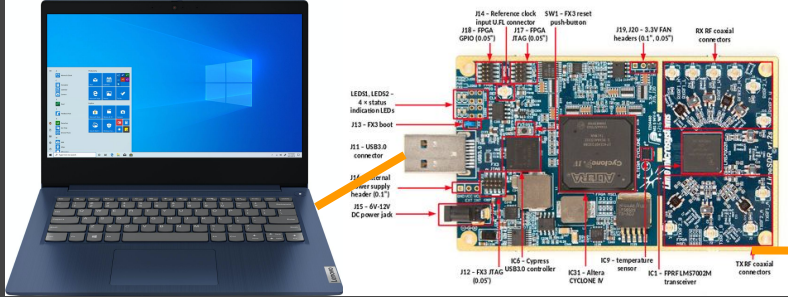




Verification and Validation

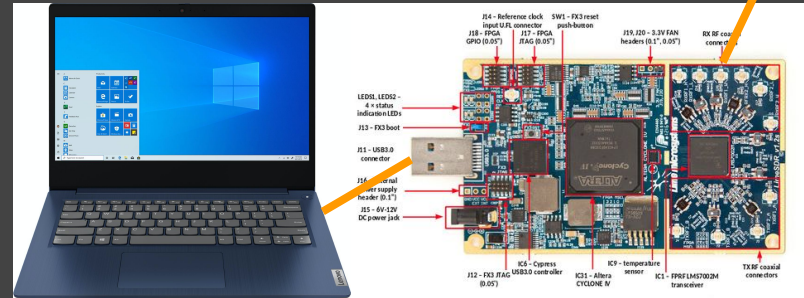


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USB Connection sends .txt as bits to LimeSDR

Wired from TX to RX sending bits



Bits converted to .txt
and compared to
initial file



Overview

Solution

CPE

DR

Risk & M

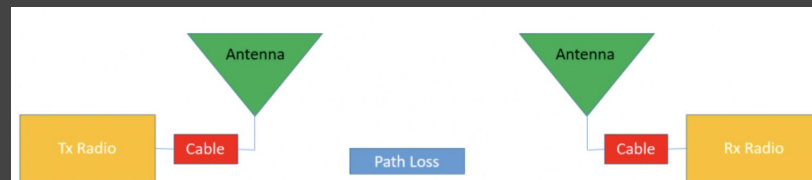
V&V

Planning

Antenna test

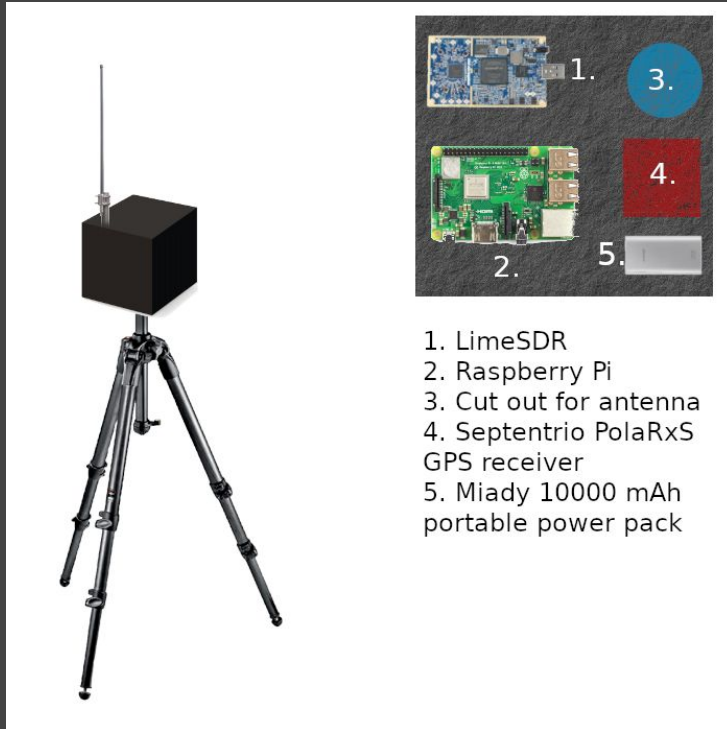
Demonstrate that the antennas used can supply the proper gain for the design

- Measure loss between a two radio/antenna setup running multiple trials at various distances
- Use least squares calculate the gain of the antennas(they are the same)



$$PL = G_1 + G_2 + 20 \log \left(\frac{\lambda}{4\pi R} \right)$$

Transmitter/Receiver Set up

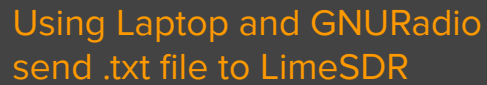


1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack

Each transmitter and receiver will feature a casing for the electronics that will hold the LimeSDR, Raspberry Pi, a battery pack, and space for wiring and a connection to the antenna. The electronics casing will be thread-mounted to a camera tripod.

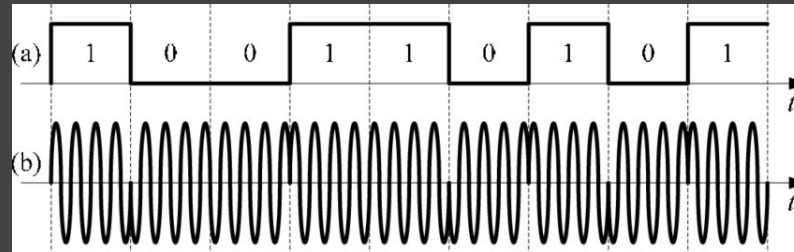


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GNURadio modbox modulates bits using BPSK modulation scheme

LimeSDR receives the modulated signal and demodulates using GNURadio demodbox



Bits are converted to .txt file
and compared to original .txt





Multiplexing and Wireless Transmission

Receiver can distinguish CDMA codes and process the desired signal

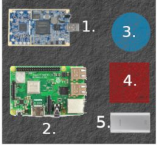
- 3 transmitters generate and transmit signals of different 100 bit .txt files
 - For each file, send preamble and .txt file using GNU radio PSK modulator
- LimeSDR receiver positioned 100m away receives the signals and listens for specific preamble
 - Based on desired preamble, receiver demodulates and converts desired signal to .txt
- Received .txt file is compared to original 3 to verify proper signal was processed



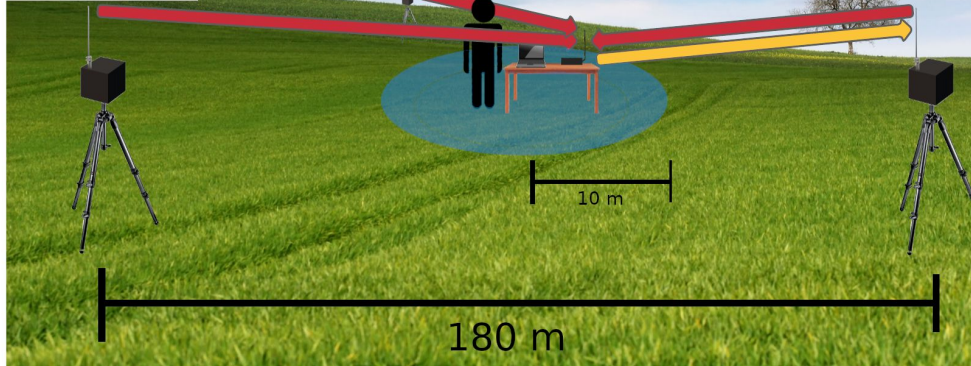
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Multiplexing & HDOP Test Setup

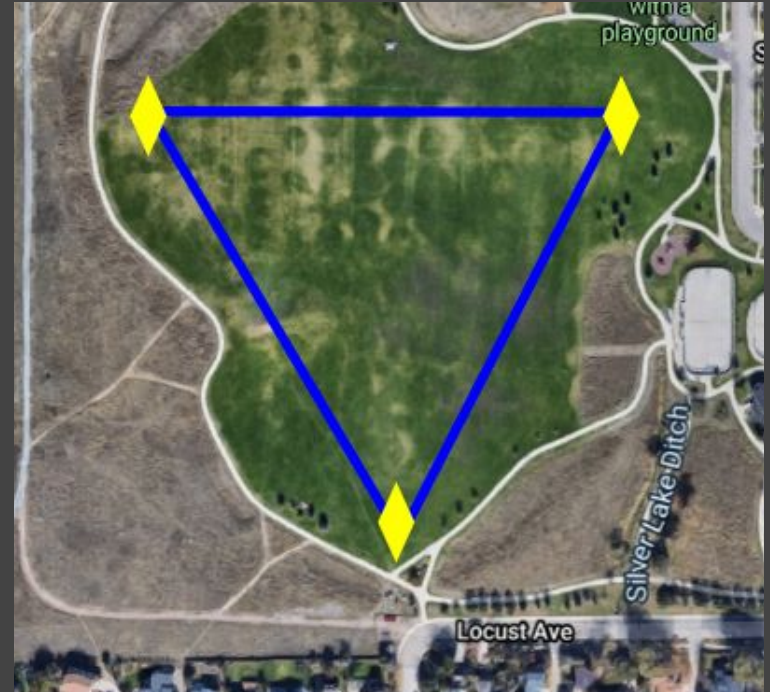
Contents of Black Boxes



1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack



- Positioning
- Comms to main pseud.
- Positioning w/in 10 m





Test Checklist

1. Walk transmitters to their positions
2. Sync transmitters and receiver with GPS
 - a. Sync up clocks to GPS (as well as to each other)
 - b. Record “true” position of each transmitter and receiver
3. Verify all stations ready
4. Run communications/positioning test



Overview

Solution

CPE

DR

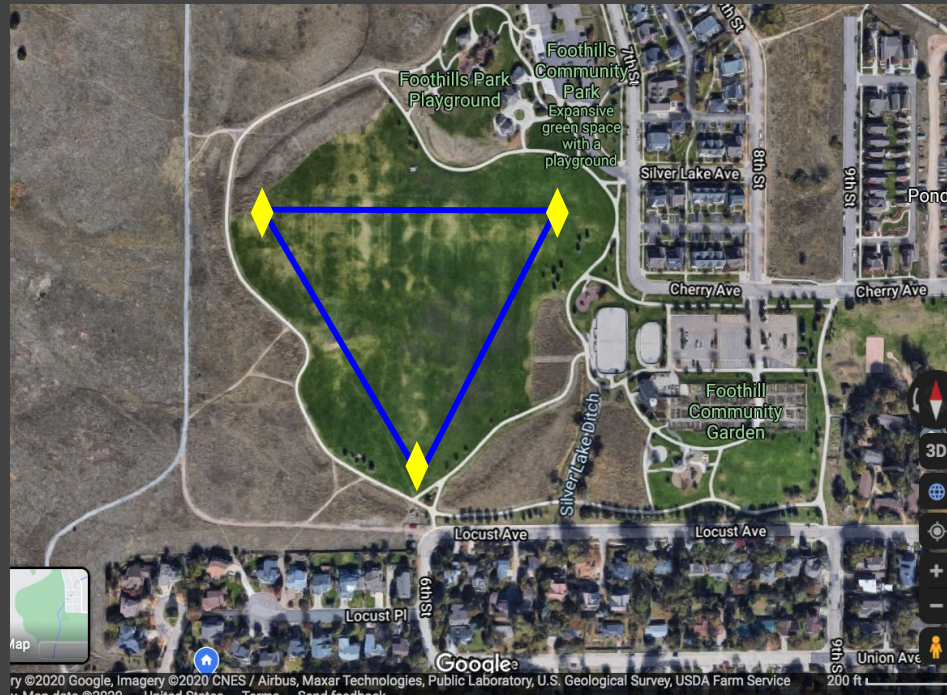
Risk & M

V&V

Planning



Testing site: Foothills Park

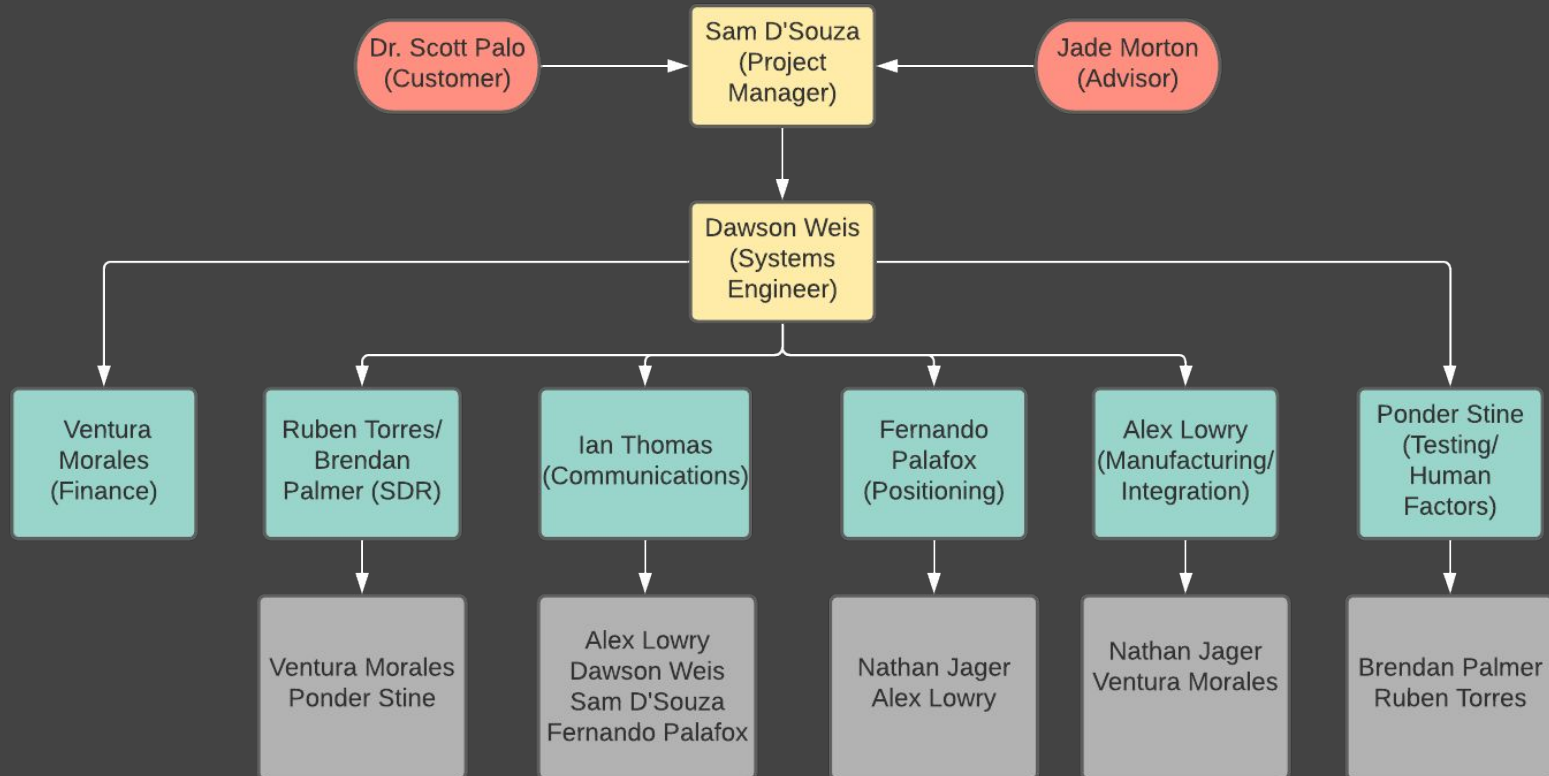


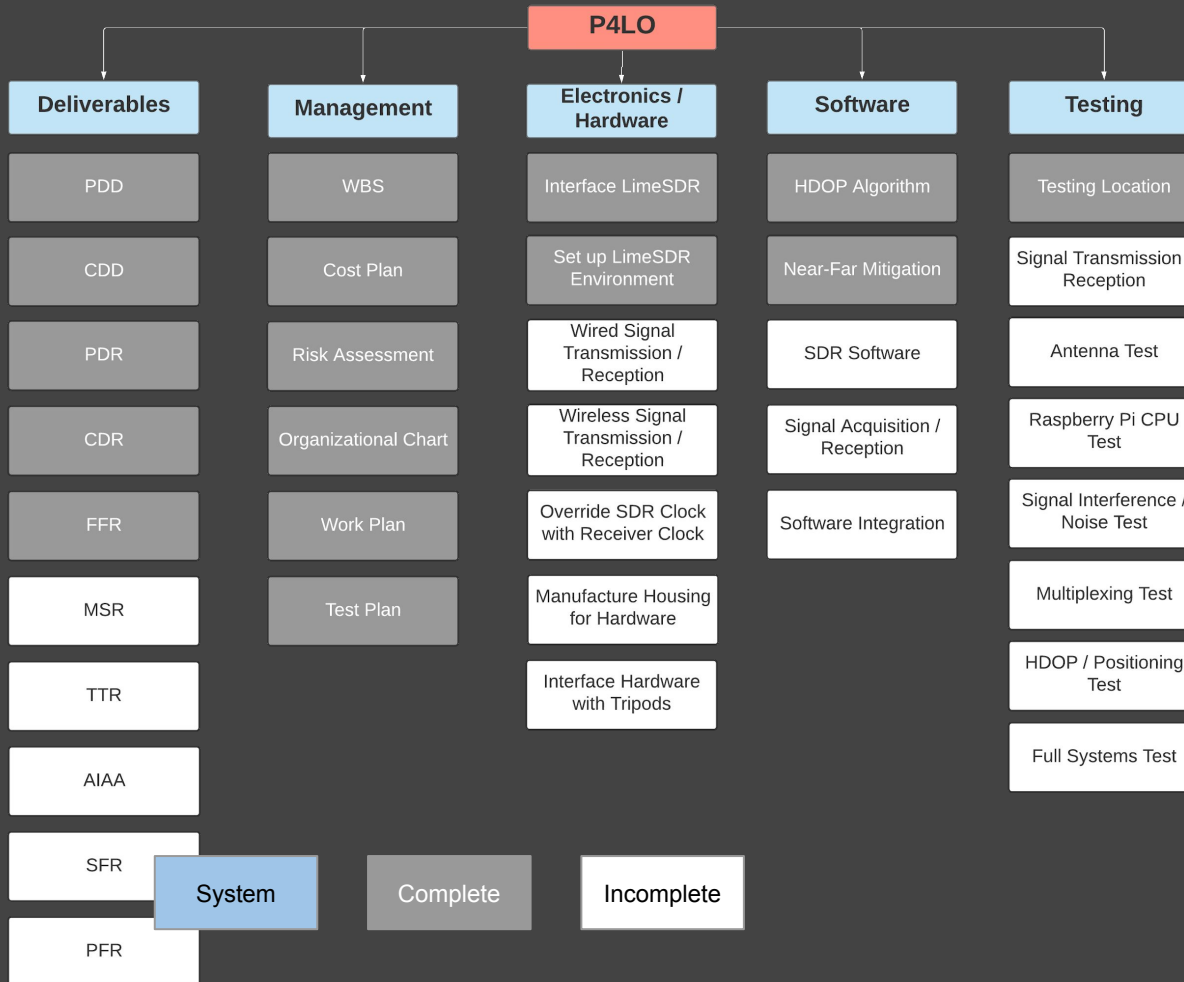


Project Planning



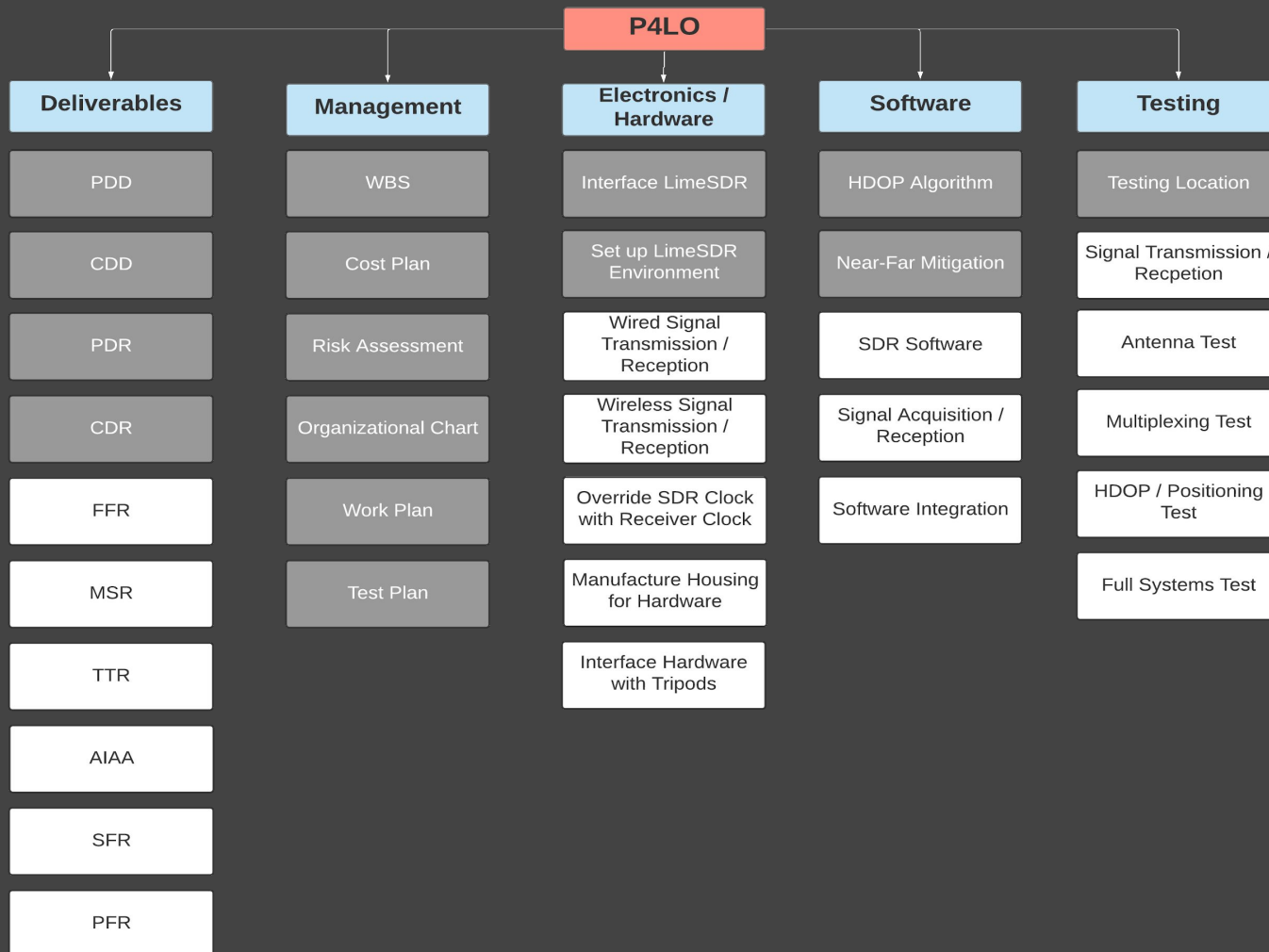
Organizational Chart







Work Breakdown Structure (WBS)



System

Complete

Incomplete







Testing Procedure	Date Scheduled	Location
Signal Reception Test	March 9th	Foothills Park
Modulation and Multiplexing Test	March 15th	Foothills Park
HDOP Test	March 23rd	Foothills Park
Full Scale Test	March 25th	Foothills Park



Budget



Overview

Solution

CPE

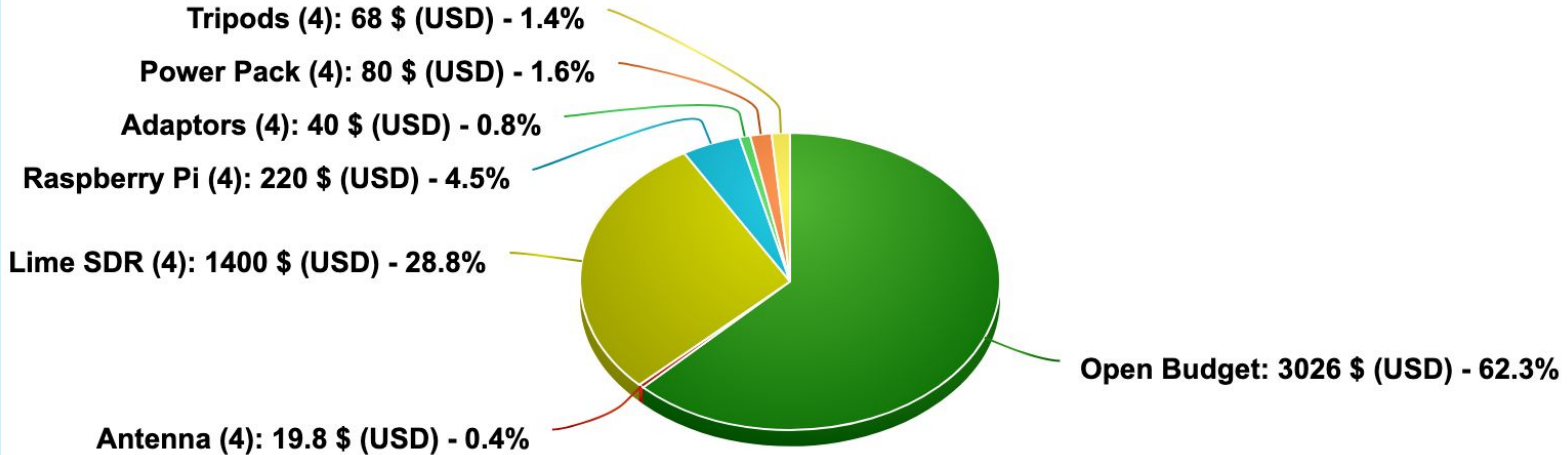
DR

Risk & M

V&V

Planning

Budget Pie Chart





Expense Budget & Margin

Team P4LO Budget					
Allowable Budget	\$5,000.00				
Communications Hardware					
Item	Individual Cost	# of Items	Subtotal Per Set	Supplier	Item Link
Antenna	\$4.95	4	\$19.80	SparkFun	https://www.sparkfun.com/products/11320
Lime SDR	\$349.95	4	\$1,399.80	SparkFun	https://www.sparkfun.com/products/15027
Raspberry Pi	\$55.00	4	\$220.00	SparkFun	https://www.sparkfun.com/products/15447
Hardware Adapators	\$9.95	4	\$39.80	SparkFun	https://www.sparkfun.com/products/9145
Power Pack	\$19.99	4	\$79.96	Amazon	https://www.amazon.com/Miady-10000mAh-Portable
Testing Tripods (6ft)	\$16.99	4	\$67.96		https://www.continentalphoto.com/itemdetails.asp?r
Budget Balance					
Sub-total Balance	\$1,827.32				
Sales Tax Total	\$146.19				
Total Cost	\$1,973.51				
Budget Balance	\$3,026.49				



"Raspberry Pi Icon #35510." *Raspberry Pi Icon #35510 - Free Icons Library*, icon-library.com/icon/raspberry-pi-icon-20.html.

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