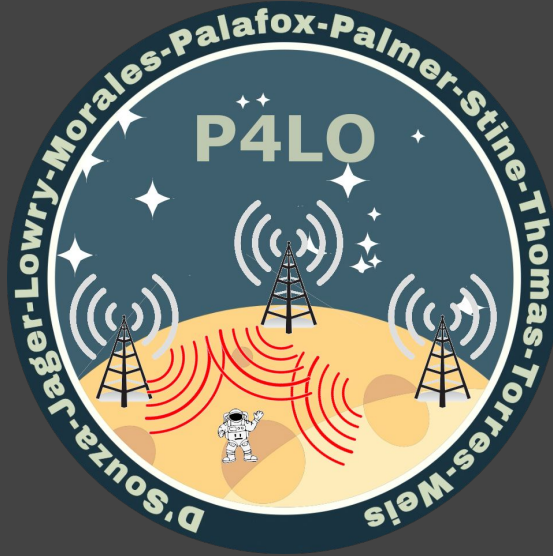


# P4LO



## Positioning For Lunar Operations

Team Advisor: Dr. Jade Morton



# Agenda

1. Project Purpose and Objectives
2. Design Description
3. Test Overview
4. Test Results
5. Systems Engineering
6. Project Management



## 3

Purpose

Design

Test

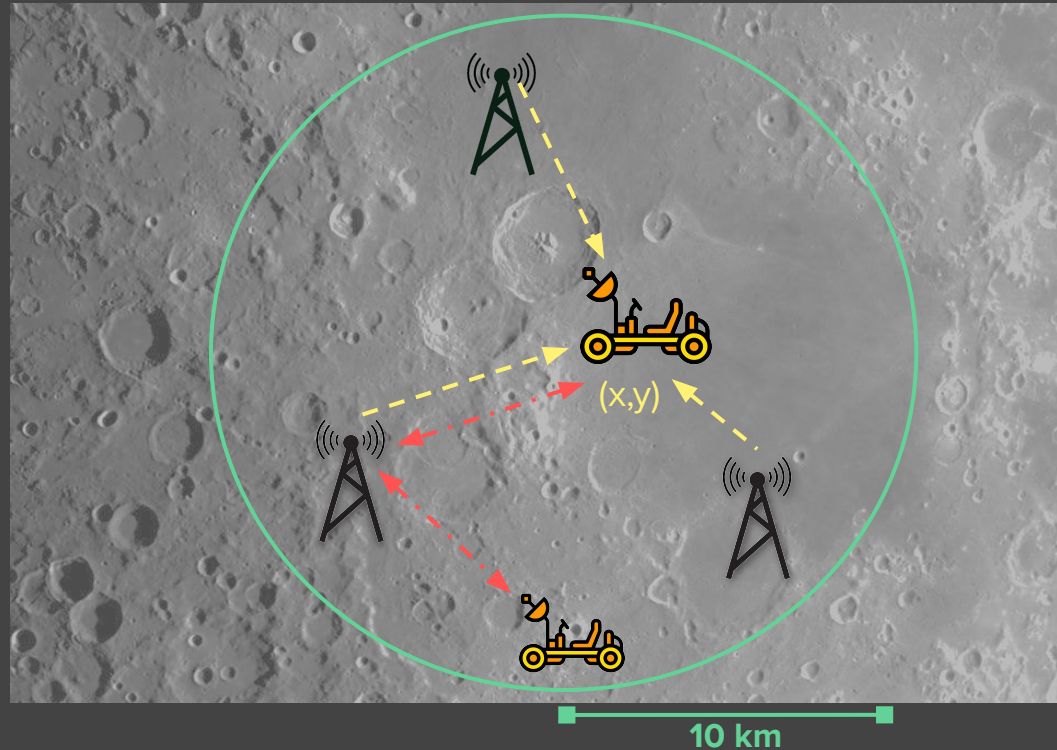
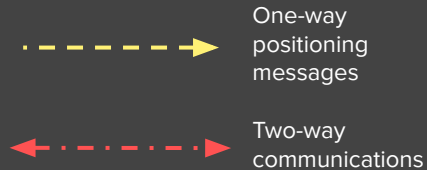
Results

Systems

Management

# Lunar CONOPS

- Provide ground-based positioning and communication to the South Pole
- 10 km radius region







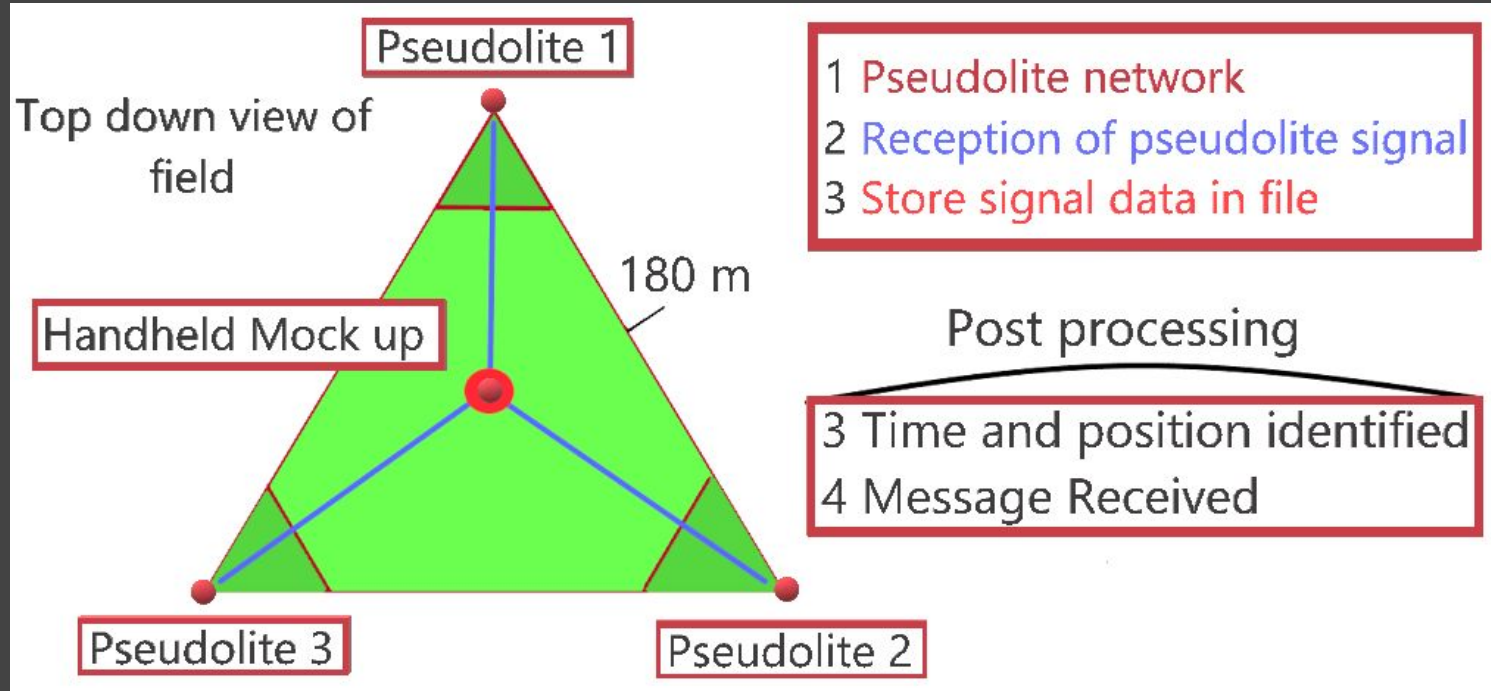
# Mission Statement

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**P4LO (Positioning For Lunar Operations)** is a prototype network of software defined radios that demonstrates an architecture for the lunar communication and positioning system LunaNet.



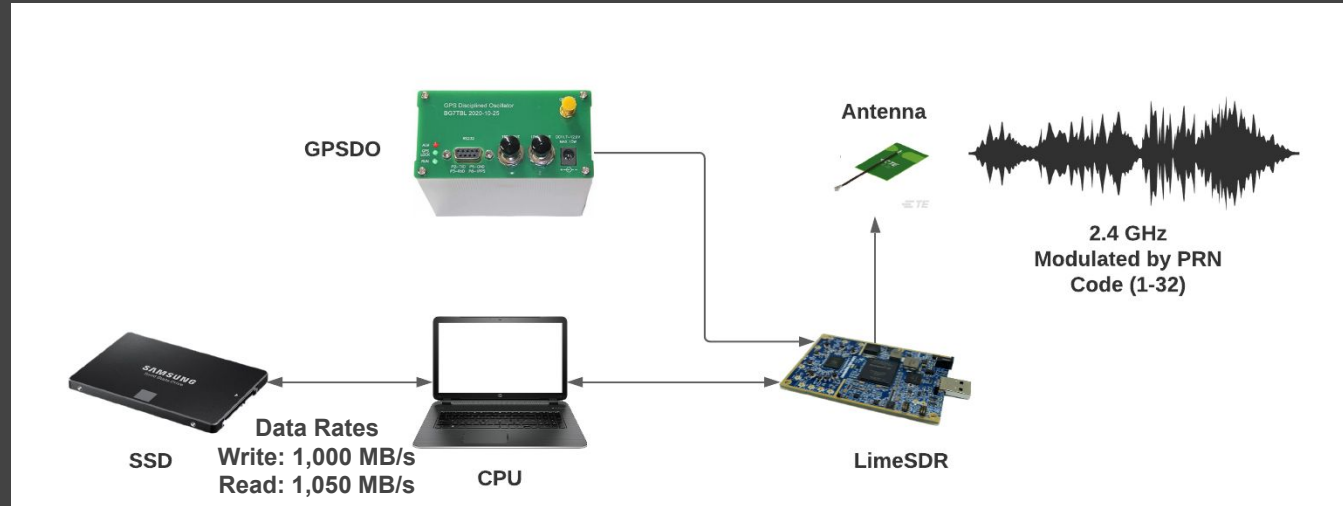
# P4LO CONOPS





# What is a Pseudolite?

- A portable ground station
  - Transmits positioning messages
  - Receives user communication messages





# Mission Success

- Key Functional Requirements
  - Scalable Positioning System
    - Sub 10 (m) positioning within specified area
    - 30 ns 1-sigma transfer time
  - Communication Network
    - Communication operating between 2.4 - 2.48 GHz
    - 1 MHz Signal Bandwidth
    - Framework allowing for up to 170 users communicating within region
    - Data rate of at least 1 kilobit per second



# Levels of Success

1	Hardware/Software interfacing	Ubuntu environment + dependencies, GNURadio, LimeSDR	Completed
2	Wireless Transmission	Wireless transmission. Getting LimeSDR to transmit over the air (simple tx/rx test)	Demonstrated
3	Acquisition	Acquiring PRN codes	Demonstrated
4	Positioning	Ranges from acquisition results + positioning algorithm to compute position solution	Capability In Place
5	Data Transmission	Transmit a .txt data file over the air at specified data rate	Capability In Place



# Design Description

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- Management

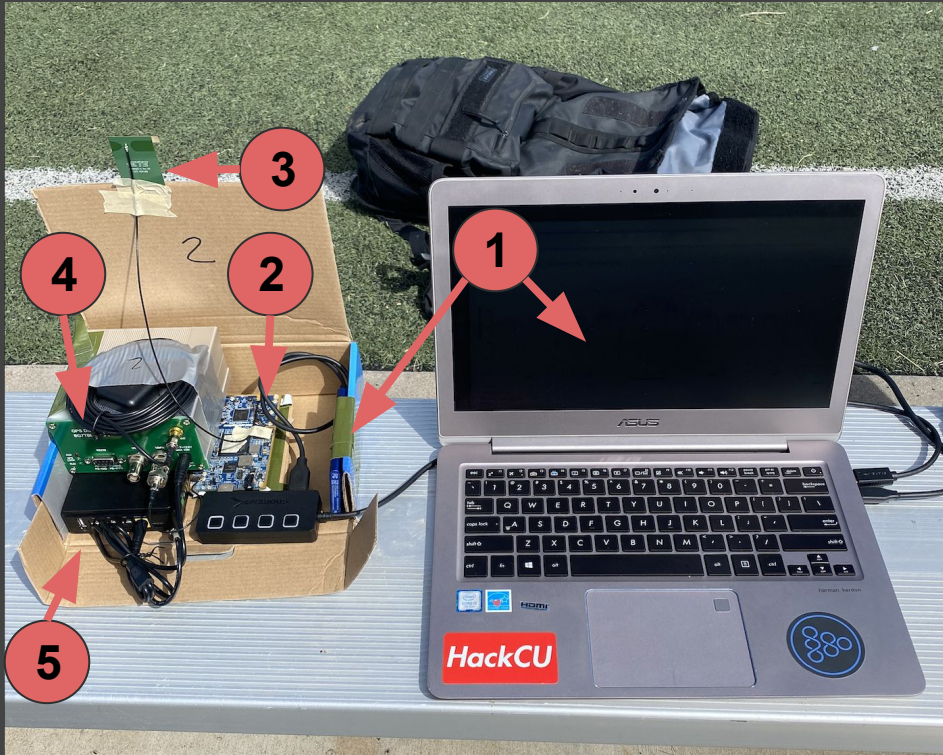








# Physical Design/Critical Project Elements



- 1 Laptop
  - GNU Radio
  - SSD (for Ubuntu OS)
- 2 LimeSDR
- 3 Antenna
  - [Range: 2.4 - 2.48 GHz]
- 4 GPS Disciplined Oscillator
  - Clock Override
- 5 Battery
  - 10,000 mAh, 12V @1.25A

Purpose

Design

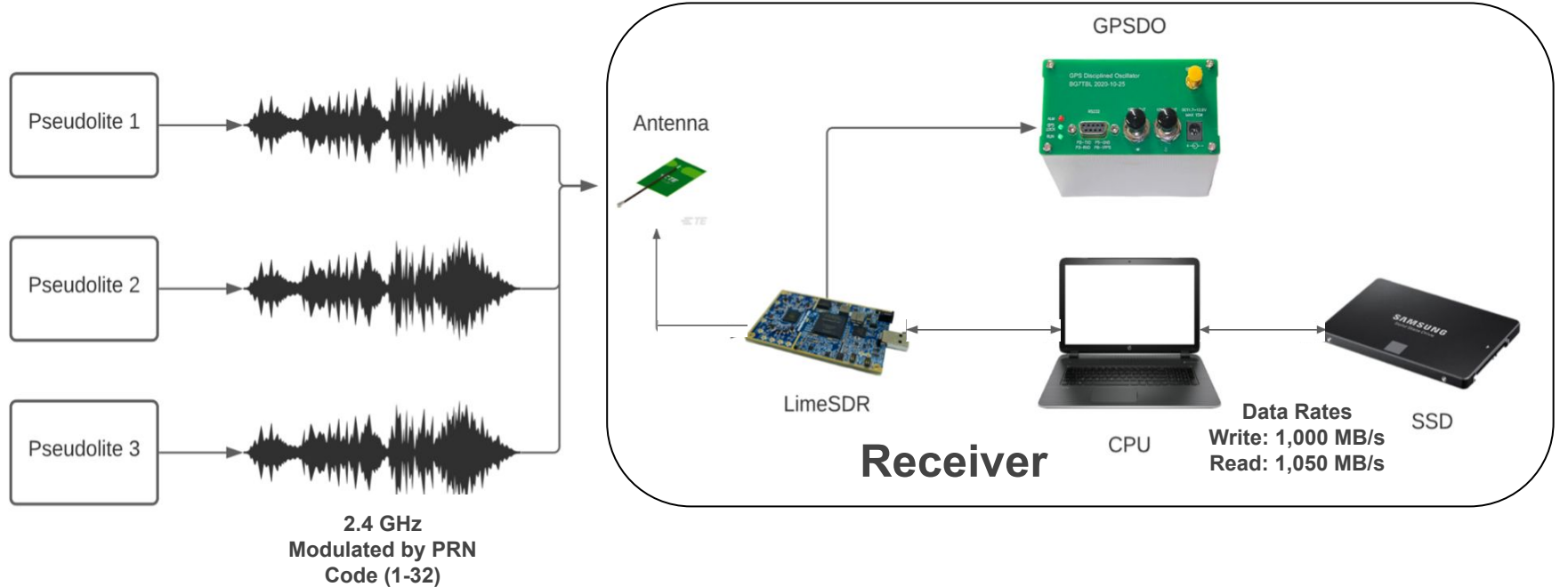
Test

Results

Systems

Management

# Hardware FBD



Purpose

Design

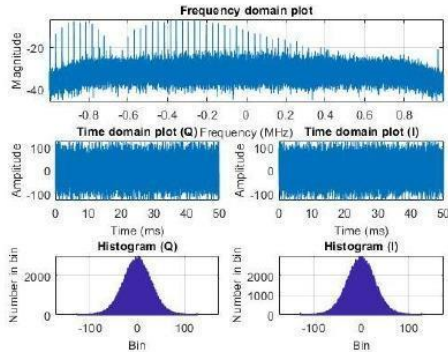
Test

Results

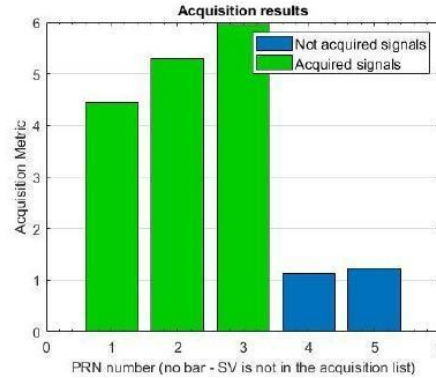
Systems

Management

# Software FBD



Raw data collection



Pseudolite synchronization  
& range determination

$$d_1 = \sqrt{(x_1 - x)^2 + (y_1 - y)^2}$$

$$d_2 = \sqrt{(x_2 - x)^2 + (y_2 - y)^2}$$

$$d_3 = \sqrt{(x_3 - x)^2 + (y_3 - y)^2}$$

Position calculation



# Test Overview

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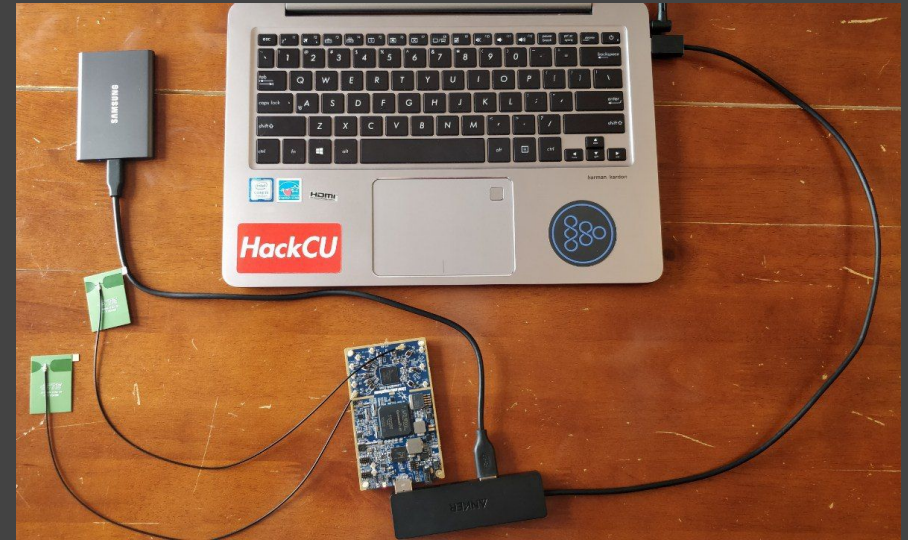
# Testing Overview

---

- Basic communication between TX and RX
- Oscillator override
- Short-range acquisition (Multi-TX)
- Long-range acquisition (Multi-TX)

# Basic Communication Test

- Integrate hardware and software
- Transmit one FM audio file from SDR transmit port and receive it on the SDR receive port, play results



# Oscillator Override

- Require a more stable oscillator for higher quality transmission and reception
- Will be using a GPS-disciplined oscillator (GPSDO)
- Verification will be done by setting receiver as a GPS receiver and comparing position solutions

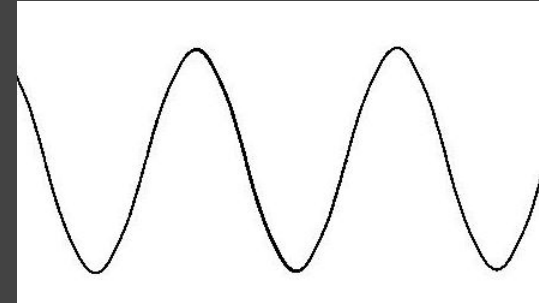


LimeSDR

+



GPSDO





Purpose

Design

Test

Results

Systems

Management

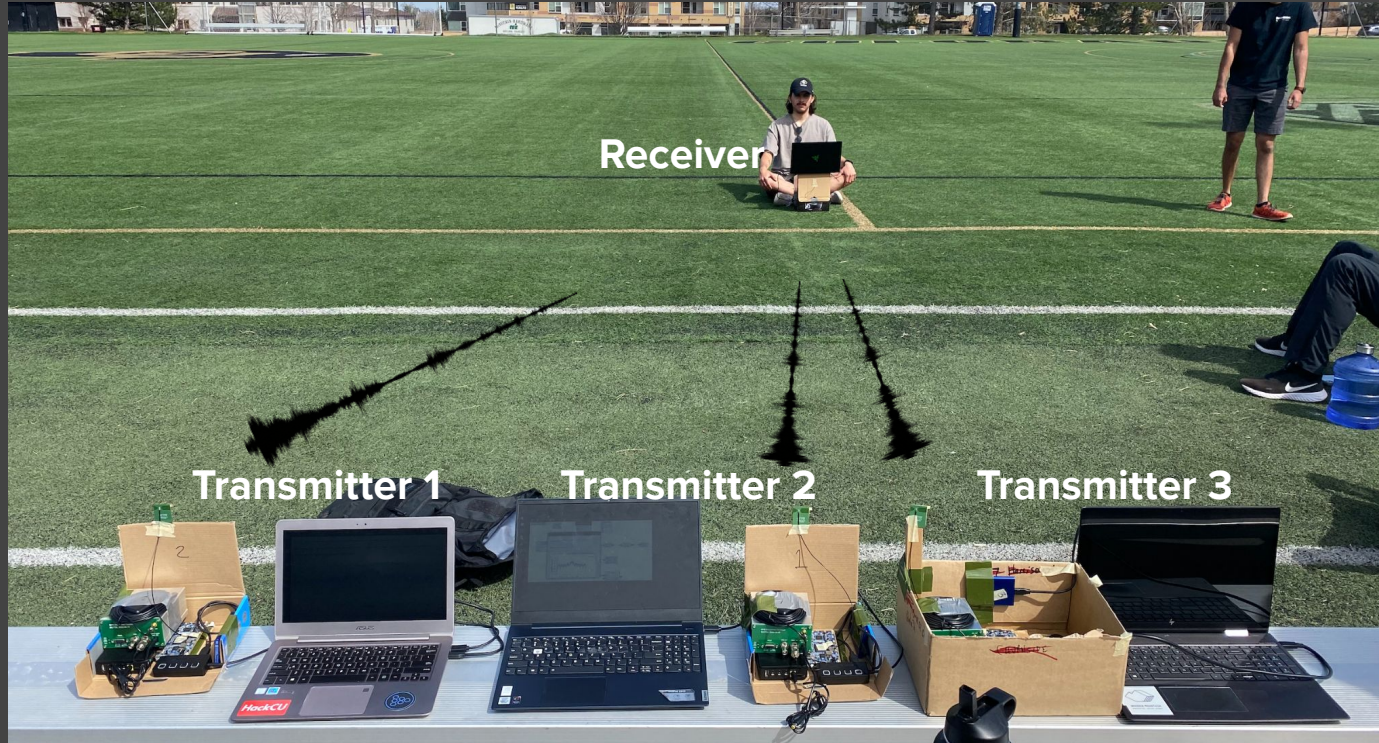


# Short range acquisition (multi-tx)

- Acquisition
  - All transmitters are sending signals on the same frequency
  - Each signal is modulated by a unique Code Division Multiple Access (CDMA) code
  - Acquisition allows for the receiver to pick out individual signals
    - Language analogy
- Test
  - Have 3 pseudolites transmitting at 2.4GHz with their unique codes
  - Receiver in same room
  - Pick out the individual signals



# Short range acquisition (multi-tx)





Purpose

Design

Test

Results

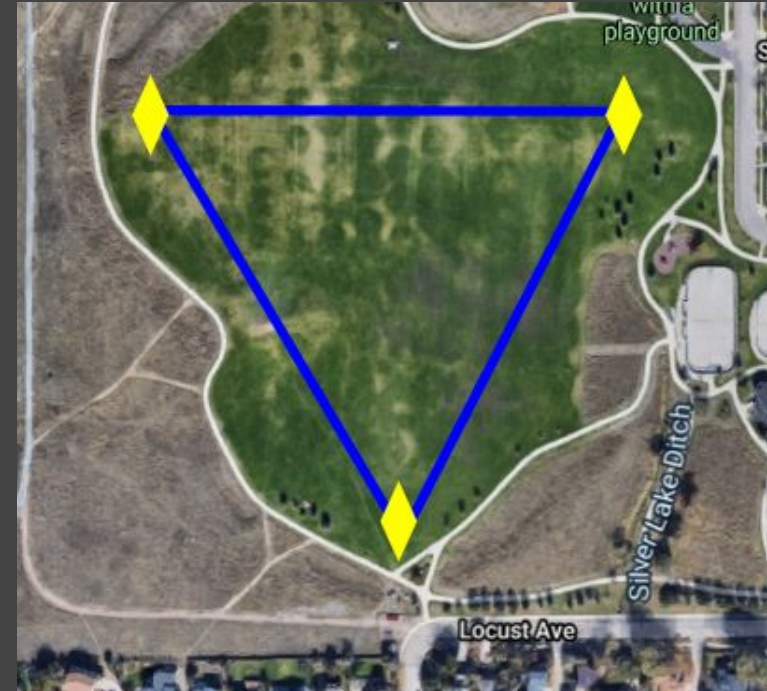
Systems

Management



# Long range acquisition (multi-tx)

- Test
  - Same setup as before, but now transmitters are positioned at a distance  $>100\text{m}$
  - Will be done outdoors
  - Right: Wonderland Park, pseudolites represented by yellow diamonds



Level of Success: 3 and 4



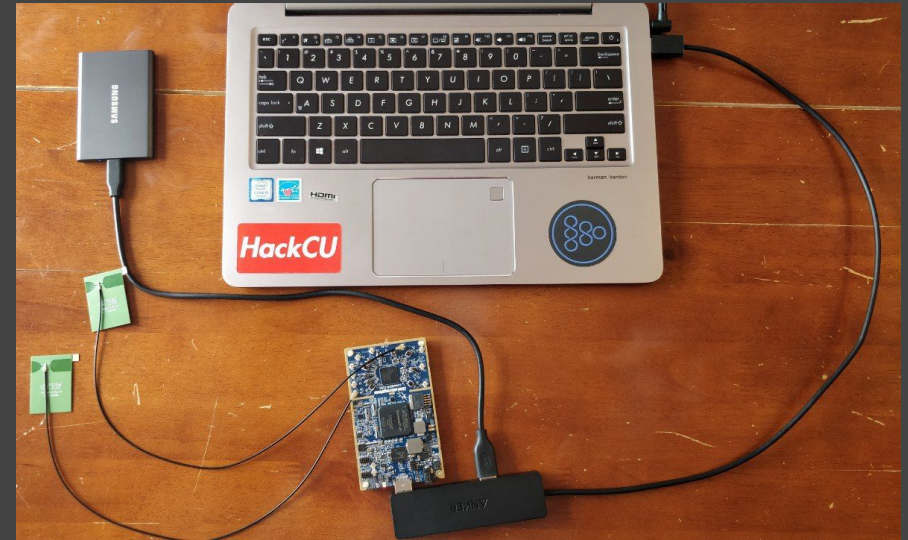
# Test Results

---

# Basic Communication Test

- Integrate hardware and software
- Transmit one FM audio file from SDR transmit port and receive it on the SDR receive port
- Verifies **FR4** and **FR5, DR1.2**

Level of Success: 1

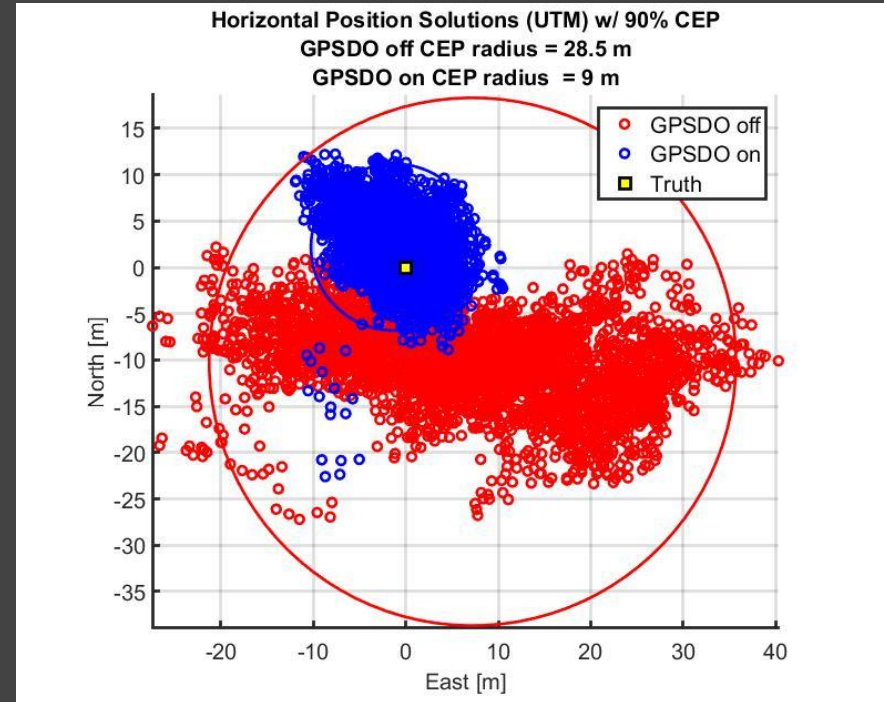




# Clock Override Test

- Compared performance of:
  - LimeSDR + onboard oscillator
  - LimeSDR + GPS-disciplined oscillator (GPSDO)
- Setup LimeSDR as a GPS receiver
- Indirect measure of clock stability/performance

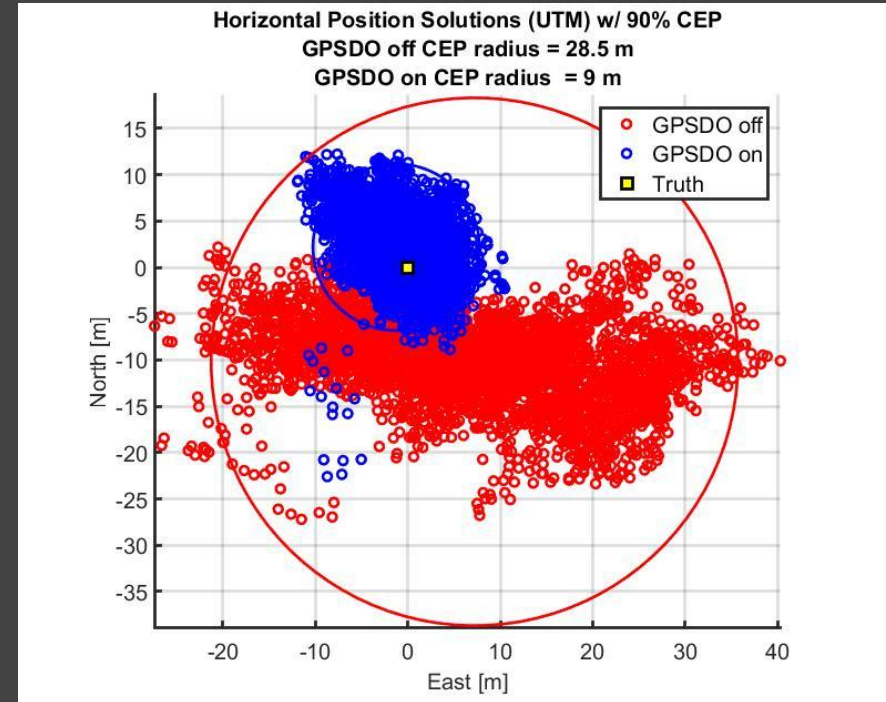
	CEP 90% radius (m)	Mean pos. Error (m)
Onboard	28.5	39.5
GPSDO	9	16.5



# Clock Override Test

- Demonstrates that positioning algorithm works
  - CEP radius satisfies <10m positioning requirement
- Using GPS satellites as opposed to the P4LO pseudolites
- Verifies **FR3, DR3.2.**

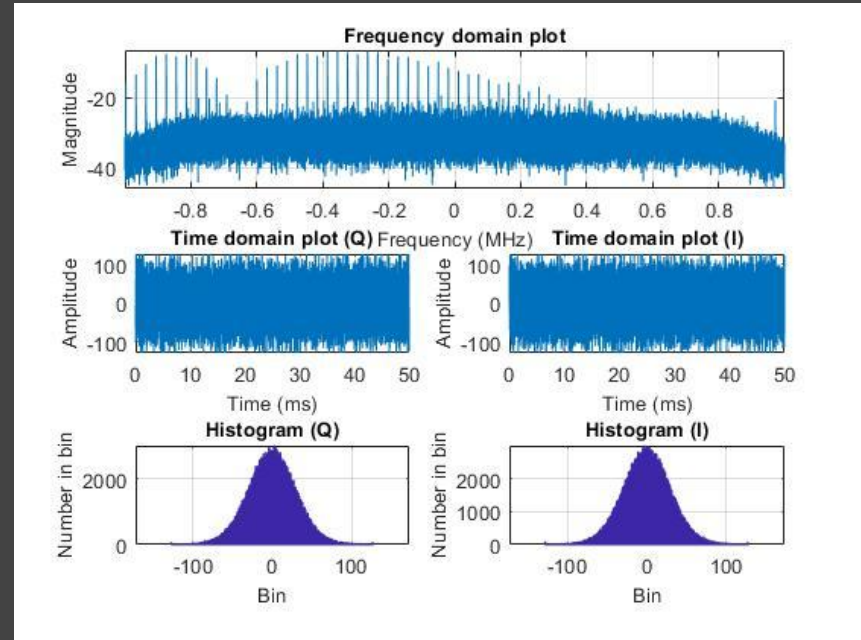
Level of Success: 2 and 5





# Short Range Acquisition (Multi-Tx)

- Transmitted PRNs 1-3
- Raw data collected by receiver
- Spectrum and time domain plots can be seen on the right,

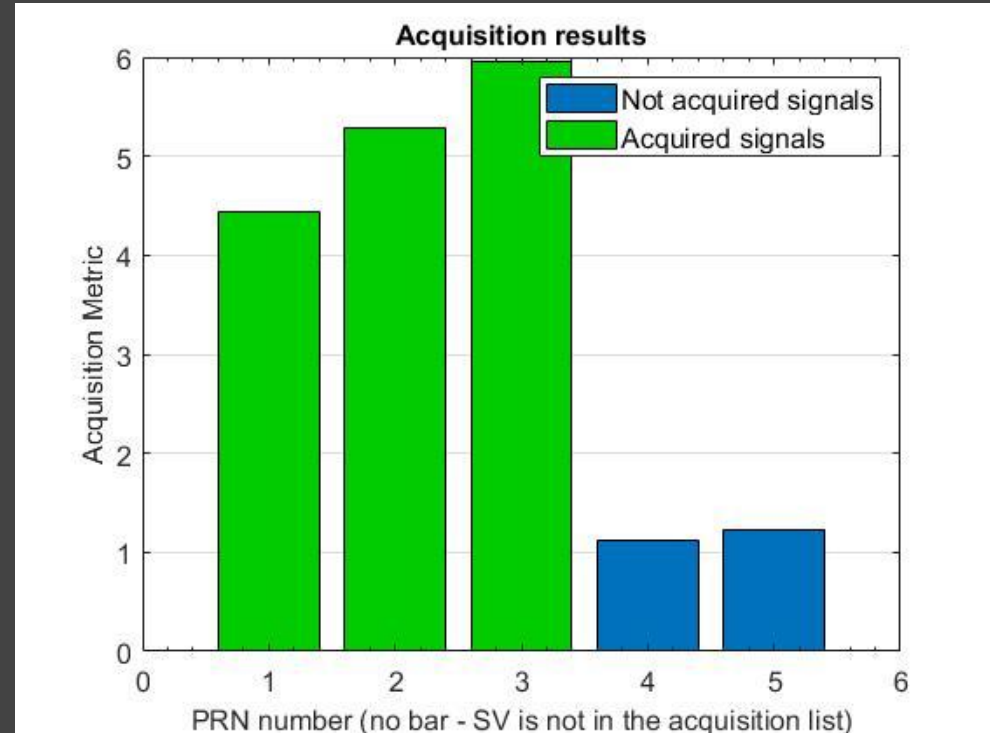




# Short Range Acquisition (Multi-Tx)

- Transmitted PRNs 1-3
- Attempted to acquire PRNs 1-5
- Acquired 1-3
- Acquisition metric: ratio between 1st and 2nd strongest correlation peak
  - A measure of SNR
  - $>2$  for successful acquisition
- Verifies **FR6** and **DR6.1**

Level of Success: 3 and 4





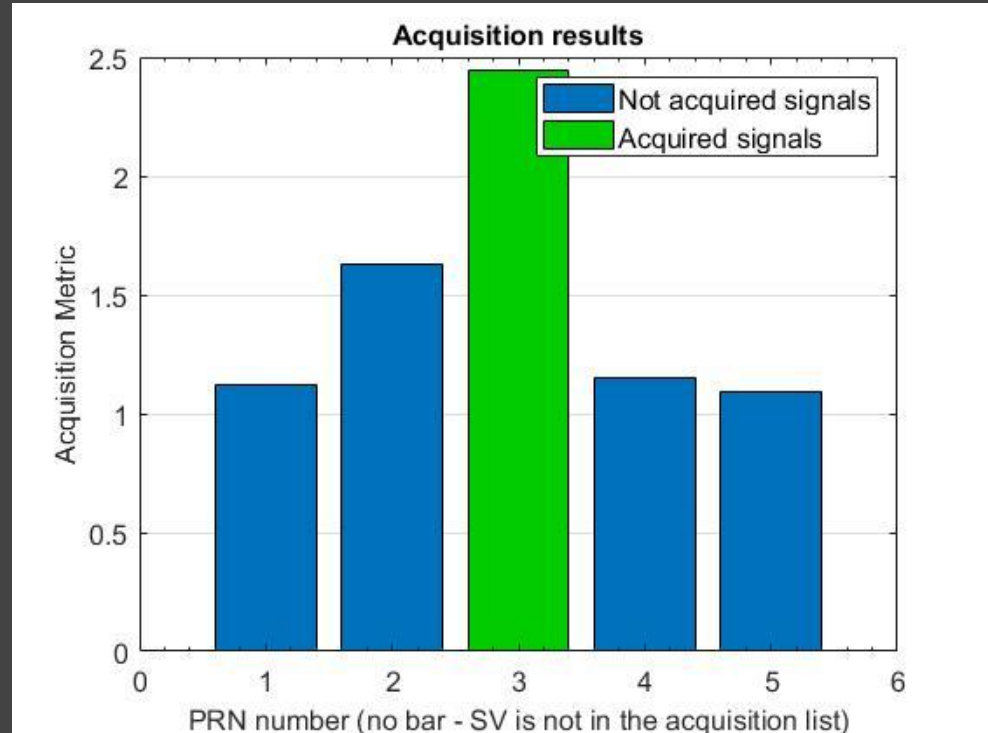


# Long Range Acquisition (Multi-Tx)

- Rx-Tx Distances  $>100\text{m}$
- Managed acquisition but with major caveats
  - Longer processing times (very noisy signals)
  - Needed better antennas and signal amplification (tx)
- Shows foundation for **DR6.2**

Level of Success: 4

Level of Success: 5





# Long Range Acquisition (Problem Solving Approach)

- **Problem**
  - Low Power
  - Noise
- **Ran the following tests**
  - Wired Power Test
  - Antenna angle test
  - Upgrade system with a low noise amplifier (LNA) and different antenna

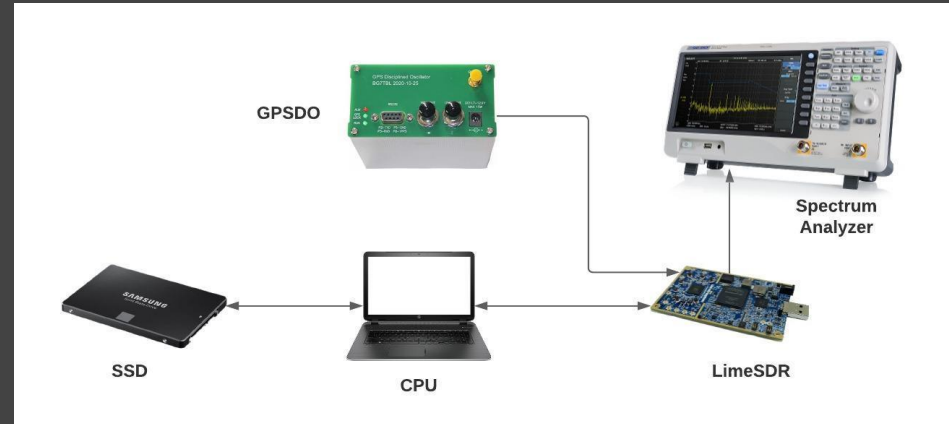


- Wired setup to test power coming out of LimeSDR



# Wired Power Test

- Wired setup to test power coming out of LimeSDR
- Transmitter -> Spectrum Analyzer
- 1mW or 0dBm
- Validates transmit power requirements
- Narrows issues to an antenna efficiency problem



Wired transmitter setup



Purpose

Design

Test

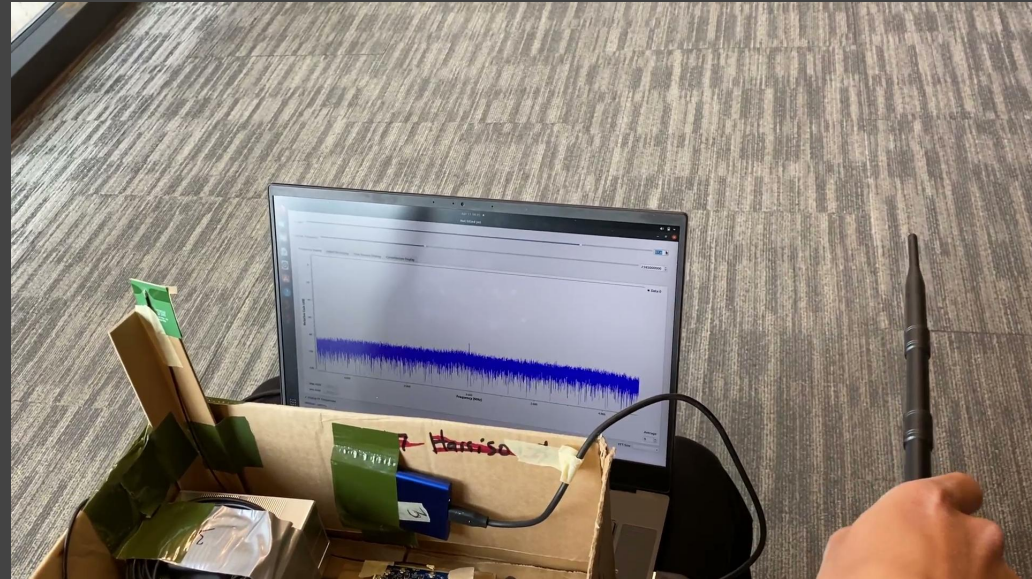
Results

Systems

Management

# Antenna Orientation Test

- Verified inconsistencies in patch antenna polarization and gain pattern
- Alignment in rx/tx antennas responsible for long-distance acquisition issues



Problem Identified: Antenna Gain Pattern

Purpose

Design

Test

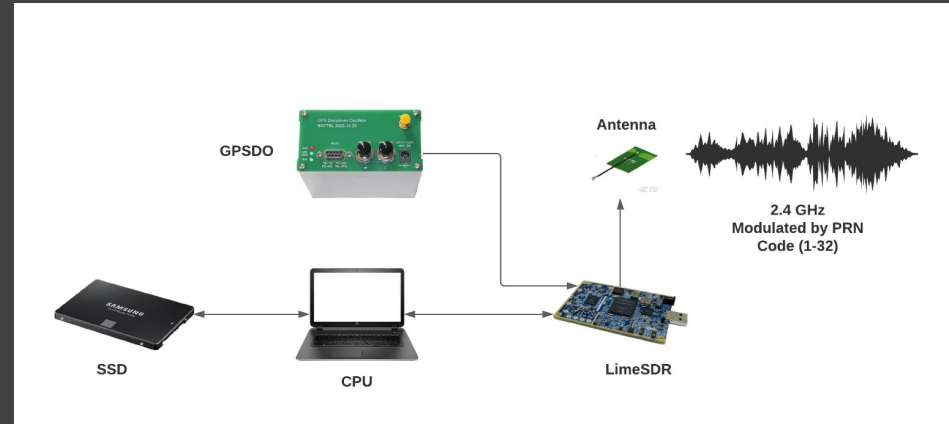
Results

Systems

Management

# Upgraded pseudolite

- More reliable antenna
  - Better documentation
  - Known polarization
  - Known gain pattern
- Low-noise amplifier (LNA)
- Upgraded  $\frac{1}{3}$  of pseudolites
- Demonstrated performance improvements
  - Increased 1m received power from -35 dBm to -15 dBm

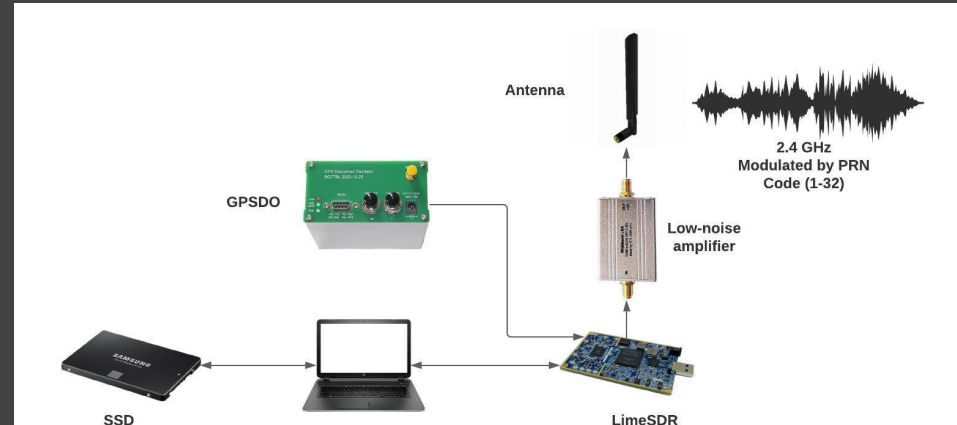


Original transmitter setup



# Upgraded pseudolite

- More reliable antenna
  - Better documentation
  - Known polarization
  - Known gain pattern
- Low-noise amplifier (LNA)
- Upgraded  $\frac{1}{3}$  of pseudolites
- Demonstrated performance improvements
  - Increased 1m received power from -35 dBm to -15 dBm



## Upgraded transmitter setup

Consistent Long Range Acquisition: ✓

## Positioning Capability in Place: ✓



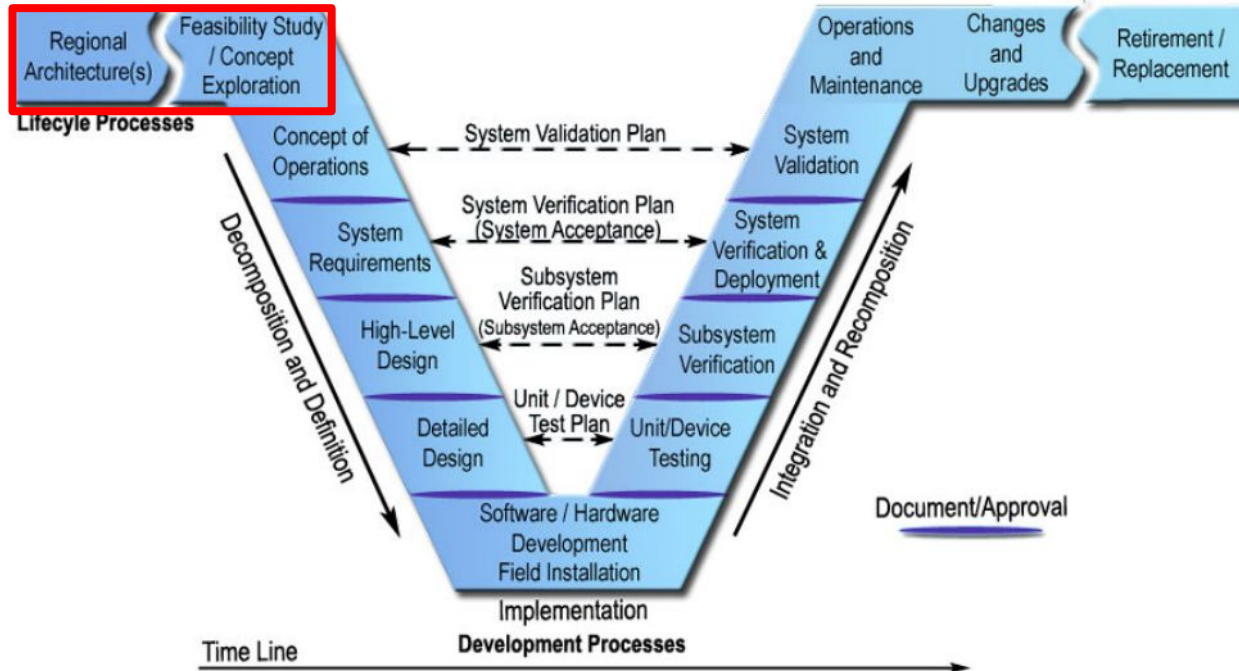
# Systems Engineering

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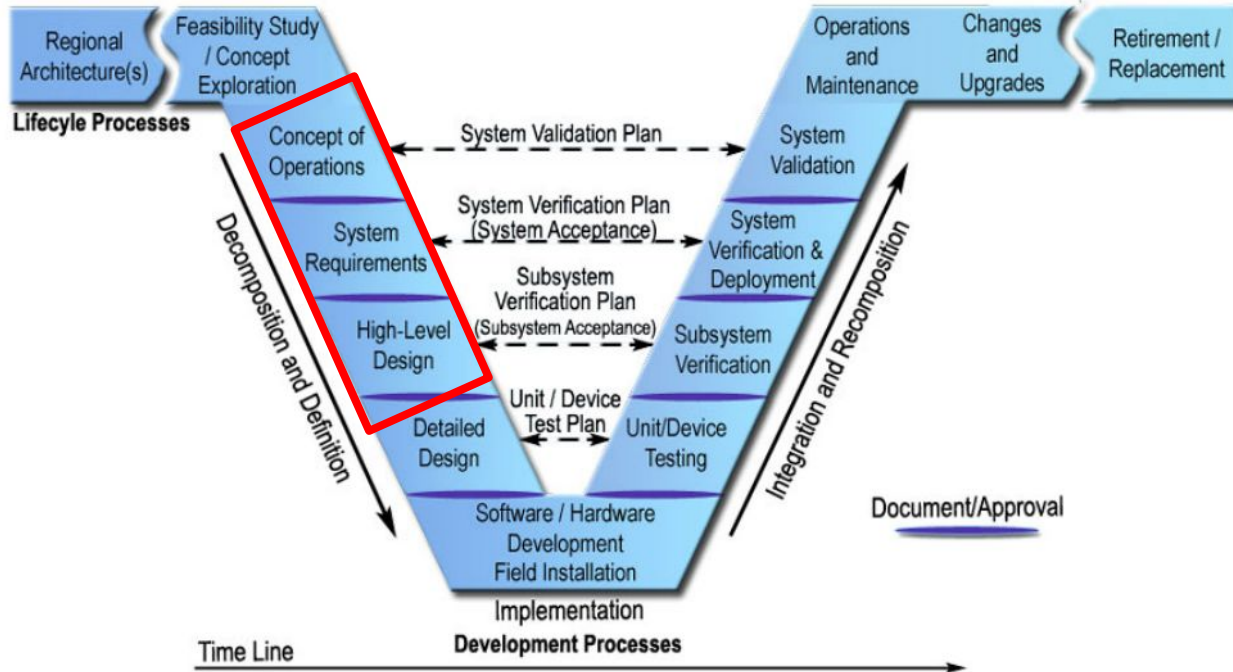
# Approach - Initial Problem Statement



- Lunar Communications and Positioning
- Narrowing Possibilities
- Feasibility Analysis
  - Risk Reduction
- Building Initial CONOPS



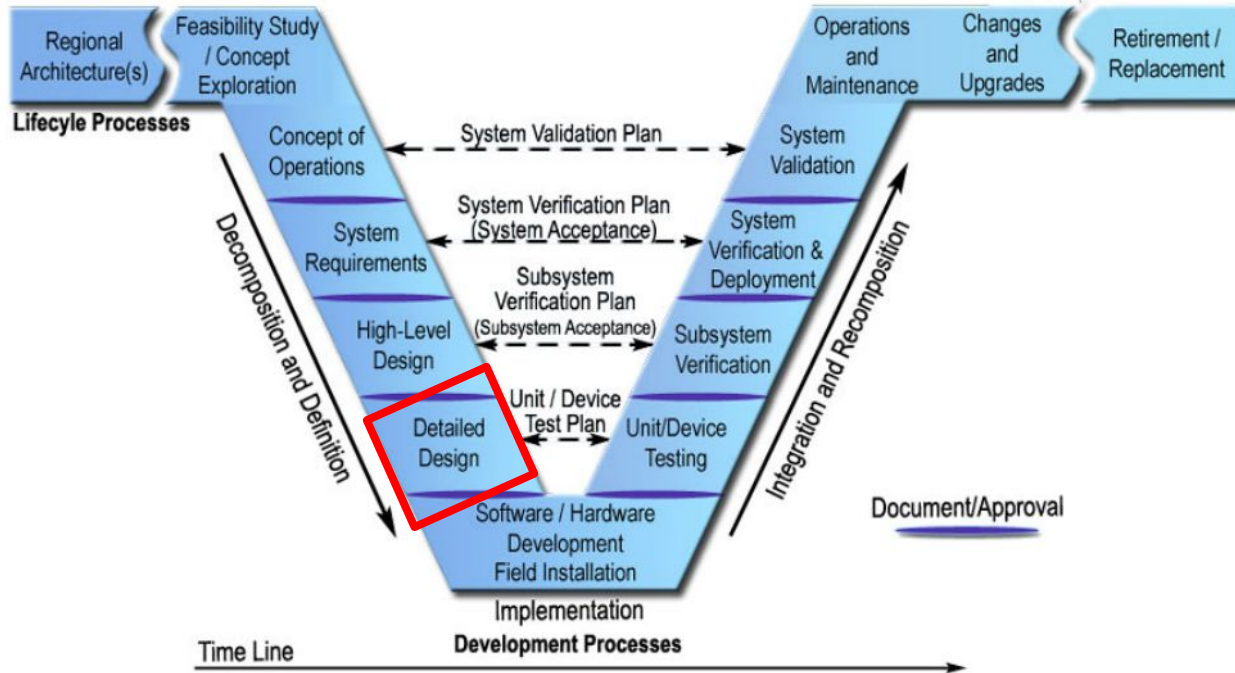
# Approach - Refining the Design



- Pseudolite Network
  - Refined CONOPS
- Derive Functional Requirements
  - Construct DR's
- Outline necessary hardware
  - SDR, Antenna, power, etc



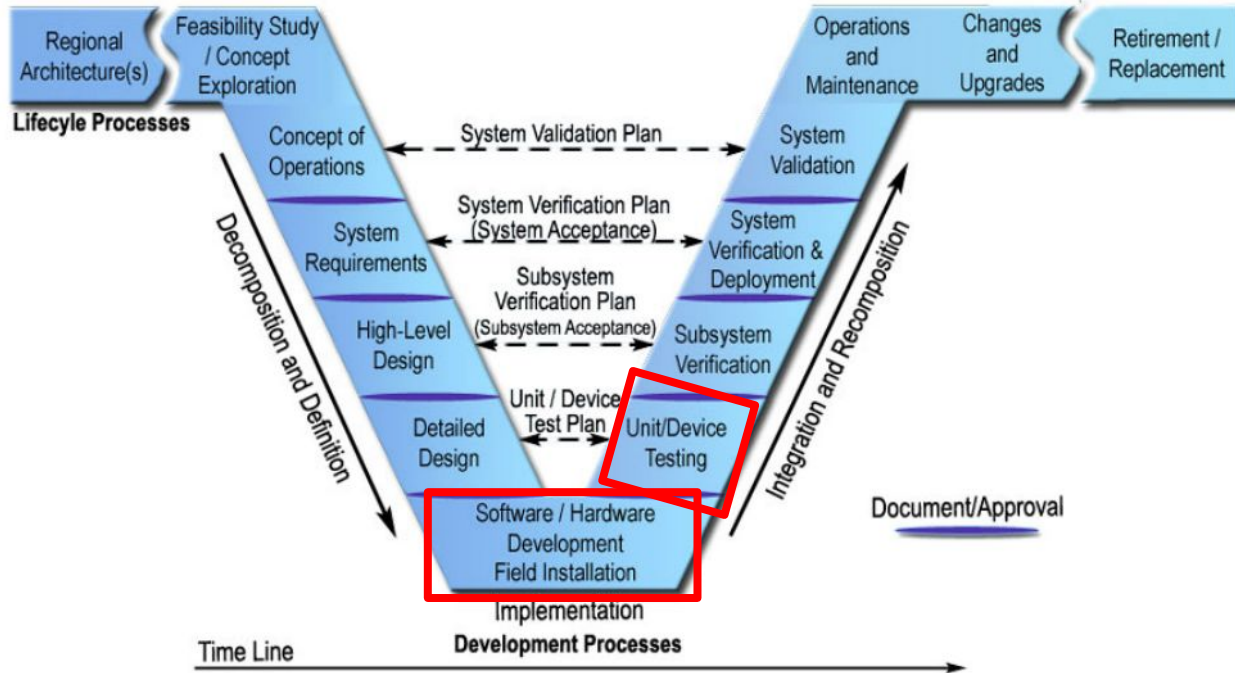
# Approach - Finalizing Design



- Trade different aspects of project
  - Software and hardware
- Development of ICD's in the form of FBD's
- Create testing plan for all aspects of project



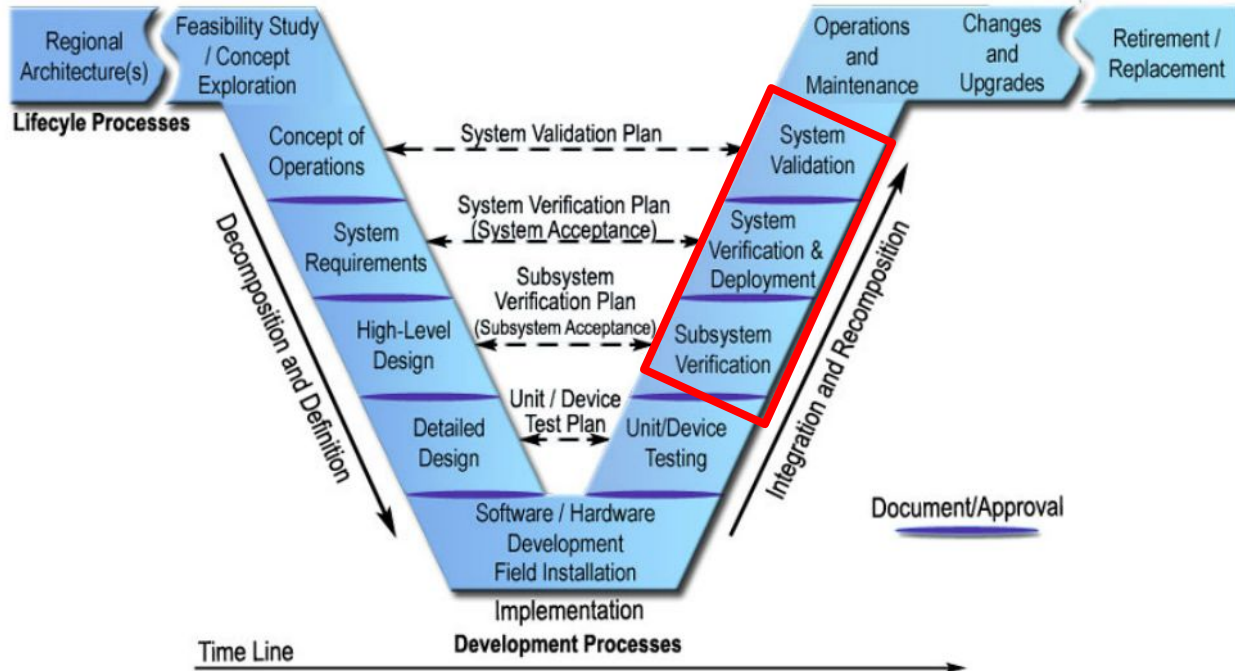
# Approach - Testing Hardware



- Get SSD's and GNU Radio set up
- Purchase and receive hardware
- Verify hardware functionality
  - Basic signal transmission and reception



# Approach - Full Scale Testing



- Test different subsystems
  - Clock Override
  - Acquisition
- Build pseudolites
- Combine subsystems into overall system
- Does system meet requirements?



# Main Issues Encountered

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- Difficulty acquiring signals consistently at distances  $>100$  m
  - Antenna/amplifying capabilities limited transmission power
  - Hardware did not meet expected performance
- Significant electrical engineering expertise required
  - CS knowledge
  - RF knowledge
  - Positioning Systems
- COVID
  - Delays in hardware shipping
  - Testing and scheduling
    - Difficulties in scheduling face-to-face meetings
    - Weather, suitable locations, hardware, etc...



## Key Lessons Learned (Systems)

- LunaNet Risk Mitigation
  - Thorough link budget estimation is required
    - Antenna
    - Amplifiers
    - Radio performance
  - Pseudolite synchronization capabilities
  - Hardware interfacing difficulties
- Design space flexibility
  - Altered requirements may change design path
- Field test efficiency
  - Field tests require proper planning and execution
  - Experience in running efficient field tests







# Project Management

## Successes

- 1 Great team chemistry
- 2 Team understanding of PNT Technologies
- 3 Hands on experience with SDR's
- 4 Team experience with GNU Radio

## Challenges

- 1 Steep Learning Curve
- 2 Risk Profile Estimation
- 3 Limited Link Budget Calculations
- 4 Hardware Interfacing Difficulties





## Hardware Addition Justification Post-CDR

Component	Purpose	Dollar Increase	Budget Percentage Increase
<b>GPSDO's</b>	Increase transmitter/receiver clock stability	\$800	16%
<b>SSD's</b>	Store data, boot programs like GNU Radio through Ubuntu	\$320	6.5%
<b>Power Packs</b>	Extend to outside testing for pseudolites (power GPS-DO, LimeSDR,etc)	\$160	3.2%
<b>Extra Adaptors</b>	USB-Extender cables and adaptors for pseudolites (needed for accommodating pseudolite housing)	\$150	3.9%



# Cost to Customer

<b>Labor Hours</b>	<b>2230</b>
<b>Labor Rate*</b>	<b>\$31.25</b>
<b>Labor Costs</b>	<b>\$69,687.5</b>
<b>Overhead (200%)</b>	<b>\$139,375</b>
<b>Materials</b>	<b>\$3097</b>
<b>Total Cost</b>	<b>\$142,472</b>

\*Assumes yearly salary of \$65,000 for 2080 hours of work



# Acknowledgements

- Dr. Jade Morton
- Dr. Scott Palo
- Dr. Dennis Akos
- Dr. Nicholas Rainville
- Sr. Instr. Trudy Schwartz
- Steve Taylor
- Dr. Kathryn Wingate
- Brian Breitsch
- Lara Buri
- Prof. John Mah



# Questions?

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# Backup Slides

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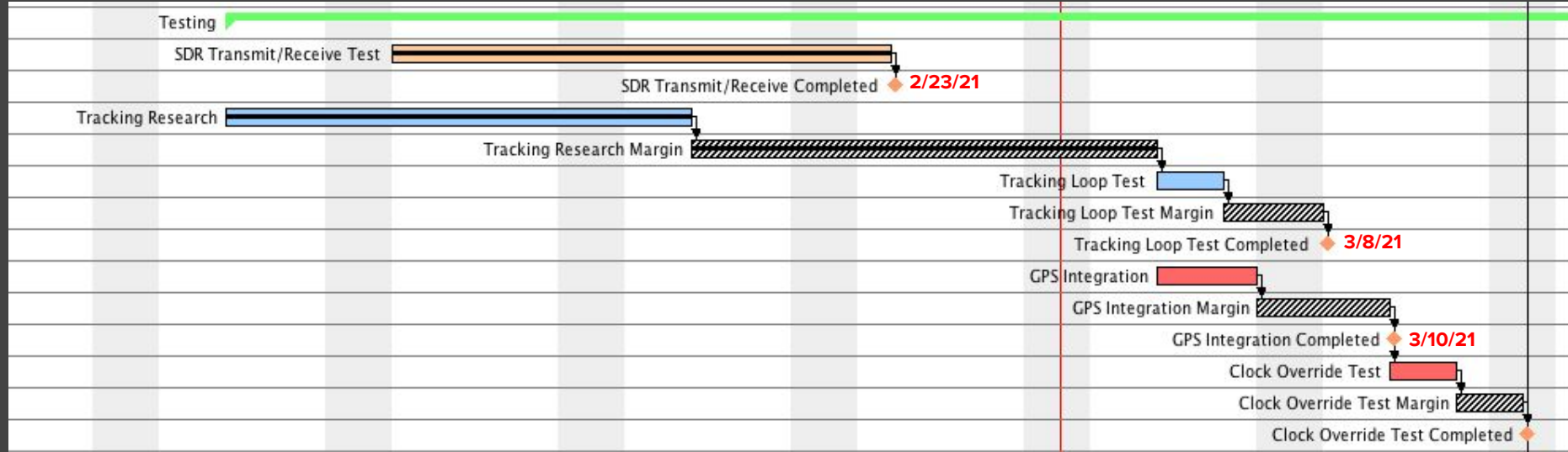


# Budget

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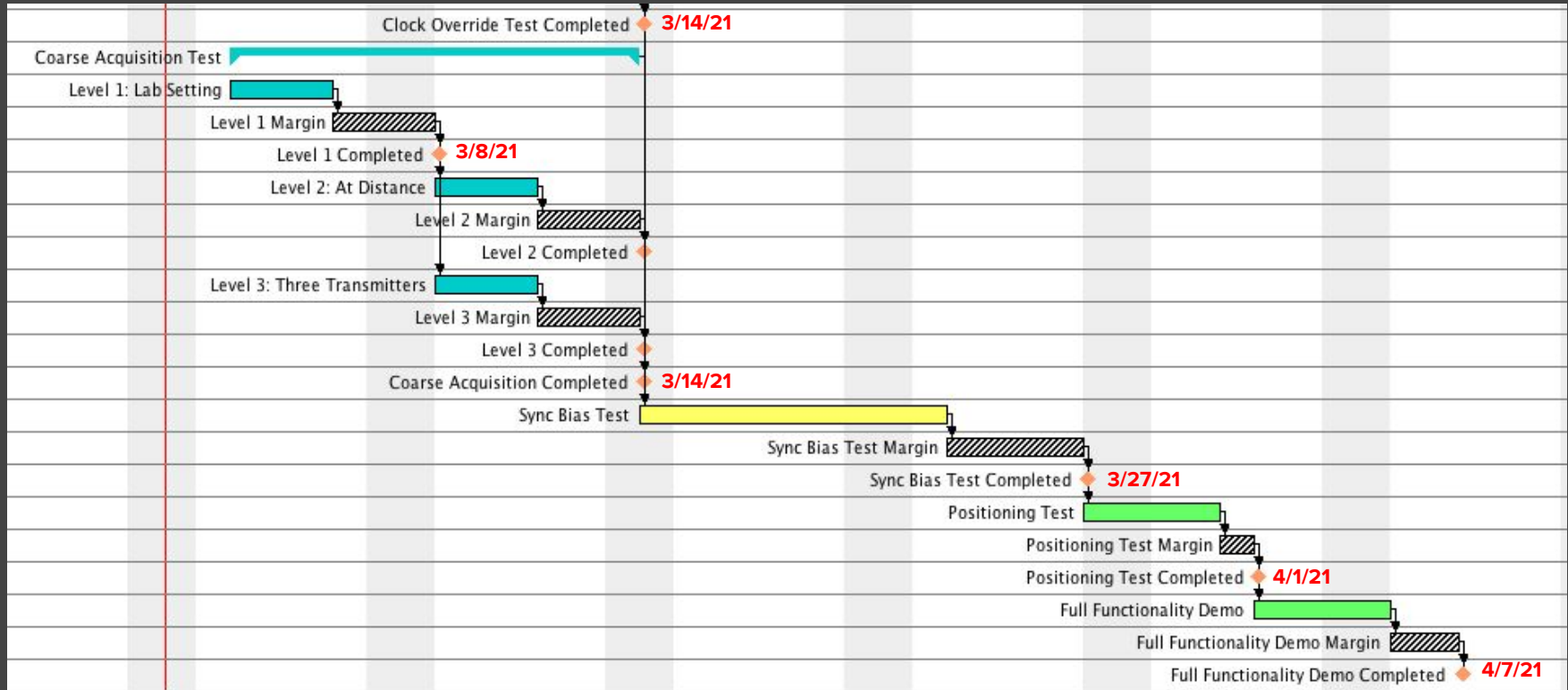


# Schedule



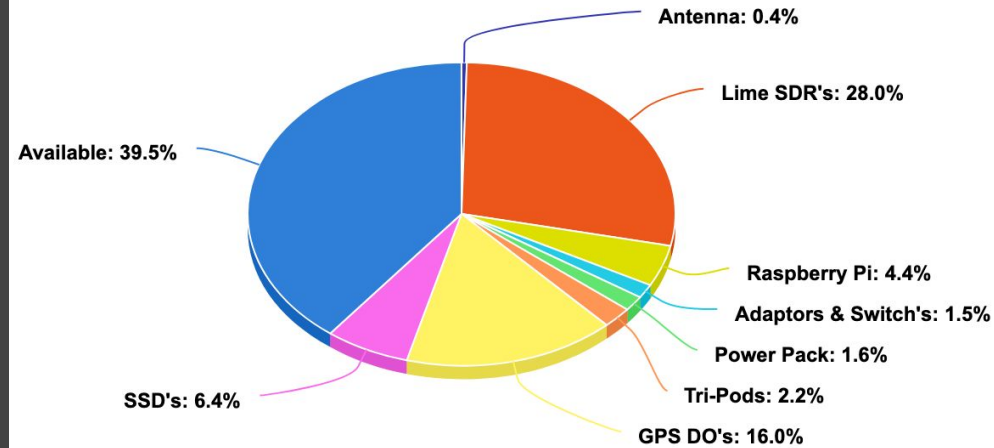


# Schedule



# Cost Plan & Margin

Group	Cost (USD)
Dual TE Connectivity Antennas (4)	\$19.80
Lime SDR's (4)	\$1,399.80
Raspberry Pi (4)	\$220.00
Adapters & Connectors (4 Sets)	\$74.65
Miadi Portable Charger (Power Pack) (4)	\$79.96
5 Feet Tripods (4)	\$111.96
GPS Disciplined Oscillator (4)	\$769.53
Samsung External SSD 500Gb(4)	\$319.96
<b>Budget Status</b>	
Projected Purchases Total	\$3026.12
Current Up to date Purchase Amount Total	\$2862.19
Available Amount / Percentage	\$1973.68 ~ 39.48%





Overview

Test Readiness

Schedule

Budget



## Major Item List & Status

Item	Status
Dual Band TE Connectivity Antenna (4)	In Possession (4)
Lime SDR's (4)	In Possession (4)
Laptops (4)	In Possession (4) Team Member Laptops
GPS Disciplined Oscillator (4)	Shipped From China (4) Max ETA (3/1)
Samsung External SSD 500Gb(4)	In Possession (4)

Obstacles: P-card access, Shipping Logistics & Chinese New Years Celebration



# References

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“Radio Waves Clip Png.” *Pngio*, [pngio.com/PNG/a104378-radio-waves-clip-png.html](https://pngio.com/PNG/a104378-radio-waves-clip-png.html).

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LunaNet: Establishing a Flexible and Extensible Lunar Exploration Communication and Navigation Infrastructure: NASA Goddard Space Flight Center.

D.J.Israele<sup>1</sup>et al., "LunaNet: a Flexible and Extensible Lunar Exploration Communications and Navigation Infrastructure," 2020 IEEE Aerospace Conference, Big Sky, MT, USA, 2020, pp. 1-14. Doi: 10.1109/AERO47225.2020.9172509 URL: <http://ieeexplore.ieee.org/stamp/stamp.jsp?tp=arnumber=9172509&number=9172248>

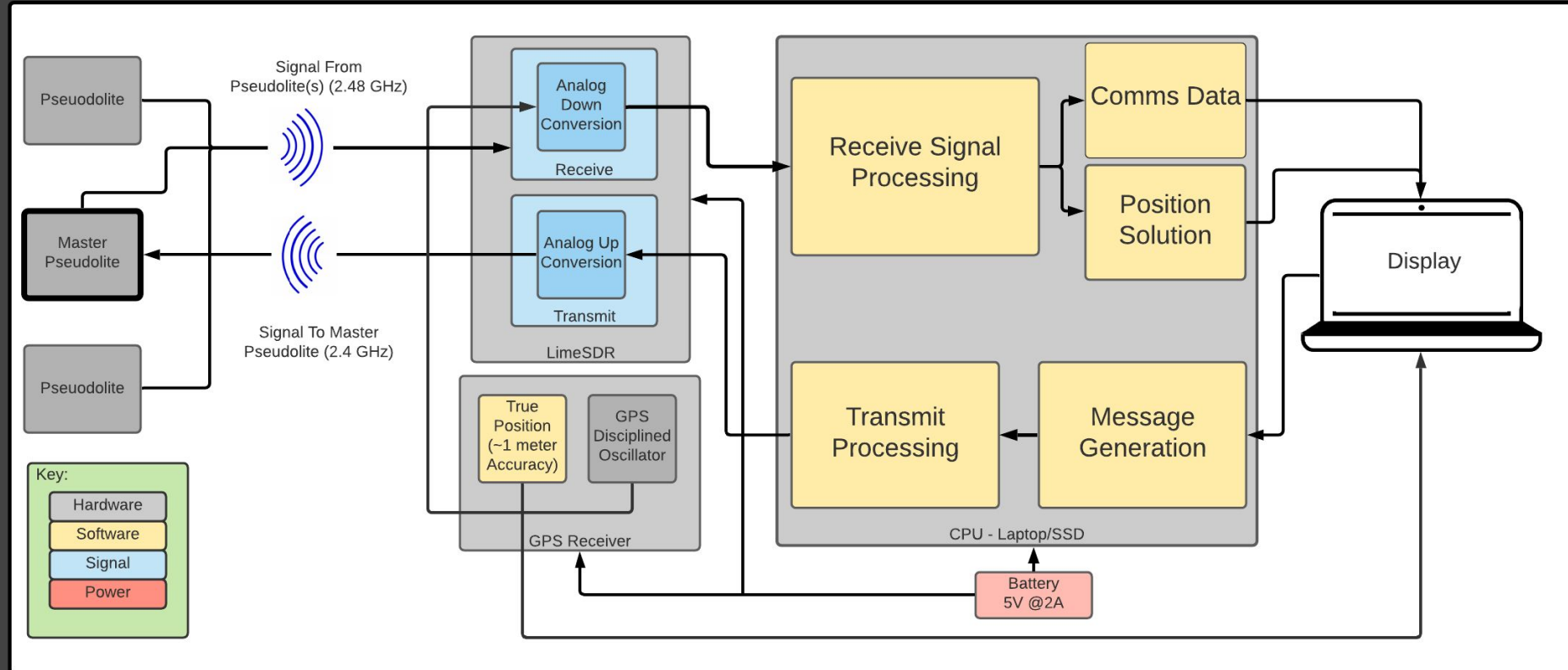
Misra, Pratap, and Per Enge. Global Positioning System: Signals, Measurements, and Performance. Ganga-Jamuna Press, 2012.





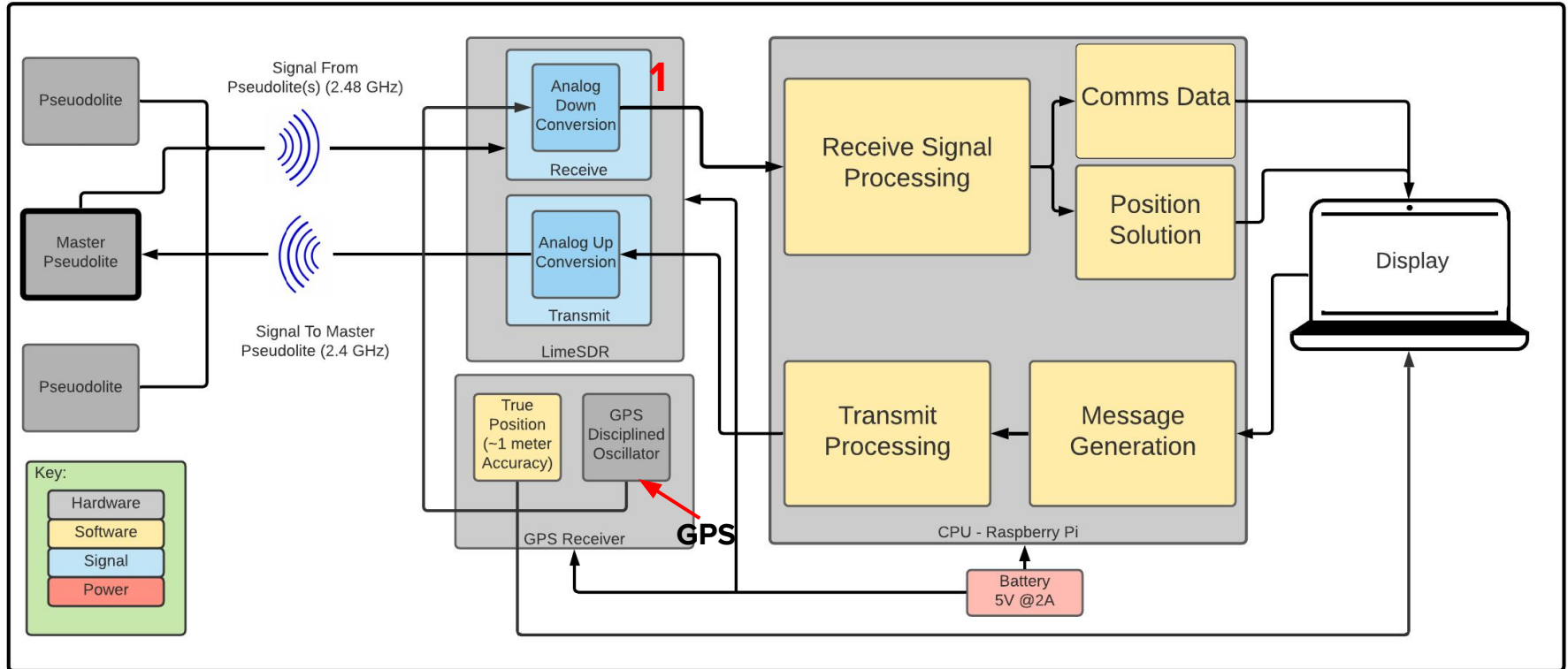
# Intro/Overview

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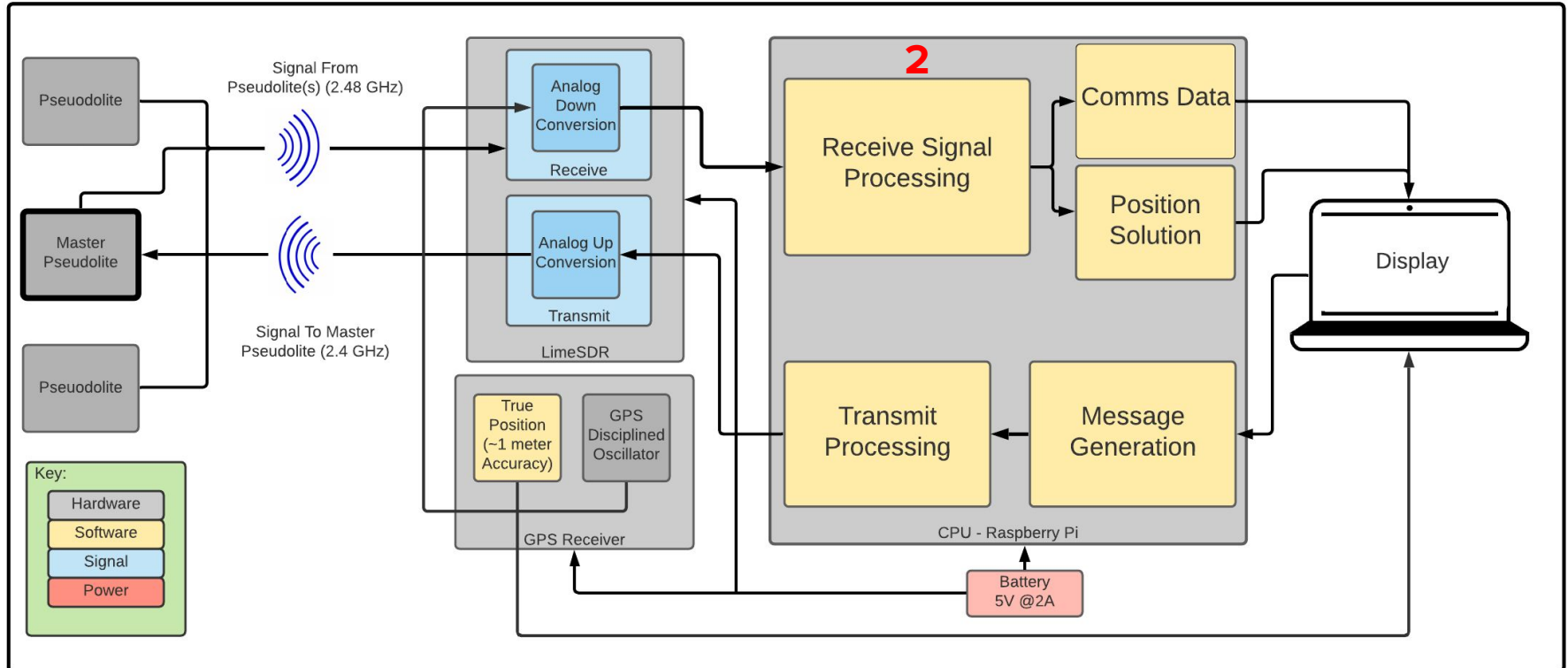




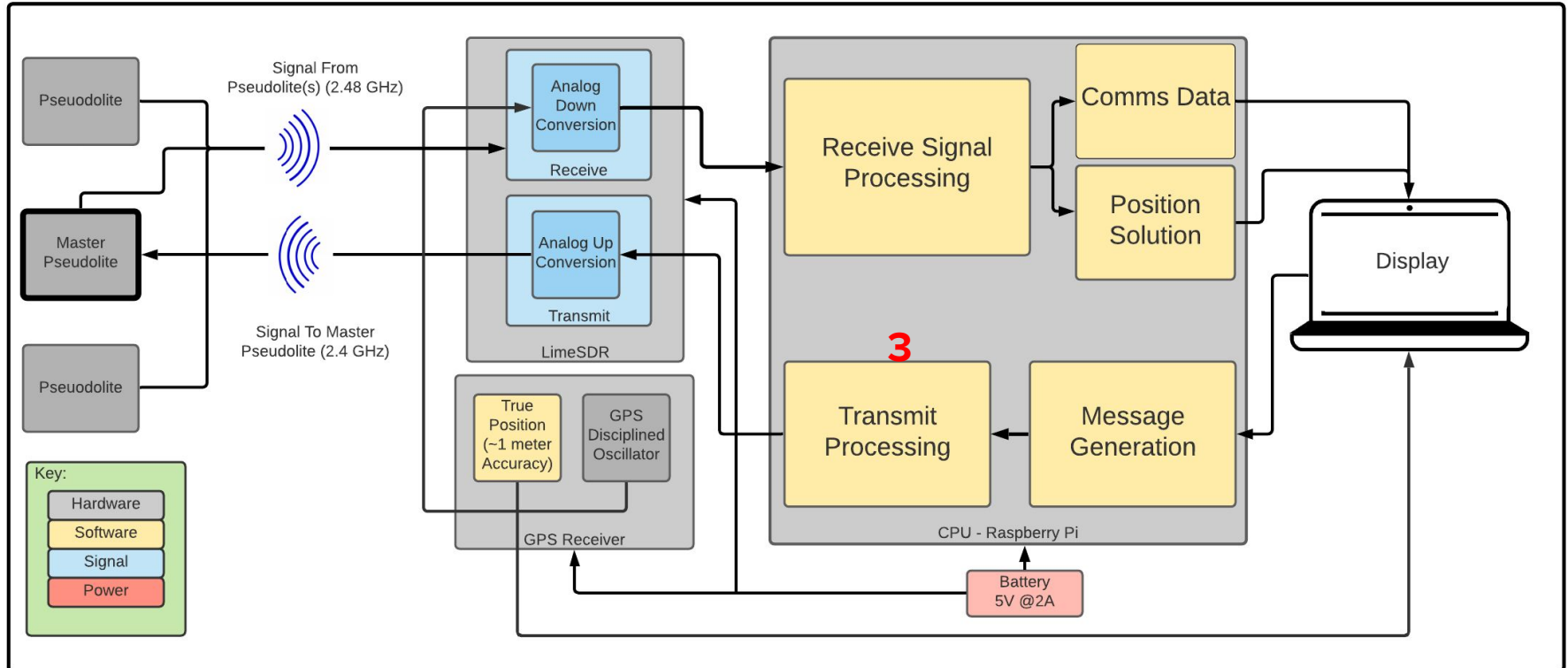
# Overall FBD



# Overall FBD



# Overall FBD





Overview

Schedule

Manufacturing

Budget

# Scaled Down Testing (Placeholder Title)

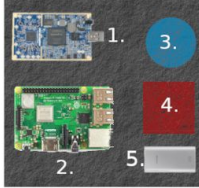
- Scale from 2.4 km to 0.18 km dist. between transmitters
- Assume 2D (only x and y, no height)
- Transmitting on Earth is at least as complicated as on the Moon
  - Atmospheric losses
  - RF Interference
  - Man-made structures/materials interfering
- Not testing environmental constraints

A scaled-down test in Boulder, CO is still able to validate the design for use on the lunar surface

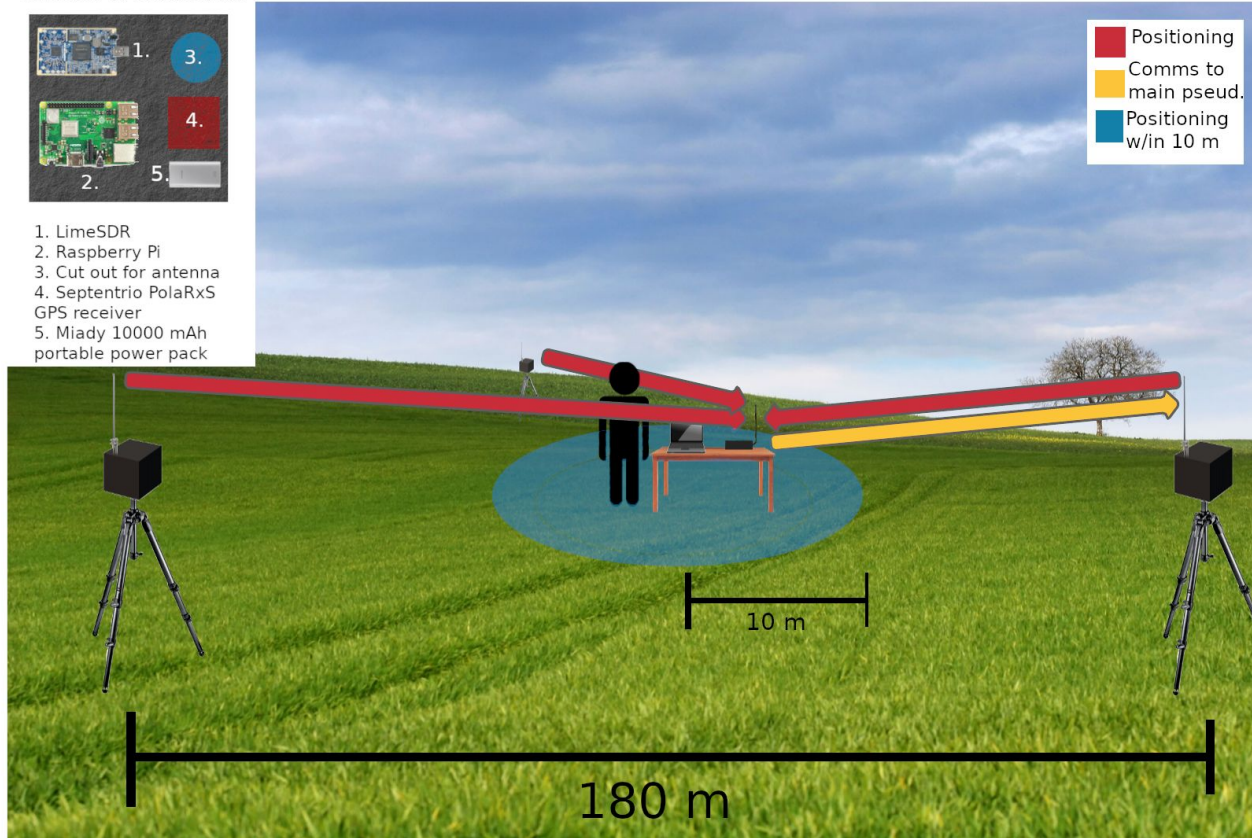


# Design Solution Set Up

Contents of Black Boxes



1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack





[Overview](#)[Solution](#)[CPE](#)[DR](#)[Risk & M](#)[V&V](#)[Planning](#)

# Critical Project Elements

Critical Project Element	Design Solution
Positioning System	One-Way Pseudolite Ranging
Antenna	Dual Band TE-Connectivity Antenna
Communication	BPSK Modulation   CDMA Multiplexing
Software Defined Radio	LimeSDR
Clock	Septentrio PolaRxS - GPS Receiver
CPU	Raspberry Pi



Overview

Schedule

Manufacturing

Budget

# Software Overview

## GNU Radio

- Reference oscillator integration
- Signal reception and transmission
  - Modulation
  - Multiplexing
  - Up/Down Conversion
  - Acquisition
  - Tracking

## Matlab

- Packet Creation
- Positioning algorithm
- For simulation
  - Modulation
  - Multiplexing
  - Up/Down Conversion
  - Acquisition
  - Tracking

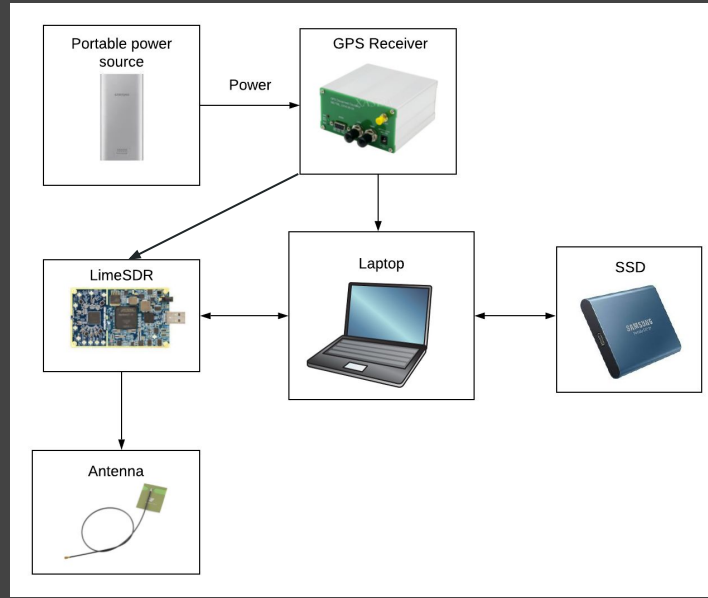
## LimeSDR

- Overwrite clock with reference oscillator
- Integrate antenna

Team Created  
Open Source Edited  
Outside Leveraged

# What is a Pseudolite?

- A portable ground station
  - Transmits positioning messages
  - Receives user communication messages



Pseudolite Hardware Block Diagram





# Positioning Test

- Design:
  - Hardware: Pseudolites, User system
  - Software: Positioning Algorithm
- Test:
  - **Feed range measurements into positioning algorithm.** Determine the position of the user to sub 10m accuracy at testing location
- Status:
  - Positioning algorithm ready to go
  - Need previous tests to be completed first
  - Pseudolites need to be setup

$$\begin{bmatrix} \delta \hat{\mathbf{x}} \\ \delta \hat{b} \end{bmatrix} = (\mathbf{G}^T \mathbf{G})^{-1} \mathbf{G}^T \delta \boldsymbol{\rho}$$

$$\mathbf{G} = \mathbf{A} = \begin{bmatrix} -\frac{x^{(1)} - x_0}{R_0^{(1)}} & -\frac{y^{(1)} - y_0}{R_0^{(1)}} & -\frac{z^{(1)} - z_0}{R_0^{(1)}} & 1 \\ -\frac{x^{(2)} - x_0}{R_0^{(2)}} & -\frac{y^{(2)} - y_0}{R_0^{(2)}} & -\frac{z^{(2)} - z_0}{R_0^{(2)}} & 1 \\ -\frac{x^{(3)} - x_0}{R_0^{(3)}} & -\frac{y^{(3)} - y_0}{R_0^{(3)}} & -\frac{z^{(3)} - z_0}{R_0^{(3)}} & 1 \\ \vdots & \vdots & \vdots & \vdots \\ -\frac{x^{(m)} - x_0}{R_0^{(m)}} & -\frac{y^{(m)} - y_0}{R_0^{(m)}} & -\frac{z^{(m)} - z_0}{R_0^{(m)}} & 1 \end{bmatrix}$$

Geometry Matrix

$$\delta \boldsymbol{\rho} = \mathbf{G} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix}$$



# Test Site Images

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# Manufacturing Slides

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- **Signal Reception: SSD + Laptops instead of Raspberry Pi**
  - Due to computation restrictions, boot Linux off external SSDs connected to laptops for additional speed and memory
- **Signal Transmission: Pseudolites equipped with a reference oscillator**
  - GPS disciplined oscillator (GPSDO)
  - Parts have been selected and ordered.



Overview

Schedule

Test Readiness

Budget

# Customer Value Proposition

- Develop testing scenario
  - Testing out a **scaled down version** of the system
  - What it takes to test the system
  - What works, what doesn't
- Develop/Integrate hardware & software
  - LimeSDR (Radio)
    - Performance metrics
    - External clock integration
  - GNU Radio v3.8 (Software)
    - Performance
    - Interfacing with LimeSDR



Overview

Schedule

Manufacturing

Budget

# Manufacturing Overview

## Purchase

- LimeSDR
- Reference Oscillator
- Antenna
- SSD
- Tripod

## Integrate

- Pseudolite electronics
  - Antenna
  - GNU Radio
  - Laptop
- Pseudolite clock
  - Override internal clock
- Calibrate system

## Develop / Manufacture

- GNU Radio script
- Communications Protocol
- Positioning Algorithm
- Pseudolite Housing



Overview

Schedule

Manufacturing

Budget

# Manufacturing Overview

## Purchase

- LimeSDR
- Reference Oscillator
- Antenna
- SSD
- Tripod

## Integrate

- Pseudolite electronics
  - Antenna
  - GNU Radio
  - Laptop
- **Pseudolite clock**
  - **Override internal clock**
- **Calibrate system**

## Develop / Manufacture

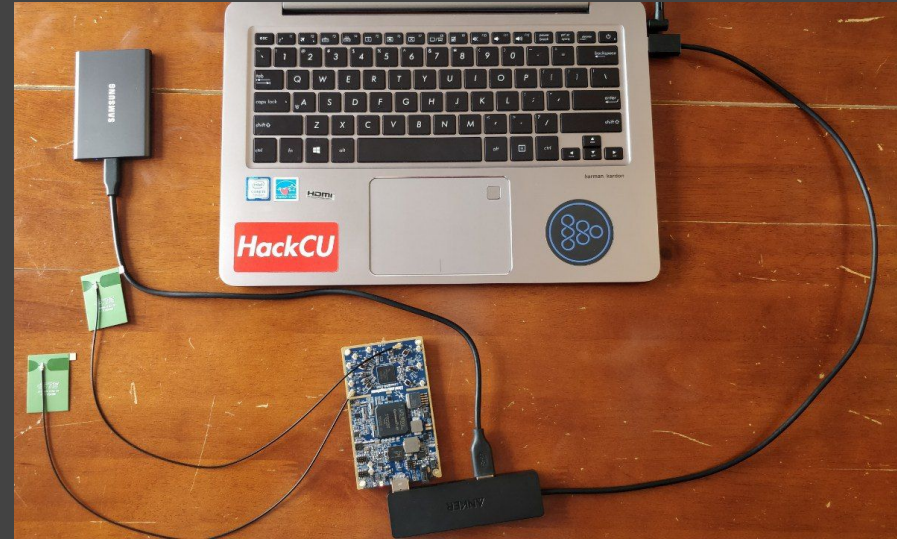
- GNU Radio script
- **Communications Protocol**
- **Positioning Algorithm**
- Pseudolite Housing



# Integrate: Pseudolite Electronics

**Task:** Send and receive test signals using pseudolite electronics

- GNU Radio 3.8 (Software)
- Ubuntu Linux booted on a Samsung T7 500 GB External SSD
  - Running GNURadio
  - 1 GBps read/write
- Dual Band TE Connectivity Patch Antennas
- LimeSDR USB 3.0
- ASUS Zenbook (Core i5 @ 3.5 GHz)
- Write samples to file



**Status:** Done



**Task:** Override internal LimeSDR clock in each pseudolite

- 

+



82



Overview

Schedule

Manufacturing

Budget



# Integrate: Calibrate System

**Task:** Measure and calibrate for total system bias (Pseudolite + GPSDO) when transmitting

**Status:** **Waiting** for overall system integration





Overview

Schedule

Manufacturing

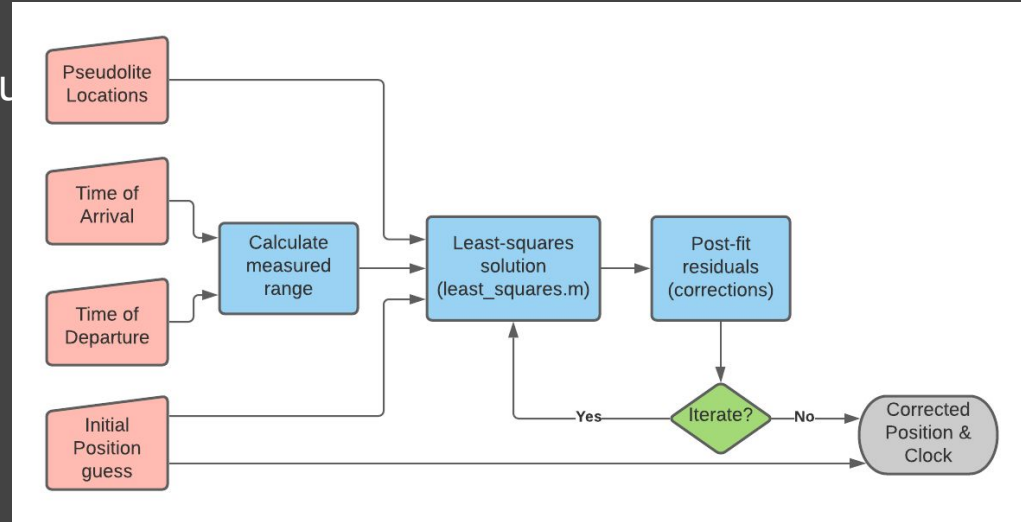
Budget

# Develop: Positioning Algorithm

**Task:** Develop code to process received samples and then compute position.

**Status:**

- Signal processing script **in progress**
- Positioning script **done**



Positioning Script Block Diagram



Overview

Schedule

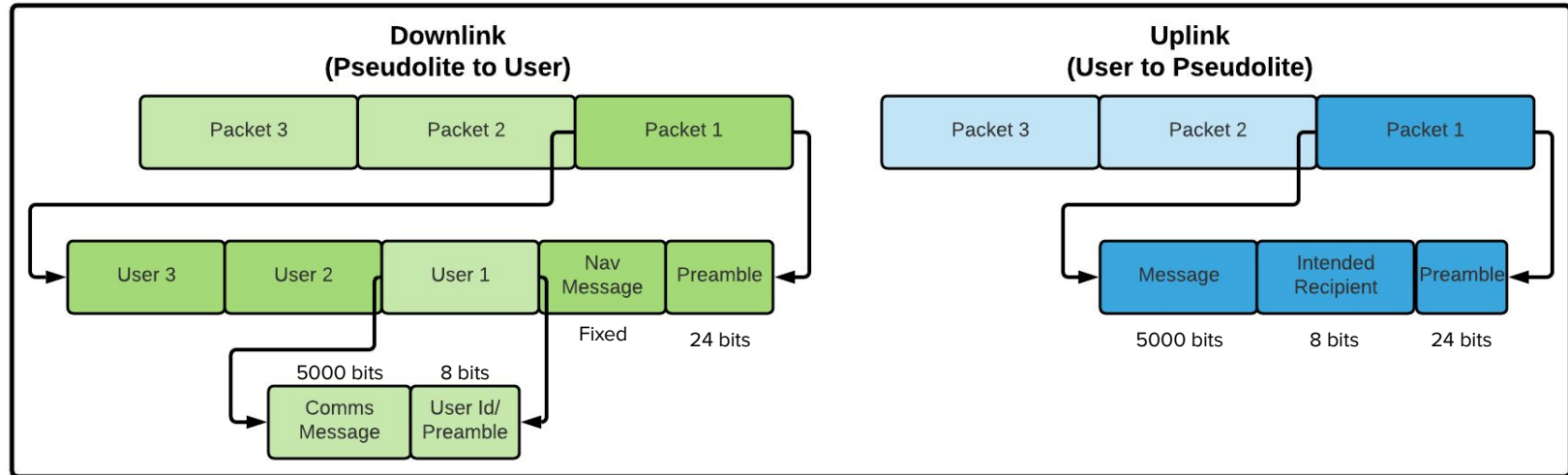
Manufacturing

Budget

# Develop: Communications Protocol

**Task:** Format communication data into packets

**Status:** Done





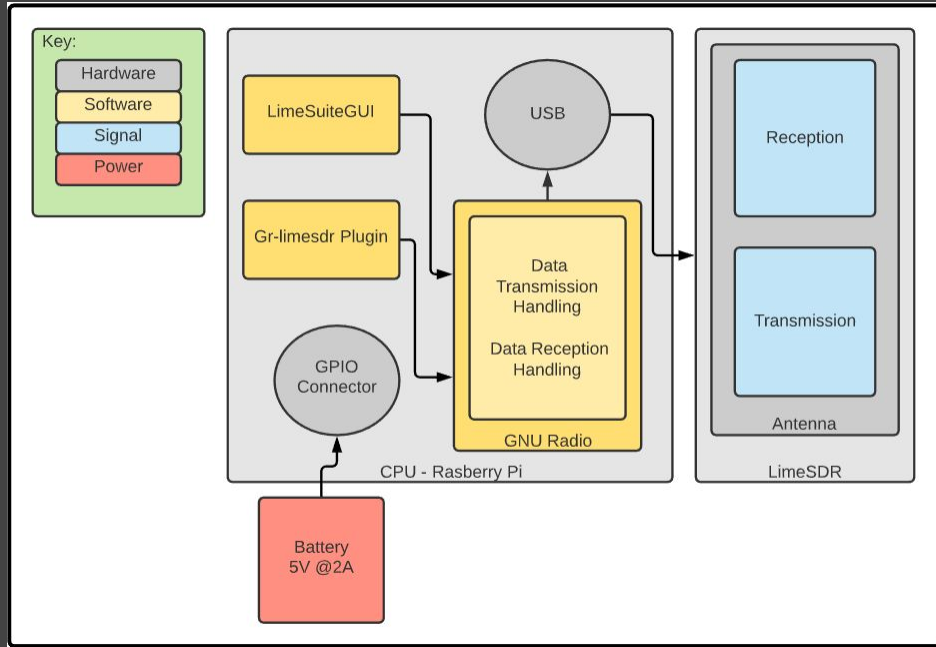
- GNU Radio system setup
- Signal detection between two transmitters
- Measured signal interference at testing location
- Preliminary CAD Housing Model Done
- Selected Referenced Oscillator



# Hardware

---

# Pseudolite Electronic Configuration



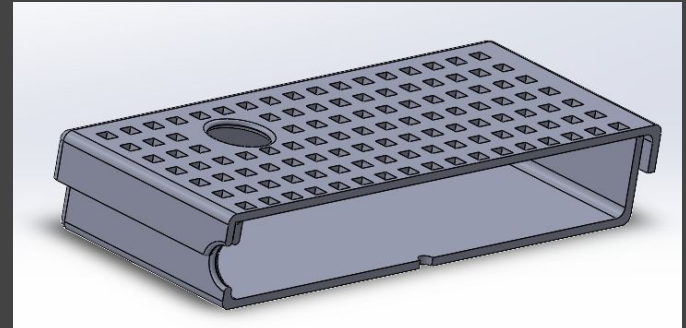
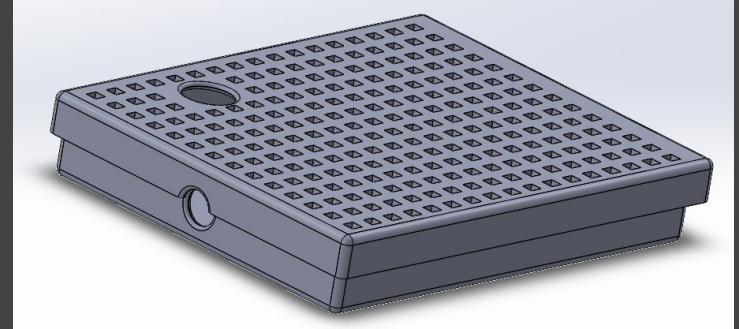
- Electronic/Hardware functional block diagram.
- DR 2.1.3: The system will operate using the LimeSDR electronics device
- The LimeSDR combined with the GNU Radio software package.
- FR1: The system must operate under a scalable LPS model
- How is this scalable?

DR 2.1.3: LimeSDR is in use



# Mechanical Hardware Integration - CAD

- CAD model of case for holding the limeSDR, GPS, power supply, and Raspberry Pi
- Holes for cable, antenna, and ventilation.
- To do: get a tripod so that the model can be made to attach





Overview

Solution

CPE

DR

Risk & M

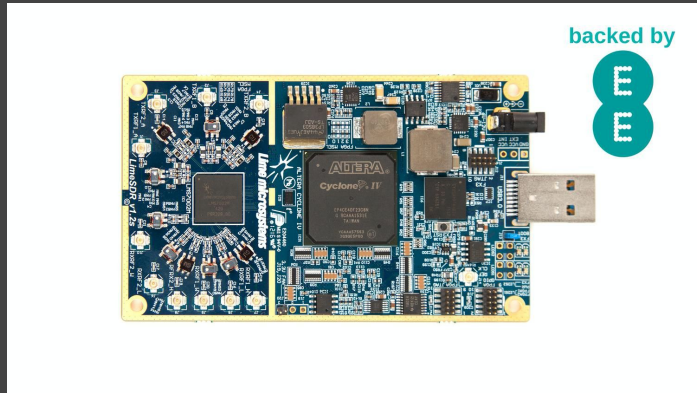
V&V

Planning

# SDR - Requirements

- FR1: The system must operate under a scalable LPS model
  - DR 2.1.3: The system will operate using the LimeSDR electronics device
- FR2: Demonstrate SMS-like communication
  - DR 2.2.1: Device must demonstrate wireless transmission and reception of data
- FR 4: System will transmit and receive data between (2.4 - 2.48 GHz )
  - DR 2.4.1: Demonstrate uplink transmission at 2.4-2.48 GHz
  - DR 2.5.1: Demonstrate downlink transmission at 2.4-2.48 GHz

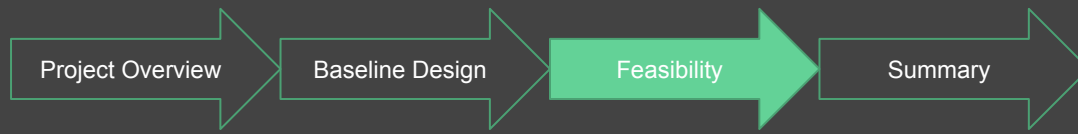
# LimeSDR (SDR - “Software Defined Radio”)



- LimeSDR (Hardware)
  - Customer requirement
  - Reception and Transmission of signal
  - Ability to work with modulator/demodulator software packages
  - Functions at 2.4-2.48 GHz
- GNU Radio (Software)
  - Suggested by customer
  - Able to implement all communication schemes and all frequency ranges
  - Able to do simulation and testing







# Antenna - Requirements

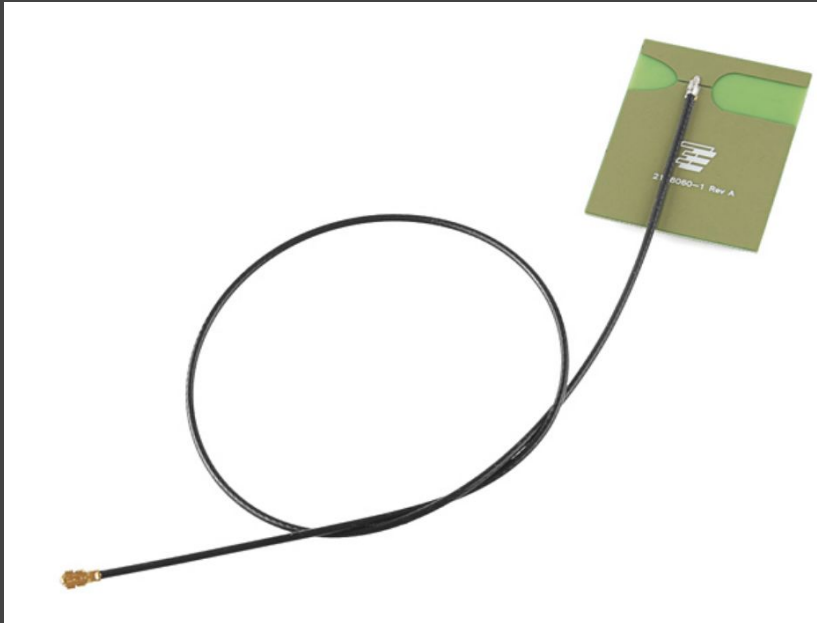
**FR4:** The system will transmit data on the S-Band frequency.

- DR 2.4.1 : Demonstrate uplink transmission at a frequency range between 2.4-2.48 GHz

**FR5:** The system will receive data on the S-Band frequency.

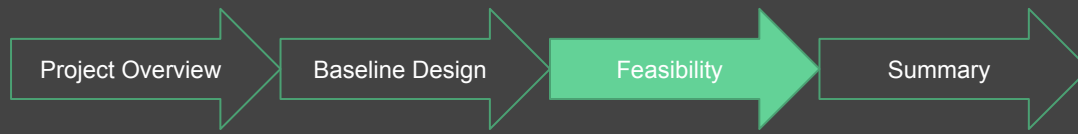
- DR 2.5.1 : Demonstrate downlink reception at a frequency range between 2.4-2.48 GHz

# Dual Band-TE Connectivity Antenna



## Dual Band-TE Connectivity

- Peak Gain 2dBi
- Embedded Antenna Style
- Freq Range: 2.4-3.8GHz
- Low Weight (<3.3gram)
- Low Cost (5 USD)
- Voltage Standing Wave Ratio (VSWR) - MAX 3:1



# Antenna - Selection Analysis

---

Customer Requirement: S-Band (due to radio silent far side of the moon)

Frequency Range	PROS	CONS
S- Band: (2-4 GHz)	<ul style="list-style-type: none"><li>• Previous communication satellites (NASA)</li><li>• Unlicensed bandwidth (2.4-2.483 GHz) good for testing</li><li>• Low Cost system</li></ul>	<ul style="list-style-type: none"><li>• Large amount of interference (Many devices at this bandwidth)</li><li>• Mainly used for large antennas (transmission)</li></ul>



Project Overview

Baseline Design

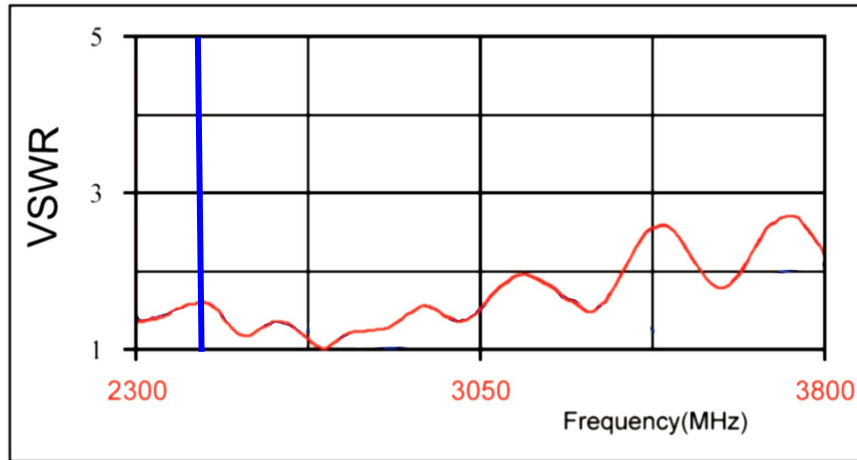
Feasibility

Summary



# Antenna - Evidence of Feasibility

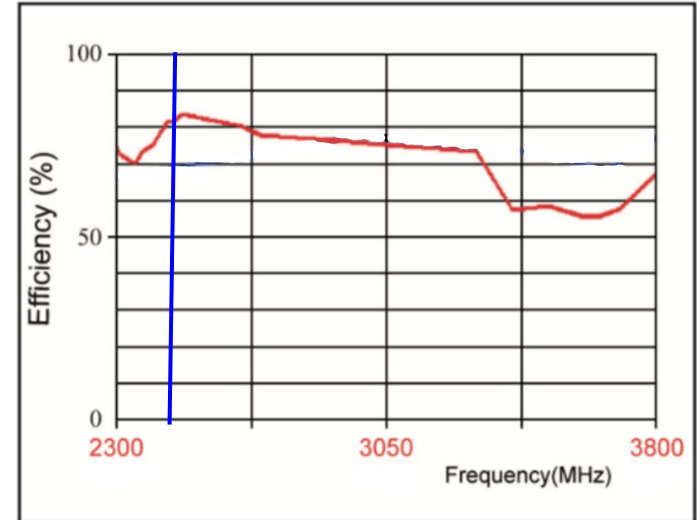
VSWR



DR 2.4.1, 2.5.1

FEASIBLE

Efficiency







Project Overview

Baseline Design

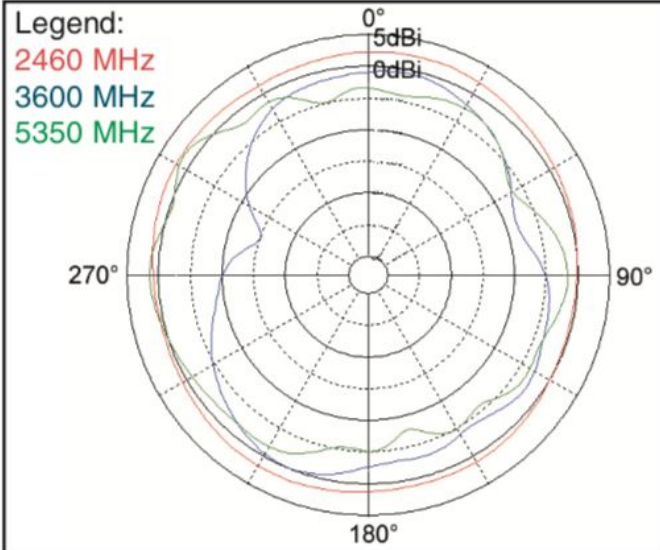
Feasibility

Summary

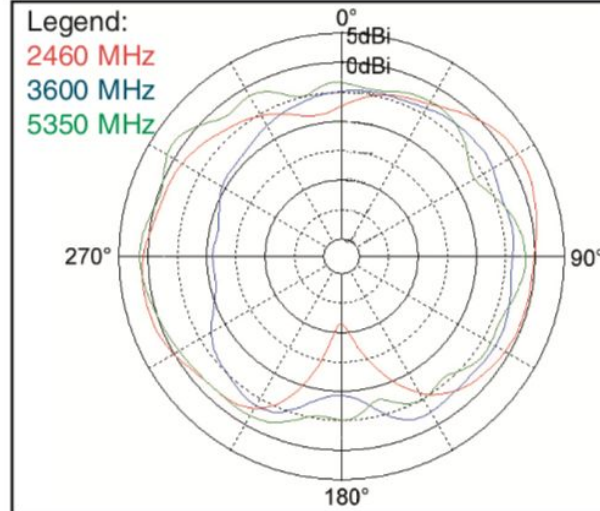


# Antenna - Evidence of Feasibility

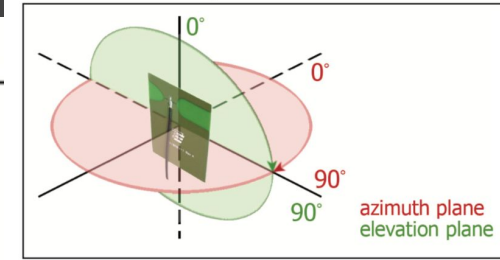
## Azimuth



## Elevation



## Test Orientation in Free Space





## 98

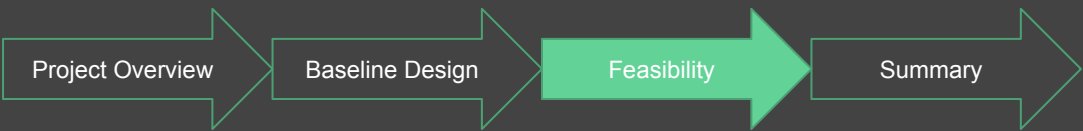
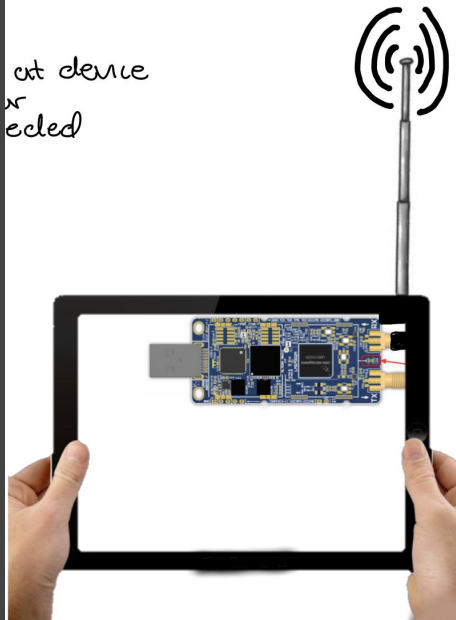


Diagram illustrating an externally mounted antenna system. A tablet holds a circuit board with an antenna labeled "Rx". A long cable, labeled "(10-30m) connecting cable", connects the board to a separate antenna at the top right.

Pros: Antenna weight is not supported by handheld device

Cons: External cable, cannot move antenna

ut device  
r  
ected







# Antenna Tradestudy BU-Slides (Table 1)

Table1.Antenna Options with performance characteristics

Antenna Model	Antenna Type	Antenna Design Configuration	Gain (dBi)	Bandwidth(GHz)
TE Connectivity Antenna	Omnidirectional-DualBand	Embedded	2	2.4-3.8 GHz, 5.150-5.870 GHz
Laird-MAF94051	External Dual-Band Omnidirectional	Attached	2	2.4-2.5
Laird OC24006H	Omnidirectional/ horizontally polarized	External	6	2.4-2.5
Argain-N2420M	Single Band embedded	Embedded	2.81	2.4-2.49



# Antenna Tradestudy BU-Slides (Table 2)

Table2.Antenna and other characteristics

Antenna Model	Weight (grams)	Unit-Cost (USD Currency)	Connector Type	Effective Area (Ae) or Effective Antenna Aperture (m <sup>2</sup> )
TE Connectivity	3.3	4.79	Cable-Side Entry	0.001828
Laird-MAF94051	113.4	7.64	RP SMA Connector with 90 degree elbow	0.001814
Laird OC24006H	260	50.20	N-Female	0.004556
Argain-N2420M	0.5	1.80	IPEX/MHF/U FL	0.002186





Table3. Point Matrix Criteria

Metric	High Score (3)	Medium Score(2)	Low Score(1)
Cost Effectiveness	low cost for a high performance and capabilities (price<5)	The antenna has an average cost for its capabilities(5-30)	The antenna has a high cost for its capabilities (Over 30)
Weight	The antenna is very light (Under 50 grams)	The antenna has a medium weight (from 50-200grams)	The antennas has a high weight value (over 200 grams)
Compatibility	The antenna can easily connect to the SDR, The user has no issues when using the antenna	The antenna is compatible with the sdr without any extra hardware but needs intervention for it to start working	The antenna is not compatible with the sdr without any extra hardware and needs intervention for it to start functioning
Performance/Specs	Omnidirectional, Operates in designated frequency, high bandwidth, good materials, Large Effective Area,etc.	The antenna has all the required specs but does not have good materials or other hardware constraints	The antenna doesn't satisfy all the needed requirements





# Antenna TradeStudy BU-Slides (Table 4)

Metric	Weighting	Antenna #1 TE	Antenna #2 Laird-1	Antenna #3 Laird-2	Antenna #4 Argain
Cost	0.15	3	2	1	3
Weight	0.30	3	2	1	3
Compatibility	0.15	3	3	2	3
Performance/Specs	0.40	3	3	3	2
Total Score	100%	100%	80%	65%	90%



# Power Budget

Component	Max Operating Power (Transmit)	Max Operating Power (Receive)
LimeSDR	4.5W (More like 2mW)	NA
Antenna	NA	3.0W
Raspberry Pi	NA	3.5W
Total	4.5W	6.5W
	or approx 2mW	

Battery Pack: Miady  
- 5V 2A spread  
evenly from two  
outputs



# Power Pack (Selected Battery)

## Miady 10000mAh Dual USB Portable Charger





# Scaling and Power

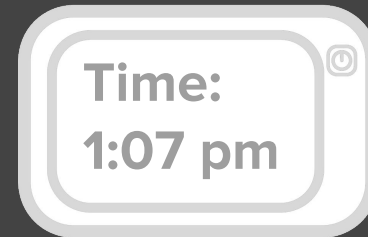
- Scaling
  - In the real-life application of this project we will use 2.4km side-lengths for the pseudolite geometry
    - Calculated using the horizon of the Moon.
    - Would also need to take into account any terrain on the Moon.
  - **Scaled down geometry** was created for **practical** reasons.
- Power
  - Current testing power is **2 mW** which is **more than enough** (even for a pseudolite configuration with 2.4km side-lengths)
  - As part of our testing we will **vary transmit power** and plot ranging accuracy vs. SNR in order to determine the **optimal transmit power**.



# Pseudolite Clocks

**FR3:** All clocks must be within **30 ns** of each other **for the duration of a test (2 hr)**

- On-board Clock (TCXO): Drifts on average 0.05ns every second
- Pseudolite Clocks: **GPS Disciplined**
  - Using clock bias to keep pseudolite clock synced
  - Reduces jitter
  - Septentrio PolaRxS
- **Expected Accuracy: 0.5 ns**





# Navigation

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Overview

Solution

CPE

DR

Risk & M

V&V

Planning



# Positioning System Requirements

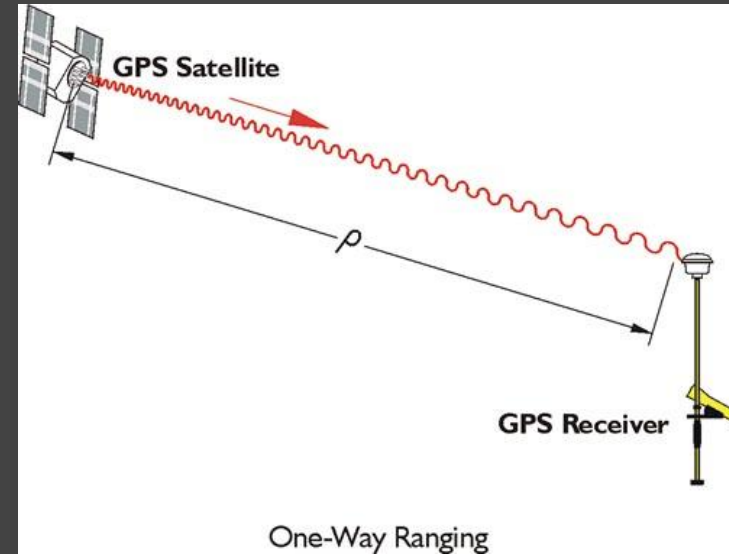
**FR3:** The system will be able to provide an architecture for a navigation solution with a 10 meter position accuracy and a 30 nanosecond 1- $\sigma$  transfer time

- **DR 2.3.2** : Provide LPS coverage to the Moon's South Pole

**Solution:** Pseudolite geometry composed of a 0.18km equilateral triangle + GPS-disciplined clocks

# One way ranging

- Requires accurate clocks on both ends to determine the distance based on time differences.
- Lots of research/documentation available supporting the algorithms and systems.
- Only satellite transmits.
- The energy and time required is low due to only one transmission of data.







- Requires clocks on both ends but does need them to be synced up with each other.
- Receiver must transmit data BACK to the transmitter - multiple times.
- **Both the satellite and the receiver know the location of the satellite.**
- The energy and processing requirements are high due to the fast data transmission requirements.



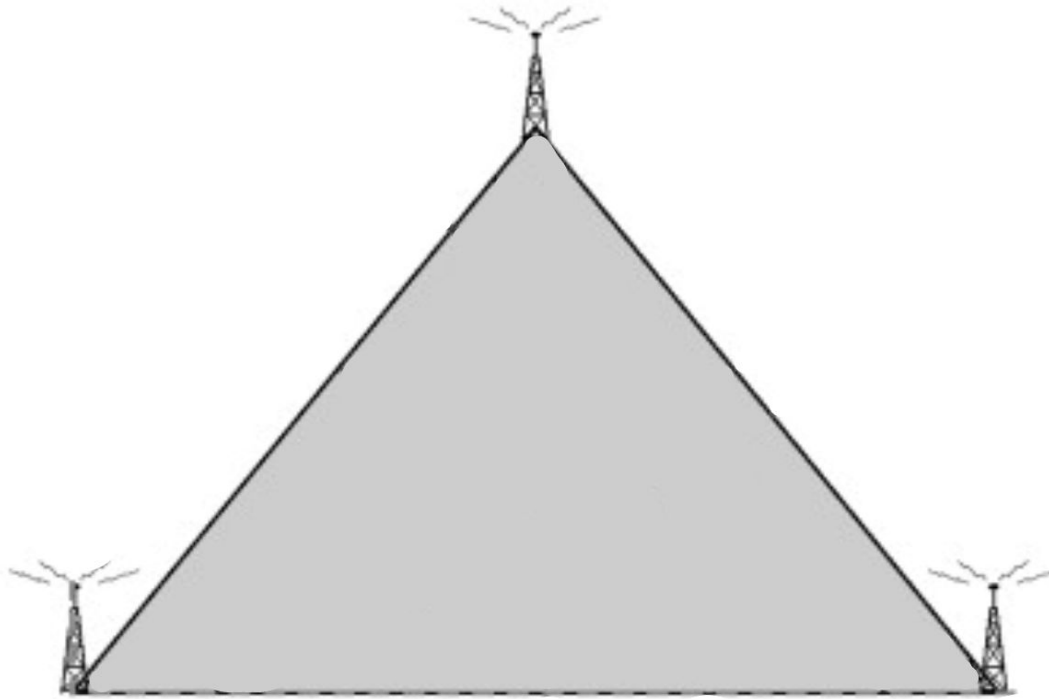


# Positioning Algorithm: Least-Squares Solution

- Reason for using least-squares method
- Position Guess and corrections

$$\begin{aligned}
 \mathbf{G} = \mathbf{A} = & \begin{bmatrix} -\frac{x^{(1)} - x_0}{R_0^{(1)}} & -\frac{y^{(1)} - y_0}{R_0^{(1)}} & -\frac{z^{(1)} - z_0}{R_0^{(1)}} & 1 \\ -\frac{x^{(2)} - x_0}{R_0^{(2)}} & -\frac{y^{(2)} - y_0}{R_0^{(2)}} & -\frac{z^{(2)} - z_0}{R_0^{(2)}} & 1 \\ -\frac{x^{(3)} - x_0}{R_0^{(3)}} & -\frac{y^{(3)} - y_0}{R_0^{(3)}} & -\frac{z^{(3)} - z_0}{R_0^{(3)}} & 1 \\ \vdots & \vdots & \vdots & \vdots \\ -\frac{x^{(m)} - x_0}{R_0^{(m)}} & -\frac{y^{(m)} - y_0}{R_0^{(m)}} & -\frac{z^{(m)} - z_0}{R_0^{(m)}} & 1 \end{bmatrix} \\
 \text{Geometry Matrix} & \\
 \delta \boldsymbol{\rho} = \mathbf{G} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix} &
 \end{aligned}$$

$$\begin{bmatrix} \delta \hat{\mathbf{x}} \\ \delta \hat{b} \end{bmatrix} = (\mathbf{G}^T \mathbf{G})^{-1} \mathbf{G}^T \delta \boldsymbol{\rho}$$



Overview

Solution

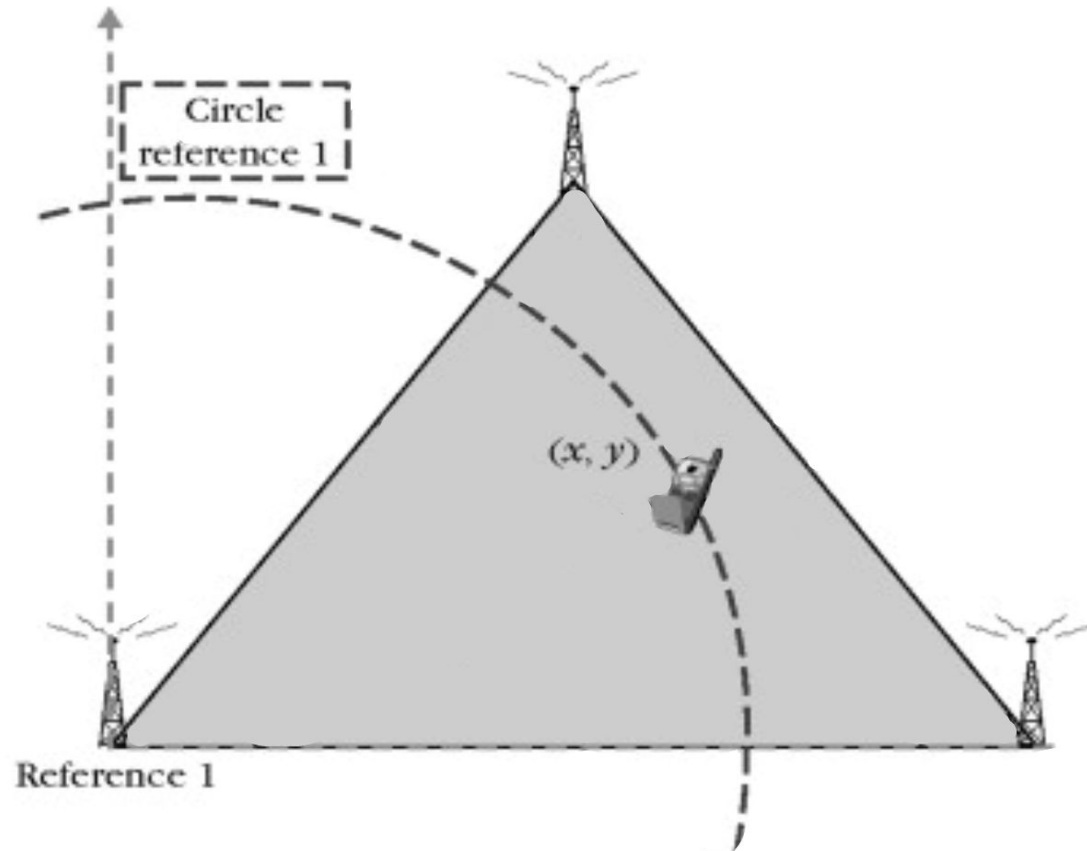
CPE

DR

Risk & M

V&V

Planning



Overview

Solution

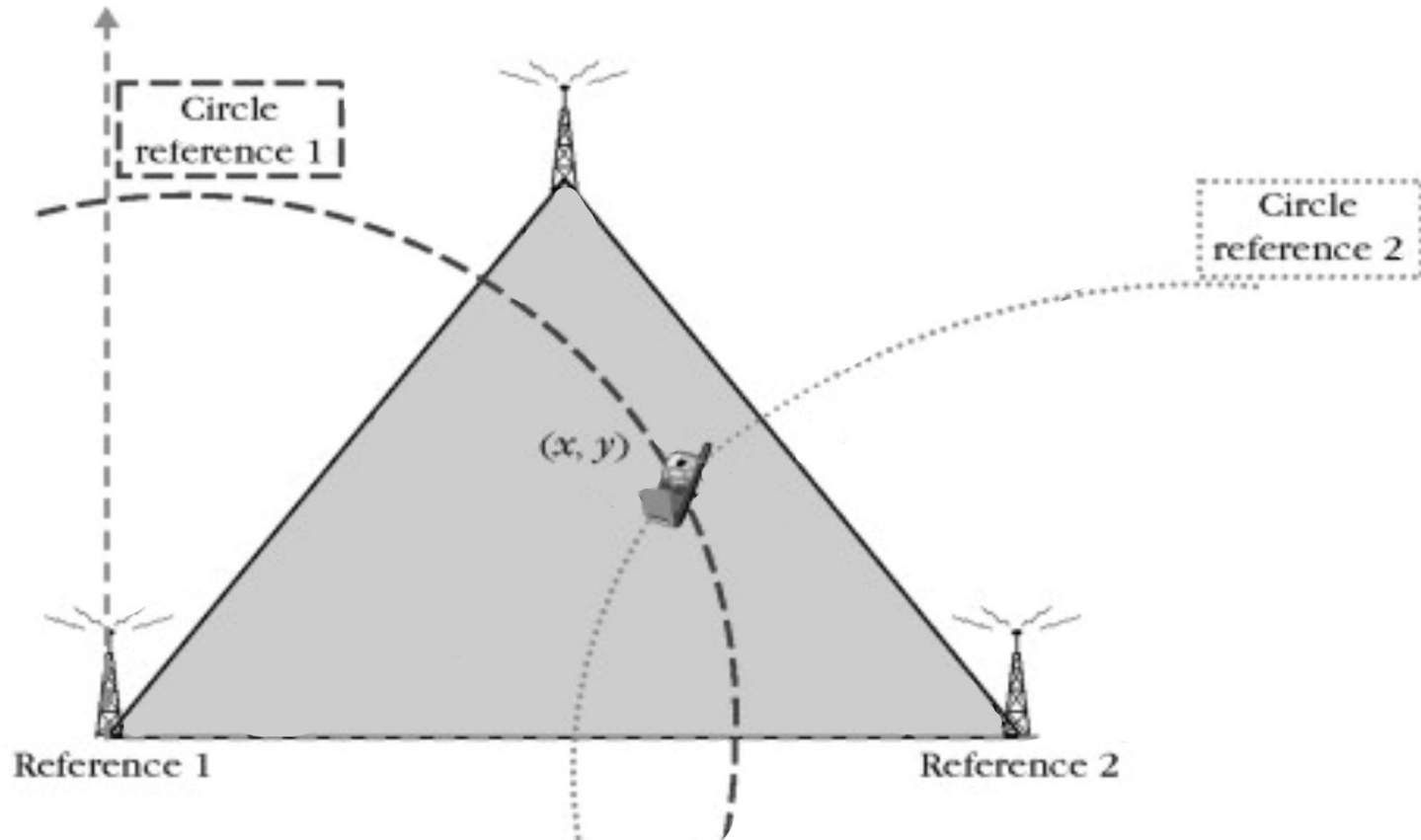
CPE

DR

Risk & M

V&V

Planning



Overview

Solution

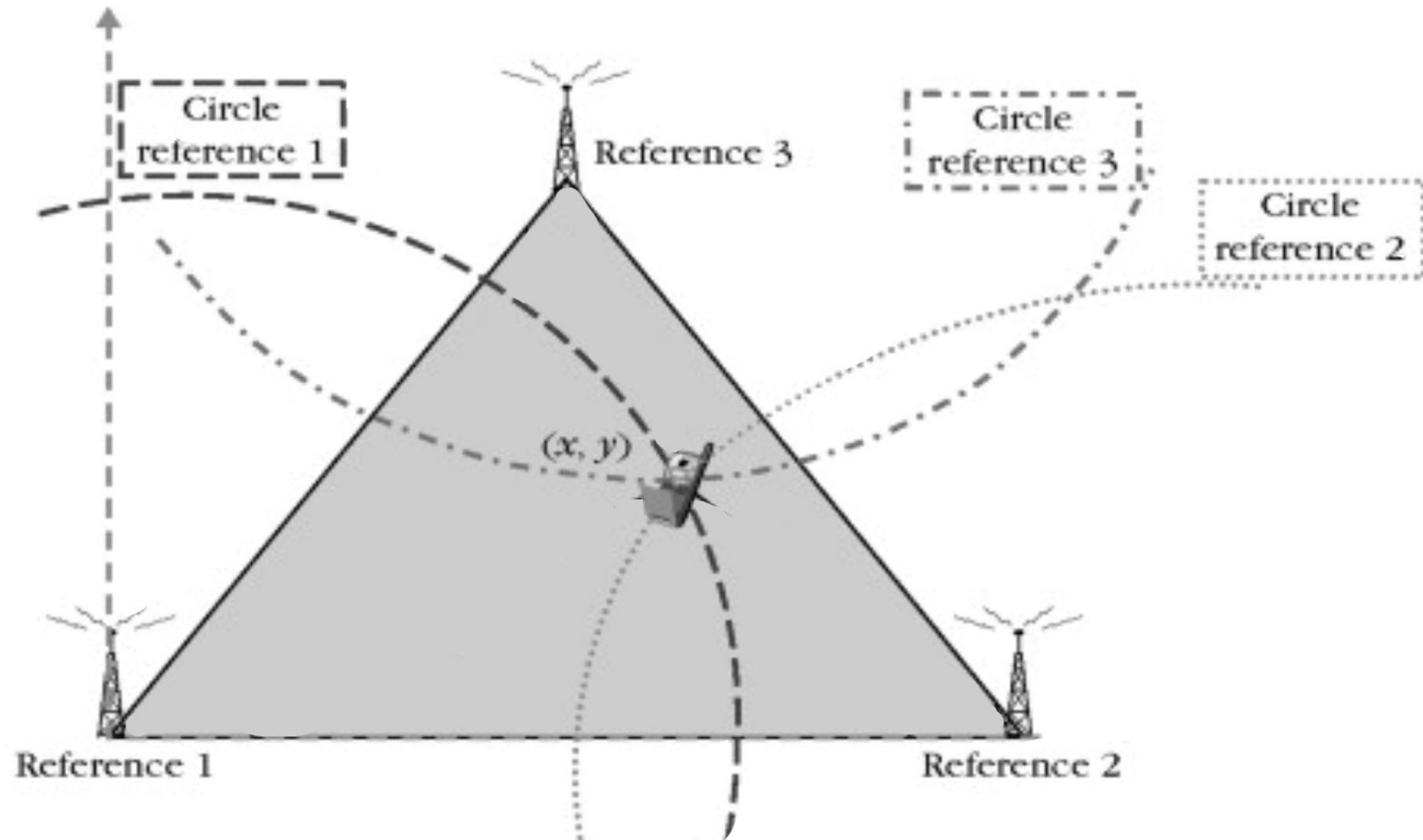
CPE

DR

Risk & M

V&V

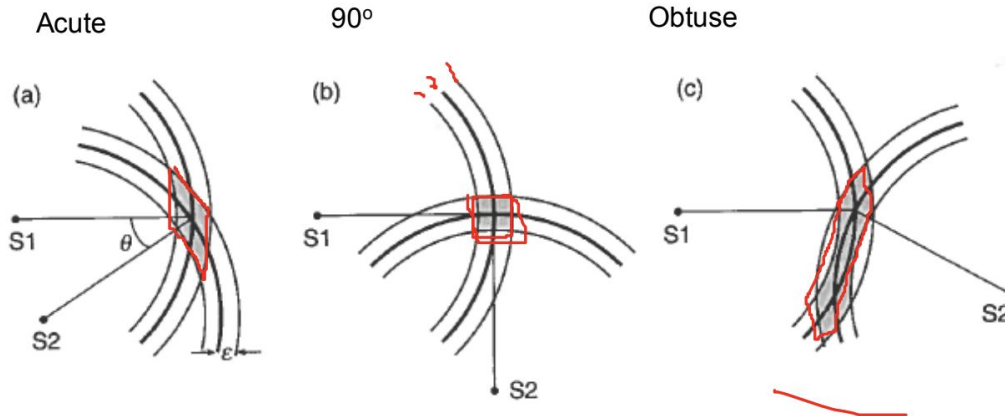
Planning



# Positioning System - HDOP

## Horizontal Dilution Of Precision

Definition: “Used to specify error propagation as a mathematical effect of navigation pseudolite geometry on positional measurement precision”.



$$\text{RMS horizontal error} = \sqrt{\sigma_E^2 + \sigma_N^2} = \sigma \cdot \text{HDOP} \quad (6.2)$$

$$\text{RMS vertical error} = \sigma_U = \sigma \cdot \text{VDOP} \quad (6.2)$$

$$\text{RMS 3-D error} = \sqrt{\sigma_E^2 + \sigma_N^2 + \sigma_U^2} = \sqrt{\sigma_x^2 + \sigma_y^2 + \sigma_z^2} = \sigma \cdot \text{PDOP} \quad (6.2)$$





# Positioning System - HDOP

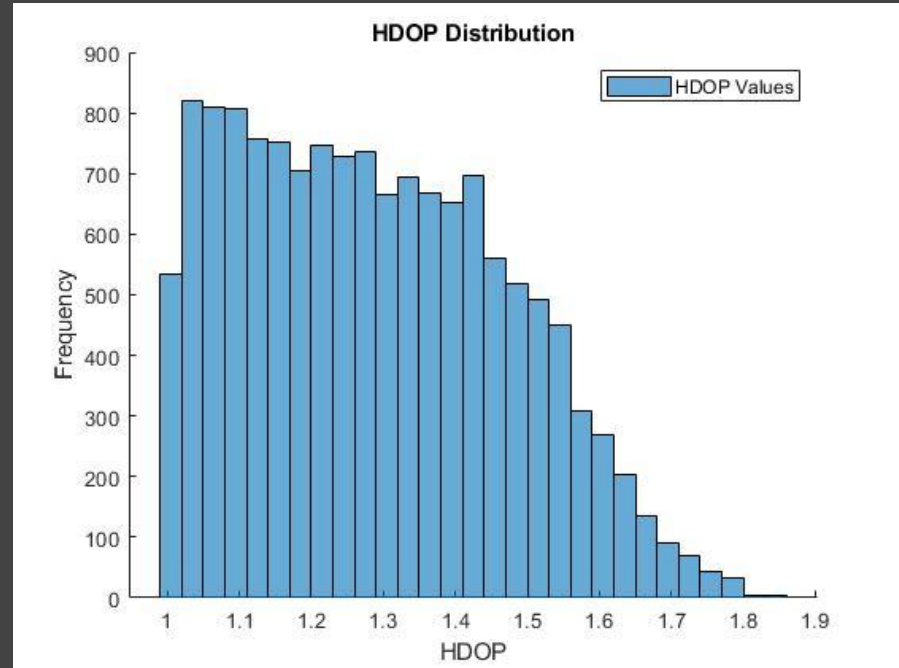
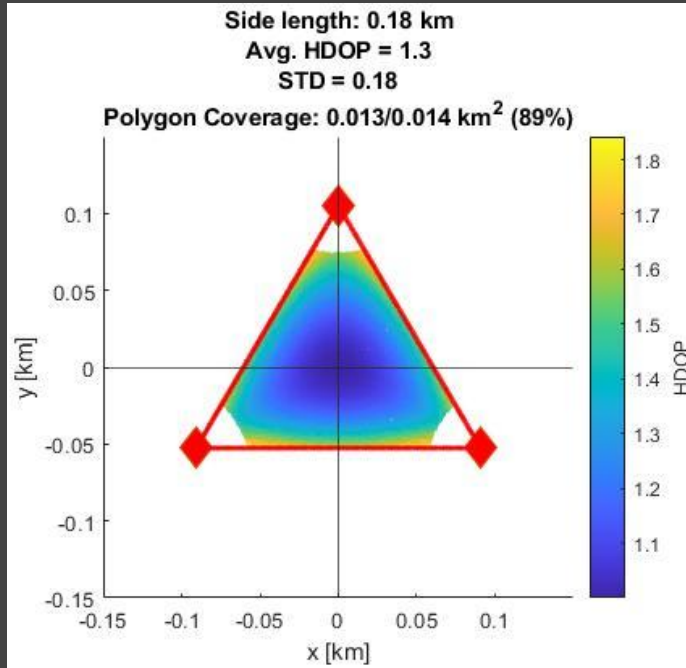
The DOPs provide a simple characterization of the user-satellite geometry. The more favorable the geometry, the lower the DOP. The lower the DOP and  $\sigma$ , the better the quality of the position estimate, in general. If a receiver is limited in the number of satellites it can track simultaneously, the user pays a price with larger rms position error due to higher DOP values.

$$\text{RMS horizontal error} = \sqrt{\sigma_E^2 + \sigma_N^2} = \sigma \cdot \text{HDOP} \quad (6.27a)$$

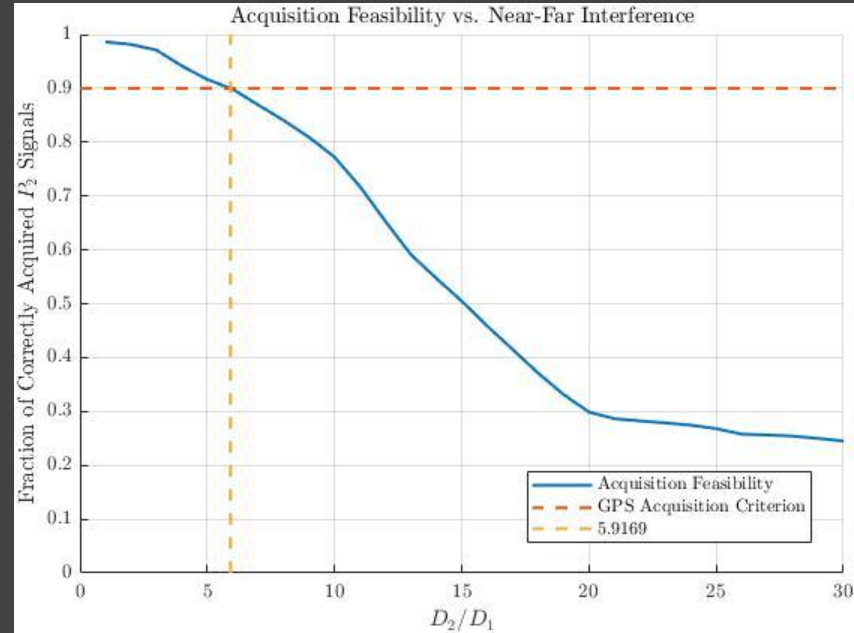
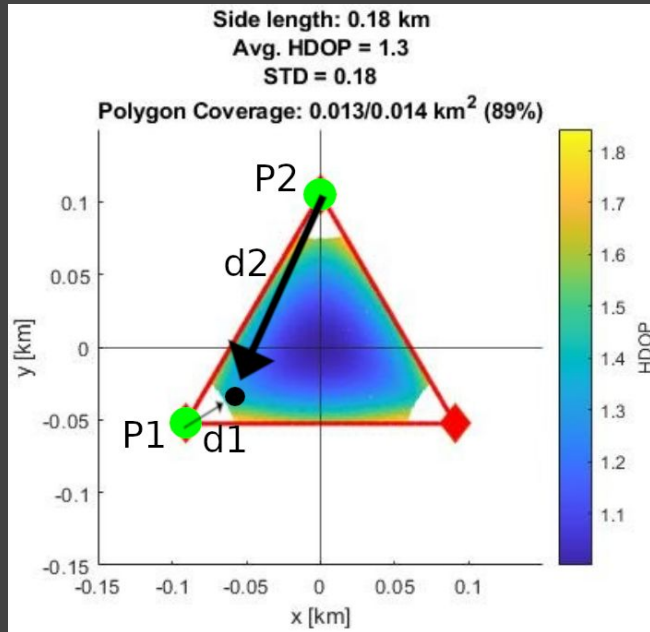
$$\text{RMS vertical error} = \sigma_U = \sigma \cdot \text{VDOP} \quad (6.27b)$$

$$\text{RMS 3-D error} = \sqrt{\sigma_E^2 + \sigma_N^2 + \sigma_U^2} = \sqrt{\sigma_x^2 + \sigma_y^2 + \sigma_z^2} = \sigma \cdot \text{PDOP} \quad (6.27c)$$

# Pseudolite Geometry

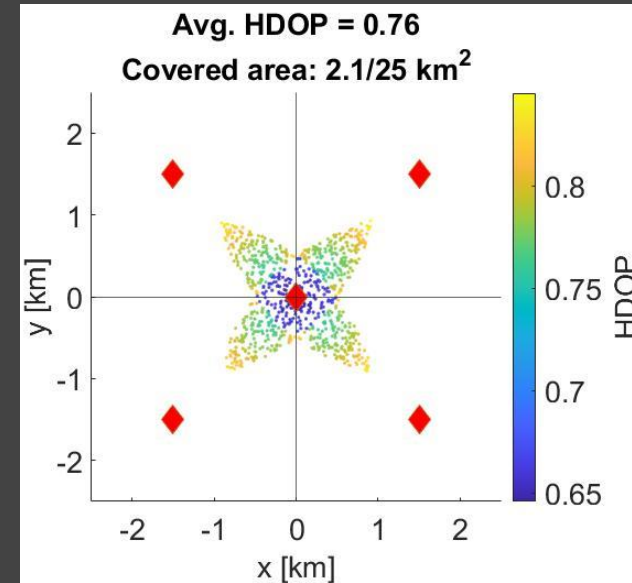
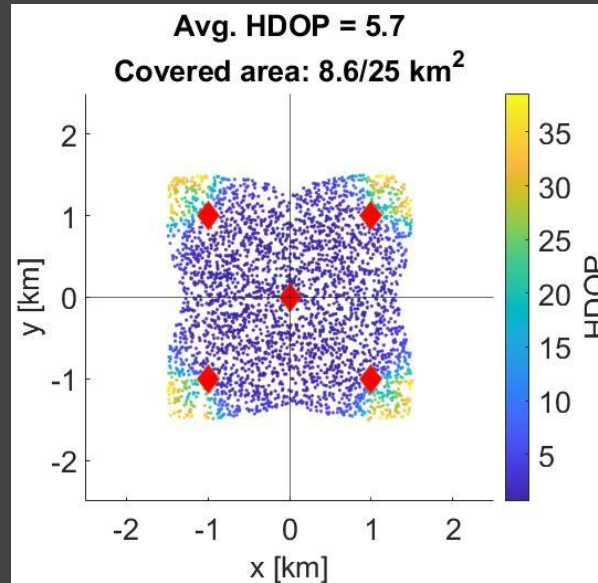
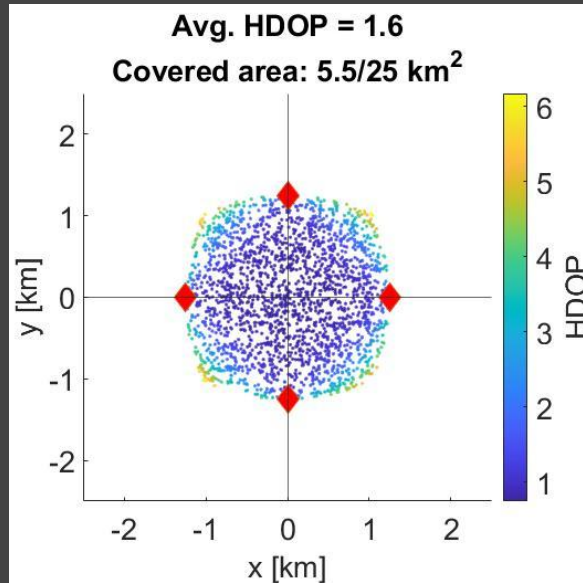


# Near-Far Problem



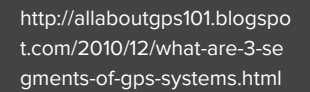


# Positioning System - Evidence of Feasibility

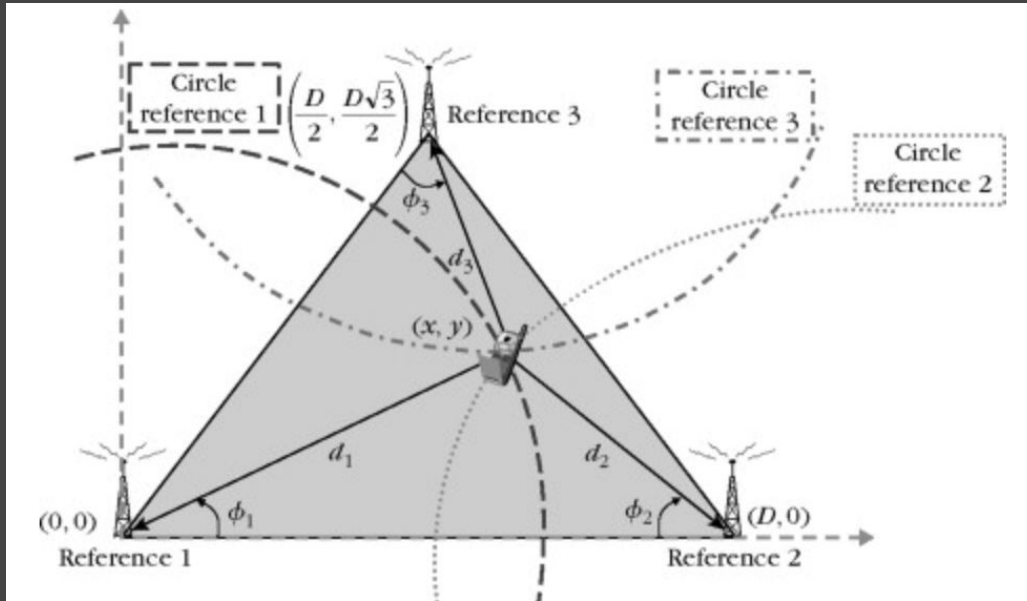


## DR 2.3.3

FEASIBLE



# Time of Arrival (TOA)



$$d_1 = c(t_1 - t_0) = \sqrt{x^2 + y^2}$$

$$d_2 = c(t_2 - t_0) = \sqrt{(D-x)^2 + y^2}$$

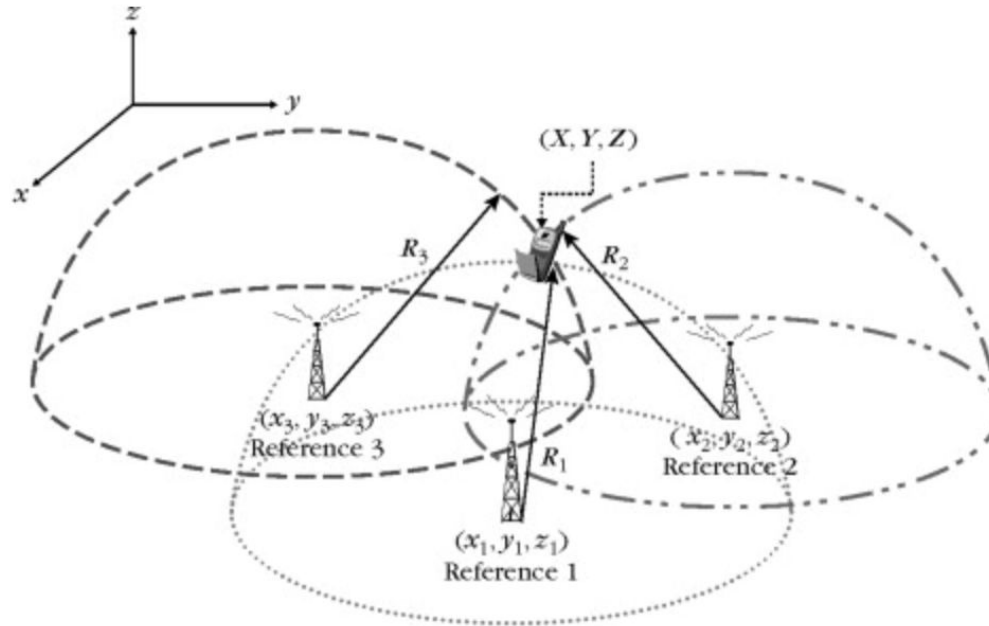
$$d_3 = c(t_3 - t_0) = \frac{1}{2}\sqrt{(D-2x)^2 + (D\sqrt{3}-2y)^2}$$

$t_i$ : TOA at reference  $i$ ;  $c$ : speed of light



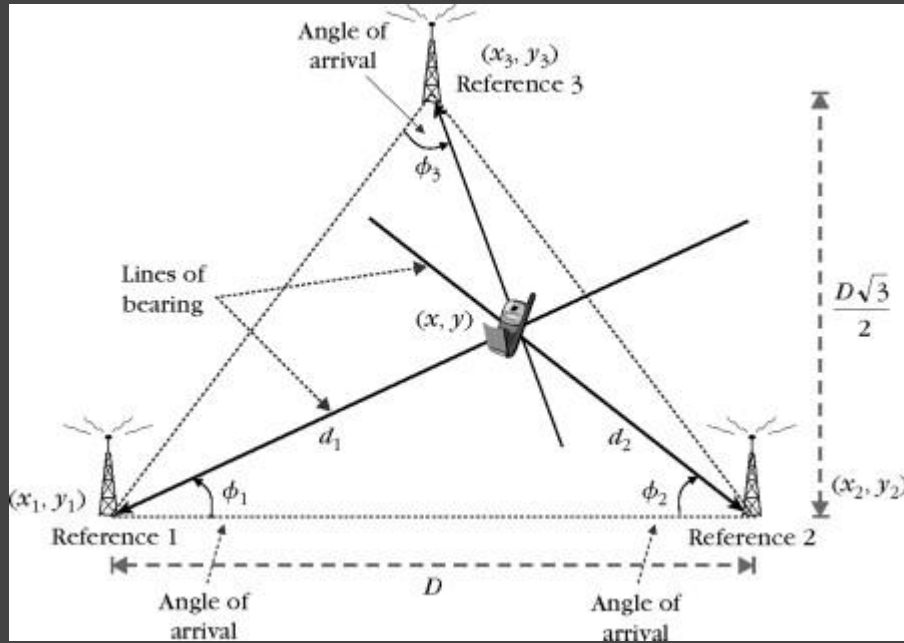
# Time of Arrival (TOA)

$$R_i^2 = (X - x_i)^2 + (Y - y_i)^2 + (Z - z_i)^2, i = 1, 2, 3.$$





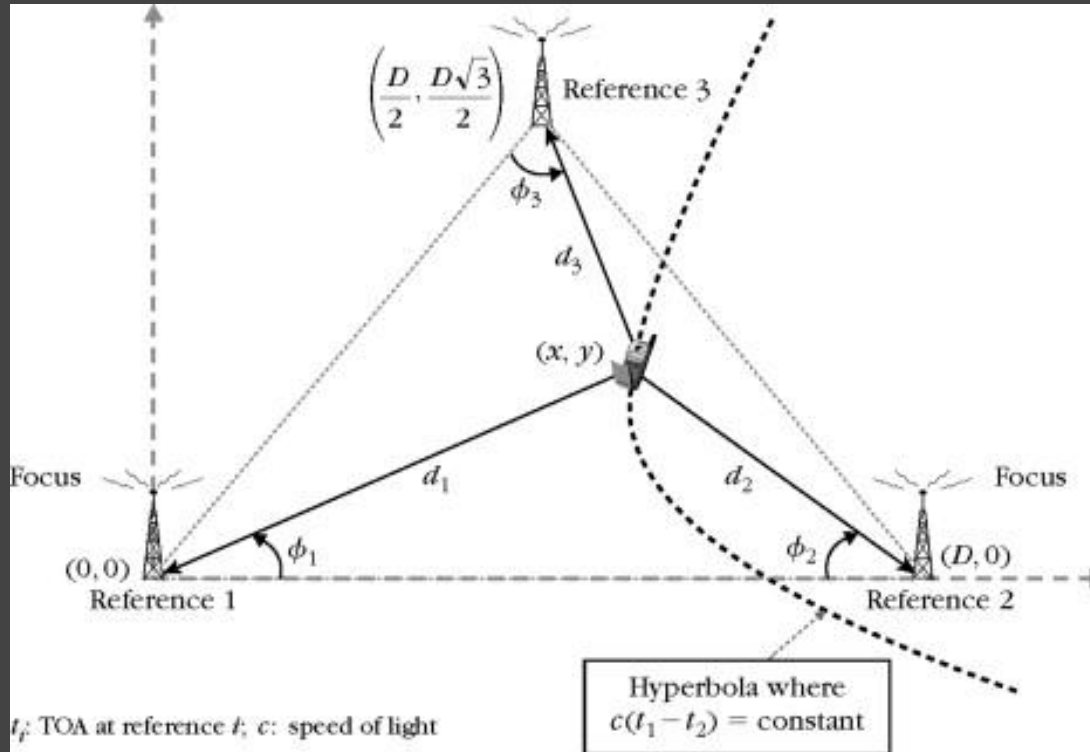
# Angle of Arrival (AOA)



$$x = d_i \cos(\phi_i) + x_i,$$

$$y = d_i \sin(\phi_i) + y_i, i = 1, 2, 3.$$

# Time Difference of Arrival (TDOA)





# Time Difference of Arrival (TDOA)

$$d_{ij} = d_i - d_j = c(t_i - t_o) - c(t_j - t_o) = c(t_i - t_j), i = 1, 2, 3, j = 1, 2, 3, i \neq j.$$

$$d_i = d_{ij} + d_j;$$

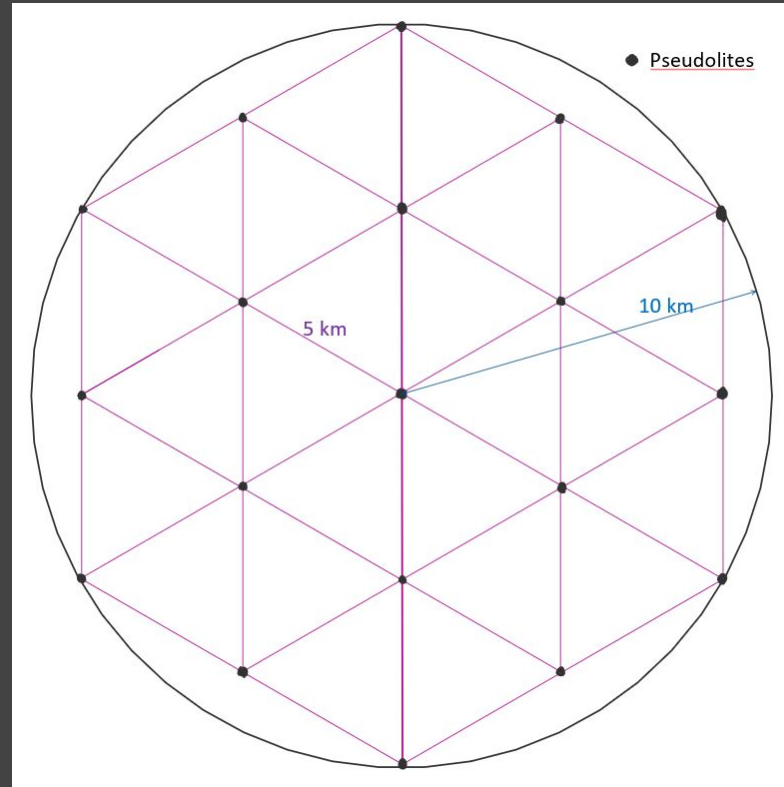
$$\begin{aligned} d_2^2 &= (d_{21} + d_1)^2 \\ &= (x_2 - x)^2 + (y_2 - y)^2 \\ &= x_2^2 - 2x_2x + x^2 + y_2^2 - 2y_2y + y^2 \\ &= x_2^2 - 2x_2x + y_2^2 - 2y_2y + d_1^2, \end{aligned}$$

$$d_1^2 = x^2 + y^2.$$

$$(d_{21}^2 - x_2^2 - y_2^2) + 2d_{21}d_1 = -2x_2x - 2y_2y,$$

$$(d_{31}^2 - x_3^2 - y_3^2) + 2d_{31}d_1 = -2x_3x - 2y_3y.$$

# Lunar Pseudolite Geometry



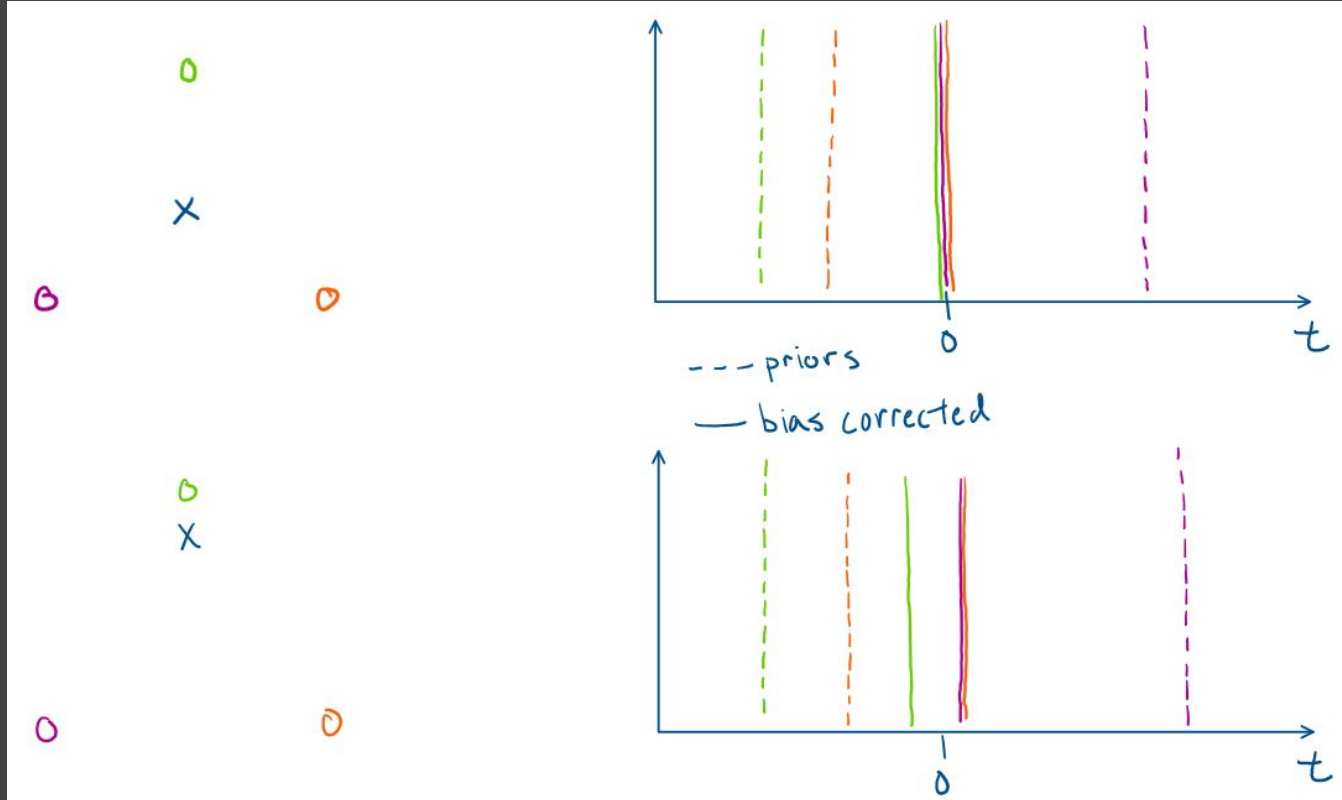


## Tentative Trade/Analysis Aspects

- 133

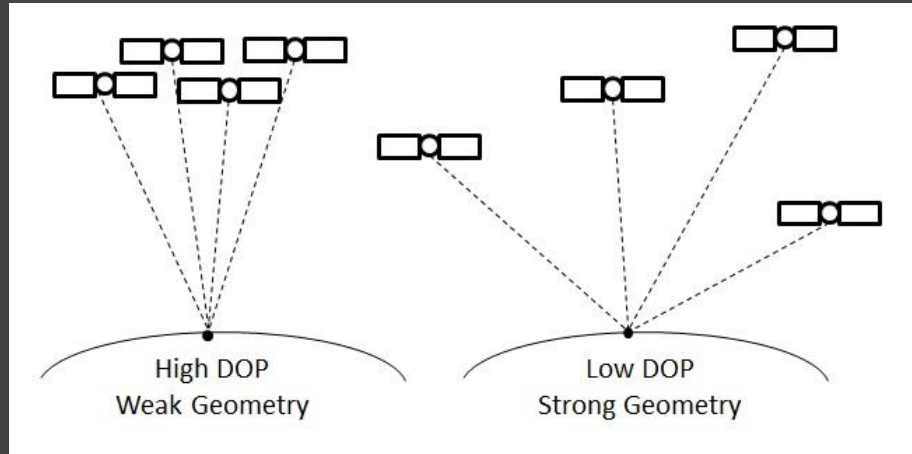


# Basic Positioning System



# GNSS: Error sources

- Error sources can be divided into two categories:
  - Ranging error: signal quality, errors in transmitter location, environmental effects, receiver design, etc....
  - Geometry: Geometric Dilution of Precision (GDOP). Measures the quality of the geometric distribution of the satellites visible to the receiver.



<https://www.polyu.edu.hk/proj/gef/index.php/glossary/dilution-of-precision/>



# GNSS: Architecture

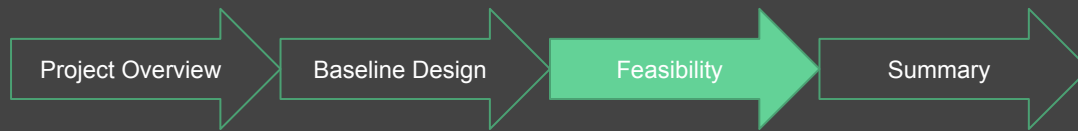
- 3 segments:
  - Ground control segment
    - Provides satellites with ephemeris data and almanac
    - Provides clock-correction factors and data on atmospheric effects
    - Keeps satellites “in check”
  - Space segment
    - Satellites which send out coded ranging signals, ephemerides, correction parameters and almanac.
  - User segment
    - Receiver (which in the case of this project, can also transmit SMS messages)
    - Acquires and track satellite signals.
    - Computes position solution.
    - For this project, THIS is the segment we’ll be working with. All other segments will be assumed as working.





# Comms

---



# Communications - Requirements

---

**FR 2:** The prototype will provide two-way SMS-like messaging

- **DR 2.1:** Device must have wireless transmission and reception of data
- **DR 2.2:** Communication data rate must be at least 200 bits/s

**FR 3:** The prototype will provide path to navigation solution with 10 meter positioning accuracy and 30 nanosecond 1-sigma transfer time

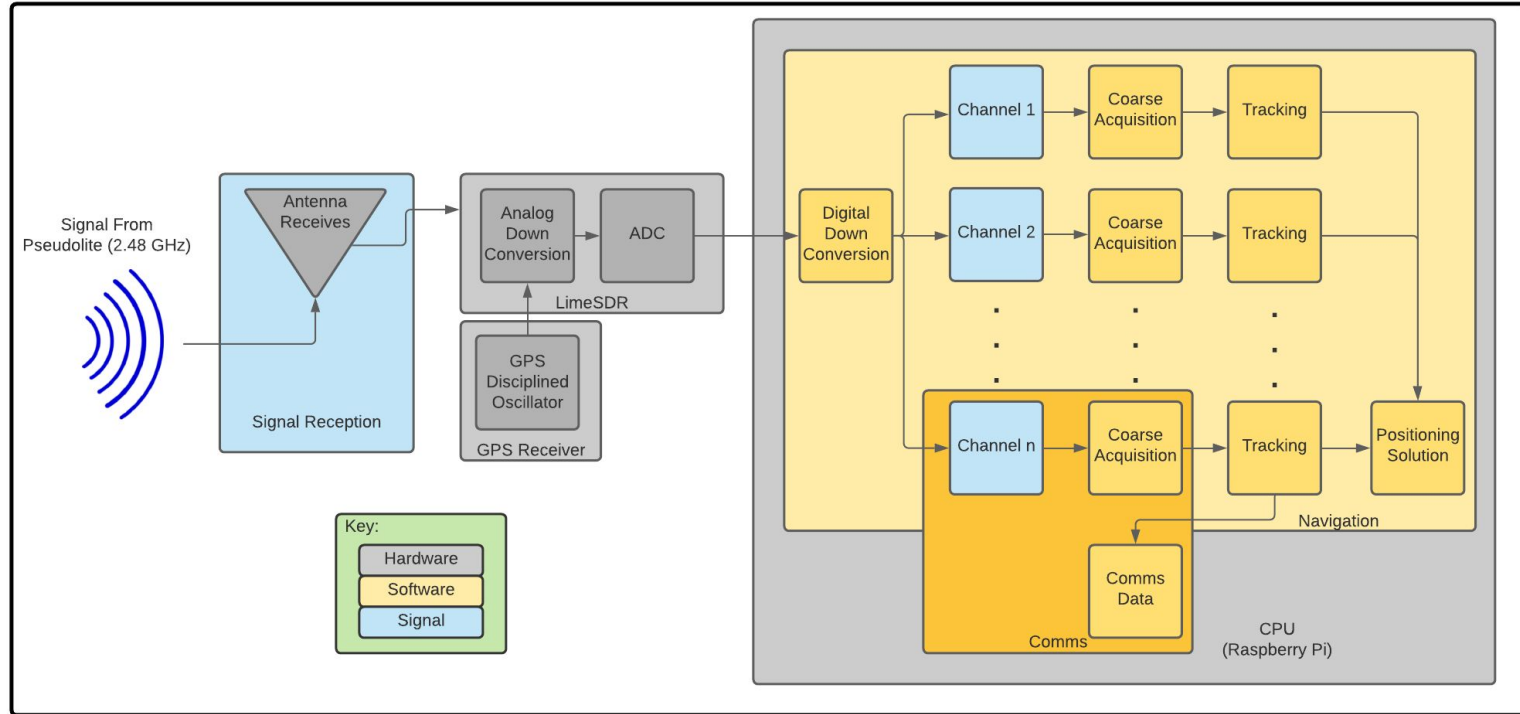
- **DR 3.1:** Receiver Signal to Noise Ratio must be at least 20 dB

**FR 4/5:** The communication link must operate at 2.4-2.48 GHz

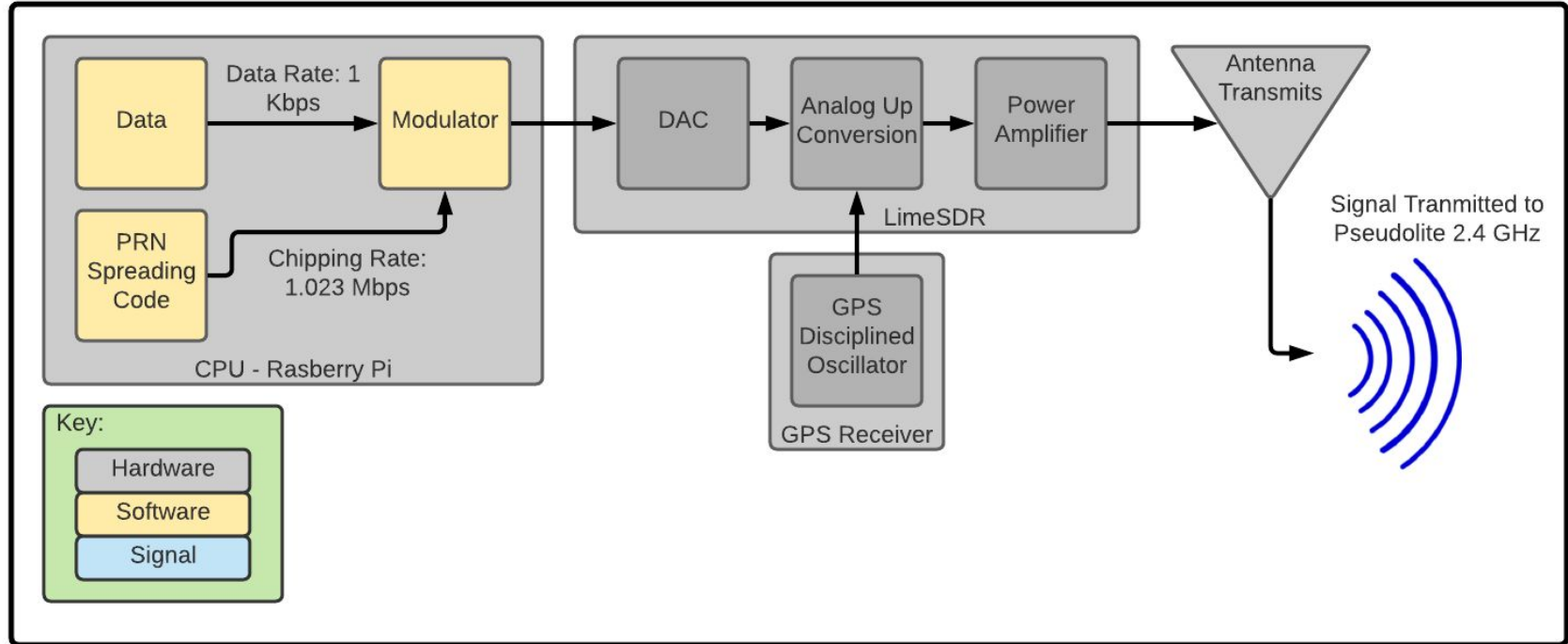
**FR 6:** The architecture must be extendable to 170 simultaneous users

**FR 7:** The communication link must have 3 dB channel bandwidth of 1 MHz

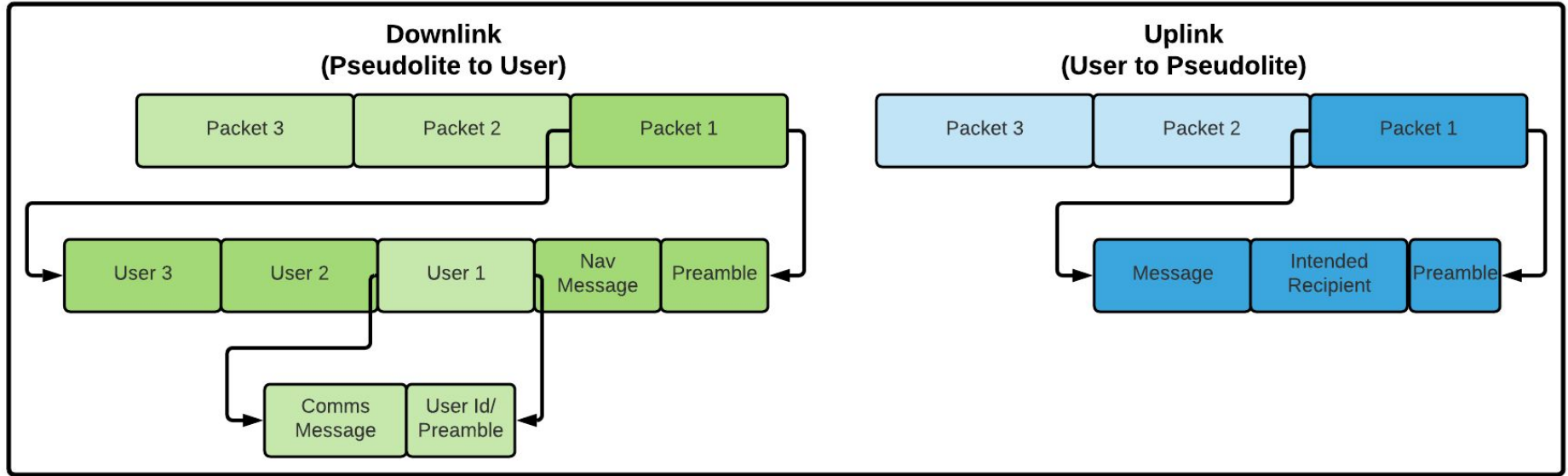
# Receive FBD



# Transmit FBD



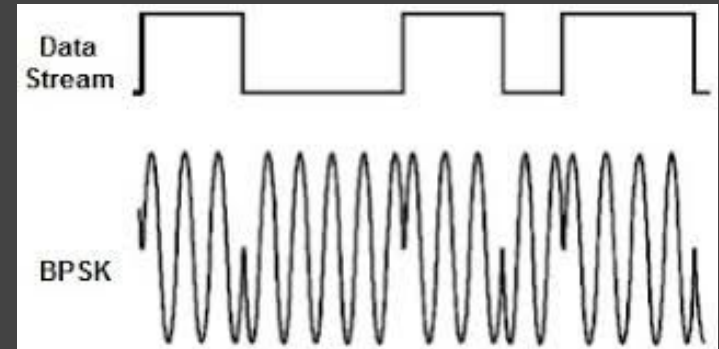
# Signal FBD





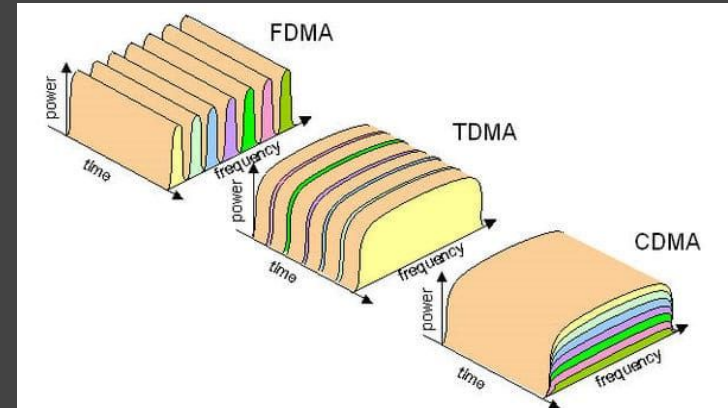
# Link Design: Modulation

- Design Choice: Binary Phase Shift Keying (BPSK)
- Reasoning:
  - Low bit error rate (BER)
  - Low complexity
  - Able to use carrier phase to get even more precise signal time of arrival
  - Currently used in GPS implementations



# Link Design: Multiplexing

- Design: Code Division Multiple Access
- Reasoning:
  - Able to use phase of spreading code to calculate time of arrival (TOA)
  - Single Frequency
  - Asynchronous CDMA
  - Low cross-interference
- Specification (Design Solution):
  - 1023 bit Gold Code PRN chipping code
  - 1.023 MHz chipping rate
  - 1 ms code period (same as data rate)





Project Overview

Baseline Design

Feasibility

Summary



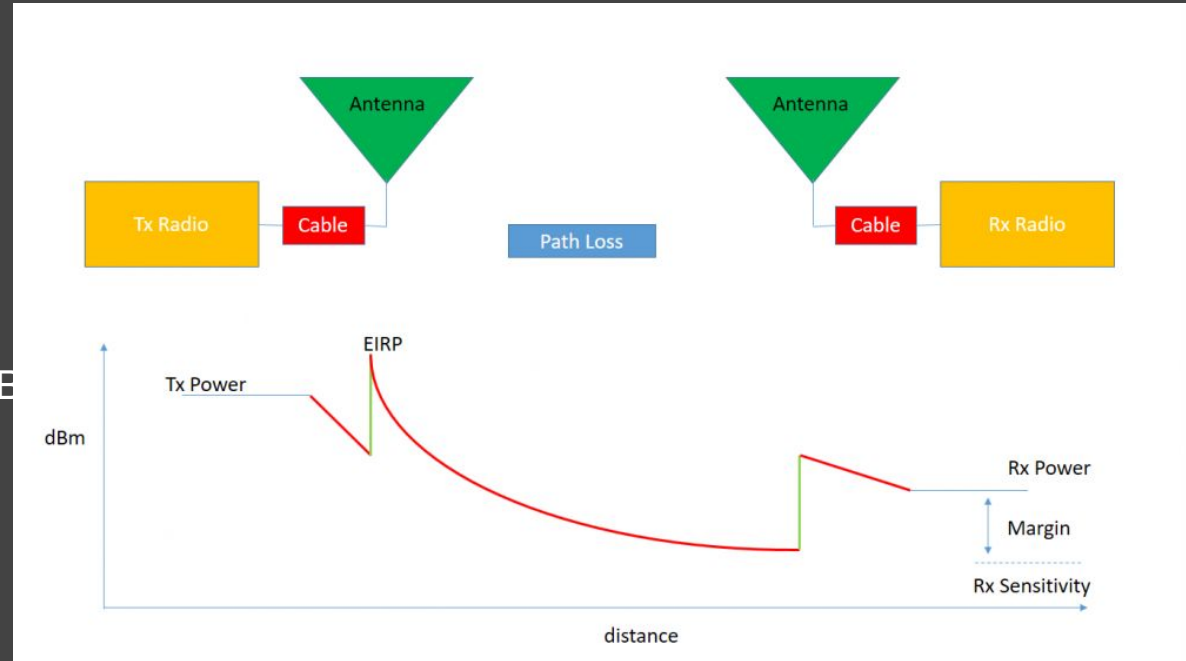
# Communications - Link Budget

## Link Budget (Single User):

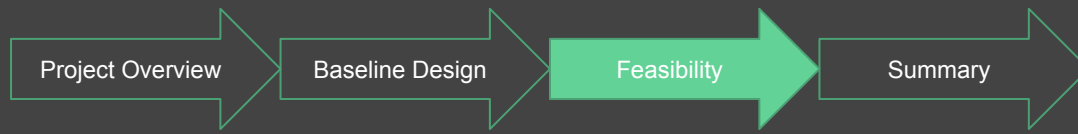
- Data Rate: 1 kbps
- Frequency: 2.4 GHz
- Modulation: BPSK
- Distance: 10 km
- Transmit Power: 2 mW
- Minimum Receive SNR: -12 dB
  - BER:  $10^{-8}$ ,  $E_b/N_0$ : 12 dB
- Link Margin: 9 dB

DR 2.1, 2.2, FR4, FR 5

FEASIBLE







- 
- Pulse Error
- Signal + Noise
- True Signal

$$\delta R = \frac{c_0}{2B\sqrt{2SNR}}$$

- FR 7: Link must operate at 1 MHz bandwidth
- Link Budget operates on a 48 dB-Hz minimum  $C/N_0$

DR 3.1, FR7

FEASIBLE



Project Overview

Baseline Design

Feasibility

Summary

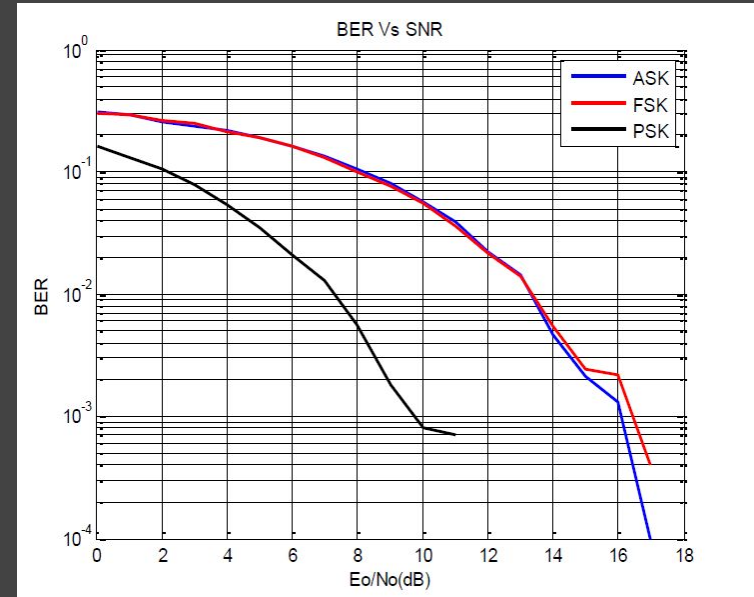


# Modulation: Phase Shift Keying

- Modulation Scheme must be compatible with SDR
  - Compatible with LimeSDR
  - Compatible with GNU Radio
- Low Bit Error Rate
- Single Frequency
- Use carrier phase for additional position accuracy
- GPS uses Phase Shift Keying

DR 2.1, 2.2, 3.1, FR4, 5, 7  
SDR Compatibility

FEASIBLE





- CDMA used in GPS
- CDMA code rate relates to first level of position accuracy
- CDMA can operate on single frequency
- Receiver can receive multiple signals simultaneously
- CDMA with 1023 bit chip code:
  - One way ranging: more end users overall, more precise timing requirements
  - Two way ranging: each user contributes to multiple access interference

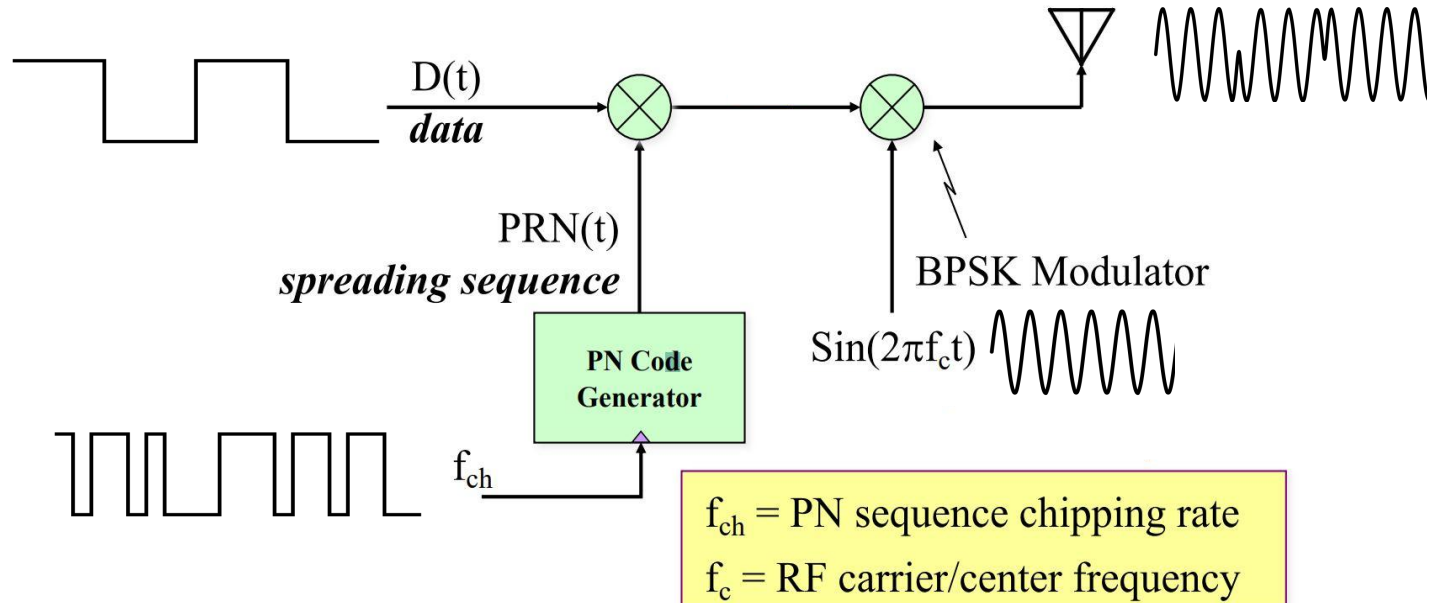


## Two Way Ranging: FR 6

DR 2.1, 3.1, FR 4, 5, 7

147

# Signal Generation



Courtesy Logan Scott



Overview

Schedule

Manufacturing

Budget

# Downlink Packet Breakdown

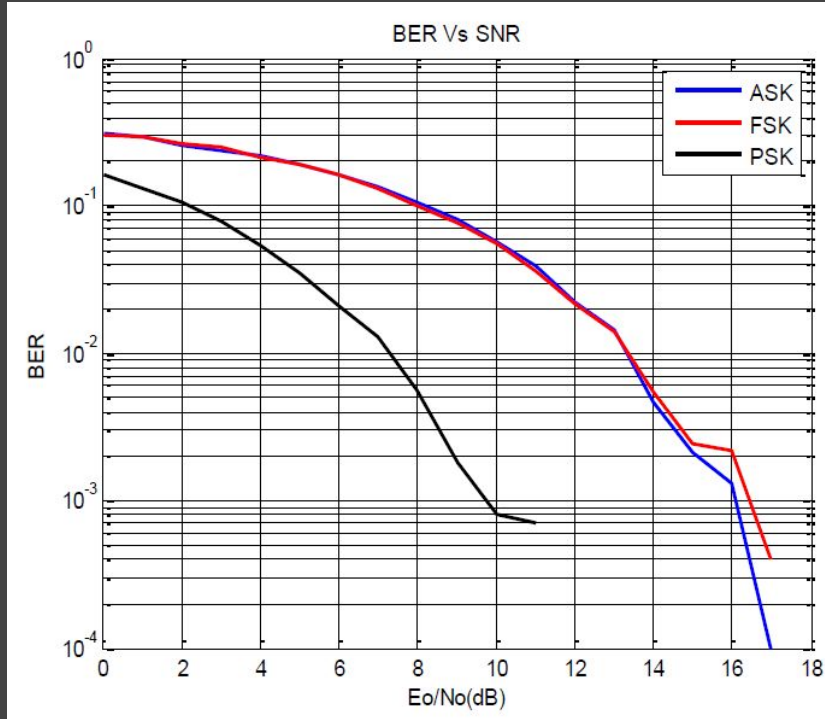
---

- Preamble
  - Unique 24 bit code sequence to allow for signal correlation
- Navigation Message
  - Element of set bit length providing timing for positioning algorithm
- User ID
  - 8 bit element containing a unique user identification sequence for each user to identify intended recipient
  - Designed to accommodate 30 user requirement
- Communications Message
  - 5000 bit text communication element broken into 8 bit characters
  - Message will be appended with ' ' (space) buffers to maintain constant size

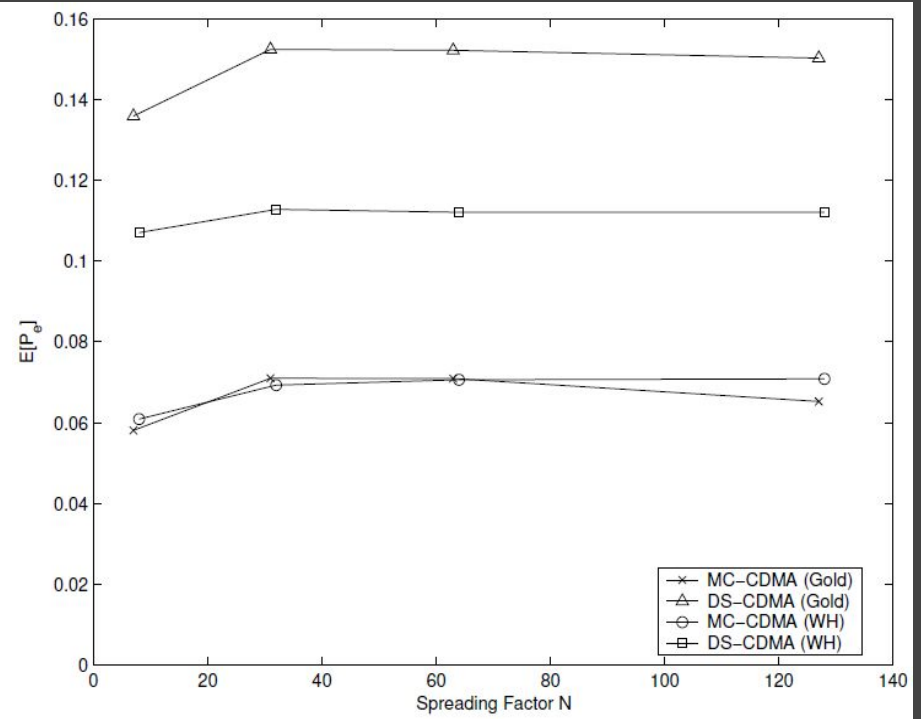


- Preamble
  - Unique 24 bit code sequence to allow for signal correlation
- User Identification
  - 8 bit element containing a unique user identification sequence for each user to identify transmitting user
  - Designed to accommodate 30 user requirement
- Communications Message
  - 5000 bit text communication element broken into 8 bit characters
  - Message will be appended with ' ' (space) buffers to maintain constant size

# Bit Error Rates of Modulation Schemes



Bharati et. al.



Carey et. al.



# Pseudolite Link Budget 1

PARAMETER	UPLINK (Receiver to Satellite)	DOWNLINK (Satellite to Receiver)	UNITS	Symbol	Reference
Speed of Light	3.0E+08	3.0E+08	m/s	$C = \lambda * f$	constant
Frequency	2.4	2.4	GHz	$f$	Input: system choice, X-band mil.com.sat
Wavelength	0.125	0.125	m	$\lambda$	
Range	10	10	km	$R$	Input: Geostationary Satellite [km]
Boltzman's Constant	1.380E-23	1.380E-23	W/(Hz-K)	$k$	constant
<b>Data Parameters</b>					
Bit Error Rate / Probability of Bit Error	Uplink 10-8	Downlink 10-8	Units [-]	Symbol BER	Reference
Data Coding Scheme	QPSK	QPSK			Input: design requirement
Required Bit Energy to Noise Ratio	12	12.0	dB	$E_b/N_0$	Input: chosen modulation (SMAD Tab.13-10)
Data Rate	1000	1,000	bps (Hz)	$R$	Lecture Pt 2, Slide 15
Carrier to Noise Ratio Density	42.00	42.00	dB-Hz	$P_r/N_0$	Input: based on mission / objective
Required Design Margin	6.00	6.00	dB		Lecture Pt 2, Slide 15
Minimum Pr/No	48.00	48.00	dB-Hz		Input: design rule (Hoffmann chap. 9.4.4)
<b>Noise (applies to receiving elements)</b>					
Receiving Antenna Noise Temperature	Uplink 400	Downlink 400	Units K	Symbol $T_a$	Reference
Receiver Cable Loss	0.9	0.9	dB	$L_c$	Lecture Pt 2, Slide 13
Receiver Noise Figure (based on receiver)	3.0	3.0	dB	$NF$	SMAD Table 13-10
Receiver Noise Factor	2.0	2.0	[-]	$F$	SMAD Table 13-10
Receiver Noise Temperature	288.6	288.6	K	$T_r$	Lecture Pt 2, Slide 10
Reference Temperature	290	290	K	$T_o$	Lecture Pt 2, Slide 10
Receiver System Noise Temperature	580.32	580.32	K	$T_s$	SMAD Eqn13-24
Receiver System Noise Power	-200.96	-200.96	dBW-Hz	No	Lecture Pt 2, Slide 13
<b>Receiver Parameters:</b>					
Receive Antenna Diameter	Uplink NA	Downlink NA	Units m	Symbol $D$	Reference
Receive Antenna Area	NA	NA	m <sup>2</sup>	$A$	Input: given geometry from spacecraft
Receive Antenna Efficiency	NA	NA	[-]	$h$	Geometry
Receive Antenna Effective Area	NA	NA	m <sup>2</sup>	$A_e$	Input: typical value
Receive Antenna Gain	2.00	2.00	Power Ratio	$G_r$	efficiency * area
Receive Antenna Beamwidth	180.0	180.0	Degrees	$q_r$	Lecture Pt 1, Slide 14
Receive Antenna Pointing Accuracy	0.0	0	degrees	$er$	Lecture Pt 1, Slide 16
Receive Antenna Pointing Loss	0.00	0.00	dB	$L_{pr}$	Input: pointing error e for chosen system
Receiver Cable Loss (see noise)	-0.5	-0.5	dB	$L_c$	SMAD 13-21
Receiver Figure of Merit	0.00	0.00	dB/K	FOM	Input: typical value
<b>Propagation Parameters:</b>					
Space Loss	Uplink -120.05	Downlink -120.05	Units dB	Symbol $L_s$	Reference
Atmospheric Attenuation (clear air)	0	0	dB	$L_a$	Lecture Pt 2, Slide 3
Polarization Loss	0	0	dB	$L_p$	Lecture Pt 2, Slide 7
					Input typical value





# Pseudolite Link Budget 2



# Satellite Link Budget 1

PARAMETER	UPLINK (Receiver to Satellite)	DOWNLINK (Satellite to Receiver)	UNITS	Symbol	Reference
Speed of Light	3.0E+08	3.0E+08	m/s	$C = \lambda * f$	constant
Frequency	2.5	2.5	GHz	$f$	Input: system choice, X-band mil.com.sat
Wavelength	0.120	0.120	m	$\lambda$	
Range	10000	10000	km	$R$	Input: Geostationary Satellite [km]
Boltzman's Constant	1.380E-23	1.380E-23	W/(Hz-K)	$k$	constant
<b>Data Parameters</b>	<b>Uplink</b>	<b>Downlink</b>	<b>Units</b>	<b>Symbol</b>	<b>Reference</b>
Bit Error Rate / Probability of Bit Error	10-6	10-6	[-]	BER	Input: design requirement
Data Coding Scheme	QPSK	QPSK			Input: chosen modulation (SMAD Tab.13-10)
Required Bit Energy to Noise Ratio	11	12.0	dB	$E_b/N_0$	Lecture Pt 2, Slide 15
Data Rate	1000	1,000	bps (Hz)	$R$	Input: based on mission / objective
Carrier to Noise Ratio Density	41.00	42.00	dB-Hz	$P_r/N_0$	Lecture Pt 2, Slide 15
Required Design Margin	3.00	6.00	dB		Input: design rule (Hoffmann chap. 9.4.4)
Minimum $P_r/N_0$	44.00	48.00	dB-Hz		
<b>Noise (applies to receiving elements)</b>	<b>Uplink</b>	<b>Downlink</b>	<b>Units</b>	<b>Symbol</b>	<b>Reference</b>
Receiving Antenna Noise Temperature	400	25	K	$T_a$	Lecture Pt 2, Slide 13
Receiver Cable Loss	0.9	0.9	dB	$L_c$	SMAD Table 13-10
Receiver Noise Figure (based on receiver)	3.0	1.0	dB	$NF$	SMAD Table 13-10
Receiver Noise Factor	2.0	1.3	[-]	$F$	Lecture Pt 2, Slide 10
Receiver Noise Temperature	288.6	75.1	K	$T_r$	Lecture Pt 2, Slide 10
Reference Temperature	290	290	K	$T_o$	SMAD Eqn13-24
Receiver System Noise Temperature	580.32	31.75	K	$T_s$	Lecture Pt 2, Slide 13
Receiver System Noise Power	-200.96	-213.58	dBW-Hz	$N_0$	Lecture Pt 2, Slide 9
<b>Receiver Parameters:</b>	<b>Uplink</b>	<b>Downlink</b>	<b>Units</b>	<b>Symbol</b>	<b>Reference</b>
Receive Antenna Diameter	NA	NA	m	$D$	Input: given geometry from spacecraft
Receive Antenna Area	NA	NA	m <sup>2</sup>	$A$	Geometry
Receive Antenna Efficiency	NA	NA	[-]	$h$	Input: typical value
Receive Antenna Effective Area	NA	0.002	m <sup>2</sup>	$A_e$	efficiency * area
Receive Antenna Gain	20.00	2.38	power Ratio	$G_r$	Lecture Pt 1, Slide 14
Receive Antenna Beamwidth	20.0	180.0	Degrees	$\theta_r$	Lecture Pt 1, Slide 16
Receive Antenna Pointing Accuracy	0.2	0	degrees	$\epsilon_r$	Input: pointing error $\epsilon$ for chosen system
Receive Antenna Pointing Loss	0.00	0.00	dB	$L_{pr}$	SMAD 13-21
Receiver Cable Loss (see noise)	-0.5	-0.5	dB	$L_c$	Input: typical value
Receiver Figure of Merit	0.03	0.08	dB/K	FOM	
<b>Propagation Parameters:</b>	<b>Uplink</b>	<b>Downlink</b>	<b>Units</b>	<b>Symbol</b>	<b>Reference</b>
Space Loss	-180.37	-180.37	dB	$L_s$	Lecture Pt 2, Slide 3
Atmospheric Attenuation (clear air)	0	0	dB	$L_a$	Lecture Pt 2, Slide 7
Polarization Loss	0	0	dB	$L_p$	Input: typical value

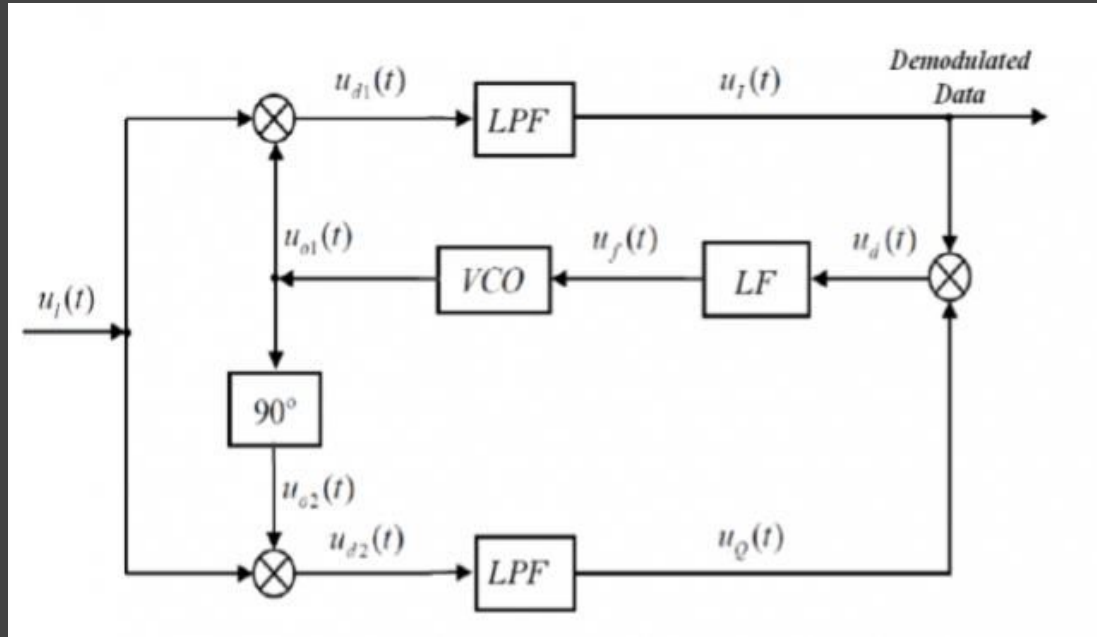


# Satellite Link Budget 2

Transmitter Parameters:	Uplink	Downlink	Units	Symbol	Reference
Transmit Antenna Diameter	NA	NA	m	D	Switch Receive
Transmit Antenna Area	NA	NA	m <sup>2</sup>	A	Switch Receive
Transmit Antenna Efficiency	NA	NA	[-]	h	Switch Receive
Transmit Antenna Effective Area	0.002	NA	m <sup>2</sup>	Ae	Switch Receive
Transmit Antenna Gain	2.38	20.00	dBi	Gt	Switch Receive
Transmit Antenna Beamwidth	180.00	20.00	degrees	qt	Switch Receive
Transmit Antenna Pointing Accuracy	0.00	0.15	degrees	et	Switch Receive
Transmit Antenna Pointing Loss	0.00	0.00	dB	Lpt	Switch Receive
Transmit Line Loss	-0.5	-0.5	dB	Lt	Input: based on chosen cable/geometry
Transmit Power	3.0	14.1	dBW	Pt	10*LOG10(Transmit power)
Transmit Power, Linear	2	25.6	W		Input: chosen transmitter
Effective Isotropic Radiated Power	4.89	33.58	dBW	EIRP	Sum of Power, Gain, and Losses in dB
<b>Link Budget:</b>	<b>Uplink</b>	<b>Downlink</b>	<b>Units</b>	<b>Symbol</b>	<b>Reference</b>
Effective Isotropic Radiated Power	4.89	33.58	dBW	EIRP	From Above
Propagation Losses	-180.37	-180.37	dB	L	Sum of Losses
Receive System Gain	19.50	1.88	dB	Gr	Sum of antenna gain and system losses
Received Power	-155.97	-144.90	dBW	Pr	Sum of Power sent out minus losses
System Noise Power	-200.96	-213.58	dBW-Hz	No	
Carrier to Noise Ratio Density	44.99	68.68	dB-Hz	Pr/No	
Minimum Pr/No	44.00	48.00	dB-Hz		
<b>Link Margin</b>	<b>0.99</b>	<b>20.68</b>	<b>dB</b>		

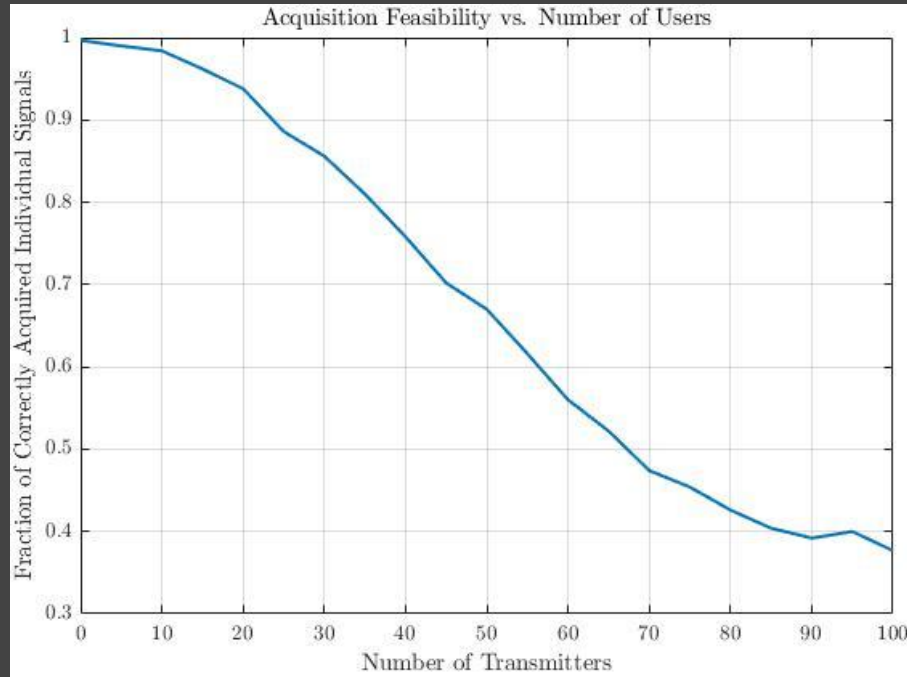


# Simple Tracking Loop (Costas Loop)





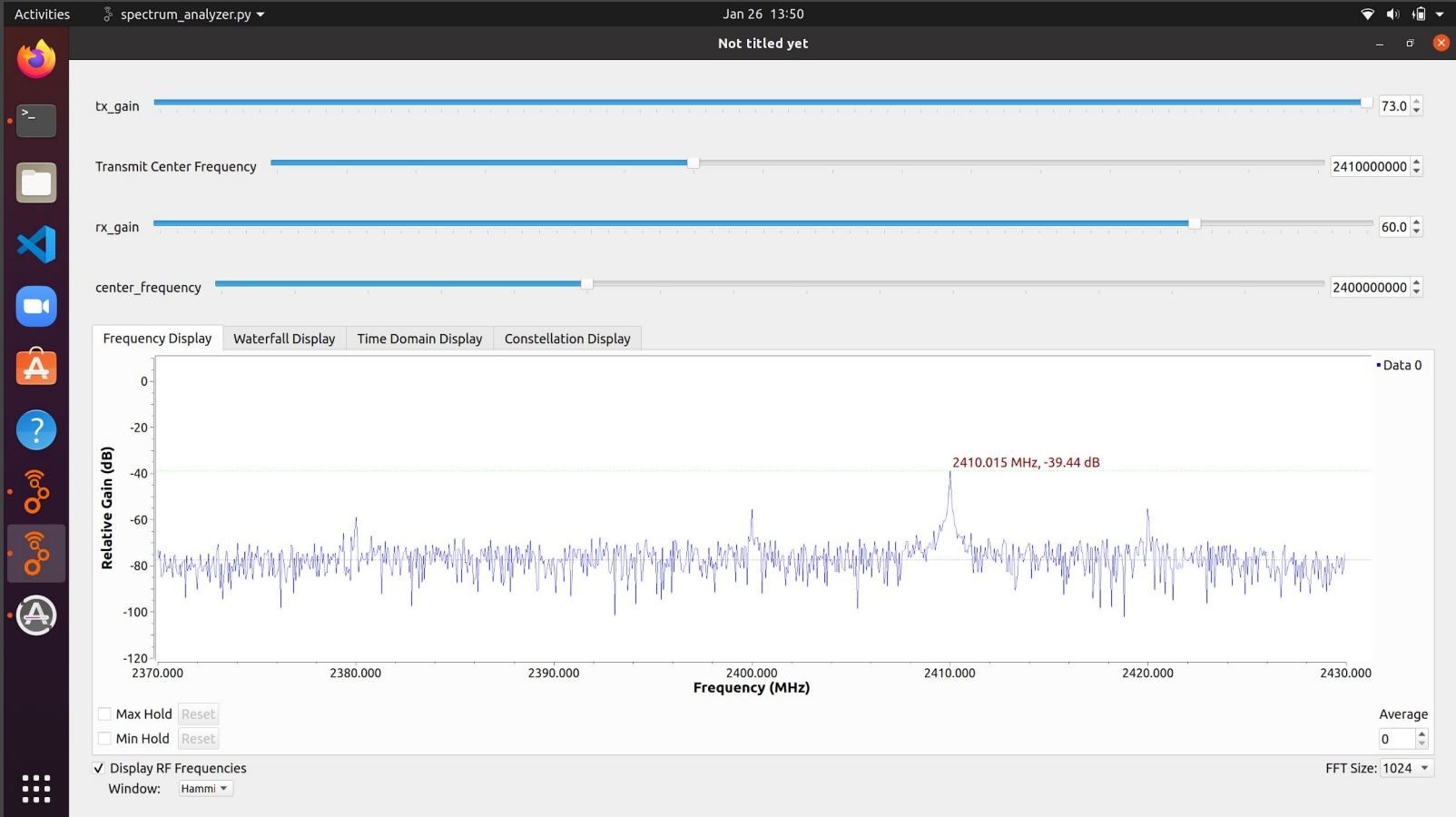
# Multiple Access Interference







# LimeSDR Hardware/Software Validation





Overview

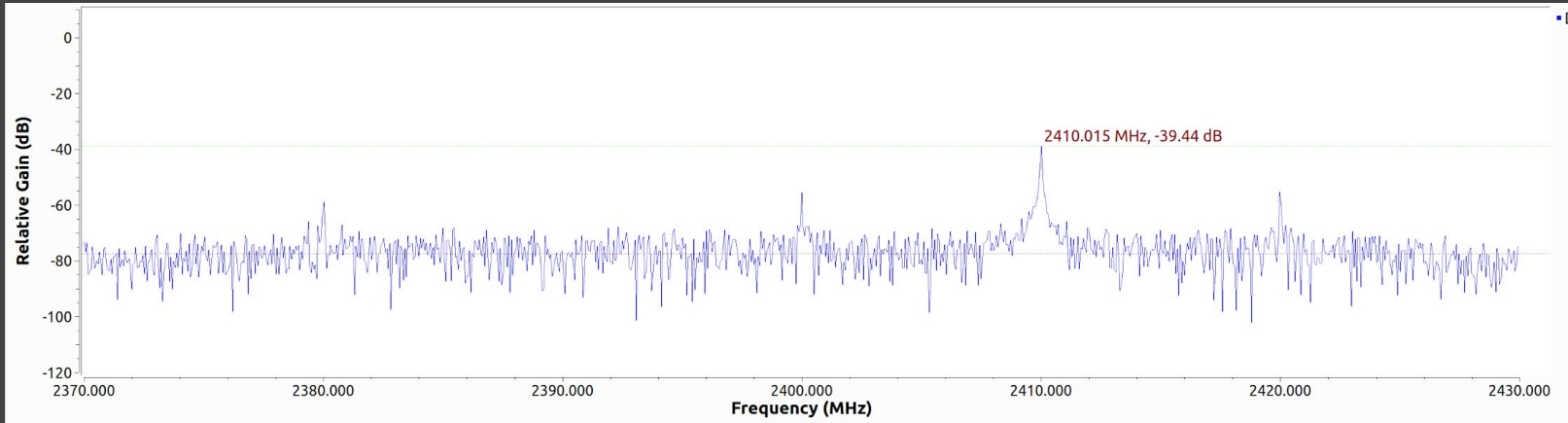
Schedule

Manufacturing

Budget

# Signal Interference (Simple Transceiver)

- Preliminary interference and verification test resulted in positive results.
- Single tone SNR:
  - At 1 foot separation 40dB
  - At 590 feet (180 Meters) -15dB (Extrapolation)







Overview

Schedule

Manufacturing

Budget

# Downlink Packet Breakdown

---

- Preamble
  - Unique 24 bit code sequence to allow for signal correlation
- Navigation Message
  - Element of set bit length providing timing for positioning algorithm
- User ID
  - 8 bit element containing a unique user identification sequence for each user to identify intended recipient
  - Designed to accommodate 30 user requirement
- Communications Message
  - 5000 bit text communication element broken into 8 bit characters
  - Message will be appended with ' ' (space) buffers to maintain constant size



Overview

Schedule

Manufacturing

Budget

# Uplink Breakdown

---

- Preamble
  - Unique 24 bit code sequence to allow for signal correlation
- User Identification
  - 8 bit element containing a unique user identification sequence for each user to identify transmitting user
  - Designed to accommodate 30 user requirement
- Communications Message
  - 5000 bit text communication element broken into 8 bit characters
  - Message will be appended with ' ' (space) buffers to maintain constant size



# Gr-LimeSDR Installation Process

The goal is to create a universal **installation script**, which a person could run on a Linux terminal and have access to a LimeSDR Development Environment.



LimeSuiteGUI





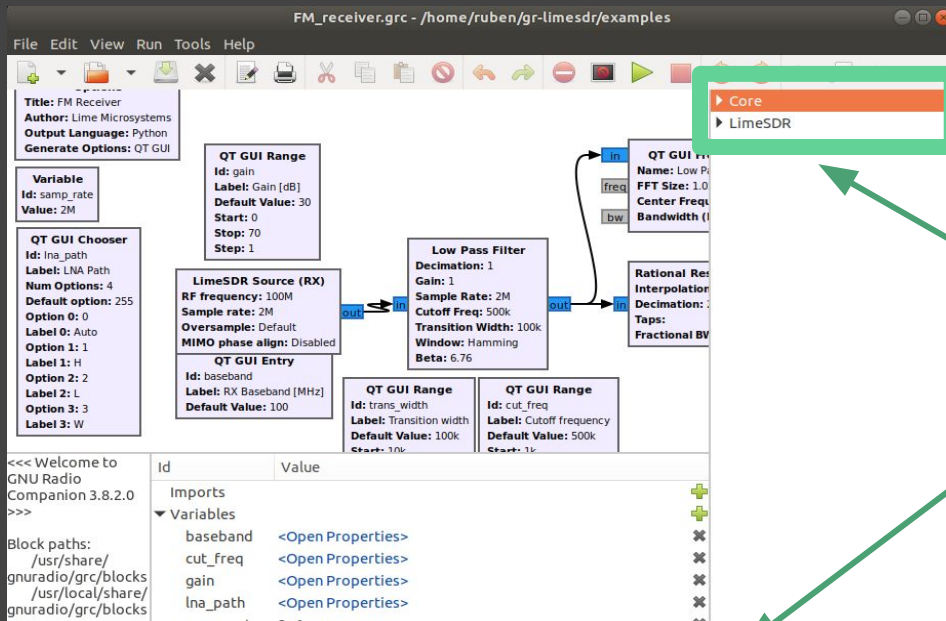
# Gr-LimeSDR Installation Process

Integrating gr-limesdr plugin package.

## Example script:

```
[3] - Gr-limeSDR Plugin Installation: (Source Code Installation)
[3.1] - Installing Dependencies:
-> sudo apt-get install libboost-all-dev swig
Just in case:
-> sudo apt-get install gnuradio-dev
[3.2] - Building gr-limesdr from source: (Only for GNU Radio 3.8!)
-> git clone https://github.com/myriadrf/gr-limesdr
-> cd gr-limesdr
-> git checkout gr-3.8
-> mkdir build
-> cd build
-> cmake ..
-> make
-> sudo make install
-> sudo ldconfig
```

# Gr-LimeSDR Development Environment



- We have a head start in the environment setup!

- We can also verify that the GNU Radio communicates with the Gr-LimeSDR plugin

- We can already verify the LimeSDR communicates with the given software via the LimeSuiteGUI

```
ASEN_4018_SeniorProjects $ LimeUtil --find
* [LimeSDR-USB, media=USB 3.0, module=FX3, addr=1d50:6108, serial=0009072C0289300F]
```



## 166



Overview

Schedule

Manufacturing

Budget

# Integration: LimeSDR + GNURadio

- gr-limesdr plugin for GNURadio
  - Provides LimeSDR Transmit and Receive blocks
    - Can specify Rx/Tx channels for multiple antennas
- Automatically finds the LimeSDR
- Configurable via .ini file
- These blocks feed samples into the well documented GNURadio sample stream

## LimeSDR Sink (TX)

**RF frequency:** 100M

**Sample rate:** 2M

**Oversample:** Default

in

## LimeSDR Source (RX)

**RF frequency:** 100M

**Sample rate:** 2M

**Oversample:** Default

**MIMO phase align:** Disabled

out



# Project Risks and Mitigation

---





Overview

Solution

CPE

DR

Risk &amp; M

V&amp;V

Planning

# Risk Identification

Risk ID	Description	Likelihood	Consequence
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources	4	5
ELE-1	Time synchronization between pseudolite and GPS receiver	3	5
SOF-1	Team inexperience handling new software. For example, GNU radio	3	4
COM-2	Signal Interference in testing environment	4	3
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly	2	5
COM-1	Multiple User Communicating	5	2
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project	2	4



Overview

Solution

CPE

DR

Risk &amp; M

V&amp;V

Planning

# Risk Matrix

Risk ID	Description
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources
ELE-1	Time synchronization between pseudolite and GPS receiver
SOF-1	Team inexperience handling new software. For example, GNU radio
COM-2	Signal Interference in testing environment
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly
COM-1	Multiple User Communicating
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project

Likelihood

Consequence

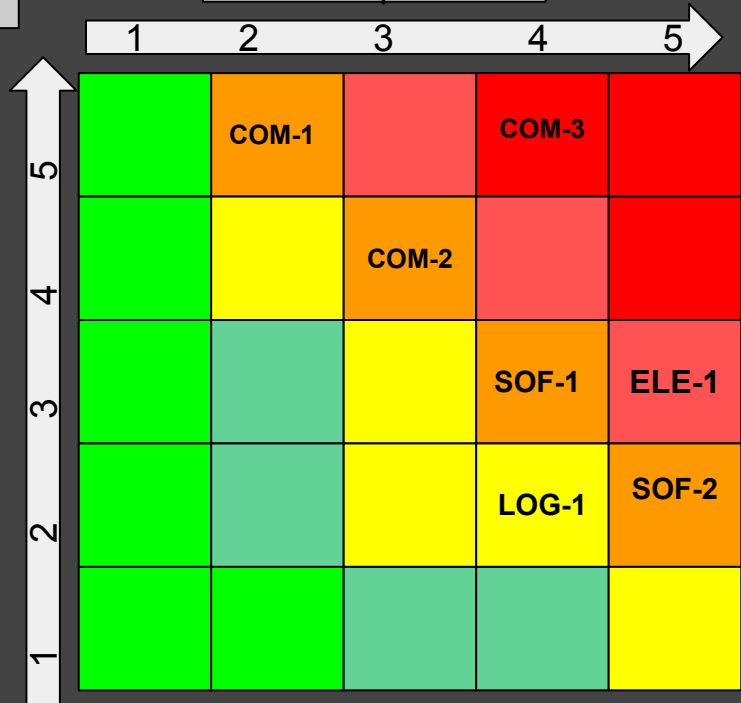
Legend

LOW RISK

Acceptable

LOW-MED  
RISK

Low Risk

MEDIUM  
RISKRequires  
MonitoringMED-HIG  
H RISKModerate  
MitigationHIGH-  
RISKImmediate  
MitigationCRITICAL  
RISKUnacceptable  
Risk



Overview

Solution

CPE

DR

Risk &amp; M

V&amp;V

Planning

# Risk Mitigation Table

Risk ID	Description	Mitigation Approach
COM-3	Near-Far problem: Difficulty to hear weak-far signals from strong-close signal sources	<ul style="list-style-type: none"><li>• Test within pseudolite geometry boundaries</li><li>• Automatic gain control</li></ul>
ELE-1	Time synchronization between pseudolite and GPS receiver	Conduct initial test to verify limeSDR clock override with Septentrio PolaRxS GPS receiver.
SOF-1	Team inexperience handling new software. For example, GNU radio	<ul style="list-style-type: none"><li>• Fall semester early development</li><li>• Online tutorials, device users manual</li><li>• External Resources: Dr Akos, Dr Rainville, Steve Taylor, Sam Holt</li></ul>
COM-2	Signal Interference in testing environment	Change testing area to back up location (Platteville), change testing time to reduce interference (overnight)
SOF-2	Lime-SDR, Raspberry Pi, and other hardware components not communicating correctly	<ul style="list-style-type: none"><li>• Fall semester early development.</li><li>• Refer to overall block diagram,</li><li>• Dive into specific functional block diagrams and debug issues (divide and conquer).</li></ul>
COM-1	Multiple User Communicating	Design an optimal CDMA code for each user
LOG-1	Hardware/Electronics don't arrive on time which sets back schedule of overall project	Change provider of electronics to avoid schedule variance (Amazon, SparkFun, BestBuy, etc)





# Verification and Validation

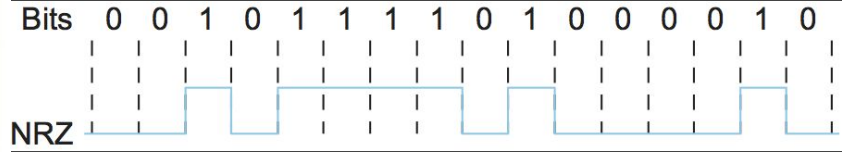
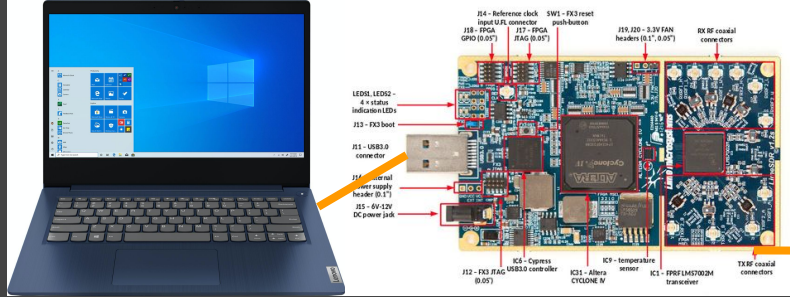
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# Signal Reception

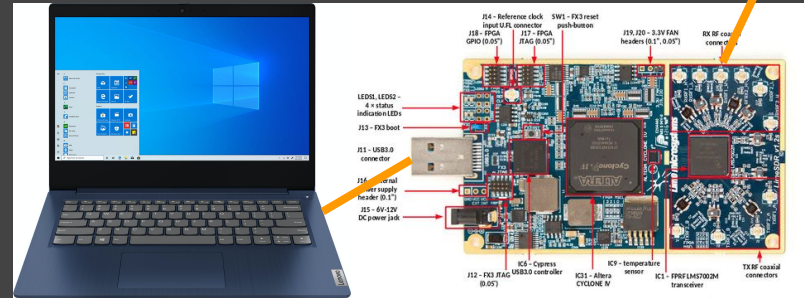
LimeSDR demonstrates reception signal through wired connection

- Laptop running GNU radio creates signal and sends to LimeSDR
  - Create 100 bit .txt file to use as transmission signal
  - Send bits to LimeSDR using USB connection
- Verify receipt of signal to the LimeSDR
  - Create .txt file from bits and compare sent and received signal



USB Connection sends .txt as bits to LimeSDR

Wired from TX to RX sending bits



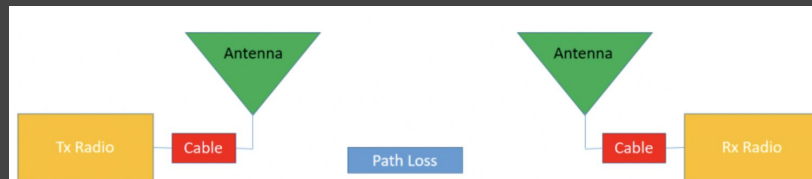
Bits converted to .txt and compared to initial file

[Overview](#)[Solution](#)[CPE](#)[DR](#)[Risk & M](#)[V&V](#)[Planning](#)

# Antenna test

Demonstrate that the antennas used can supply the proper gain for the design

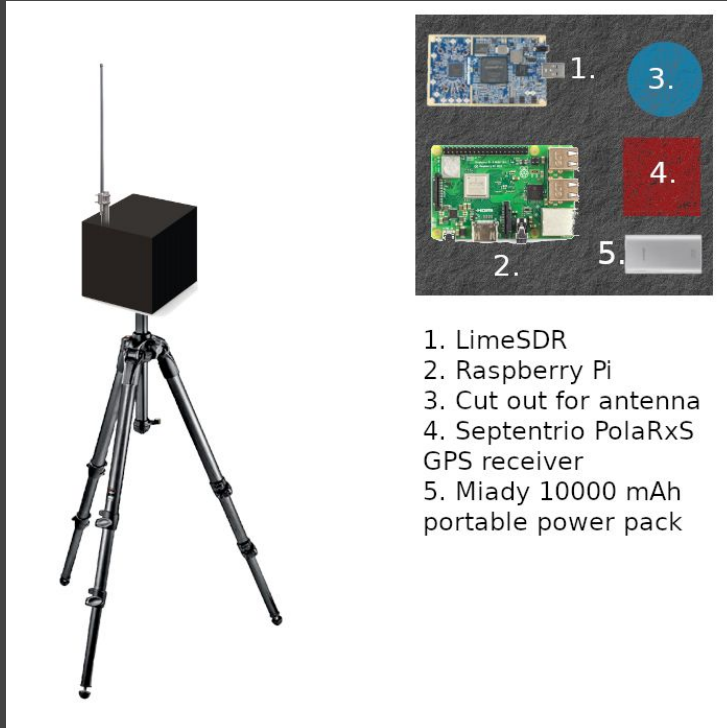
- Measure loss between a two radio/antenna setup running multiple trials at various distances
- Use least squares calculate the gain of the antennas(they are the same)



$$PL = G_1 + G_2 + 20 \log \left( \frac{\lambda}{4\pi R} \right)$$



# Transmitter/Receiver Set up



1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack

Each transmitter and receiver will feature a casing for the electronics that will hold the LimeSDR, Raspberry Pi, a battery pack, and space for wiring and a connection to the antenna. The electronics casing will be thread-mounted to a camera tripod.



# Modulation Scheme

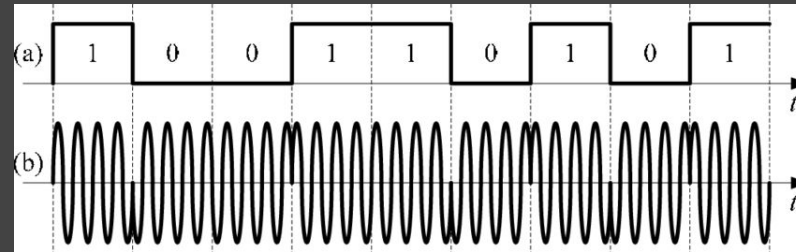
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LimeSDR receiver positioned 100m away from LimeSDR transmitter

- Transmitter modulates 100 bit .txt file to 2.48 GHz carrier frequency using GNU Radio PSK modulator
- LimeSDR receiver receives signal and demodulates it using GNU radio PSK demodulator
- Signal is re-output as .txt file and compared to initial file



Using Laptop and GNURadio  
send .txt file to LimeSDR



GNURadio modbox modulates  
bits using BPSK modulation  
scheme

LimeSDR receives the  
modulated signal and  
demodulates using GNURadio  
demodbox



Bits are converted to .txt file  
and compared to original .txt





# Multiplexing and Wireless Transmission

Receiver can distinguish CDMA codes and process the desired signal

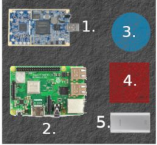
- 3 transmitters generate and transmit signals of different 100 bit .txt files
  - For each file, send preamble and .txt file using GNU radio PSK modulator
- LimeSDR receiver positioned 100m away receives the signals and listens for specific preamble
  - Based on desired preamble, receiver demodulates and converts desired signal to .txt
- Received .txt file is compared to original 3 to verify proper signal was processed



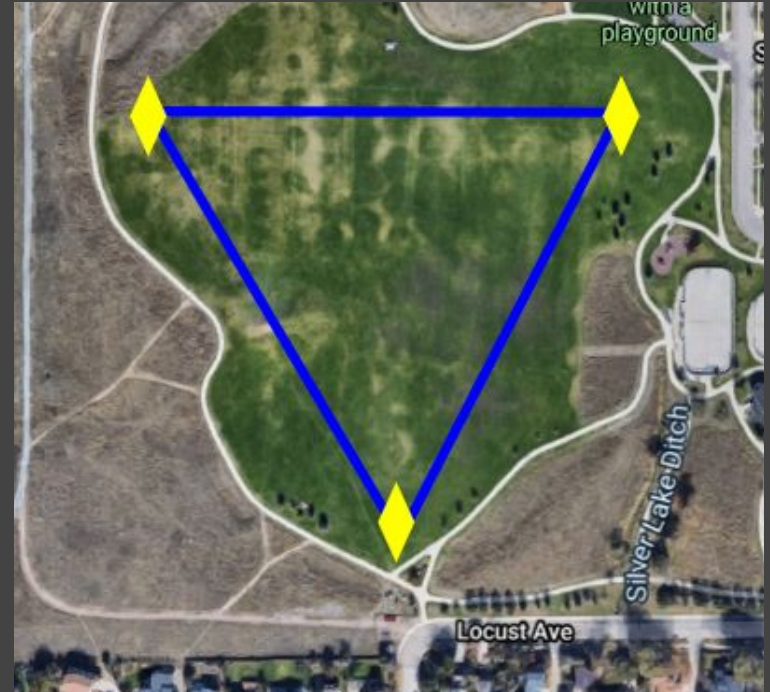
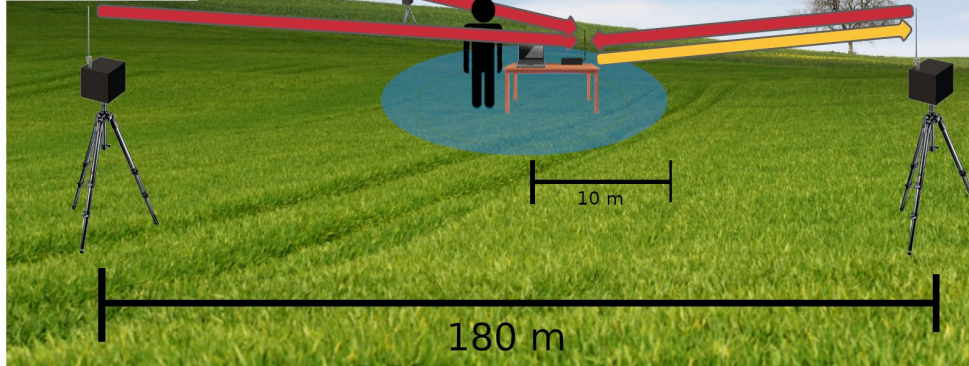
## 181

# Multiplexing & HDOP Test Setup

Contents of Black Boxes



1. LimeSDR
2. Raspberry Pi
3. Cut out for antenna
4. Septentrio PolaRxS GPS receiver
5. Miady 10000 mAh portable power pack



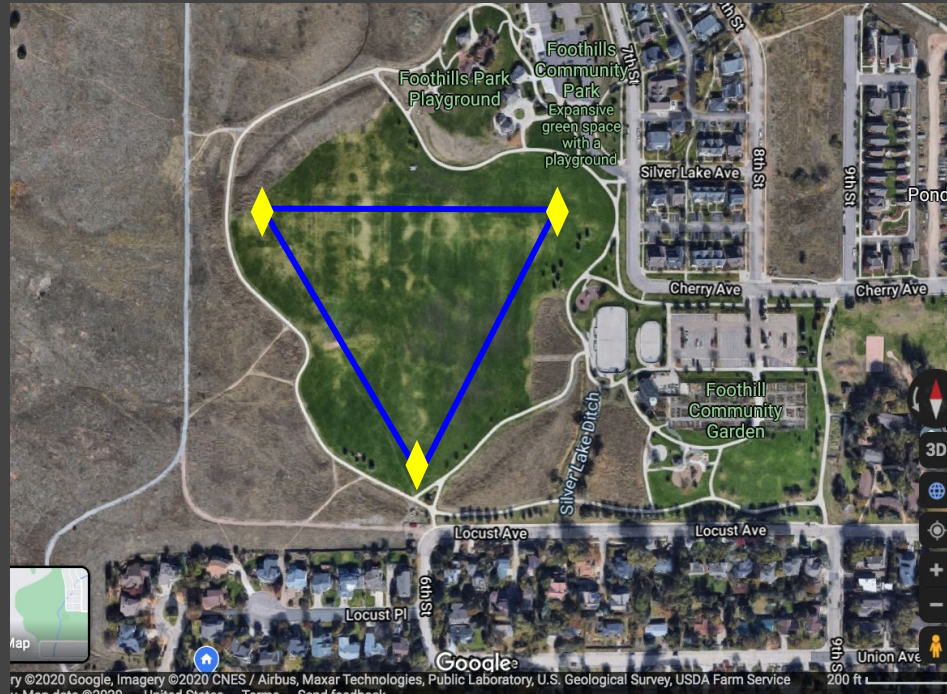


# Test Checklist

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1. Walk transmitters to their positions
2. Sync transmitters and receiver with GPS
  - a. Sync up clocks to GPS (as well as to each other)
  - b. Record “true” position of each transmitter and receiver
3. Verify all stations ready
4. Run communications/positioning test









# Back-Up Testing Area



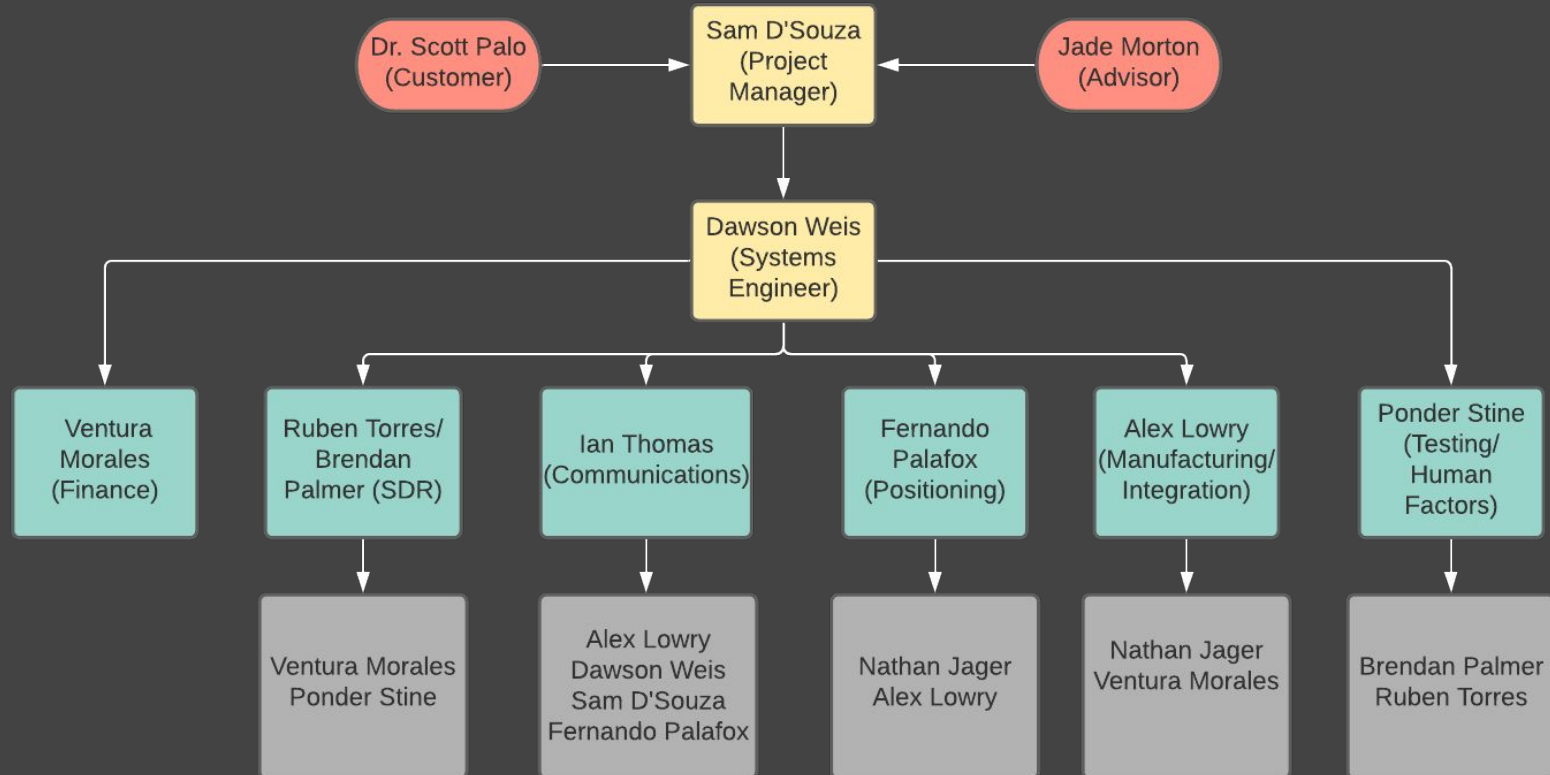


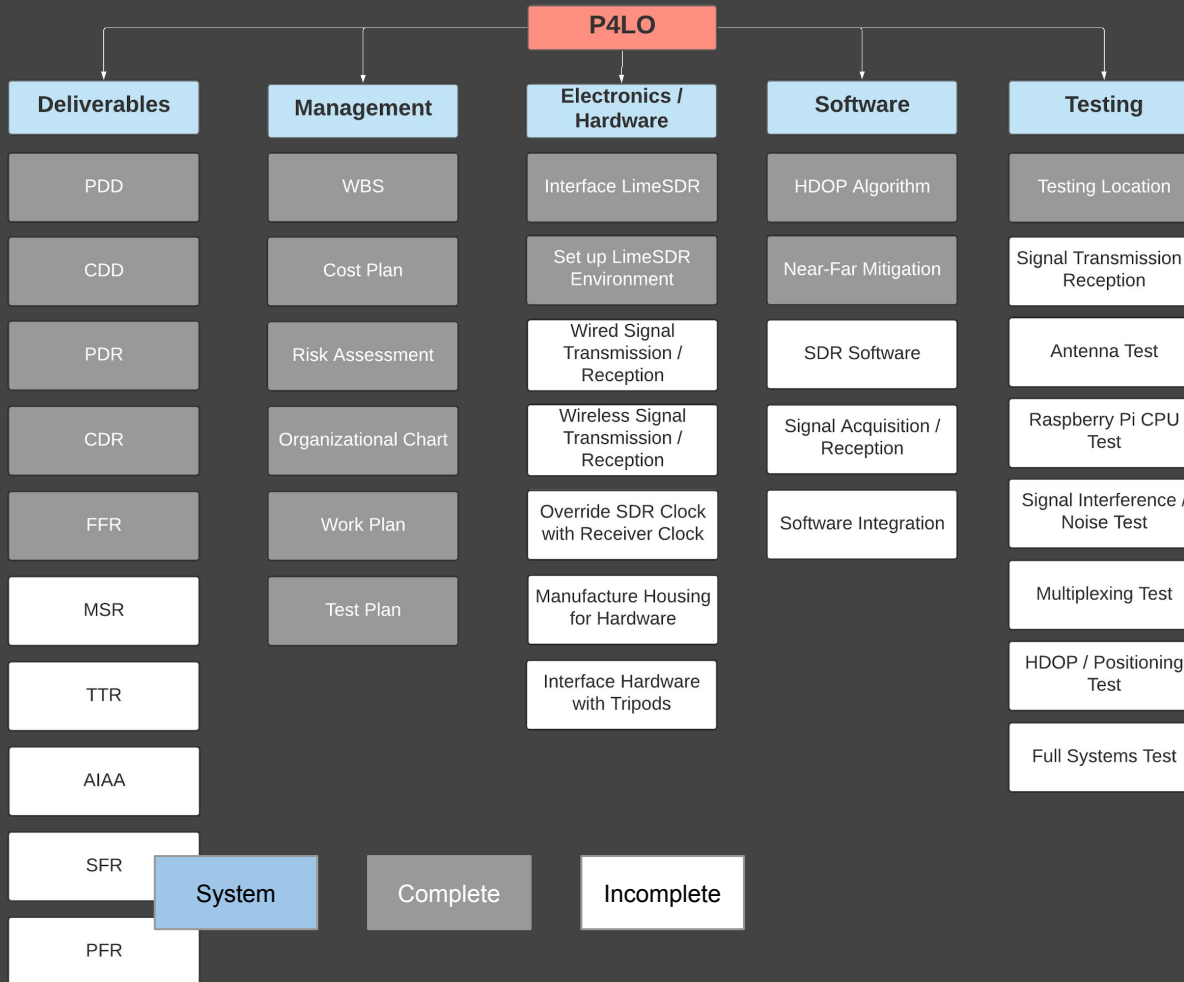
# Project Planning

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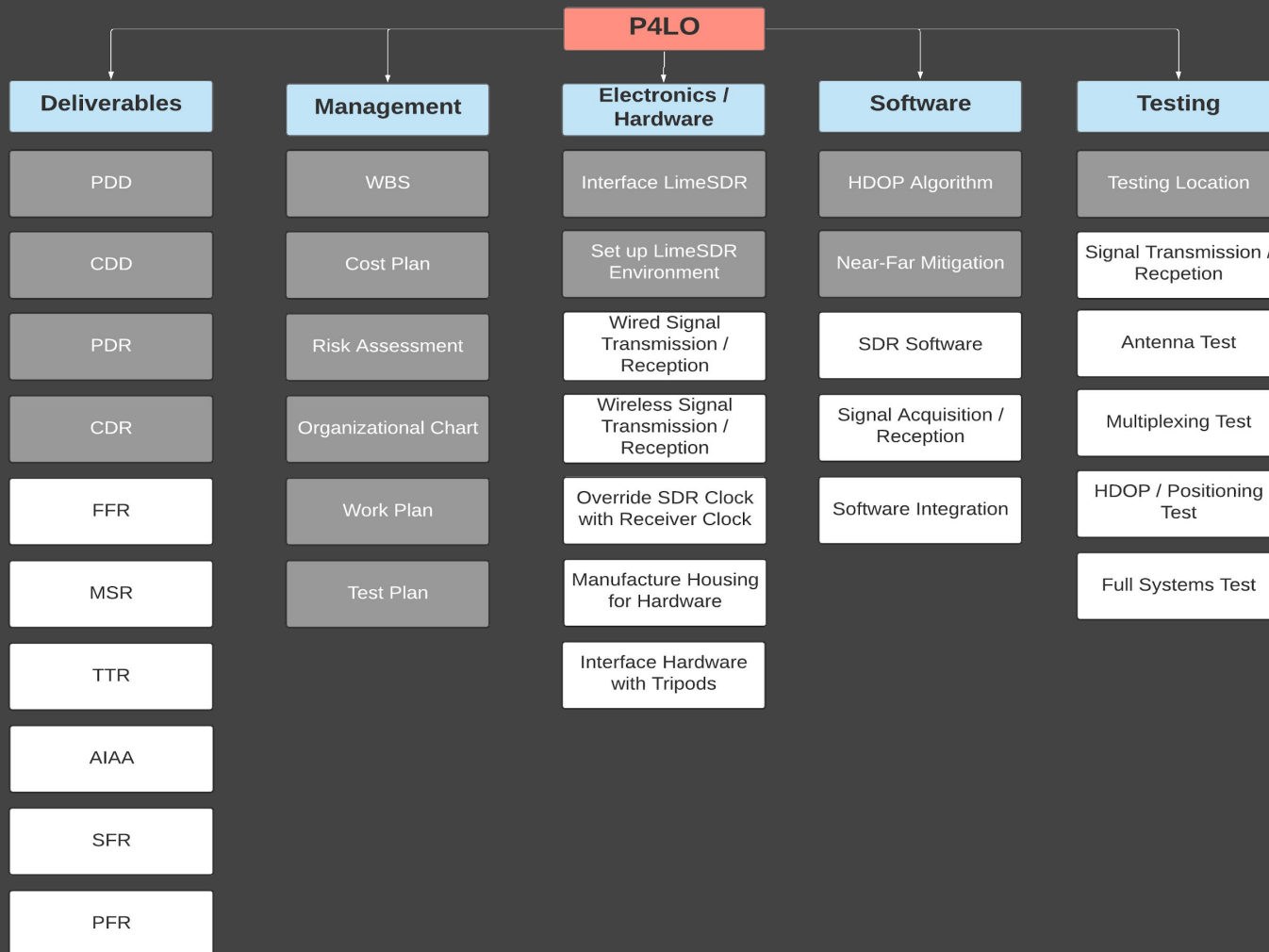
# Organizational Chart







## Work Breakdown Structure (WBS)



System

Complete

Incomplete







Testing Procedure	Date Scheduled	Location
Signal Reception Test	March 9th	Foothills Park
Modulation and Multiplexing Test	March 15th	Foothills Park
HDOP Test	March 23rd	Foothills Park
Full Scale Test	March 25th	Foothills Park





# Budget

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Overview

Solution

CPE

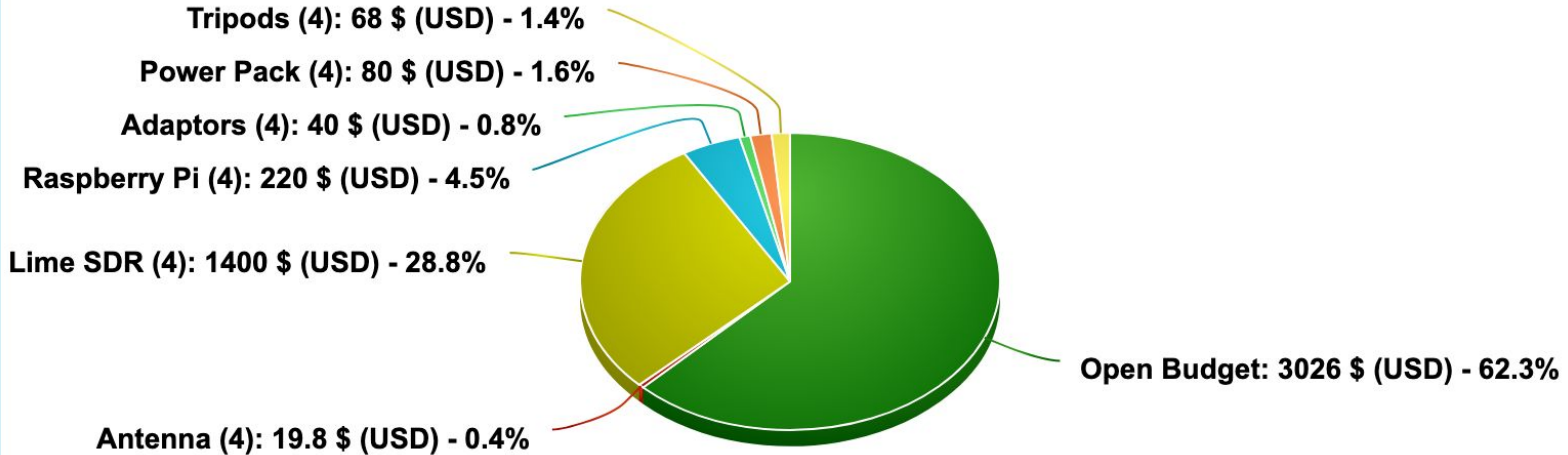
DR

Risk & M

V&V

Planning

# Budget Pie Chart





# Expense Budget & Margin

Team P4LO Budget					
Allowable Budget	\$5,000.00				
Communications Hardware					
Item	Individual Cost	# of Items	Subtotal Per Set	Supplier	Item Link
Antenna	\$4.95	4	\$19.80	SparkFun	<a href="https://www.sparkfun.com/products/11320">https://www.sparkfun.com/products/11320</a>
Lime SDR	\$349.95	4	\$1,399.80	SparkFun	<a href="https://www.sparkfun.com/products/15027">https://www.sparkfun.com/products/15027</a>
Raspberry Pi	\$55.00	4	\$220.00	SparkFun	<a href="https://www.sparkfun.com/products/15447">https://www.sparkfun.com/products/15447</a>
Hardware Adapators	\$9.95	4	\$39.80	SparkFun	<a href="https://www.sparkfun.com/products/9145">https://www.sparkfun.com/products/9145</a>
Power Pack	\$19.99	4	\$79.96	Amazon	<a href="https://www.amazon.com/Miady-10000mAh-Portable">https://www.amazon.com/Miady-10000mAh-Portable</a>
Testing Tripods (6ft)	\$16.99	4	\$67.96		<a href="https://www.continentalphoto.com/itemdetails.asp?r">https://www.continentalphoto.com/itemdetails.asp?r</a>
Budget Balance					
Sub-total Balance	\$1,827.32				
Sales Tax Total	\$146.19				
Total Cost	\$1,973.51				
Budget Balance	\$3,026.49				



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