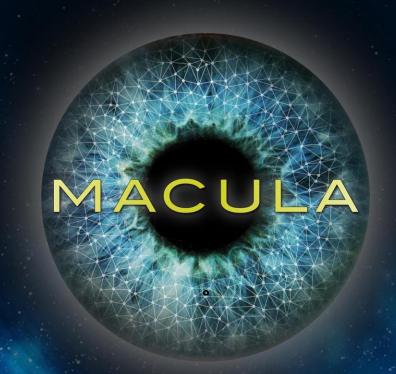
#### Mapping Architecture Concept for Universal Landing Automation



#### CRITICAL DESIGN REVIEW

Customer: Jeffrey Thayer & Brian Argrow, University of Colorado Boulder AES

Faculty Advisor: Jay McMahon

#### Team Members:

Trevor Arrasmith, Brett Bender, Chris Brown, Nick Dawson, David Emmert, Bryce Garby, Russell Gleason, Matthew Hurst, Jared Levin, Ansel Rothstein-Dowden



# Agenda



Project Purpose / Objectives	Brett
Design Solution	Brett
Critical Project Elements	Russell
Design Requirements & Satisfaction	Trevor, Russell
Project Risks	David
Verification & Validation	Jared
Project Planning	David



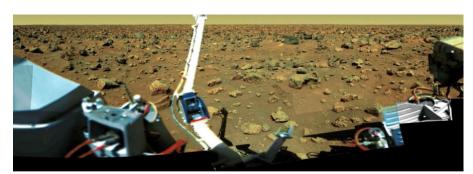


# PROJECT PURPOSE AND OBJECTIVES



## Motivation

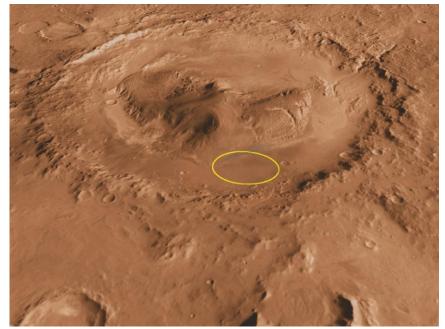




**Rocks on the Martian surface** 

http://geology.isu.edu/wapi/Geo\_Pgt/Mod09\_Mars/images/VIEWFRMLANDER2VLFMOS21.gif

Landing zones for spacecraft must be pre-determined as "safe," and can be far from areas of scientific interest



Curiosity's error ellipse on Mars (20 km minor, 25 km major axis)

 $http://www.nasa.gov/images/content/573652 main\_pia14294-anno-43\_946-710.jpg$ 



# Project Objectives



**Design**, **manufacture**, and **test** a **proof-of-concept** light detection and ranging (lidar) **scanning system** for a landing spacecraft

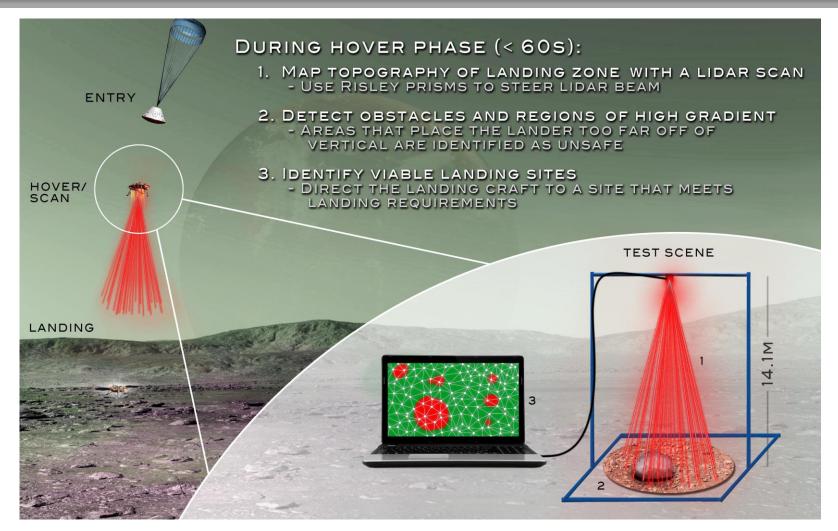
#### **Success Levels:**

- 1. Lidar sensor and scanning mechanism, mounted on a stationary platform, shall **record correlated range and attitude measurements** at a 0.1 m spatial resolution from a nadir distance of 14.1 m with a maximum 20° off nadir
- 2. System shall scan a known test scene and **project measurements into** a 3D point cloud
- 3. System shall scan a landing-zone mockup and **analyze the 3D point** cloud for hazards
- 4. System shall **select a safe landing zone**; if no safe landing zone is found, hazard definition will be loosened until a landing zone is found



# Concept of Operations







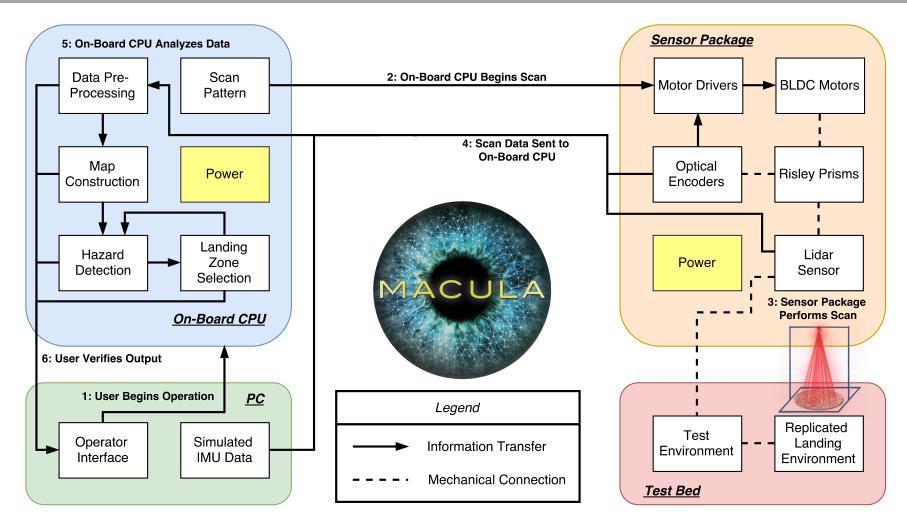


## DESIGN SOLUTION



## Functional Block Diagram



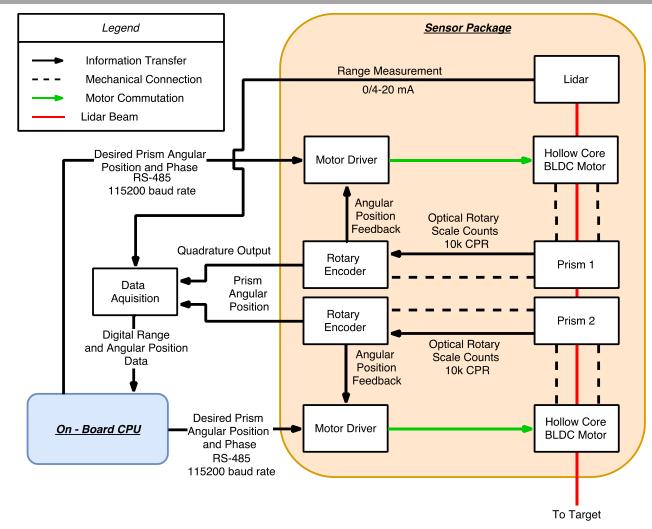




#### Hardware Architecture Diagram



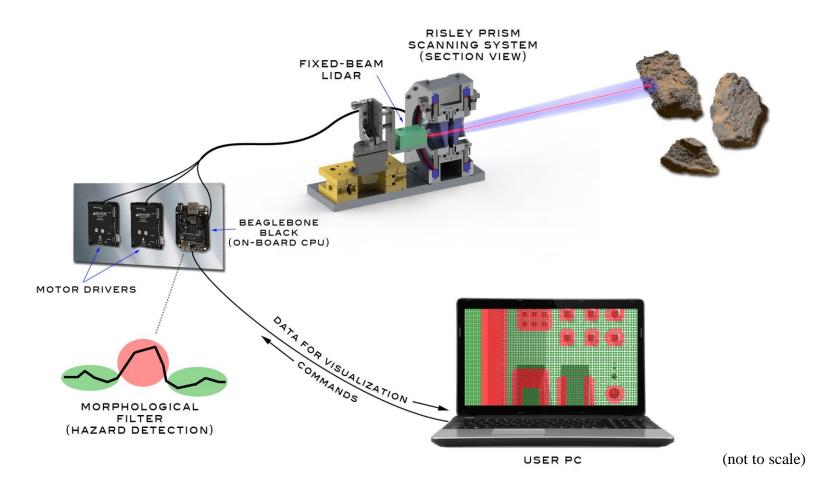
9





# Final Design Overview

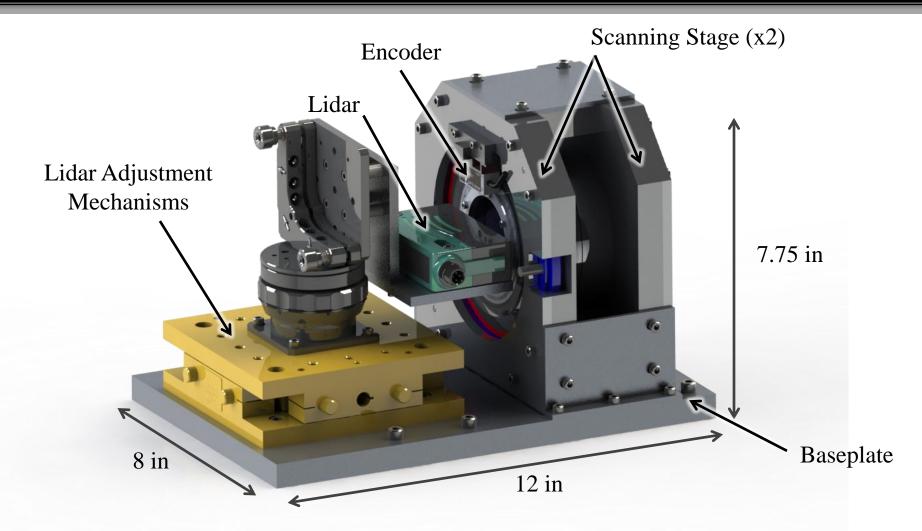






# Full Scanning System

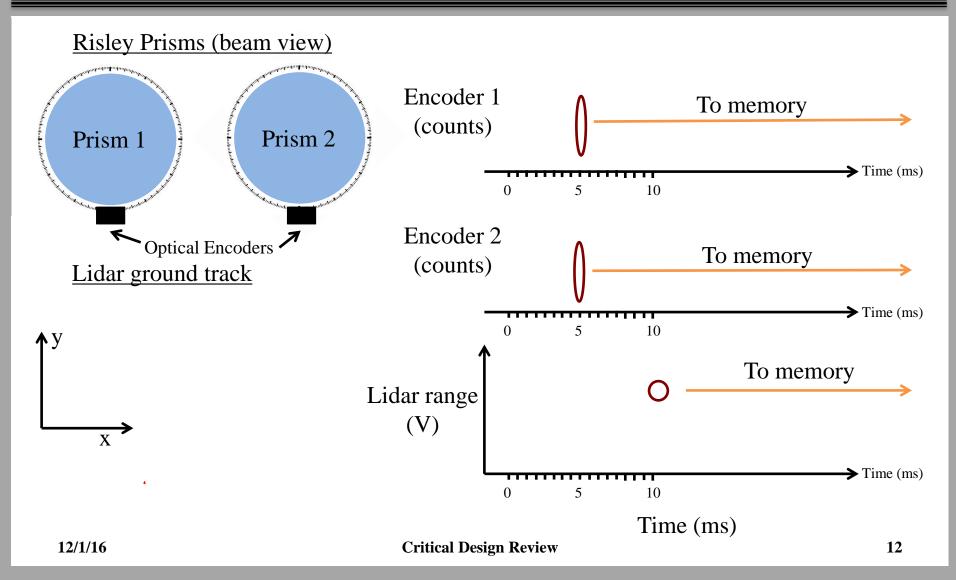






# Timing

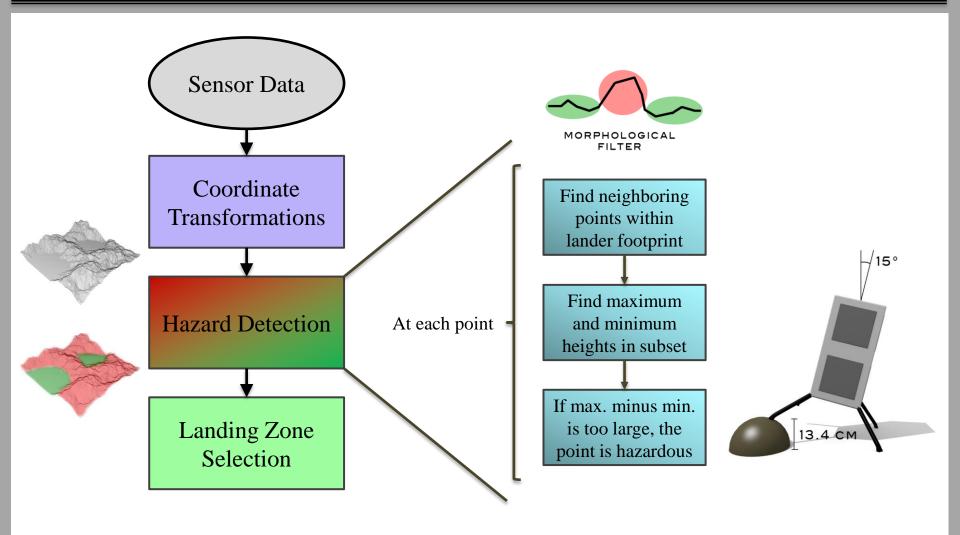






## Software









# CRITICAL PROJECT ELEMENTS



# Critical Project Elements



- 1. Optics (direct laser pulse and obtain range measurement)
  - Lidar
  - Risley Prisms
- 2. Risley Prism Control (achieve desired scan pattern)
  - Scan Pattern
  - Motors
  - Motor Drivers
  - Encoders
- 3. Embedded System (sensor communication)
  - Microcontroller + Motor Drivers/Encoders/Lidar





# DESIGN REQUIREMENTS AND THEIR SATISFACTION



# Functional Requirements

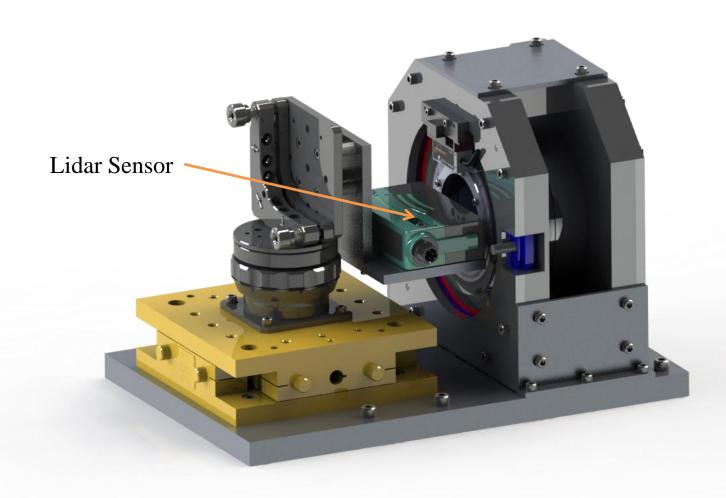


- **FR 1:** The system shall analyze a potential landing zone for a 12U cubesat.
- **FR 2:** The on-board processor (OBP) shall receive commands and data from a user-operated PC (UPC).
- **FR 3:** The OBP shall command the sensor package (SP).
- **FR 4:** The SP shall use fixed-beam lidar sensor to obtain range measurements.
- **FR 5:** The SP shall have control over the direction of the lidar beam direction using two Risley prisms.
- **FR 6:** The OBP shall receive data from the SP.
- **FR 7:** The OBP shall project the SP data into a 3D point cloud.
- FR 8: The OBP shall analyze the 3D point cloud to identify hazardous locations.
- FR 9: The OBP shall select an acceptable landing site.
- FR 10: The OBP shall generate output readable by the UPC.



## DRs: Lidar Sensor







#### DRs: Lidar Sensor



#### Pepperl+Fuchs VDM28

• **FR 4:** The SP shall use fixed-beam lidar sensor to obtain range measurements.

	Required	Pepperl+Fuchs VDM28
Range	12 m - 15 m	0.2 m - 15 m
Range Error	<0.05 m	0.025 m
Cross Range Error	<0.045 m	0.0080 m

Cost: ~\$500 (donated)

Sampling Frequency: 100 Hz

Wavelength: 660 nm



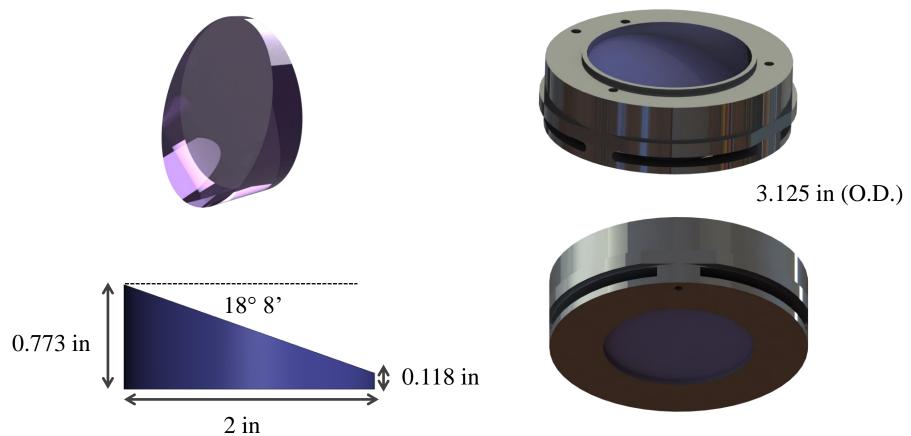


# DRs: Risley Prisms



**FR 5:** The SP shall have control over the direction of the lidar beam direction using two Risley prisms.

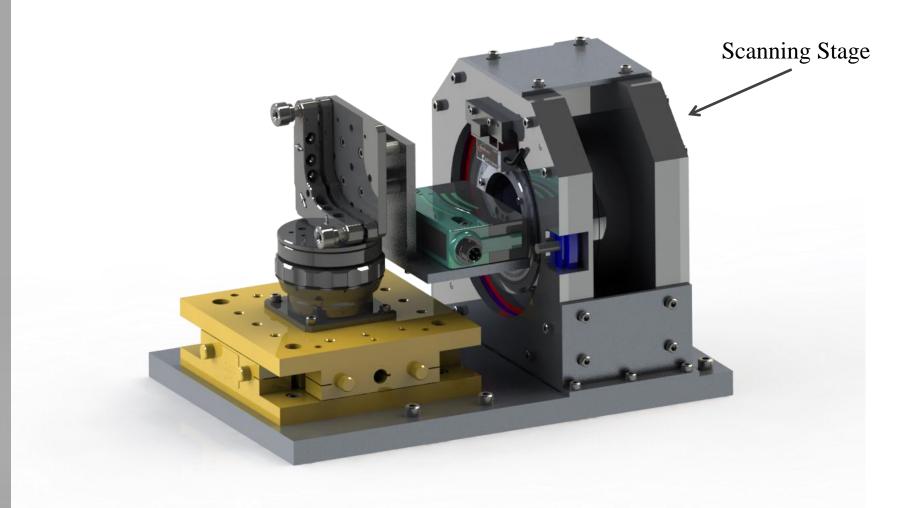
**DR 5.1.1:** 20° scan angle, **DR 5.2:** Individually controlled, **DR 5.4:** Shall not inhibit lidar sensor





# DRs: Risley Prisms

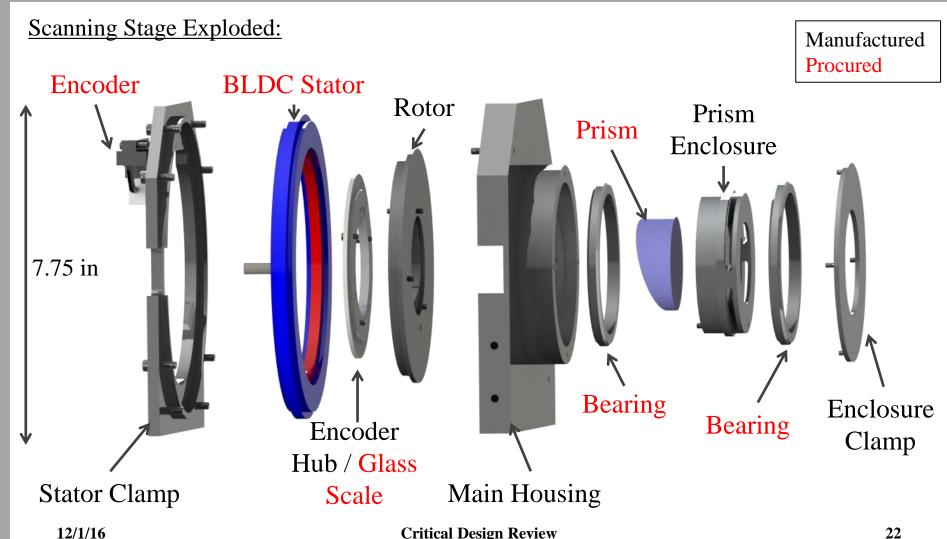






# DRs: Risley Prisms





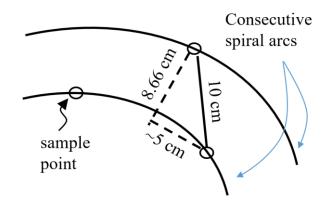


#### DRs: Scan Pattern

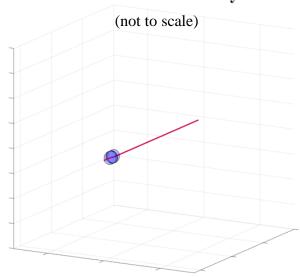


**DR 1.1-1.4:** 10 cm spatial resolution at 14.1 m nadir range with 20° maximum scan angle

Spiral with 59.2 revolutions spaced 8.66 cm apart



#### **Animation of Scan with Risley Prisms**



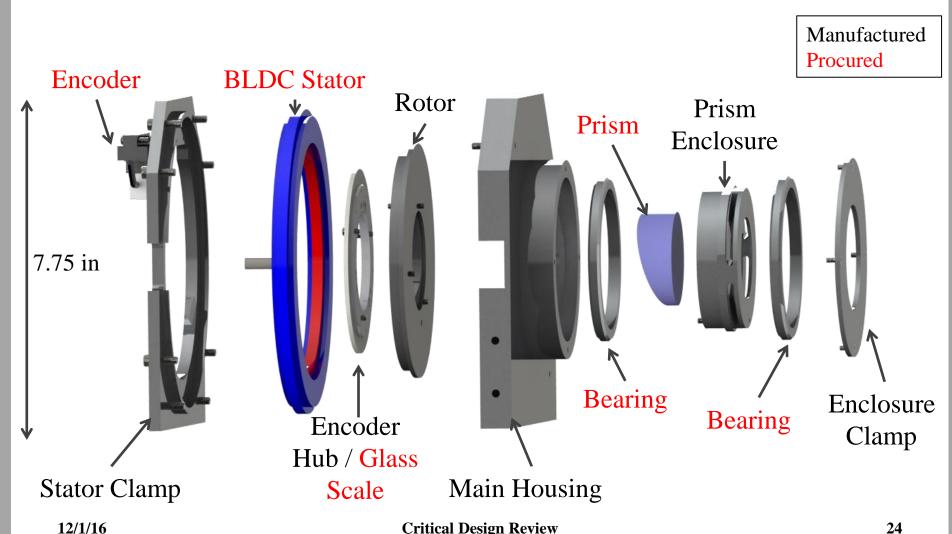
What will this design choice mean for the Risley prism actuation system?

	Time Test	Resolution Test
Max Speed [rad/s]	8.46	0.55
Min Speed [rad/s]	7.34	0.48
Max Acceleration [rad/s <sup>2</sup> ]	14.36	0.05



#### DRs: Motors







### **DRs: Motors**



**DR 5.2.1**: 15 rad/s<sup>2</sup>, **DR 5.2.2**: 0.45 rad/s - 10 rad/s

Inertia of rotating components: 9.5x10<sup>-4</sup> kg m<sup>2</sup>

#### Direct Drive:

Most mechanically simple solution that satisfies requirements

#### Brushless DC (BLDC) Motor:

Large hollow core required for optics Fine continuous control of pointing or speed

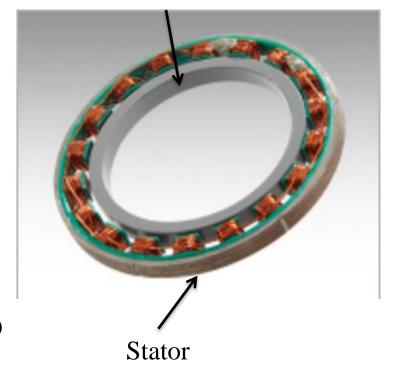
#### ULT-165-A-12-A-x-00x:

Up to 5000 RPM (~523 rad/s)

Max continuous torque of 1.255 Nm (~1200 rad/s²)

4.4 in. rotor inner diameter

#### Rotor





#### DRs: Motor Drivers



#### Advanced Motion Controls DPRALTE (~\$600 each)

**DR 3.3**: The OBP shall send commands to the motor drivers.

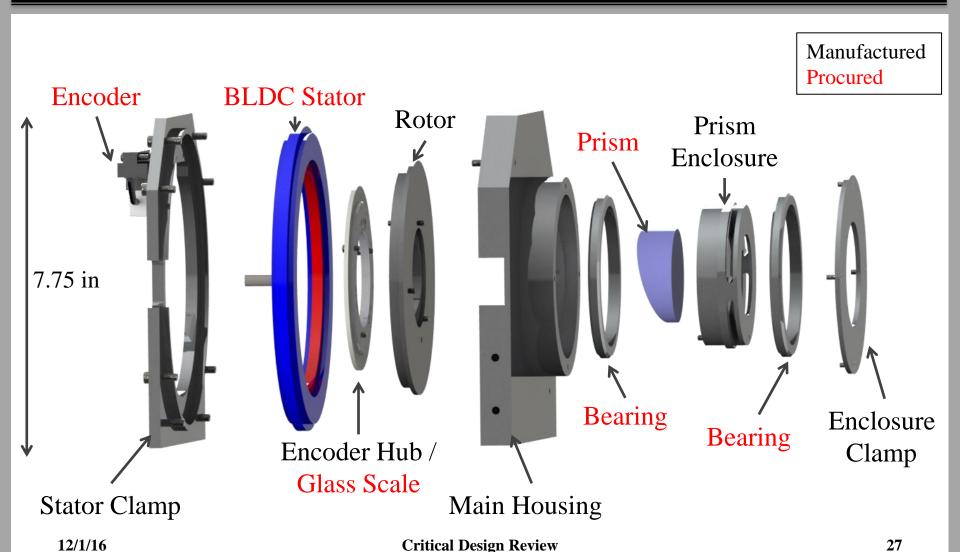
	Design Req.	DPRALTE
Commanding	Shall be commanded from OBP at least once every 10 ms	RS485 and step input
Motor Rates	4.3 – 96 RPM (0.45 – 10 rad/s)	4 – 120 RPM (0.42 – 12.6 rad/s)





## DRs: Angular Position Sensor







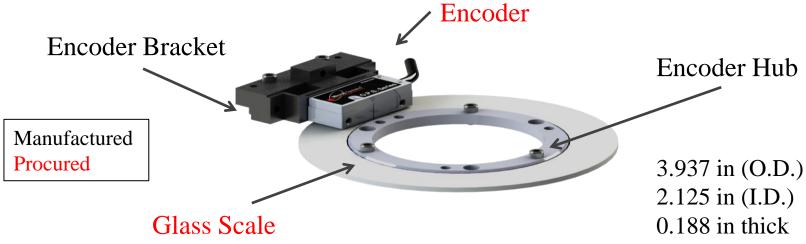
## DRs: Angular Position Sensor



#### Celera Motion OPS-SM-40 (~\$600 each)

Prism orientation measurements are required for motor control and for beam attitude knowledge.

	Design Req.	OPS-SM-40
Angular Uncertainty	Stackup	± 0.009°
Measurement Time	≤ 1 µs	5 ns



12/1/16

**Critical Design Review** 



## DRs: Microcontroller (OBP)



#### BeagleBone Black (~\$55)

FR 2: The OBP shall receive commands and data from a user-operated PC.

**FR 3:** The OBP shall command the sensor package.

**DR 4.6:** Lidar range and beam attitude measurements shall be taken within one microsecond.

**FR 6:** The OBP shall receive data from the sensor package.

FR 7-9: Analyze sensor measurements to determine a safe landing site.

FR 10: The OBP shall generate output readable by the PC.

	Design Req.	BeagleBone Black
Communication with PC	Receive commands & data, update status, write results	UART, USB, Ethernet
Communication with sensor package	Command motor drivers	UART <b>←→</b> RS485, GPIO
Measurement time	≤ 1 µs	~600 ns (full ADC conversion)
Crunch data	FPU→10 second computation time	~ capable of 1 Gigaflop





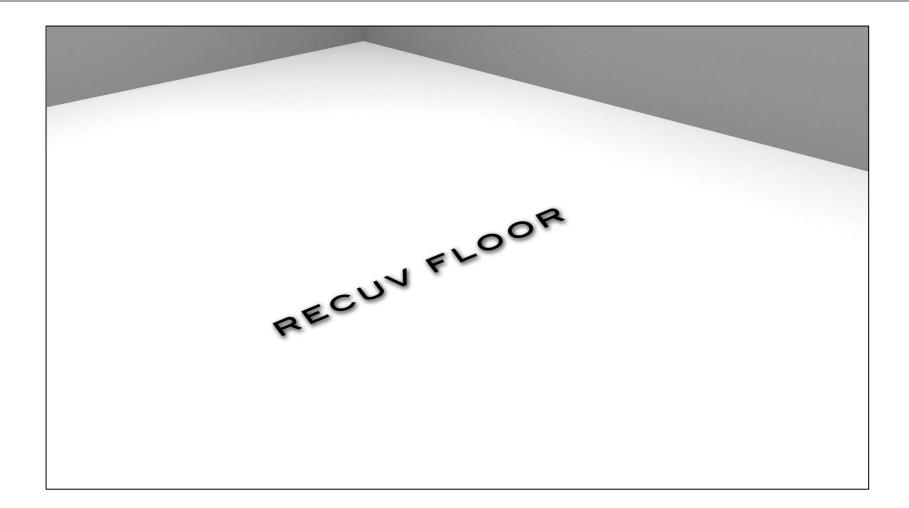
#### Calibration Plan



- This calibration test plan determines the alignment of the lidar and prisms to within a tolerance that meets our pointing knowledge requirement
  - DR: Error  $\leq$  5 cm in the plane of the ground
- Most of the error in the beam steering is systematic and can be accounted for in software
- Reassembling the system after calibration introduces uncertainties that will not be calibrated

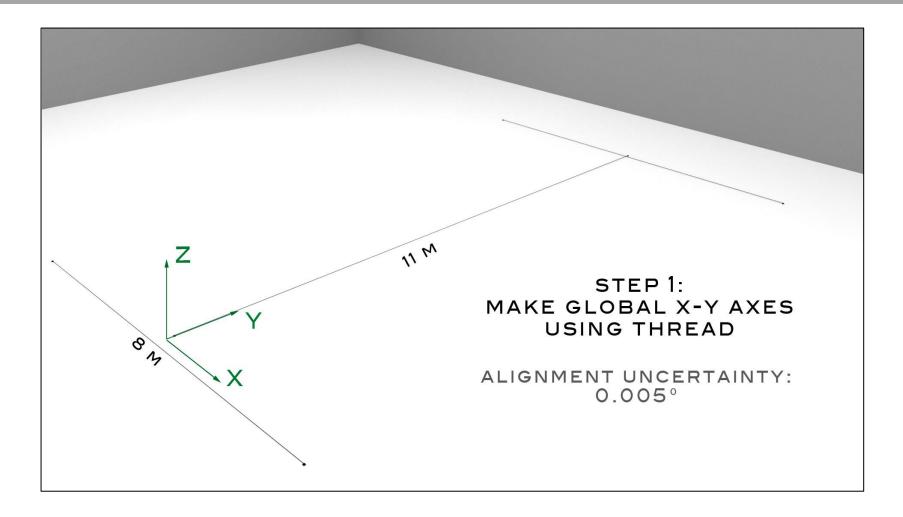






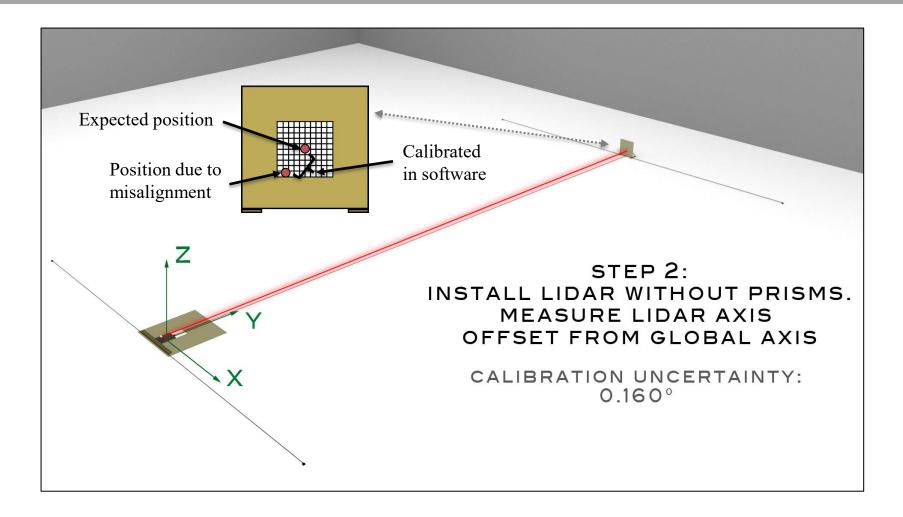






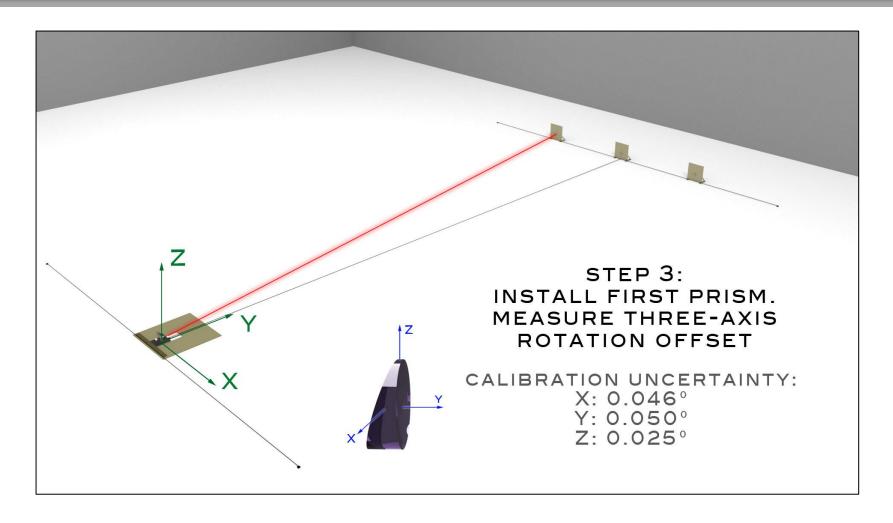






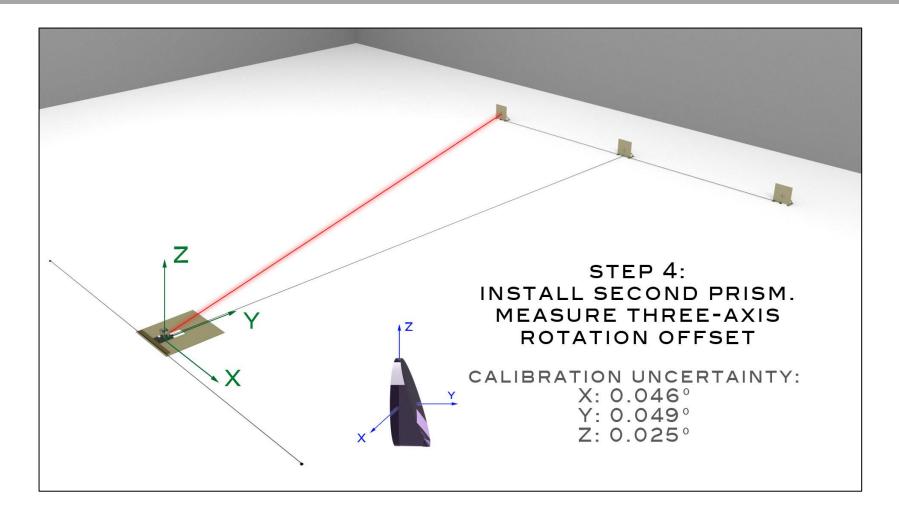








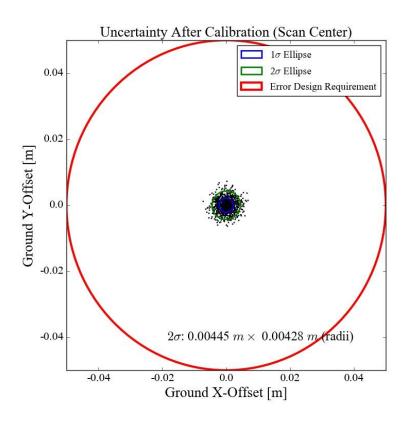


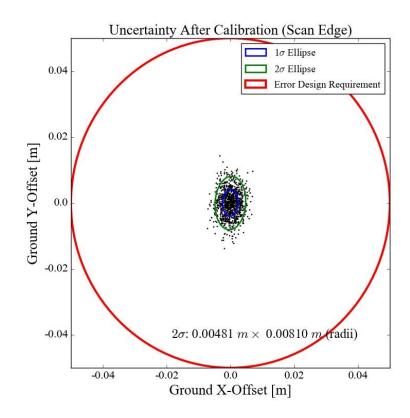






• This shows that the system does not need to be permanently fixed after our calibration tests









## PROJECT RISKS



# Risk Summary



R1	Reflection off of prisms causes false return
R2	Encoder mount damage occurs during shipping, glass scale installation, or storage
R3	Limited machine shop availability due to schedules and use by other students
R4	Risley prism damage occurs during storage or handling

		Severity				
		1	2	3	4	5
	5					
po	4		R3			
Likelihood	3					R1
	2				R2/4	
	1					



# Risk Summary



R5	RECUV test facility unavailable at necessary times and/or necessary durations
R6	Delays during manufacturing process
R7	Loss of calibration resulting from moving system around and/or adjusting system during testing

		Severity				
		1	2	3	4	5
	5					
Likelihood	4					
	3			<b>R6/R7</b>		
	2			R5		
	1					



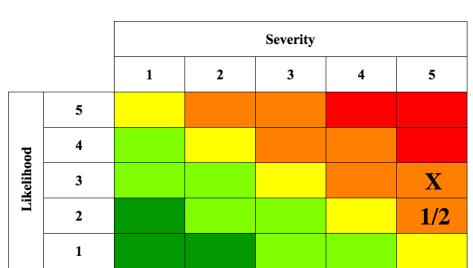
# Mitigation



R1	Reflection off of prisms causes false return						
	1. Add broadband a	1. Add broadband antireflective coating to prisms (for 400-700 nm wavelength)					
		Reduces likelihood of risk					
	Test using two different lidar sensors (diffuse & specular detectors)  Reduces likelihood of risk						
,							

#### Type of Risk

- □ Budget
- ☑ Technical
- □ Safety
- □ Schedule





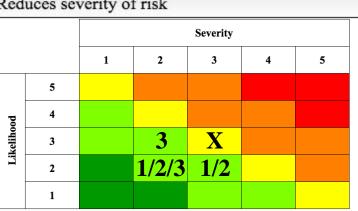
# Mitigation



R6	Delays during manufacturing process					
	Utilize full extent of machining resources available, included multiple shops     (Aero, ITLL) and full time employees who will machine parts					
	Reduces likelihood of risk					
	2. Manufacture parts in parallel using all available team members					
	Reduces likelihood of risk					
	3. Purchase extra stock that can be used if parts need to be remade					
	Stock materials can be bought with 20% budget margin					
	Reduces severity of risk					
	Savarity					

#### Type of Risk

- **☑** Budget
- □ Technical
- □ Safety
- ☑ Schedule





# Mitigation



R7	Loss of calibration resulting from moving system around and/or adjusting system during testing					
	1. Place covers over adjustment knobs to prevent accidental adjustments					
	Reduces likelihood of risk					
	2. Include margin time during testing to allow for calibration					
	Reduces severity of risk					

#### Type of Risk

- □ Budget
- ☑ Technical
- □ Safety
- ☑ Schedule

S						
		1	2	3	4	5
	5					
Likelihood	4					
	3		2	X		
	2		1/2	1		
	1					





# VERIFICATION AND VALIDATION

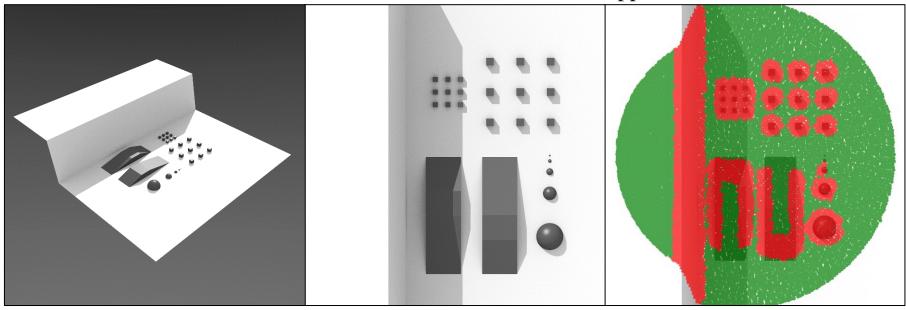


## Predicted Performance



- The lidar-generated point cloud will be compared against the physical dimensions of the test bed
- Hazard detection results will be compared against simulated scans

Hazard detection with all uncertainties applied

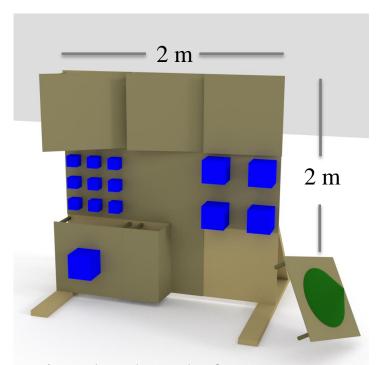




# Test Setup



- Construct a landing zone mockup with known dimensions
- Nine modular test panels
- Characterization tests
  - Flat wall
  - Safe and hazardous slopes
  - Safe and hazardous cubes
  - Cliff
  - Cone
  - Rocky shapes
- Can measure dimensions manually or with handheld 3D scanner
  - Compare lidar-generated point cloud to point cloud made from measurements to determine accuracy





#### **Test Process**



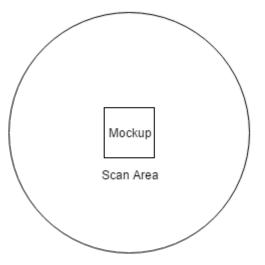
- Two system tests must be conducted to meet requirements
  - Resolution requirement
    - 12.5 minutes
    - Must accurately identify hazards
  - Time requirement
    - 50 seconds for complete scan
    - 59.2 spirals

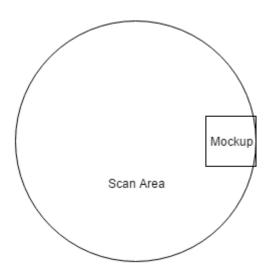


#### **Test Process**



- Place mockup centered at nadir
  - Verify mockup is correctly represented by a full scan
- Place mockup with edges aligned with 20° outer scan radius
  - Verify mockup is correctly represented by a full scan
- Producing accurate results at these locations verifies correct operation at extremities
  - Verifies full capabilities





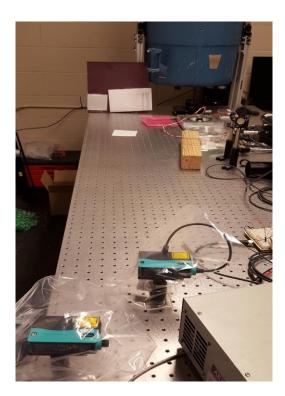


## **Facilities**



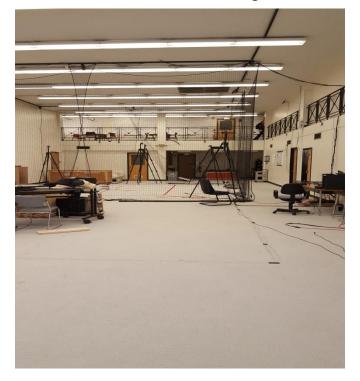
#### • <u>ARSENL</u>

- Optics bench
- Component testing



#### • <u>RECUV Lab</u>

- Length = 21.7 m
- Height = 9.5 m
- Width = 9.2 m
- Full-scale testing





# Equipment



- Wood
  - Physical landing zone mockup
- Measuring devices
  - Tape measure
    - Measure large distances
    - Construct landing zone mockup
    - $\pm 3.5 \text{ mm}$
  - Calipers
    - Construct landing zone mockup
    - $\pm 0.02 \text{ mm}$
  - Theodolite
    - Accurately determine angles and heights for calibration tests
    - $\pm 2$ -9 arcseconds  $\approx \pm 1$  mm height measurement
- Optics bench
  - Component-level testing







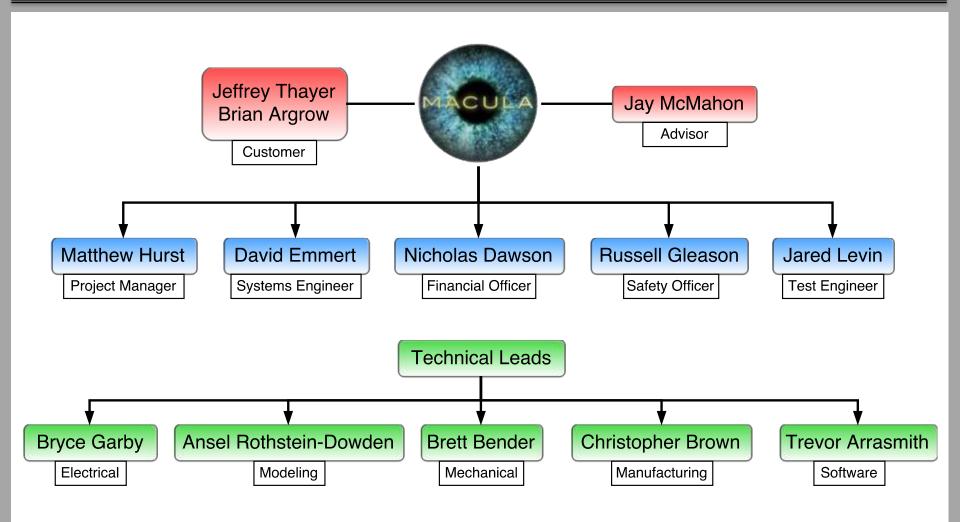


## PROJECT PLANNING



# Organizational Chart

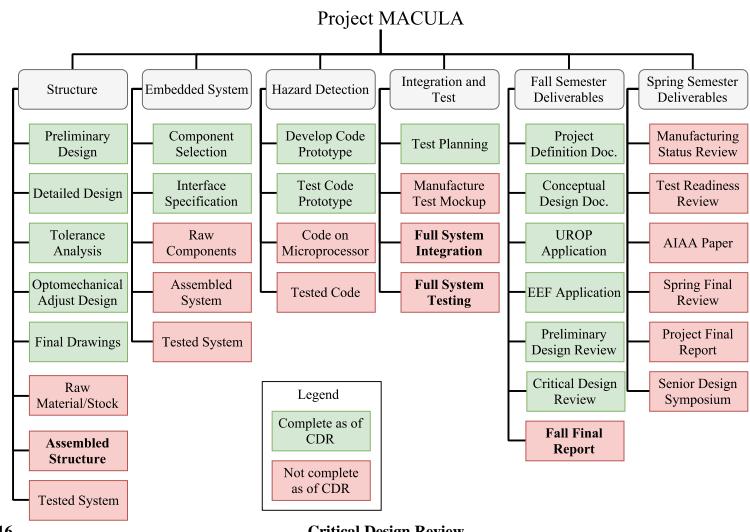






#### Work Breakdown Structure

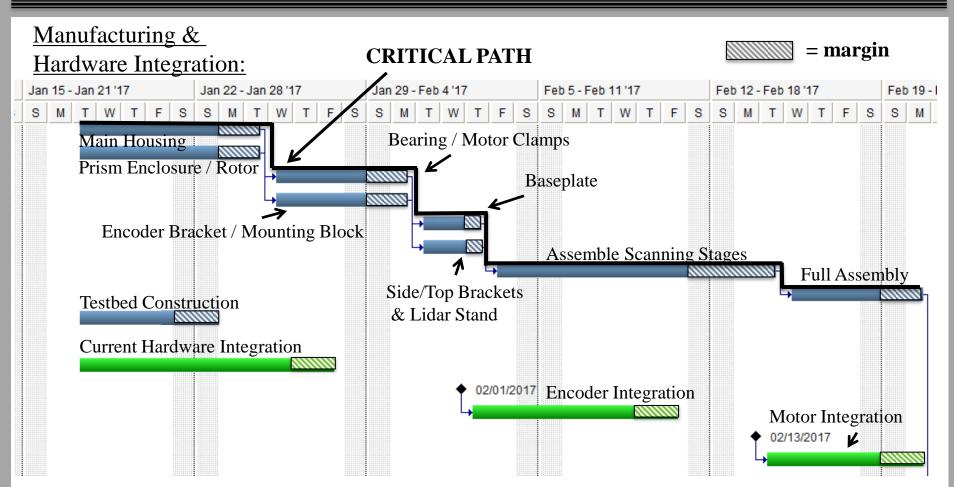






#### Work Plan





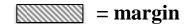
• Note: Encoder Hub will be manufactured before winter break

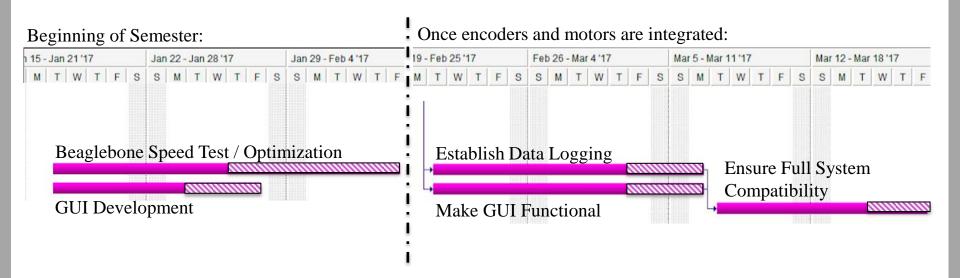


## Work Plan



#### Software:







## Cost Plan



		C . (11CD)	D: (USD)	T (116D)
Item	Quantity	Cost per (USD)	Discount (USD)	Total cost (USD)
Lidar:				
Pepperl-Fuchs VDM28-15-L-IO/73c/110/122	1	470.17	470.17	0
Motors:				
ULT-165-A-12-A-x-00x	1	1033		1033
Encoders:				
OPS read head (OD 3.937" ID 2.756")	2	560		1120
Bearings:				
VA030CP0 Thin Section Bearing 3"x3 1/2"x1/4"				
inch Open	4	81.77		327.08
Microprocessor:				
BeagleBone Black Rev C.1	1	55		55
Reflective Tape				
Oralite (Reflexite) R99 Rail Microprismatic				
Retroreflective Conspicuity Tape: 4 in. x 15 ft.	9	36.22		325.98
Motor Drivers:				
DPRALTE - 020B080	2	605	363	847
Risley prism:				
P-WRC059 coated with BBAR 400-700 nm	2	108		216



# Cost Plan

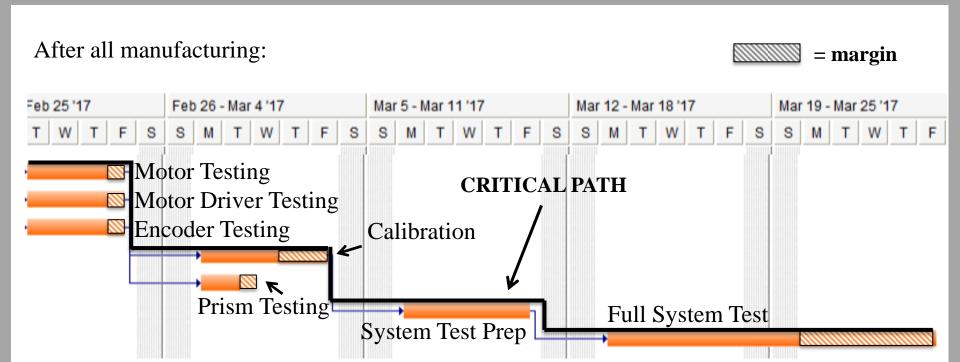


Item	Quantity	Cost per (USD)	Discounts (USD)	Total cost (USD)
Materials				
Mat: Misc electronics				300
Mat: Fasteners				166
Mat: Stock				0
Mat: Tooling				812
Shipping (assuming \$15 per item)				330
Total (Cost)				6381.83
Budget			CU Aerospace	5000
			UROP	1000
			Customer	2300
Total (Budget)				8300
Percent Margin				23.11%



#### Test Plan







# Acknowledgements



Advisor: Jay McMahon

PAB: James Nabity, Kaley Pinover, Brian Argrow, Bobby Hodgkinson, Matt Rhode, Trudy Schwartz, Bob Marshall, Josh Stamps, Jelliffe Jackson

Active Remote Sensing Lab: Jeff Thayer, Rory Barton-Grimley, Bobby Stillwell

Computational and Mechanical Geometry Lab: John Evans, Luke Engvall, Joseph Benzaken

Dale Lawrence

Blue Canyon Technologies: Steve Steg, Matt Carton, Bryce Peters

Pepperl+Fuchs: Michael Turner









# QUESTIONS?



# Backup Master



#### <u>Main:</u>

Purpose and Objectives

**FBDs** 

**Design Overview** 

**CPEs** 

<u>FRs</u>

<u>Lidar</u>

**Prisms** 

Scan

**Motors** 

**Motor Drivers** 

Encoders

**Controllers** 

Microcontroller

Calibration

Risks

Verification and Validation

**Planning** 

#### **Backup:**

References

**Cubesat Lander Concept** 

**Requirements** 

**Lidar Sensor** 

Retroreflection

**Risley Prisms** 

**Prism Mounting** 

Prism Positions: Forward

**Problem** 

**Prism Positions:** 

**Backward Problem** 

System Assembly

**Drawings** 

**Material Analysis** 

Risk Analysis

Hardware Trades

**Prism Control** 

**Calibration Testing** 

Software

**BeagleBone** 

**Measurements** 

**Encoder Integration** 

Communication

Power

Connections

Verification and Validation

**Budget** 



### References



- "AMG Series Optical Mounts & Gimbals." Aerotech. Aerotech Inc., n.d. Web. 22 Sept. 2016. <a href="https://www.aerotech.com/product-catalog/gimbals-and-optical-mounts/amg.aspx">https://www.aerotech.com/product-catalog/gimbals-and-optical-mounts/amg.aspx</a>.
- Berg, Mark . Computational Geometry: Algorithms and Applications. Berlin: Springer, 2008. Print.
- By FAS March 23, 2016. "Federation Of American Scientists -." Federation Of American Scientists. N.p., n.d. Web. 24 Sept. 2016.
- Cryan, Scott, and Christian, John A. "A Survey of LIDAR Technology and its Use in Spacecraft Relative Navigation." AIAA Guidance, Navigation, and Control (GNC) Conference, Boston, MA, 19-22 Aug. 2013. Web.
- Dunbar, Brian. "Lidar Atmospheric Sensing Experiment (LASE): Measuring Water Vapor, Aerosols and Clouds." NASA. NASA, 21 Nov. 2004. Web. 24 Sept. 2016.
- Folger, Jean. "Why Curiosity Cost \$2.5 Billion." Investopedia, LLC. 5 Sept. 2012. Web.
- Hughes, T. J. R, J. Cottrell A., and Y. Bazilevs. "Isogeometric Analysis: CAD, Finite Elements, NURBS, Exact Geometry and Mesh Refinement." Computational Methods in Applied Mechanics and Geometry 194.39-41 (2005): 4135-195. Web.
- Gonzales, Rafael C. Digital Image Processing. 3rd ed. N.p.: Pearson Education International, n.d. Print.
- "LeddarTech Launches LeddarVu, a New Scalable Platform Towards High-Resolution LiDAR." PR Newswire Association LLC. Quebec City, 7 Sept. 2016. Web.



### References



"LiDAR, Laser Scanners and Rangefinders." RobotShop, Inc. 2016. Web. <a href="http://www.robotshop.com/en/lidar.html">http://www.robotshop.com/en/lidar.html</a> >

Lynch, Andrew, and Kai Focke. "Beam Manipulation: Prisms vs. Mirrors." Photonik International (n.d.): n. pag. Mar. 2009. Web. 20 Sept. 2016. <a href="http://www.edmundoptics.com/globalassets/resources/articles/beam-manipulation-prisms-vs-mirrors-en.pdf">http://www.edmundoptics.com/globalassets/resources/articles/beam-manipulation-prisms-vs-mirrors-en.pdf</a>.

"Puck LiDAR: Our Lightest Yet." Velodyne LiDAR, Inc., 2016. Web. <a href="http://velodynelidar.com/vlp-16-lite.html">http://velodynelidar.com/vlp-16-lite.html</a>

"Round Wedge Prisms." Round Wedge Prisms. THORLABS, n.d. Web. 20 Sept. 2016. <a href="https://www.thorlabs.com/NewGroupPage9.cfm?ObjectGroup\_ID=147">https://www.thorlabs.com/NewGroupPage9.cfm?ObjectGroup\_ID=147</a>.

Saniee, Kamron. "A Simple Expression for Multivariate Lagrange Interpolation." SIAM (2007): n. pag. Web.

Schwarze, Craig. "A New Look At Risley Prisms." Photonics Spectra (n.d.): n. pag. Photonics Media, June 2006.
Web. 20 Sept. 2016.

Shan, Jie, and Charles Toth K. Topographic Laser Ranging and Scanning: Principles and Processing. Boca Raton: CRC/Taylor & Francis Group, 2009. N. pag. Print.

Shewchuk, J. "What is a good linear finite element? interpolation, conditioning, anisotropy, and quality measures (preprint)." University of California at Berkeley 73 (2002).



#### References



"Technology Overview." Advanced Scientific Concepts, Inc. 2015. Web. <a href="http://www.advancedscientificconcepts.com/technology/technology.html">http://www.advancedscientificconcepts.com/technology/technology.html</a> >

Wilsenack, Frank. "Defense & Security." Detecting and Tracking Thin Aerosol Clouds. SPIE, 12 Oct. 2012. Web. 24 Sept. 2016.

Y. Bazilevs et al., Isogeometric analysis using T-splines, Comput. Methods Appl. Mech. Engrg. (2009), doi:10.1016/j.cma.2009.02.036

Riemersma, Thiadmer. "Candela, Lumen, Lux: the equations." CompuPhase, 2016. Web.

"Standard Test Method for Coefficient of Retroreflection of Retroreflective Sheeting Utilizing the Coplanar Geometry." ASTM International, Designation E 810-03, 2016. PDF.

"Pepperl+Fuchs VDM 28 Photoelectric Sensor Datasheet." Pepperl+Fuchs, 2016. PDF.

"Reflexite Daybright V92 Conspicuity Sheeting." Reflexite America, 2006. PDF.





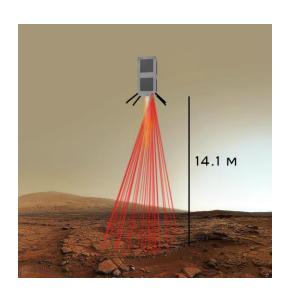
# BACKUP: GENERAL CONCEPT

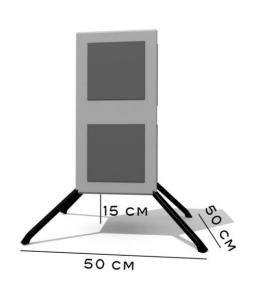


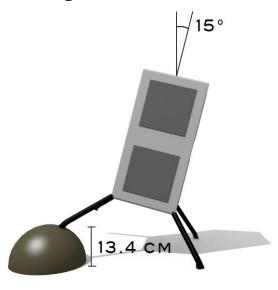
# CubeSat Lander Concept



- Landing hazard definition based on hypothetical CubeSat lander dimensions
- Hazards (obstacles and gradients) identified where the lander could land more than 15° off of vertical
- Scanning resolution of 10 cm selected to detect ~98% of potential hazards







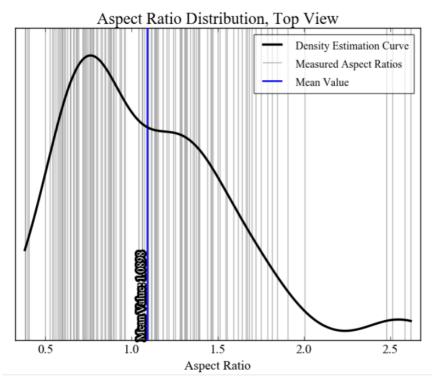


### Top View Martian Rock Analysis





**Top view of Martian surface** 

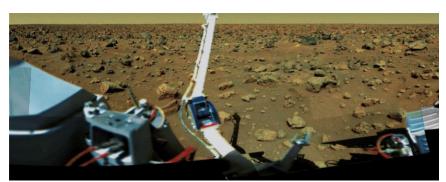


Aspect ratio distribution

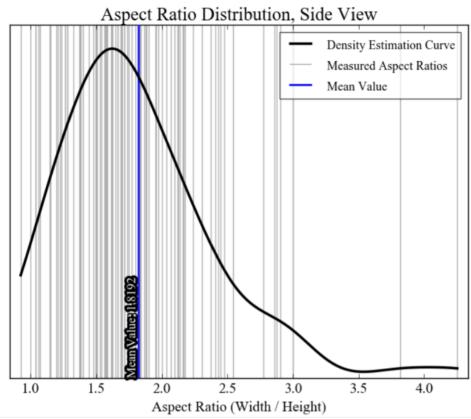


#### Side View Martian Rock Analysis





**Side view of Martian surface** 



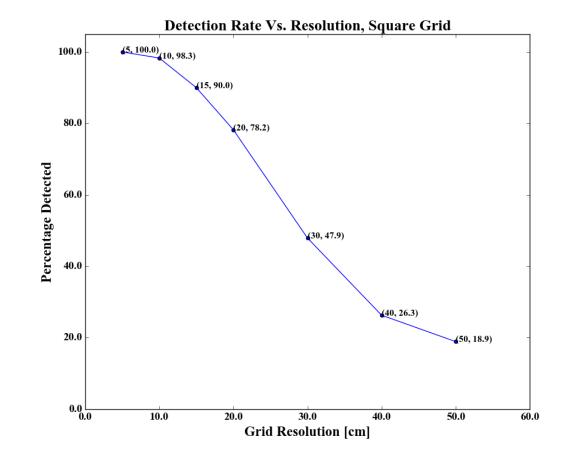
Aspect ratio distribution



# Resolution Requirement



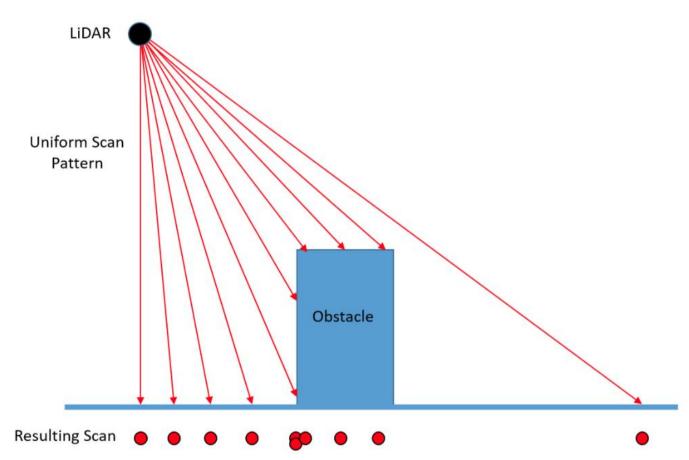
- Statistical analysis of rock size/aspect ratios on Mars
- Created a software map of a characteristic landing surface
- Monte Carlo simulation with different scan resolutions
- Determine probability of aliasing over a hazard (failure) vs. scan resolution





# Shadowing



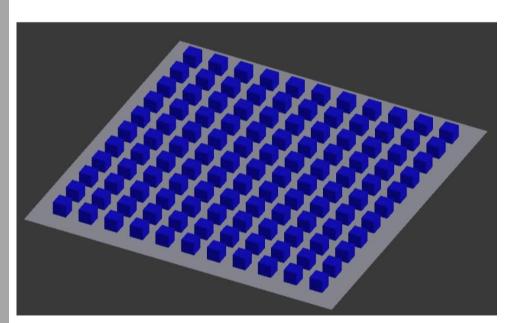


Visualization of shadowing effects

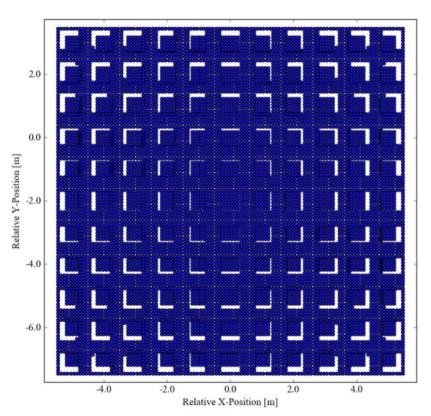


# Maximum Scan Angle





Blender visualization of a test scene



Resulting points with lidar scan





# BACKUP: DESIGN REQUIREMENTS



#### FR 1



- 1. The system shall analyze a potential landing zone for a 12U cubesat.
  - 1.1. The system shall scan up to a half-angle of 20° off of nadir.
  - 1.2. The system shall scan from a nadir range of 14.1 m.
  - 1.3. The system shall scan with a resolution of better than 0.1 m.
    - 1.3.1. The error in this resolution shall be less than 0.05 m in the plane of the scan area.
  - 1.4. The system shall complete the scan and analysis in less than 60 seconds.
    - 1.4.1. The system shall complete the scan in less than 50 seconds.
    - 1.4.2. The system shall complete the analysis in less than 10 seconds.





- 2. The on-board processor (OBP) shall receive commands and data from a user-operated PC (UPC).
  - 2.1. The OBP shall execute a main driver routine.
    - 2.1.1. While executing the main driver, the OBP shall receive a "ready" command from the UPC.
    - 2.1.2. After a "ready" command is received, the OBP shall receive a "start command from the UPC.
    - 2.1.3. During operation (after a "start" command) the system shall receive a "stop" command from the UPC.
      - 2.1.3.1. Upon receiving a "stop" command, the system shall stop operation (cut power to the lidar and the motors) within 1 second.





- 2. The on-board processor (OBP) shall receive commands and data from a user-operated PC (UPC).
  - 2.2. Outside of the main driver, the OBP shall receive and store data from the UPC.
    - 2.2.1. The OBP shall receive and store in memory a list of Risley Prism orientations for the desired scan.
    - 2.2.2. The OBP shall receive and store in memory a list of simulated IMU values for the spacecraft.
      - 2.2.2.1. These IMU data shall be a Direction Cosine Matrix (DCM) for the system at each scan time, relative to the scan surface.
    - 2.2.3. The OBP shall retain memory while powered and unpowered.
    - 2.2.4. The OBP shall be programmable so that stored data and routines can be modified through the interface with the UPC.





- 3. The OBP shall command the sensor package (SP).
  - 3.1. The OBP shall control power (on or off) to the lidar sensor.
  - 3.2. The OBP shall control power (on or off) to the motors.
  - 3.3. The OBP shall send commands to the motor drivers.
    - 3.3.1. The OBP shall read the data file of Risley prism orientations to send to the motors.
    - 3.3.2. The desired prism orientations must be updated at least once every 10 ms.





- 4. The SP shall use a fixed-beam lidar sensor to obtain range measurements.
  - 4.1. The lidar shall operate within a range of 12 m 15 m.
  - 4.2. The lidar shall have a range error with a standard deviation of less than 2.5 cm at all ranges between 12 m and 15 m.





- 5. The SP shall have control over the lidar beam direction using two Risley prisms.
  - 5.1. The Risley prisms shall be capable of actuating the beam across the entire scan area.
    - 5.1.1. The Risley prisms together shall be capable of deflecting the lidar beam by at least 20° from nadir.
  - 5.2. The Risley prisms shall be individually controlled in order to direct the lidar beam.
    - 5.2.1. The Risley prism actuation system shall be capable of producing sufficient torque to achieve 15 rad/s^2.
    - 5.2.2. The Risley prism actuation system shall be capable of producing angular rates between 0 rad/s and 10 rad/s.
  - 5.3. After system calibration, the lidar shall have a cross-range error with a standard deviation of less than 2.5 cm for all locations in the scan area.
    - 5.3.1. The sum of two standard deviations plus the radius of the beam spot shall not exceed 5 cm at any point in the scan area.
    - 5.3.2. The Risley prism orientations shall be known to within 0.1° about the axis of rotation.





- 5. The SP shall have control over the lidar beam direction using two Risley prisms.
  - 5.4. The SP shall not inhibit the lidar sensor from receiving a return signal.
    - 5.4.1. The Risley prism receiver field of view shall be less than 50% obscured.
    - 5.4.2. The transmissivity of the Risley prisms shall allow for a beam return of at least 90% strength, assuming a perfect specular retroreflection from the target.
      - 5.4.2.1. The Risley prisms shall be covered with an anti-reflective coating appropriate for the lidar wavelength.
    - 5.4.3. The Risley prism actuation system shall not impede the optical path of the lidar beam for any orientation within the scan area.





- 6. The OBP shall receive data from the SP.
  - 6.1. The OBP shall read and save the lidar range measurement to memory every 10ms.
    - 6.1.1. The output of the lidar sensor shall be converted into a voltage.
    - 6.1.2. The voltage shall be readable by the OBP Analog to Digital Converted (ADC).
      - 6.1.2.1. The ADC shall have a resolution of at least 12 bits.
  - 6.2. The OBP shall read the prism orientation measurements.
    - 6.2.1. The OBP shall read the quadrature output of the each encoder continuously to translate into a count.
      - 6.2.1.1. Each count shall be translated into an absolute angular position of each prism.
    - 6.2.2. Each prism orientation shall be saved to memory every 10ms.
  - 6.3. The lidar range measurement and prism orientations shall be correlated such that prism orientations from t=0 match with lidar ranges from t=5ms.





- 7. The OBP shall project the SP data into a three-dimensional (3D) point-cloud.
  - 7.1. The OBP shall translate the prism orientations into a location (origin) for the outgoing lidar beam.
  - 7.2. The OBP shall translate the prism orientations into a direction vector for the outgoing lidar beam.
  - 7.3. The OBP shall project the range measurement along the computed direction vector, then add this to the computed origin to find a point in an intermediate cartesian frame relative to the lidar emitter.
  - 7.4. The OBP shall rotate the point in the intermediate frame into an inertial frame using the simulated IMU data.
  - 7.5. These calculations shall occur as the scan is being completed.





- 8. The OBP shall analyze the 3D point-cloud to identify hazardous locations.
  - 8.1. The OBP shall begin analysis once the scan points have reached a distance of 0.45 meters from nadir.
  - 8.2. The OBP shall process a scan point by finding all points within the error-compensated lander footprint range, then computing the maximum height difference between of all these points. A safe point is one where this difference does not exceed the error-compensated hazard height.
  - 8.3. The points shall be analyzed in the order in which they arrive.
  - 8.4. A point shall be analyzed if and only if it is the next point in the queue and the distances from nadir of the most recently found points has exceeded the sum of the distance of the queued point from nadir and its error-compensated lander footprint range.





- 9. The OBP shall select an acceptable landing site.
  - 9.1. The OBP shall identify the first computed safe point as the acceptable landing site.





- The OBP shall generate output readable by the UPC.
  - 10.1. The OBP shall generate health and status information readable by the PC in real time.
    - 10.1.1. The OBP shall provide a status message to the UPC once per second while the system is driver is running.
      - 10.1.1.1. The status message shall be "off" if the system is running the driver but is not ready or running.
      - 10.1.1.2. The status message shall be "warming up" if the system has received the "ready" command but has not yet completed the ready sequence.
      - 10.1.1.3. The status message shall be "ready" if the system has completed the ready sequence after receiving the "ready" command.
      - 10.1.1.4. The status message shall be "running" if the system is executing the scan. This message shall be time-stamped relative to the receipt of the "start" command.
      - 10.1.1.5. The status message shall be "analyzing" if the system has completed the scan but not the analysis. These messages shall be time-stamped with relative to the receipt of the "start" command.
      - 10.1.1.6. The status message shall be "complete" if the system has completed the scan and analysis. This message shall be time-stamped relative to the receipt of the "start" command.
      - 10.1.1.7. After a "complete" message is displayed, the system status shall be reset to "off."
      - 10.1.1.8. The status message shall be "stopped" if the operation was terminated with the "stop" command. This status shall remain in effect until the driver is restarted.





- The OBP shall generate output readable by the UPC.
  - 10.1. The OBP shall generate health and status information readable by the PC in real time.
    - 10.1.2. The health and status information shall appear on the terminal of the UPC once per second.
  - The OBP shall save raw sensor outputs to memory.
    - The OBP shall save lidar sensor measurements to memory.
    - 10.2.2. The OBP shall save prism orientation measurements to memory.
  - The OBP shall save translated beam attitudes to memory.
  - 10.4. The OBP shall save x, y, and z coordinates for each point to memory.
  - The OBP shall save a SAFE/UNSAFE designation to memory for each point.
  - 10.6. The OBP shall save the coordinates of the selected landing site to memory.
  - 10.7. The saved data for each point shall be correlated.
  - 10.8. The saved data shall be readable on the UPC outside of the driver routine.



## Lidar Sensor Sampling



#### Pepperl+Fuchs VDM28:

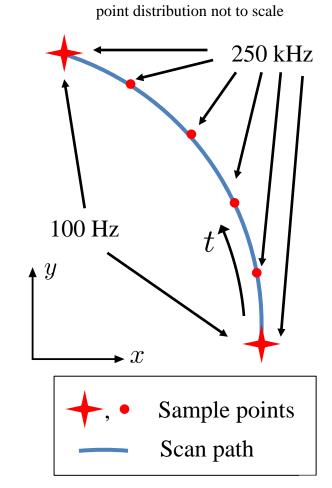
 COTS sensor that meets requirements and budget constraints

#### Sensor shortcomings:

- 100 Hz sampling frequency
- Time-averages over 10 ms (takes 2500 samples in that interval)

#### Possible solution:

- Custom-built sensor with higher sampling frequency and no timeaveraging
  - Cost estimate: ~\$10,000





## Lidar Sensor Sampling

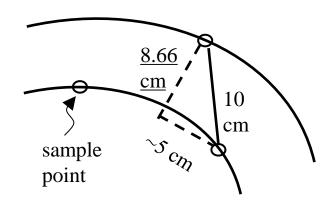


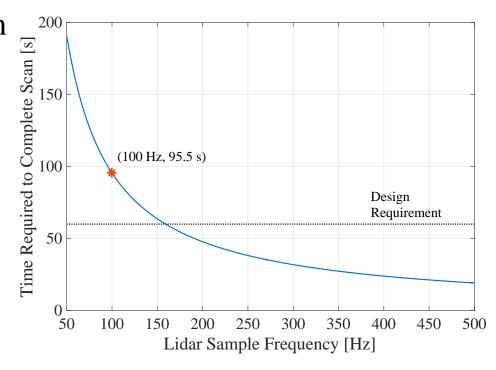
#### Resolution

- Spiral Spacing: 8.66 cm
- Arc-point Spacing: 10 cm

#### Minimum frequency

- Total points: 9,550
- $f_{min} = 159 \text{ Hz}$







### Scan Time vs. Scan Resolution



0.1 m spatial resolution at 14.1 m nadir range with 20° maximum scan angle



Scan process completed in less than 60 seconds

**Problem**: These two closely coupled objectives cannot be completed concurrently due to financial limitations on the lidar sensor



# Lidar Wavelength



#### Feasibility for MACULA

 Test surface can be constructed with white diffuse paint or white retroreflective tape

#### Why this sensor was selected

- Meets budget and accuracy constraints
- Test surface can be constructed to fit sensor

#### Benefits of Using 660 nm

• Visible spectrum (verification)



## Lidar Wavelength



- Per **FR1**, MACULA is proof-of-concept system for CubeSat lander
  - Wavelength can be selected for custom-built sensors
  - Implemented systems will choose wavelength based upon landing surface



http://pics-about-space.com/asteroid-surface?p=1



http://pics-about-space.com/planet-mars-surface?p=1



### Laser Wavelength vs. Cost



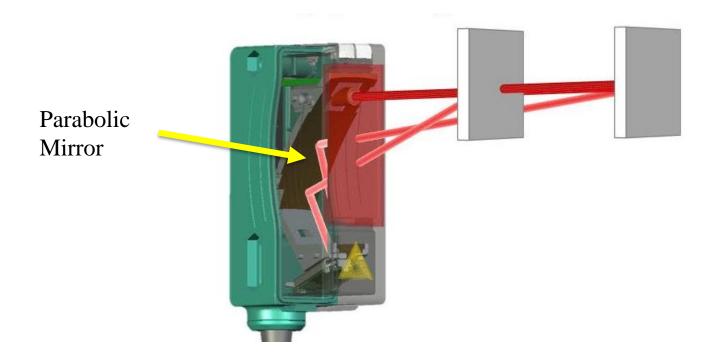
- Red lasers are the most common and cheapest to manufacture
- Laser colors other than red require specialized crystals with rare-earth elements such as Neodymium
  - These extra components can drive up the cost of other color lasers (yellow, blue, green) to dozens of times the cost of a red laser
- These colors can have better reflection on certain surfaces, but do not provide a general advantage over red lasers



## Detector Functionality



- Parabolic mirror to collect diffuse returns
- Specular returns do not disperse



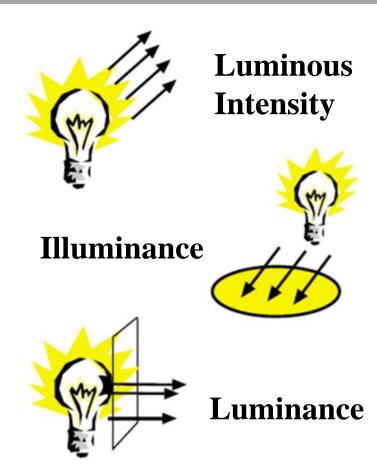


### Retroreflection



- Luminous Intensity
   [candela] Quantity of
   luminous flux in given
   direction
- Illuminance [lux] Measure of concentration of luminous flux falling on surface
- Luminance [candela/m²]

   Measure of flux emitted from or reflected by a uniform surface



http://www.konicaminolta.com/instruments/knowledge/light/concepts/04.html

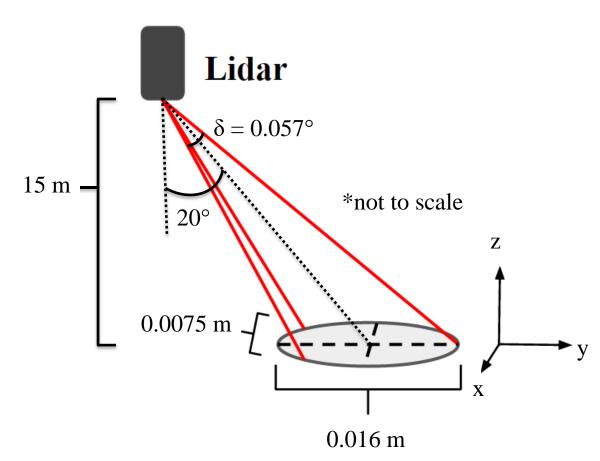


### Retroreflection



#### Laser Emitter

- Pulse: < 4 nJ
- Pulse length: 5 ns
- Beam divergence:
  - $-\delta = 0.057^{\circ}$
- Luminous Intensity:
  - 4.24e7 candela
- Illuminance on surface (15 m, 20° from nadir)
  - 1.89e5 lux





### Retroreflection

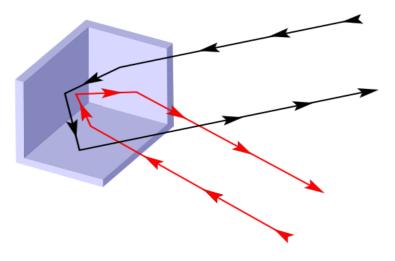


#### Reflexite Daybright V92

Observation	Entrance	
Angles	Angles	White
0.2 °	-4 °	460
	30°	250
0.5 °	-4 °	100
	30 °	65

- Luminance of return:
  - 1.23e7 candela/m<sup>2</sup>
- Luminance of Pepperl+Fuchs datasheet tests (90% Kodak White):
  - 1.70e5 candela/m<sup>2</sup>

#### **Prismatic Retroreflector**



https://en.wikipedia.org/wiki/Retroreflector



## Risley Prism Specs



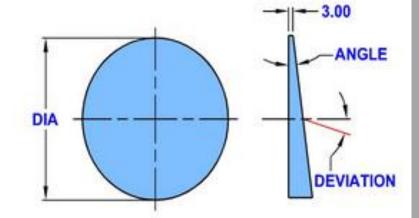
- Suitable for wide range of wavelengths
  - 450 nm 2000 nm
- Coatings available for 660 nm
  - Reflectance of about 1%, resulting in above 90 % transmissivity
- Cost:
  - \$100 each uncoated
  - Additional \$5 for coated



# Prism Specifications



- Ross Optical P-WRC059
  - Diameter: 5.08 cm
  - 10° Maximum Beam Deviation (per prism)
  - Wedge Angle: 18° 8′
  - Angle Error:  $\pm$  30 arc seconds
  - Material: N-BK7 Grade A fine annealed
    - Transmission: 91% at 660 nm
    - Density: 2.51 g/cm<sup>3</sup>
    - Thermal Expansion:  $7.1 \times 10^{-6} \text{ K}^{-1}$
  - Thi.. LJge Thickness: 3mm
  - Dimensional Tolerance  $\pm 0.1$  mm

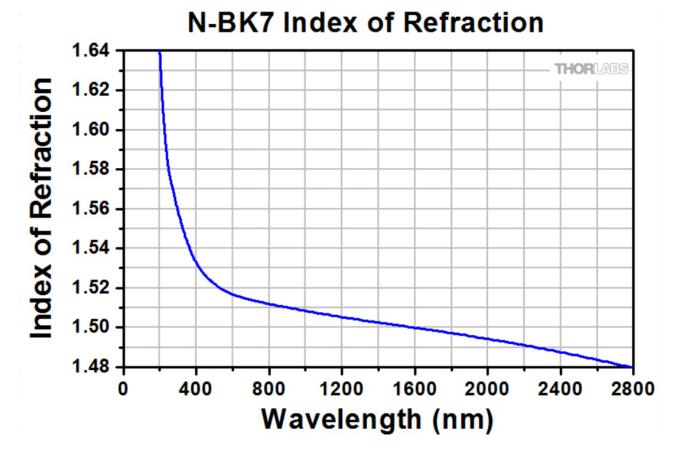




### Index of Refraction



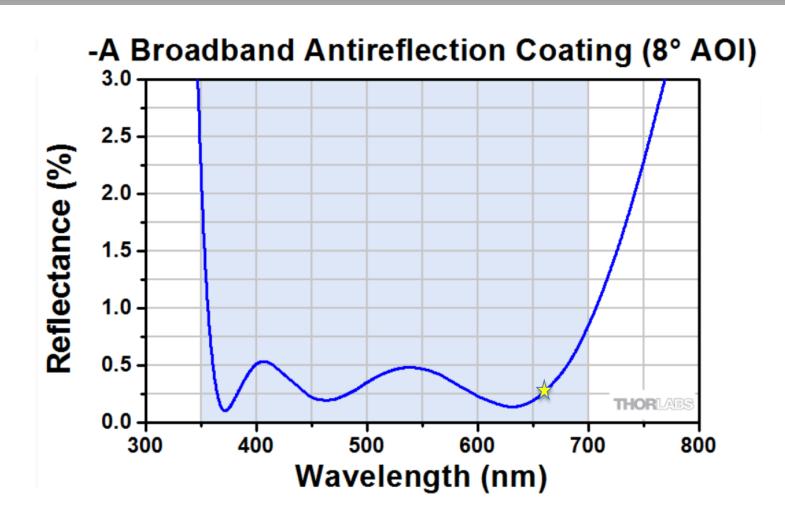
N-BK7 has variable index of refraction





## Coating Reflectance



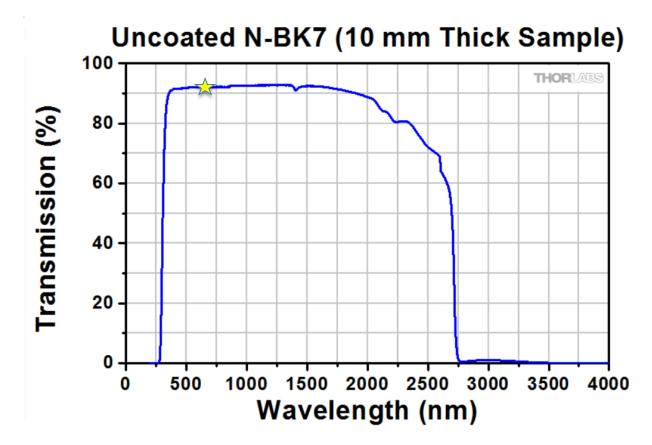




### Prism Attenuation



• Material: N-BK7 Grade A fine annealed

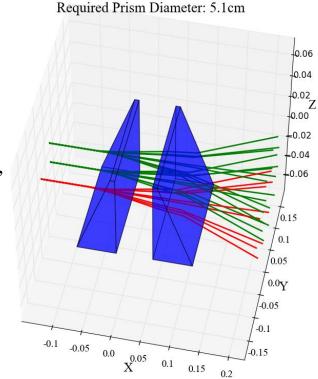




### Prism Diameter



- Beam lines calculated for eight rotations of the prisms (rotated together to produce maximum deflection angle)
- Transmitter and two points on the edge of the receiver are projected straight and their refractions are calculated for each of the prism rotations
  - This is only part of the receiver field of view. The lidar is placed to maximize what the receiver can see, without clipping the transmitter
- Prism diameter based on the farthest point from the center axis for any beam on any prism face
- Resulting distance is divided by 0.9 to produce the prism diameter (for best refraction results from the prism)
- Modeled as blocks for ease of plotting only.
   Reported size is the diameter

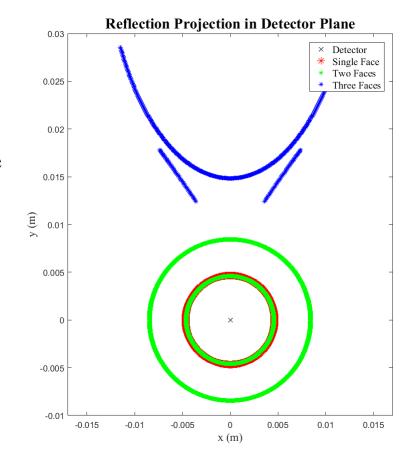




# Reflection Analysis



- Possible Risk: Reflections off prisms trigger lidar false returns.
- Reflections were analyzed to determine if they would hit detector.
- This risk can be mitigated by moving lidar away from prisms.

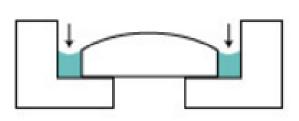




### UV Cured Epoxy



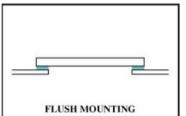
- Upon curing, surface of epoxy exposed to UV light shrinks forming a meniscus
- Various techniques can be utilized to minimize stress upon shrinkage.
- Internal Barrel Button Bond method selected. Allows for slip fit and minimal shrinkage stress.

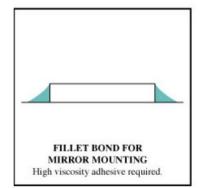


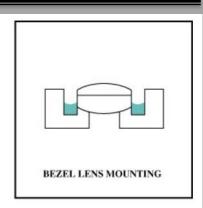
Epoxy Shrinkage shown on the right. Epoxy mounting techniques shown on the left.

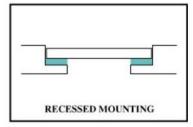
https://www.norlandprod.com/techrpt s/techniques.html















## **Epoxy Selection**



#### Selected UV Cured Epoxy: Thorlabs NOA81

- Shrinkage: 1.5 %
- Tensile Strength: 4000 psi
- Glass to Metal bond strength: Excellent
- Cost: \$33.5
- Amount per bottle: 1 oz
- Recommended Curing Intensity:

 $>2 \text{ mW/cm}^2$  @ 365 nm





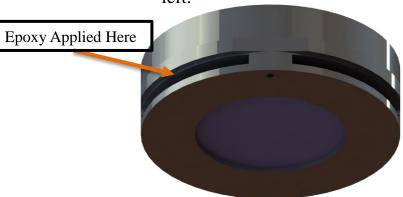
# Mounting Stress Analysis



- Allowable Shear Stress: 1000 psi
- Estimated Area Exposed to Epoxy: 0.47 in<sup>2</sup>
- Allowable Torque: 470 lb-in or 53.10 Nm
- Maximum Torque Supplied by Motors: 1 Nm
- Very low chance of failure

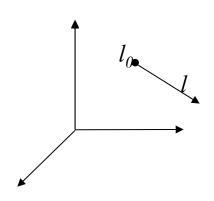


Epoxy Shrinkage shown on the right. Epoxy mounting techniques shown on the left.

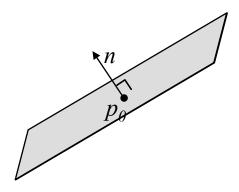








Point  $l_0$  is associated with direction l



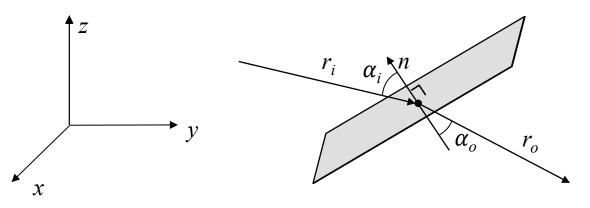
Point  $p_0$  and normal n define a plane

- Line l intersects the plane p at  $l_0 + \frac{n \cdot (p_0 l_0)}{l \cdot p} * l$
- The distance travelled between  $l_0$  and  $p_0$  is  $\frac{n \cdot (p_0 l_0)}{l \cdot p} * ||l||$





#### Snell's Law in 3 dimensions



$$T = (\hat{n} \quad v \quad n \times v), \qquad v = -(r_i - r_i \cdot (-n))$$

$$R = \begin{pmatrix} \cos(\Delta\alpha) & \sin(\Delta\alpha) & 0 \\ -\sin(\Delta\alpha) & \cos(\Delta\alpha) & 0 \\ 0 & 0 & 1 \end{pmatrix},$$

coordinates such that  
our new 
$$x$$
 and  $y$  lie in  
the plane formed by  $r_i$   
and  $n$ . The equations  
are shown below

we transform

This reduces to Snell's

Law in 2 dimensions if

$$R = \begin{pmatrix} \cos(\Delta\alpha) & \sin(\Delta\alpha) & 0 \\ -\sin(\Delta\alpha) & \cos(\Delta\alpha) & 0 \\ 0 & 0 & 1 \end{pmatrix}, \qquad \Delta\alpha = \alpha_o - \alpha_i = \sin^{-1}\left(\frac{n_i}{n_o}\sin(\alpha_i)\right) - \alpha_i$$
$$\alpha_i = \cos^{-1}\left(\frac{r_i \cdot n}{||r_i|| \cdot ||n||}\right)$$

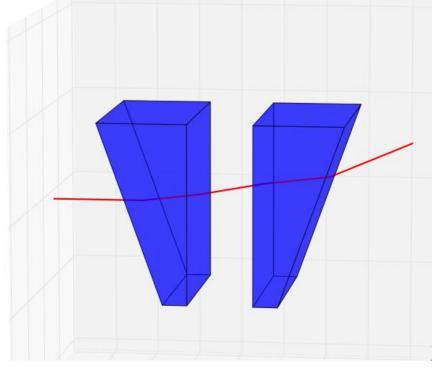
$$r_o = T \cdot R \cdot T^{-1} r_i$$





By propagating in between prism faces and through material interfaces we can trace the ray

path







#### Topographic Map

$$R = r - e - n(d'_1 + d'_2) - \frac{d_p}{\cos \beta_{1_o}}$$

$$P_{xyz} = R \cdot l_{2_o} + p_{2_o}$$

$$P_n = \begin{pmatrix} 0 & 1 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & -1 \end{pmatrix} P_{xyz}$$

$$P = \text{DCM}_{\text{IMU}} \cdot P_n + \begin{pmatrix} 0 \\ 0 \\ 15\cos(20^\circ) \end{pmatrix}$$

- *r* Lidar return
- e Distance between lidar and first prism
- $d_p$  Prism separation
- n Prism refractive index
- d' Distance travelled within prism
- l Direction vector
- p Position vector
- $\beta$  Angle between beam and optical axis

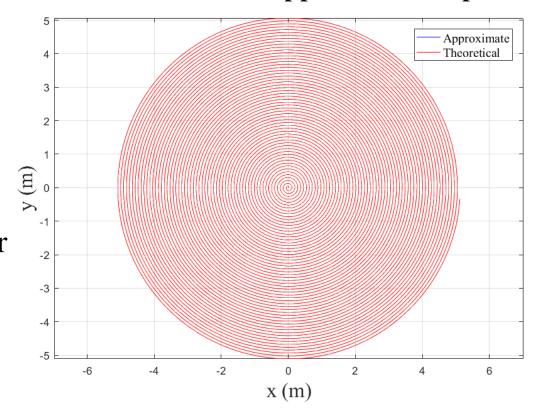


## Scan Pattern to Prism Angles



- Radial component of points determined by prism offset angle
- Once offset angle is determined, theta component is found by rotating prisms together
- Prism angles can be found numerically for all points in scan

### Theoretical vs. Approximated Spiral



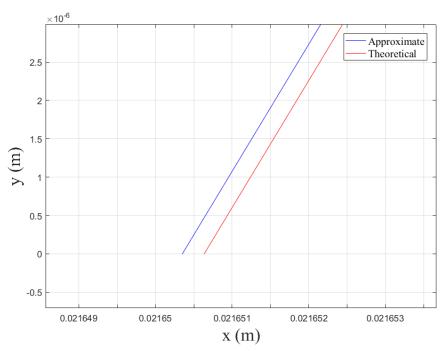


## Scan Pattern Approximation

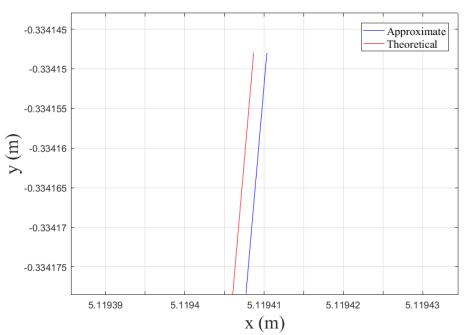


#### Time Scan:

#### Scan Center:



### Scan Edge:



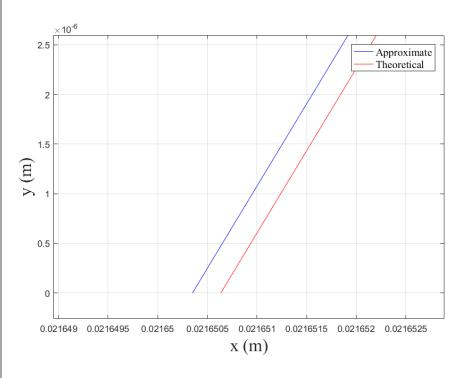


## Scan Pattern Approximation

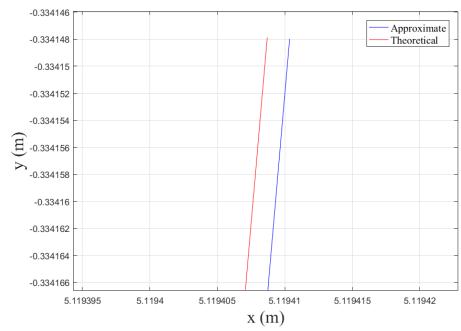


#### **Resolution Scan:**

#### Scan Center:



#### Scan Edge:

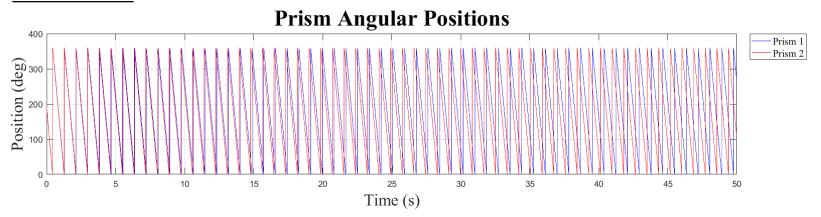




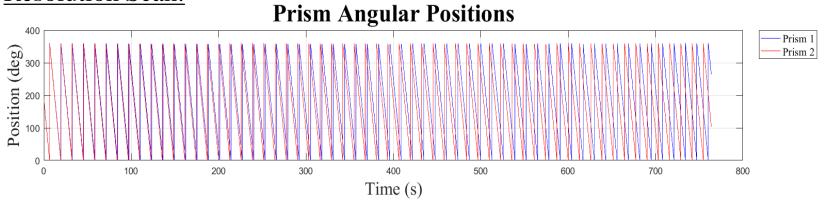
## **Prism Positions**



## Time Scan:



#### **Resolution Scan:**





## **Motor Rates**



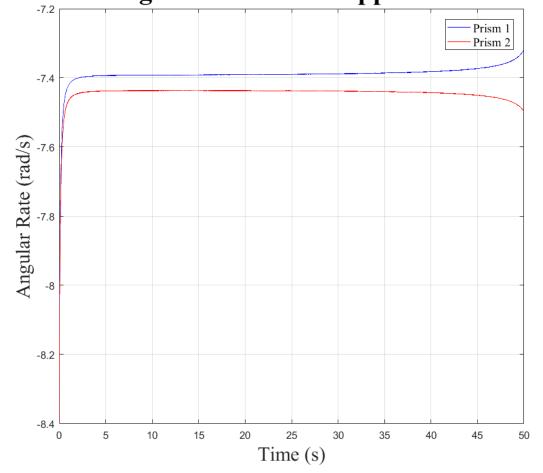
#### Time Scan:

• Min Rate: 7.34 rad/s

• Max Rate: 8.46 rad/s

• Max Accel: 14.36 rad/s<sup>2</sup>

## **Prism Angular Rates over Approximate Scan**





## Motor Rates



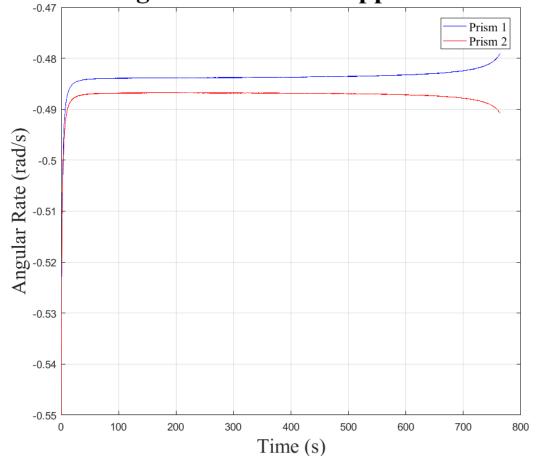
#### Resolution Scan:

• Min Rate: 0.48 rad/s

Max Rate: 0.55 rad/s

• Max Accel: 0.047 rad/s<sup>2</sup>

Prism Angular Rates over Approximate Scan





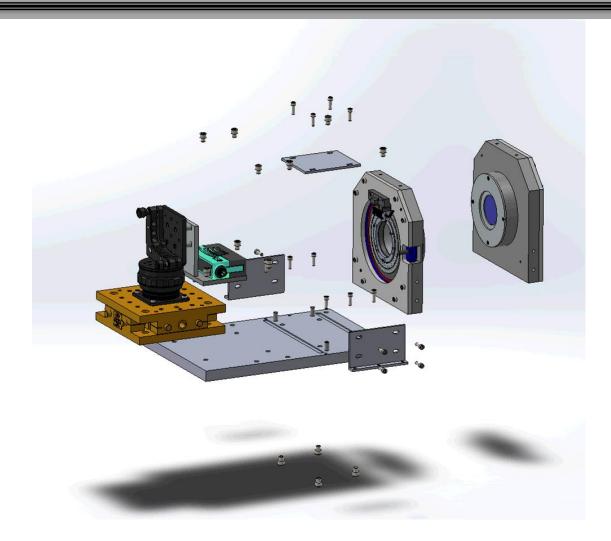


# BACKUP: SYSTEM ASSEMBLY



# Movie

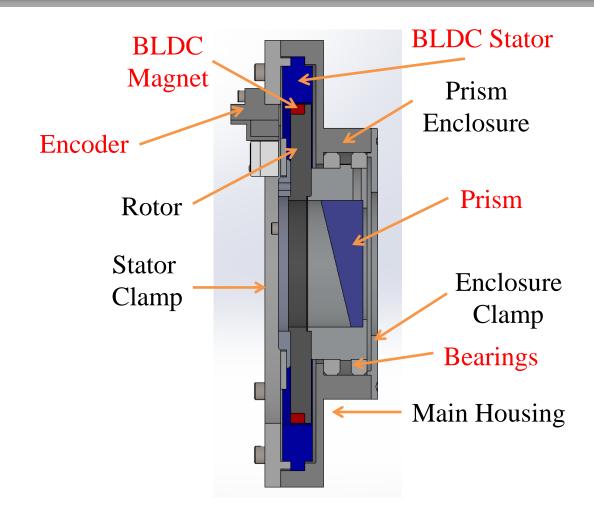


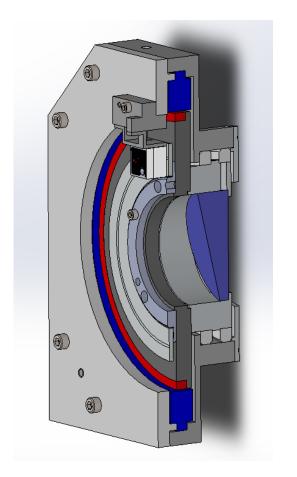


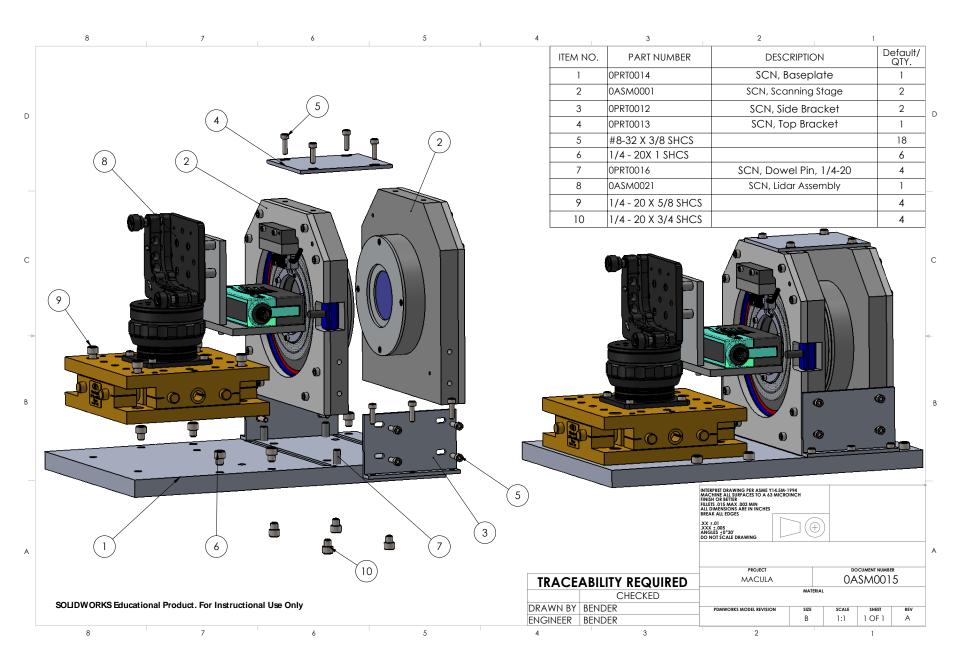


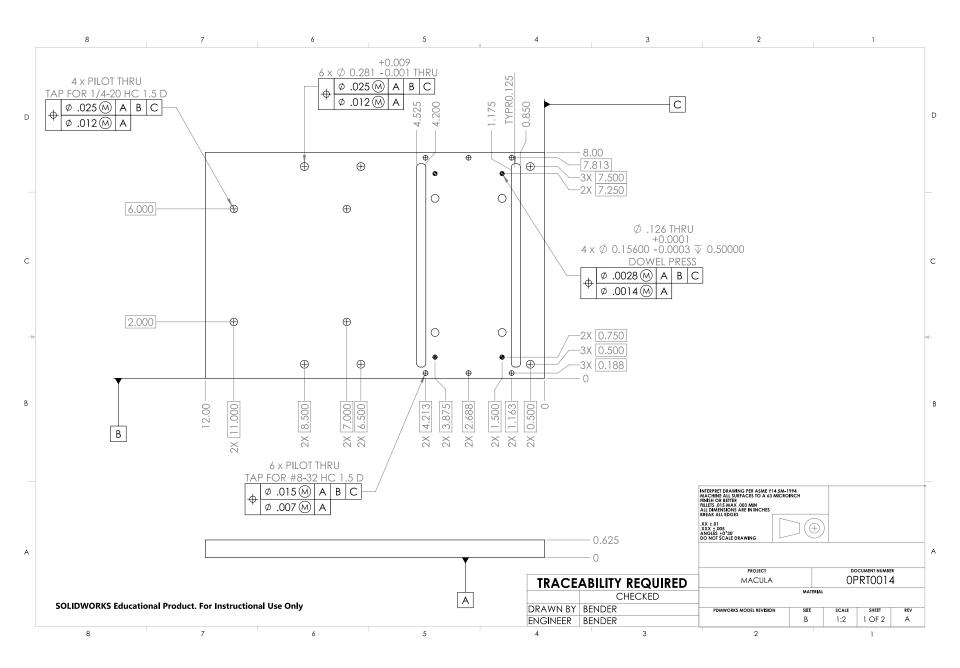
# Scanning Stage

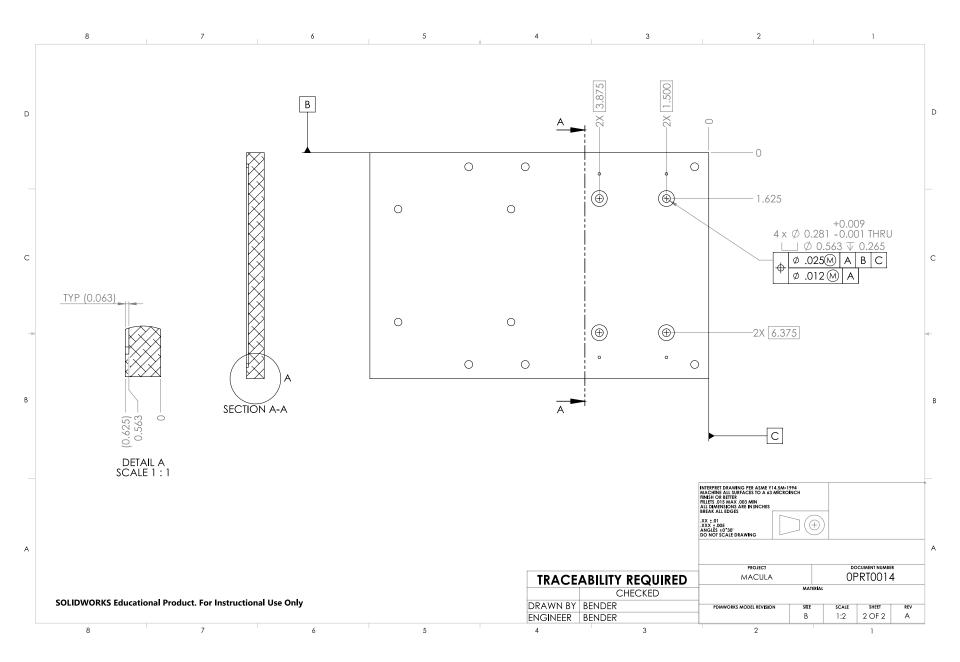


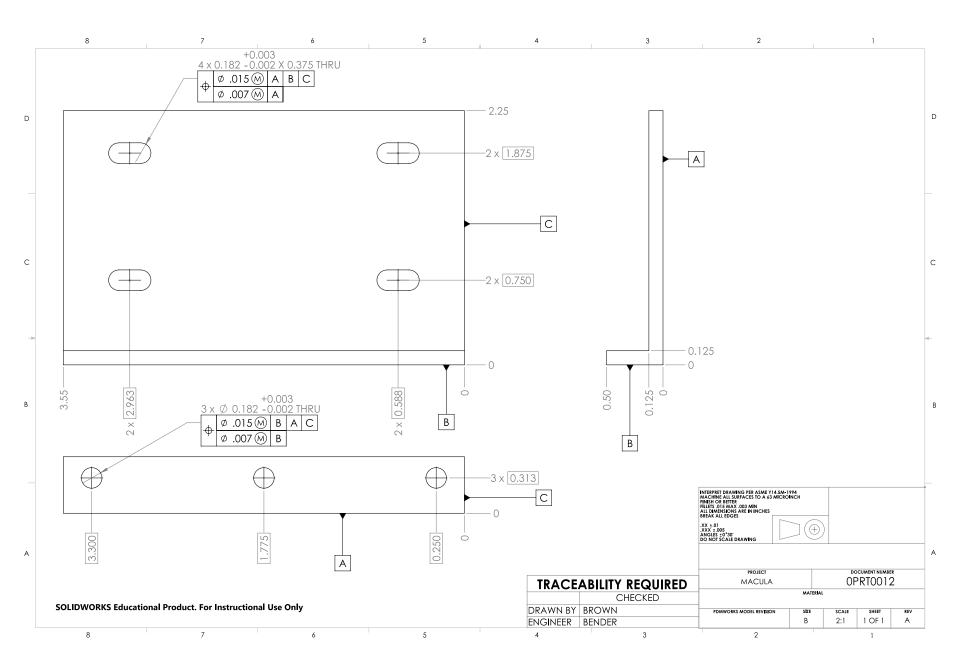


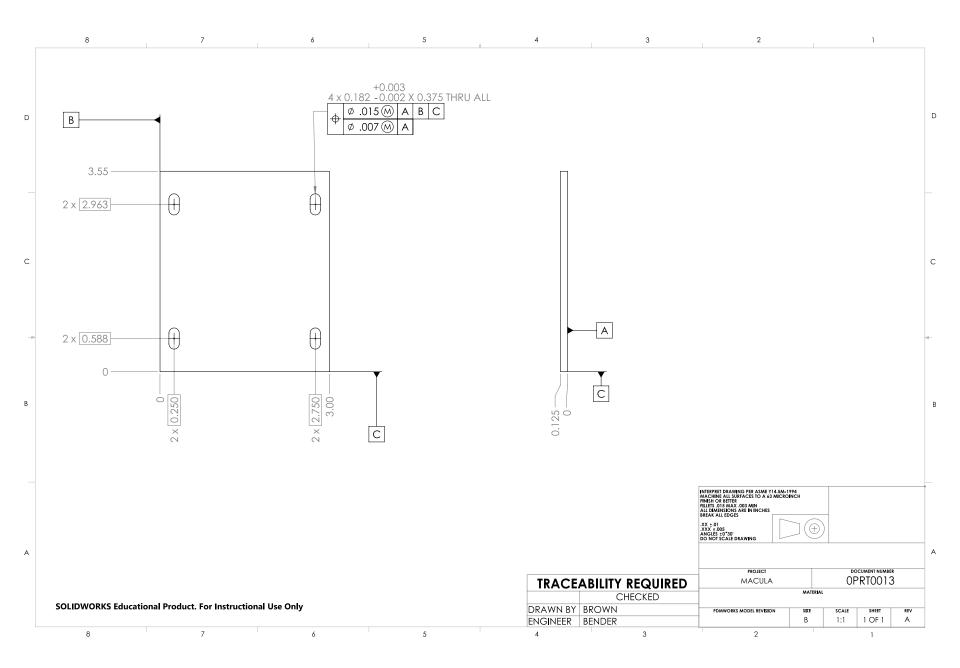


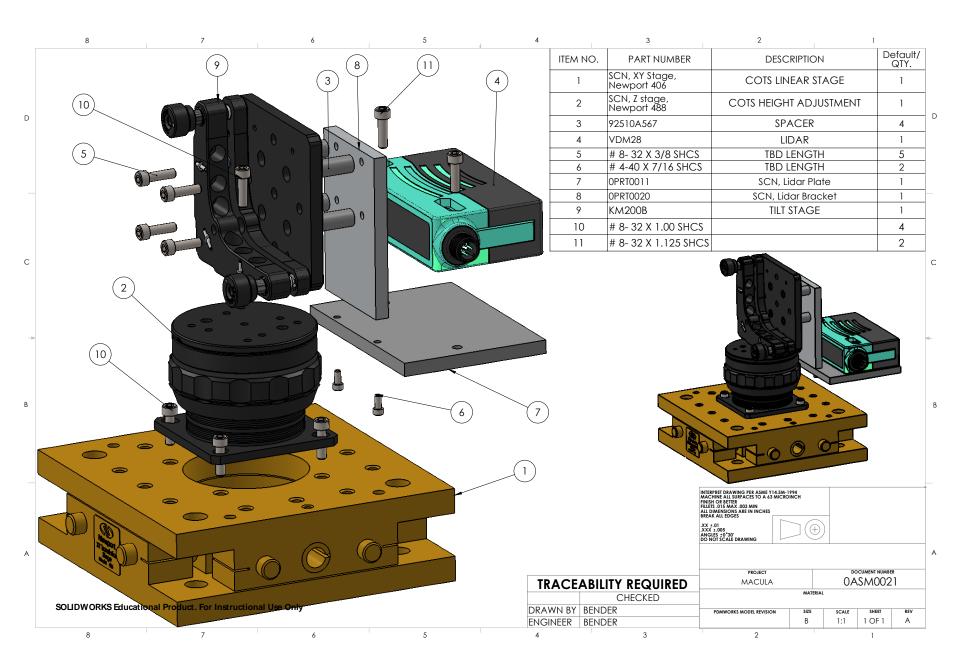


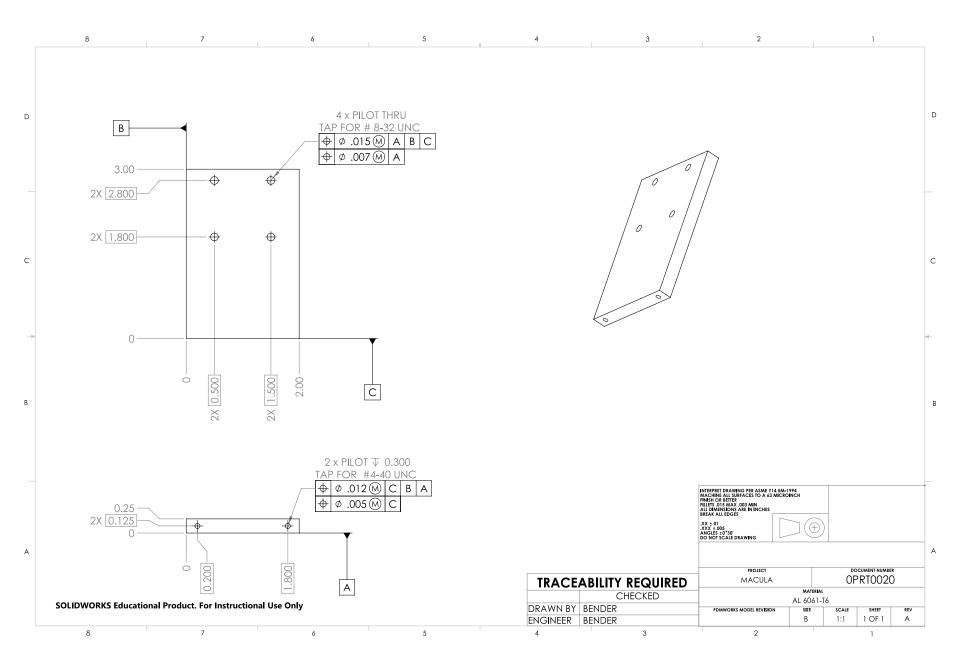


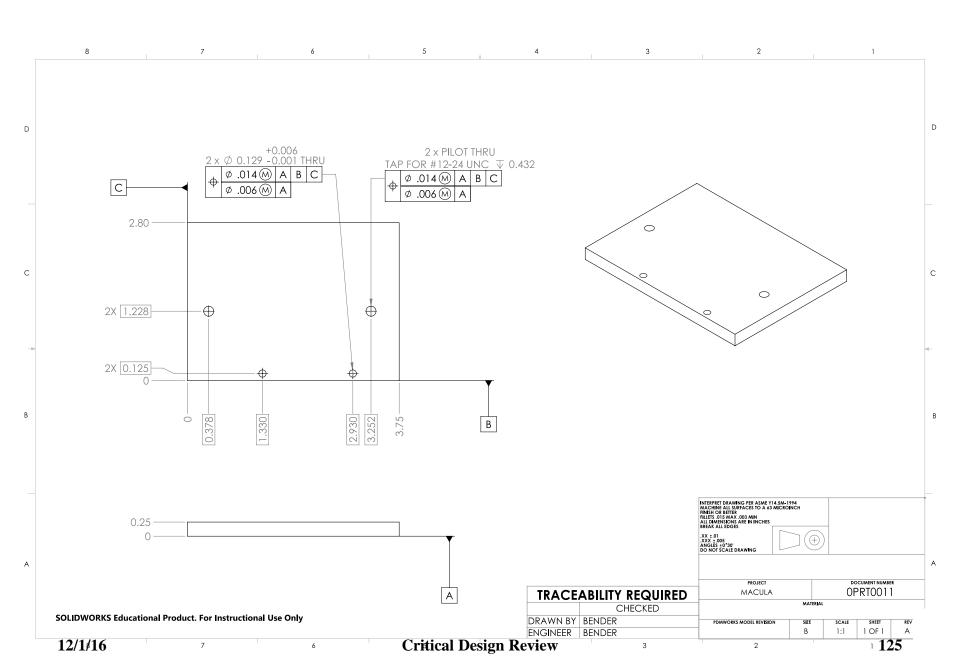


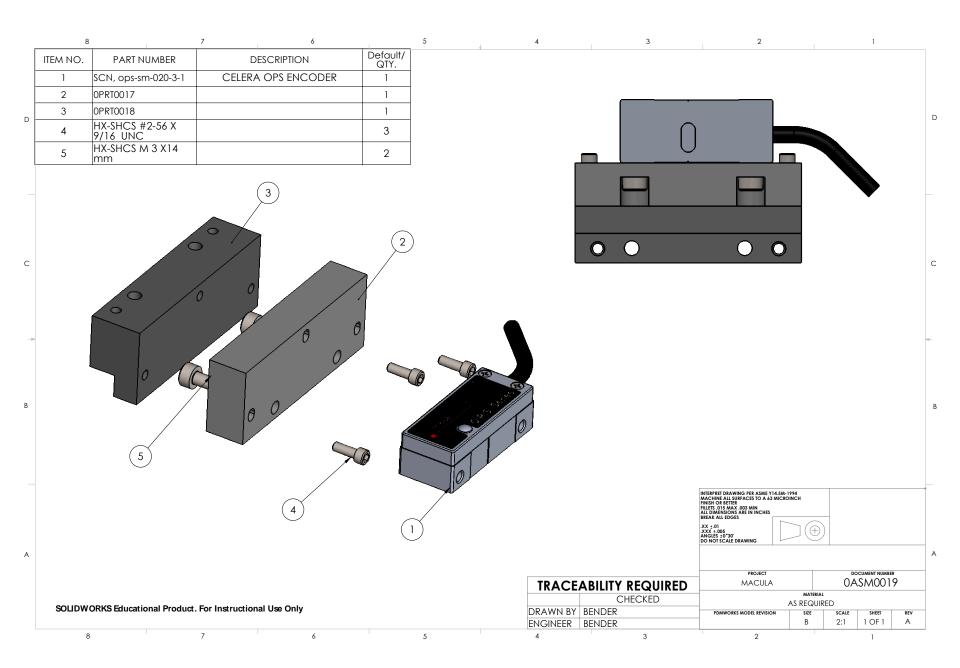


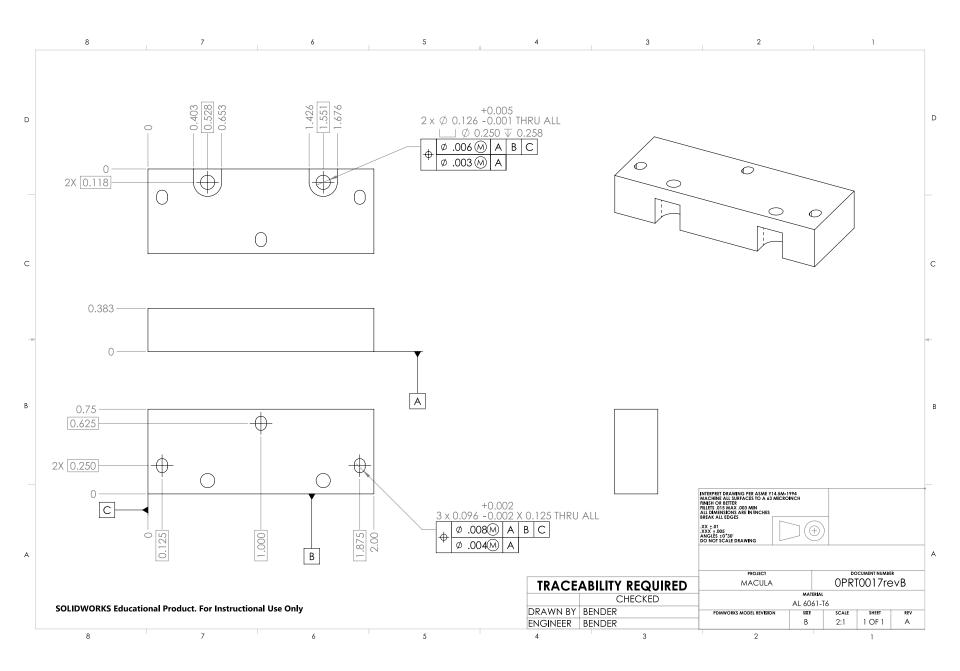


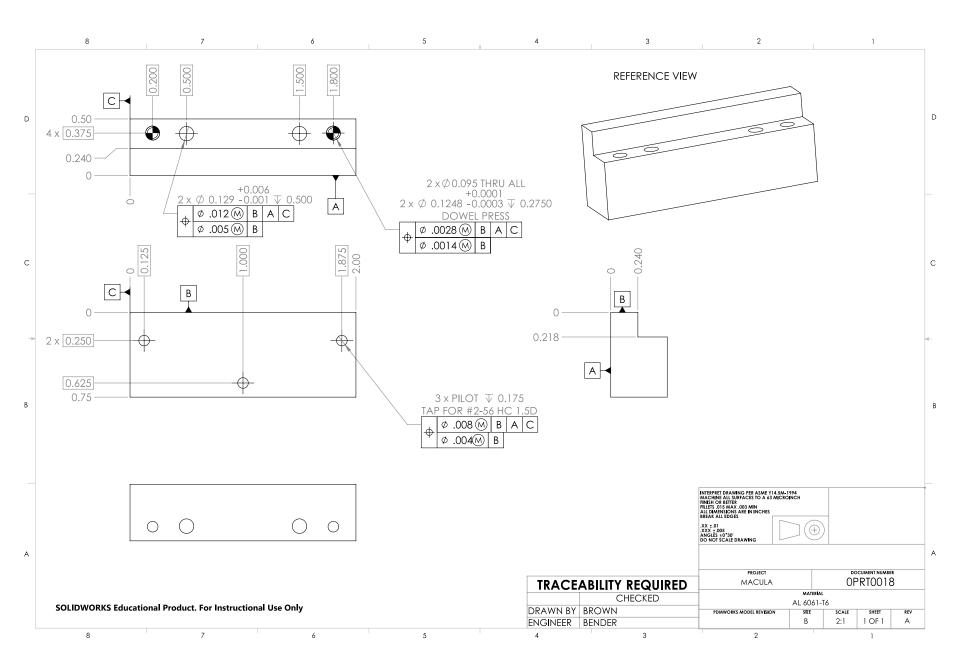


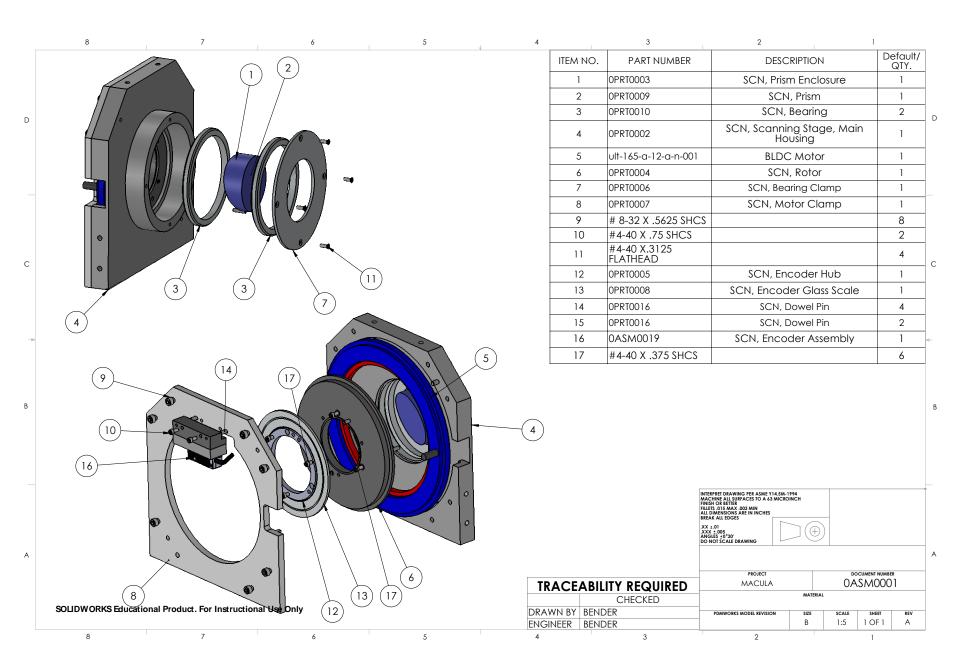


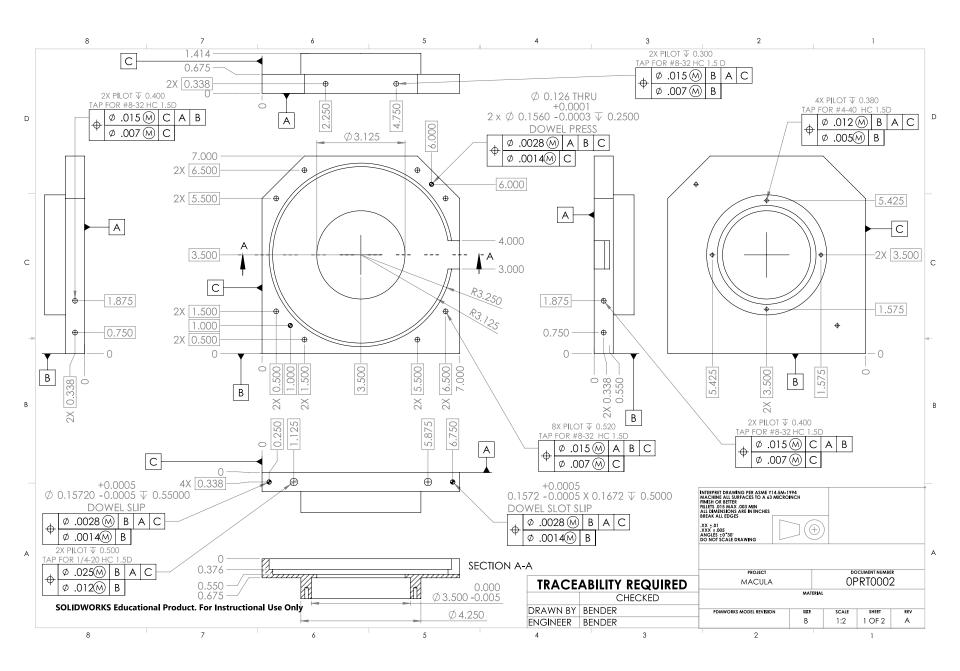


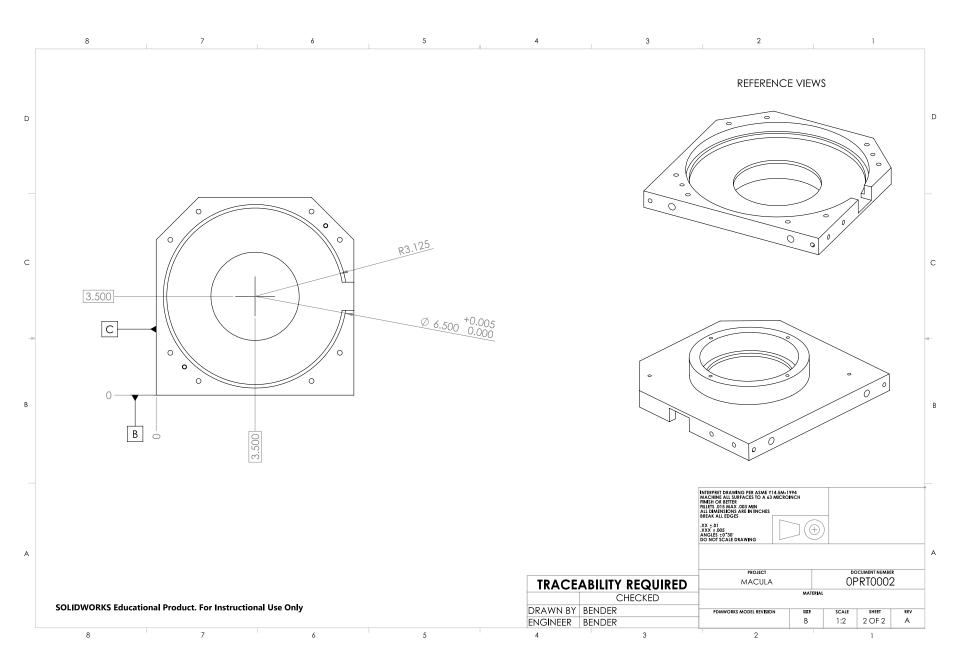


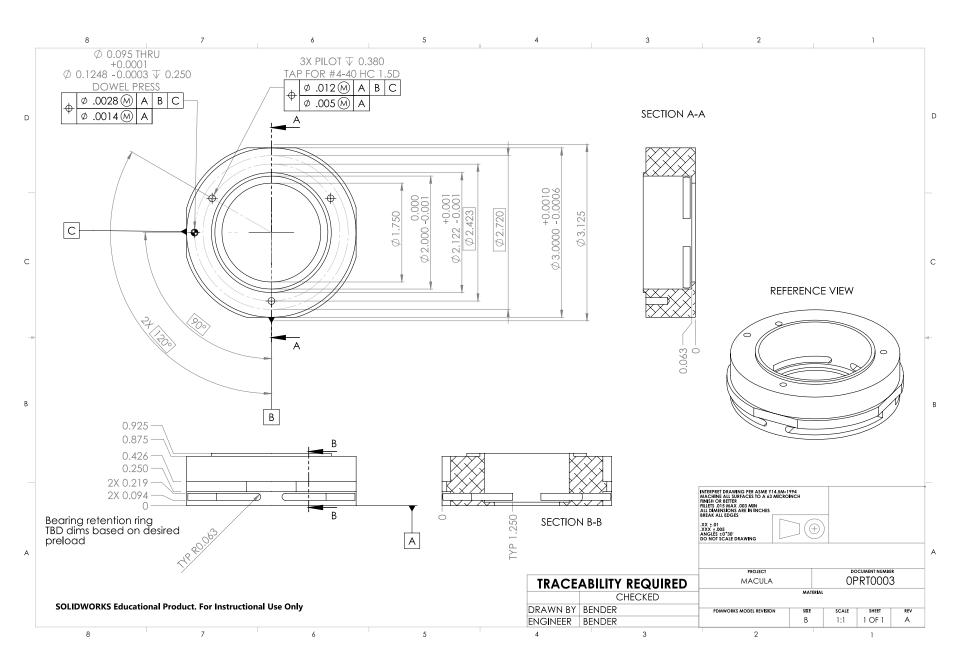


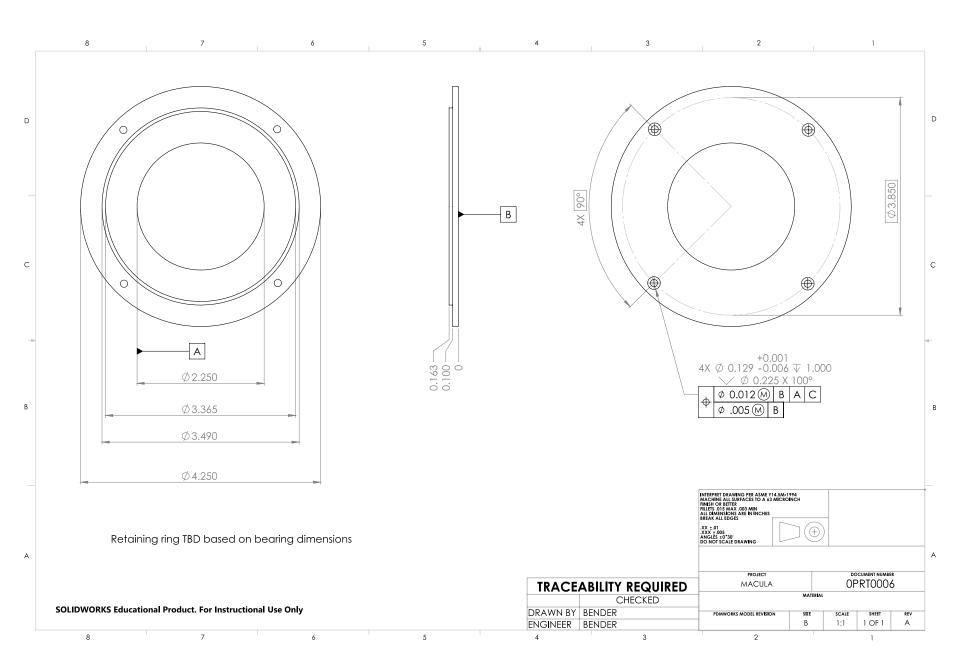


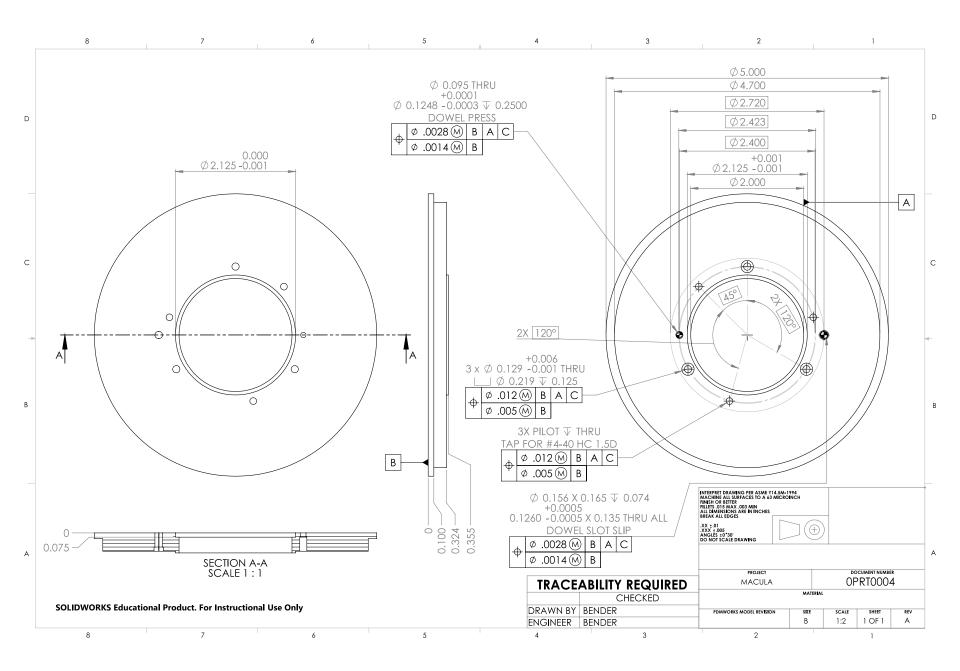


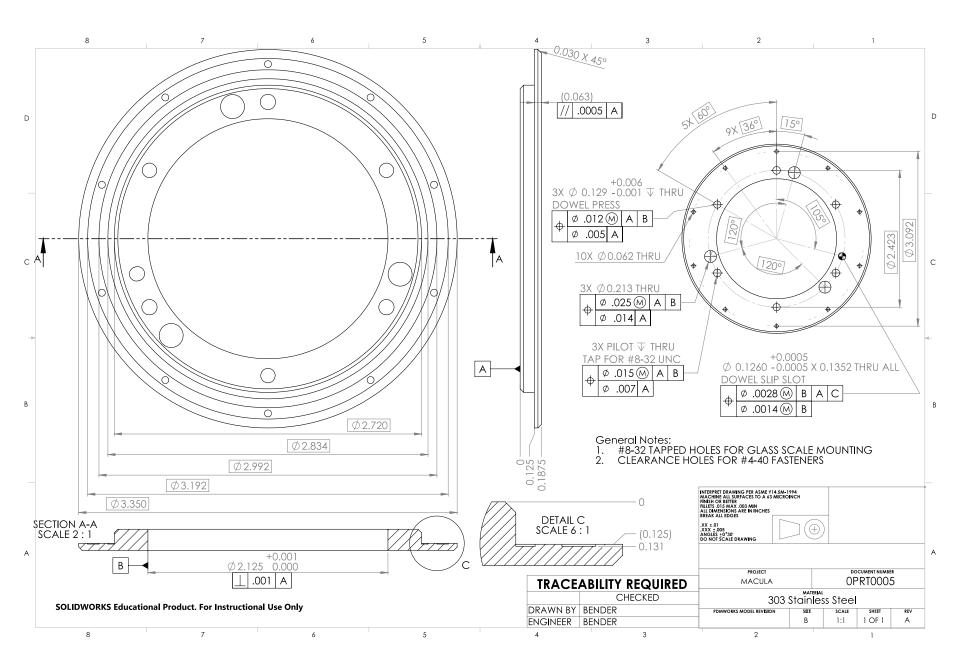


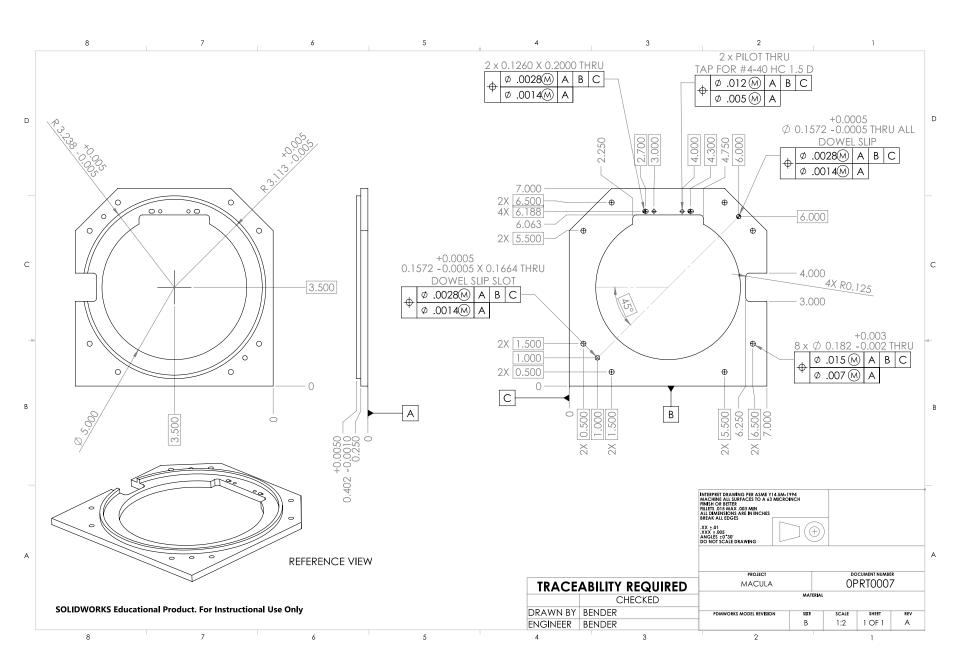
















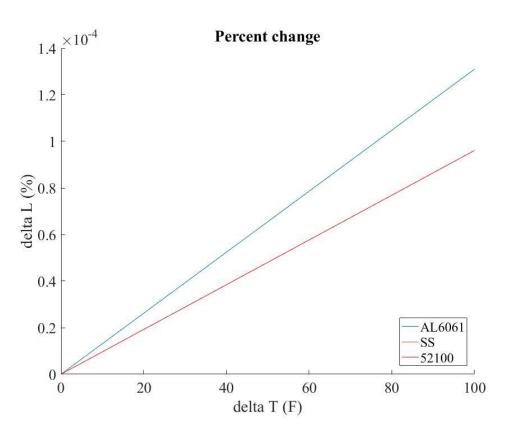
# BACKUP: MATERIAL ANALYSIS



## **Thermal**



## • Linear Thermal Expansion: $\Delta l = l_0 \alpha (T_f - T_o)$



## **Thermal Expansion Coefficients**

• 
$$\alpha_{al}=$$
 13.1  $\frac{\mu in}{inF}$ 

• 
$$\alpha_{ss}=$$
 6.61  $\frac{\mu in}{inF}$ 

• 
$$\alpha_{52100} = 9.61 \frac{\mu in}{inF}$$

#### Max error (5°F)

- 0.003806°
- 2.5% of error budget



## Mechanical



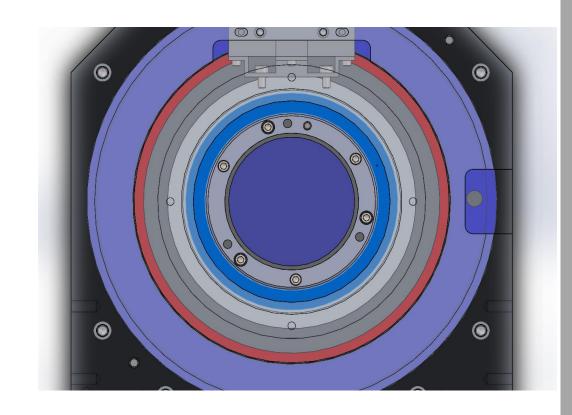
## • Steel dowel pin

$$-V_{max} = 7117 \text{ N}$$

$$-\tau_{allow}=246~\mathrm{Nm}$$

$$-\tau_{max} = 1.26 \text{ Nm}$$

$$-FOS = 202$$







## BACKUP: RISKS

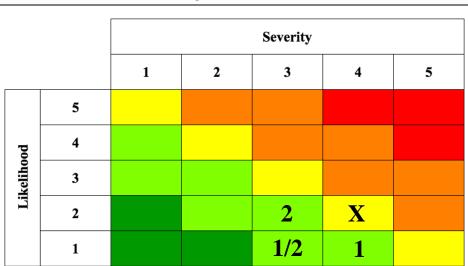




R2	Encoder mount damage occurs during shipping, glass scale installation, or storage					
	<ol> <li>Ensure safe packing of encoder hubs when shipped out to company</li> </ol>					
		Reduces likelihood of risk				
	2. Manufacture additional encoder hub and buy additional glass scale					
		Stock materials can be bought with 20% budget margin				
		Reduces severity of risk				

## Type of Risk

- □ Budget
- ☑ Technical
- □ Safety
- ☑ Schedule







R3	Limited machine sh	nop availabili	ty d	ue to s	chedule	es and 1	use by (	other s	tudents
	Go over manufacturing plans ahead of time to ensure efficient use of time in shop								
	Reduces likelihood of risk								
	2. Utilize full extent of machining resources available, included multiple shops								
	(Aero, ITLL) and full time employees who will machine parts								
	Reduces severity of risk								
	Severity Severity								
	Type of Risk				1	2	3	4	5
	□ Budget			5					
	□ Technical		poo	4	2	X			
	□ Safety		Likelihood	3	1/2	1			
	•		Γ	2					
	☑ Schedule			1					

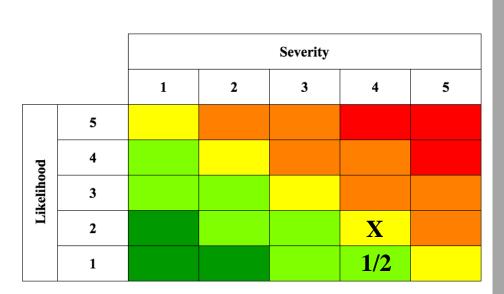




R4	Risley prism damage occurs during storage or handling				
	1. Ensure proper pa	dding / protection is in place when storing and moving prisms			
		Reduces likelihood of risk			
	2. Limit contact with prisms before and during mounting in system				
		Reduces likelihood of risk			

## Type of Risk

- **☑** Budget
- ☑ Technical
- □ Safety
- □ Schedule



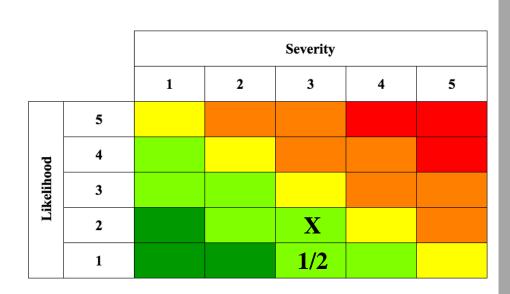




R5	RECUV test facility unavailable at necessary times and/or necessary durations				
	1. Coordinate testing times far in advance to ensure availability				
	Reduces likelihood of risk				
	2. Run through test procedures before testing on location for efficient use of time				
	Reduces likelihood of risk				

## Type of Risk

- □ Budget
- □ Technical
- □ Safety
- ☑ Schedule







# BACKUP: HARDWARE TRADE STUDIES



### Position Sensor Trade Study



Metric	1	2	3	4	5
Resolution	≥ 0.15°	≥ 0.075°	≥ 0.0375°	≥ 0.0187°	≥ 0.0094°
Interface/Decoding	Not a common on chip peripheral	-	-	-	Common on chip peripheral
Cost per	≥ \$600	≥ \$500	≥ \$400	≥ \$300	≥ \$200

	Weight	Absolute	Incremental	Sin Cos In- cremental	Rotary Poten- tiometer	Resolver
Resolution	50%	3	4	5	1	3
Interface	40%	1	5	1	5	1
Cost	10%	2	2	2	5	1
Sum	100%	2.1	4.2	3.1	3.0	2.0



### Encoder Trade Study



Metric	e 1		3	4	5
Cost	≥ \$1400	≥ \$1200	≥ \$1000	≥ \$800	≥ \$600
Immunity to Environment	Knocked out by a slight breeze	Constant Interference	Innately exposed scanning head	Innately shrouded scanning head	Impervious
Design Flexibility	Any change requires replacing both scanning head and scale	Diameter change requires replacing both scanning head and scale	Scale change requires replacing both scanning head and scale	-	Scanning head and scale may be changed in- dependently of each other, diameter only affects scale

	Weight	OPS Series	IncOder Series	Lika SMR Series	Heidenhain
Cost	40%	5	3	5	1
Immunity to Environment	30%	3	5	2	4
Design Flexibility	30%	5	2	5	5
Sum	100%	4.4	3.3	4.1	3.1



## Motor Driver Trade Study



Metric	1	2	3	4	5
Cost	≥ \$500	≥ \$400	≥ \$300	≥ \$200	≥ \$100
Development Time	Build board, create control law, tuning	-	Create control law, then tune	-	Just tuning
Modes of control	Torque / Current		Velocity, Torque / Current	-	Position, Velocity, Torque / Current
Commutation / Feedback	Hall effect	-	Sensor-less	-	Encoder Feedback

	Weight	Custom	AMC Servo Drivers	TI BLDC Motor Controllers	ST Eval Boards
Cost	10%	5	2	5	5
<b>Development Time</b>	40%	1	5	3	3
Modes of control	20%	5	5	3	5
Commutation / Feedback	30%	5	5	3	5
Sum	100%	3.4	4.7	3.2	4.2



#### Microcontroller Trade Study



Metric	Metric 1 2 3		3	4	5
Cost	≥ \$500	≥ \$400	≥ \$300	≥ \$200	≥ \$100
Development Time	Build board from scratch	-	Bare Metal C	Has an OS or a third party application to ease development	OS, can use previously written Python scripts
Peripherals	Does not have any needed	Has only ADC	Has ADC, and UART	Has ADC, UART, USB/Ethernet	Has ADC, UART, USB/Ethernet, quadrature decoders
Design Flexibility	Component changes requires a different mi- crocontroller	-	Component changes requires additional chips to handle them, but can interface with microcon- troller	Program FPGA to adapt	Easy to adjust to component changes

	Weight	Custom	BeagleBone Black	Pi Series	MyRio	ZedBoard
Cost	15%	4	5	5	1	3
<b>Development Time</b>	30%	1	5	5	4	5
Peripherals	40%	5	5	3	5	5
Design Flexibility	15%	5	3	3	4	3
Sum	100%	3.65	4.7	3.9	3.95	4.4





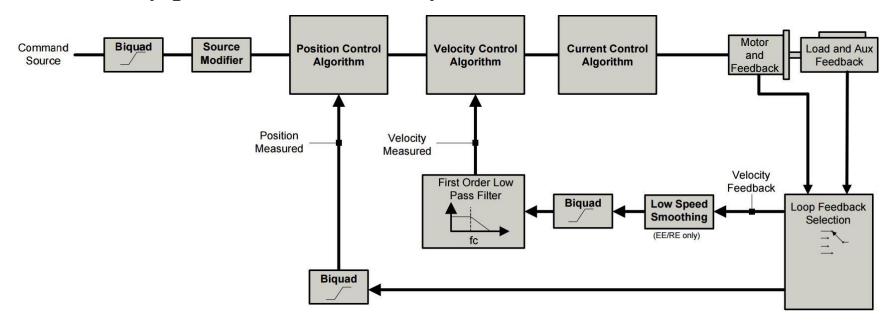
# BACKUP: MOTOR/PRISM CONTROL



#### Control Loop



- Built in feedback control
- Primary positional or velocity control



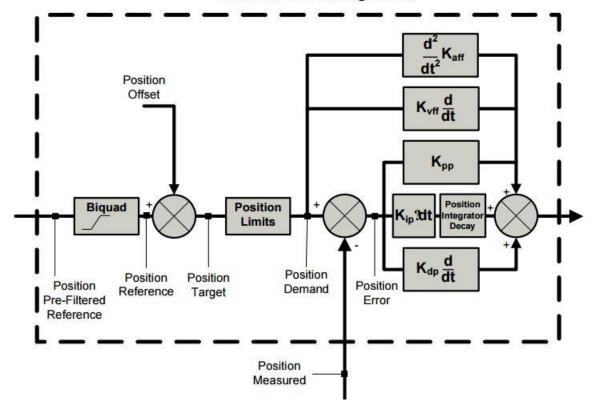


#### **Position Control**



- P.I.D control
- Feedforward acceleration and velocity

#### **Position Control Algorithm**

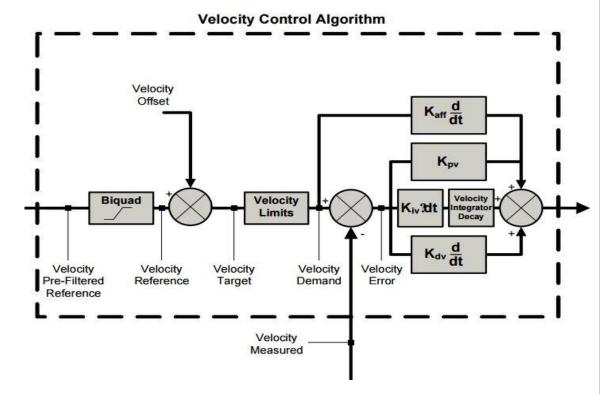




### Position around Velocity



- Used when position trajectory must be tracked closely
- Feed forward gains added for better tracking

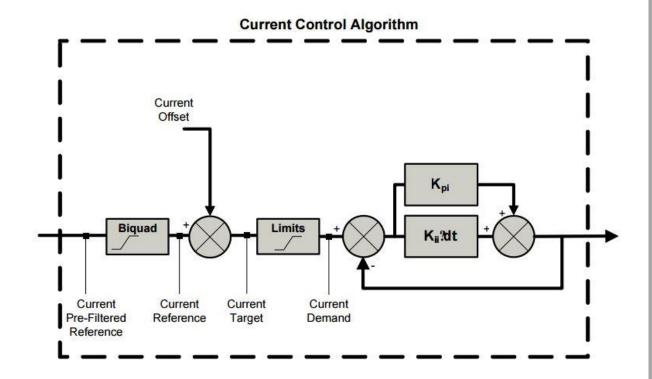




### Position around Torque



Point-to-point applications





## Tuning



- Performed with a motor installed into system
- Oscilloscope
  - 1-3Hz square wave
  - Channel one: Position Target
  - Channel two: Position Measured
- Gains initialized to zero

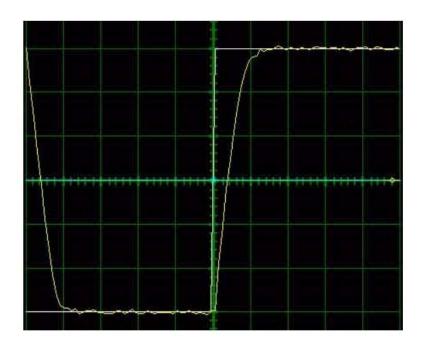
$$-0 \le Kp \le 0.5$$

$$-0 \le Ki \le 9.766$$

$$-0 \le Kd \le 0.0008$$

$$-0 \le Kv \le 0.0008$$

$$-0 \le Ka \le 8x$$





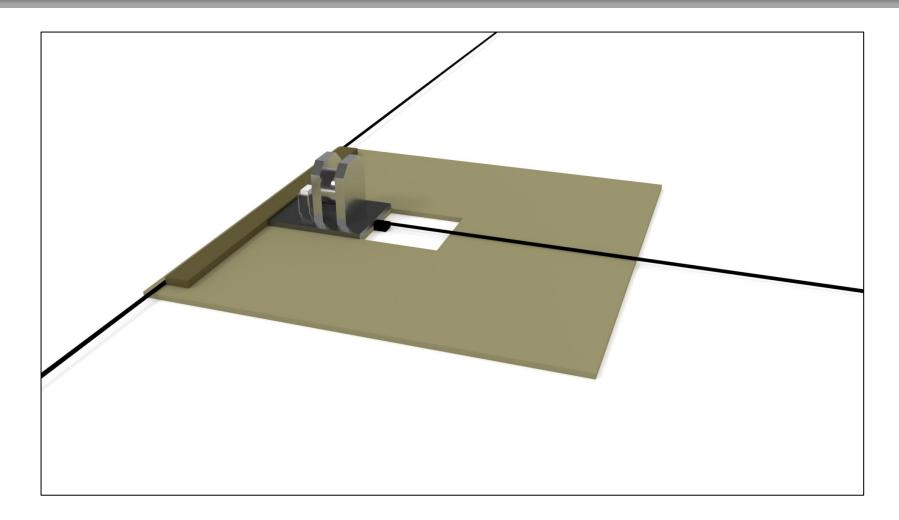


#### BACKUP: CALIBRATION



#### Test Plan: Calibration (Close-up)







#### Theodolite



#### What does it do?

 Determines vertical and horizontal angles of surveyed

#### How does it work?

- Plumb bobs ensures that it is vertical relative to the surveying point
- Internal bubble level ensures that it is level relative to the horizon
- Graduated circles allow for horizontal and vertical angles of surveyed object to be measured





#### Sources of Error

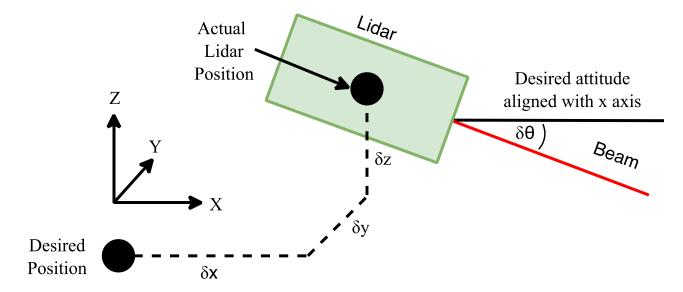


#### Sources of Error

Must be Calibrated System Inherent

#### Lidar:

- Translational deviations
- Rotational deviations
- Beam divergence





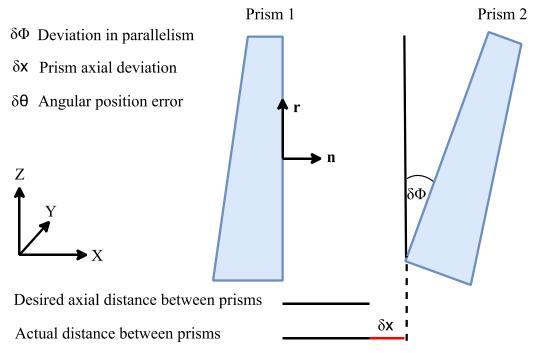
#### Sources of Error

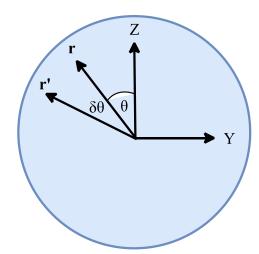


#### Prisms:

- Uncertainty in wedge angle
- Uncertainty in index of refraction

Acceptable / Easily Mitigated
Must be Calibrated





Prism normal vector ideally aligned with x axis



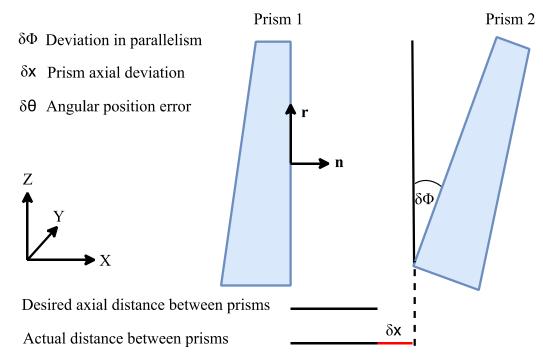
#### Sources of Error

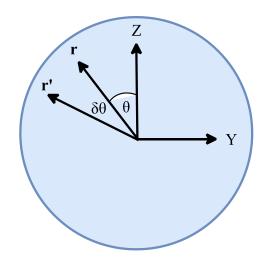


#### Prisms:

- Uncertainty in angular position
- Deviation from parallelism
- Translational deviations

Acceptable / Easily Mitigated
Must be Calibrated





Prism normal vector ideally aligned with x axis





# BACKUP: SOFTWARE/ALGORITHM



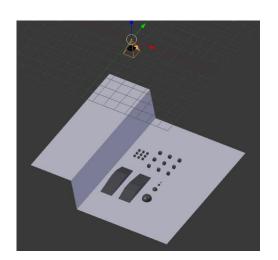
#### Blender Lidar Simulator



- Blender is an open-source program for 3D modeling
- Projects points onto any arbitrary face or object to simulate a lidar scan
- Can extract 3D data by running Python scripts within Blender



**Example Blender artwork** 

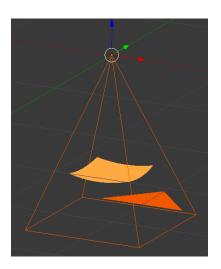


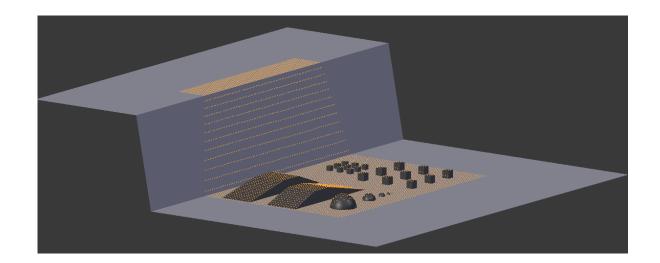


#### Blender Lidar Simulator



- Scan pattern is defined on the plane of the ground, and projected backward onto a sphere centered on the lidar (Blender camera)
- The pattern is then projected outward from the lidar location onto the modeled map
- A Python script exports the point cloud to a CSV file





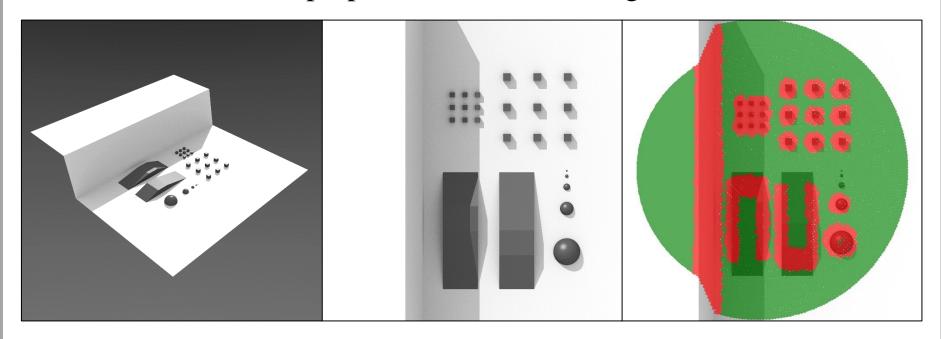


#### Hazard Detection Algorithms



#### Morphological Filter

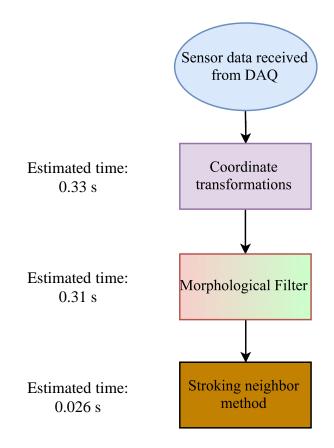
- Identifies hazards by height differences between neighboring points
- Time to run on laptop: 0.31 sec for 10 cm grid





#### Time Estimates





Estimated total time for software elements when run in Python on a personal laptop: about **0.666 s** 

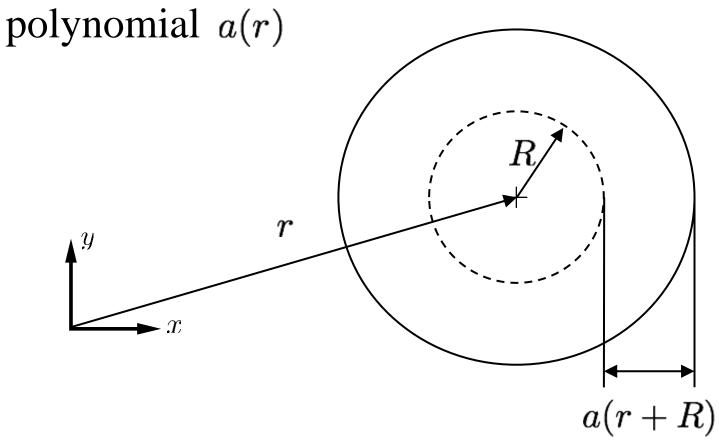
Analysis shows that the BeagleBone will run ~10.24 times slower (**6.83 s**). Given our 10 s margin, we will be well within the time requirement even after porting to the microprocessor. More computationally expensive functions may be written in C for speed improvements.



### **Error Compensation**



We fit the semimajor axis of the error ellipse to a







#### BACKUP: BEAGLEBONE



#### BeagleBone Black Rev C.1





	F	eature						
		I3358BZCZ100						
Processor	1GHz, 2000 MIPS							
Graphics Engine	SGX530 3D, 20M Polygons/S							
SDRAM Memory	512MB D	DR3L 800MHZ						
Onboard Flash	4GB, 8bit 1	Embedded MMC						
PMIC	TPS65217C PMIC regu	lator and one additional LDO.						
Debug Support		oin CTI JTAG, Serial Header						
Power Source	miniUSB USB or DC Jack	5VDC External Via Expansion Header						
PCB	3.4" x 2.1"	6 layers						
Indicators	1-Power, 2-Ethernet,	4-User Controllable LEDs						
HS USB 2.0 Client Port	Access to USB0, C	lient mode via miniUSB						
HS USB 2.0 Host Port	Access to USB1, Type A Socket, 500mA LS/FS/HS							
Serial Port	UART0 access via 6 pin 3.3\	TTL Header. Header is populated						
Ethernet	10/3	100, RJ45						
SD/MMC Connector	micro	oSD, 3.3V						
		et Button						
User Input		ot Button						
		ver Button						
Video Out	16b HDMI, 1280x1024 (MAX) 1024x768.1280x720.1440x900 .1920x1080@24Hz							
Video Out		ID Support						
Audio		Interface, Stereo						
	Power 5V, 3.3V	7 , VDD_ADC(1.8V)						
	3.3V I/C	on all signals						
Expansion Connectors	McASP0, SPI1, I2C, GPIO(69 n	nax), LCD, GPMC, MMC1, MMC2, 7						
Expansion Connectors		mers, 4 Serial Ports, CAN0,						
	EHRPWM(0,2),XDMA Interrupt, Power button, Expansion Board ID							
	(Up to 4 can be stacked)							
Weight	1.4 oz (39.68 grams)							
Power	Refer to Section 6.1.7							



## Expansion Header P8 Pinout



PIN	PROC	NAME	MODE0	MODE1	MODE2	MODE3	MODE4	MODE5	MODE6	MODE7
			mobeo	mode:	MODEL	GND				
1,2 3	R9	GPIO1_6	gpmc_ad6	mmc1_dat6						gpio1[6]
4 5 6 7 8	T9	GPIO1_7	gpmc_ad7	mmc1_dat7						gpio1[7]
5	R8	GPIO1_2	gpmc_ad2	mmc1_dat2						gpio1[2]
6	T8	GPIO1_3	gpmc_ad3	mmc1_dat3						gpio1[3]
/	R7 T7	TIMER4 TIMER7	gpmc_advn_ale		timer4 timer7					gpio2[2]
9	T6	TIMER7	gpmc_oen_ren gpmc_be0n_cle		timer/ timer5					gpio2[3] gpio2[5]
10	U6	TIMERS	gpmc_beun_cle gpmc_wen		timer6					gpio2[5]
11	R12	GPIO1 13	gpmc_wen	lcd data18	mmc1 dat5	mmc2 dat1	eQEP2B in		pr1_pru0_pru_r30_15	gpio2[4]
12	T12	GPIO1 12	gpmc ad12	Lcd data19	mmc1_dat4	Mmc2_dat0	Egep2a in		pr1_pru0_pru_r30_14	gpio1[12]
13	T10	EHRPWM2B	gpmc ad9	lcd_data22	mmc1 dat1	mmc2 dat5	ehrpwm2B		prograduo de constituir de con	gpio0[23]
14	T11	GPIO0_26	gpmc_ad10	lcd_data21	mmc1_dat2	mmc2_dat6	ehrpwm2_tripzone_in			gpio0[26]
15	U13	GPIO1_15	gpmc_ad15	lcd_data16	mmc1_dat7	mmc2_dat3	eQEP2_strobe		pr1_pru0_pru_r31_15	gpio1[15]
16	V13	GPIO1_14	gpmc_ad14	lcd_data17	mmc1_dat6	mmc2_dat2	eQEP2_index		pr1_pru0_pru_r31_14	gpio1[14]
17	U12	GPI00_27	gpmc_ad11	lcd_data20	mmc1_dat3	mmc2_dat7	ehrpwm0_synco			gpio0[27]
18	V12	GPIO2_1	gpmc_clk_mux0	lcd_memory_clk	gpmc_wait1	mmc2_clk			mcasp0_fsr	gpio2[1]
19	U10	EHRPWM2A	gpmc_ad8	lcd_data23	mmc1_dat0	mmc2_dat4	ehrpwm2A			gpio0[22]
20 21	V9 U9	GPIO1_31 GPIO1_30	gpmc_csn2	gpmc_be1n	mmc1_cmd mmc1_clk			pr1_pru1_pru_r30_13	pr1_pru1_pru_r31_13	gpio1[31]
22	V8	GPIO1_30 GPIO1_5	gpmc_csn1 gpmc_ad5	gpmc_clk mmc1 dat5	mmc1_cik			pr1_pru1_pru_r30_12	pr1_pru1_pru_r31_12	gpio1[30] gpio1[5]
23	U8	GPIO1_3 GPIO1_4	gpmc_ad5	mmc1_dat4						gpio1[4]
24	V7	GPIO1_1	gpmc_ad1	mmc1_dat1						gpio1[1]
25	U7	GPIO1 0	gpmc ad0	mmc1 dat0						gpio1[0]
26 27 28	V6	GPIO1 29	gpmc csn0	_						gpio1[29]
27	U5	GPI02_22	lcd_vsync	gpmc_a8				pr1_pru1_pru_r30_8	pr1_pru1_pru_r31_8	gpio2[22]
28	V5	GPI02_24	lcd_pclk	gpmc_a10				pr1_pru1_pru_r30_10	pr1_pru1_pru_r31_10	gpio2[24]
29	R5	GPI02_23	lcd_hsync	gpmc_a9				pr1_pru1_pru_r30_9	pr1_pru1_pru_r31_9	gpio2[23]
30	R6	GPIO2_25	lcd_ac_bias_en	gpmc_a11	0504 : 1		.5. 1		.5 .	gpio2[25]
31	V4	UART5_CTSN	lcd_data14	gpmc_a18	eQEP1_index	mcasp0_axr1	uart5_rxd		uart5_ctsn	gpio0[10]
32 33	T5 V3	UART5_RTSN UART4_RTSN	lcd_data15 lcd_data13	gpmc_a19 gpmc_a17	eQEP1 strobe eQEP1B in	mcasp0_ahclkx mcasp0_fsr	mcasp0_axr3 mcasp0_axr3		uart5_rtsn uart4_rtsn	gpio0[11] gpio0[9]
34	U4	UART3 RTSN	lcd_data13	gpmc_a17	ehrpwm1B	mcasp0_isi	mcasp0_axr2		uart3 rtsn	gpio2[17]
35	V2	UART4 CTSN	lcd_data11	gpmc_a16	eQEP1A in	mcasp0_arickr	mcasp0_axr2		uart4 ctsn	gpio0[8]
36	U3	UART3 CTSN	lcd_data10	gpmc a14	ehrpwm1A	mcasp0_axr0	modopo_dxi2		uart3 ctsn	gpio2[16]
37	U1	UART5 TXD	lcd_data8	gpmc a12	ehrpwm1 tripzone in	mcasp0 aclkx	uart5 txd		uart2 ctsn	gpio2[14]
38	U2	UART5 RXD	lcd data9	gpmc a13	ehrpwm0 synco	mcasp0_fsx	uart5 rxd		uart2 rtsn	gpio2[15]
39	T3	GPIO2_12	lcd_data6	gpmc_a6		eQEP2_index		pr1_pru1_pru_r30_6	pr1_pru1_pru_r31_6	gpio2[12]
40	T4	GPI02_13	lcd_data7	gpmc_a7		eOFP2_strobe	pr1_edio_data_out7	pr1_pru1_pru_r30_7	pr1_pru1_pru_r31_7	gpio2[13]
41	T1	GPIO2_10	lcd_data4	gpmc_a4		eQEP2A_in		pr1_pru1_pru_r30_4	pr1_pru1_pru_r31_4	gpio2[10]
42	T2	GPIO2_11	lcd_data5	gpmc_a5		eQEP2B_in		pr1_pru1_pru_r30_5	pr1_pru1_pru_r31_5	gpio2[11]
43	R3	GPIO2_8	lcd_data2	gpmc_a2		ehrpwm2_tripzone_in		pr1_pru1_pru_r30_2	pr1_pru1_pru_r31_2	gpio2[8]
44 45	R4 R1	GPIO2 9 GPIO2 6	lcd data3 lcd data0	gpmc a3		ehrpwm0 synco ehrpwm2A		pr1_pru1_pru_r30_3	pr1_pru1_pru_r31_3	gpio2[9]
46	R2	GPIO2_6 GPIO2_7	lcd_data0	gpmc_a0 gpmc_a1		ehrpwm2B		pr1_pru1_pru_r30_0 pr1_pru1_pru_r30_1	pr1_pru1_pru_r31_0 pr1_pru1_pru_r31_1	gpio2[6] gpio2[7]
40	R2	GFIOZ_I	icu_uata i	gpmc_a1	l	enipwinzo		pr1_pru1_pru_r30_1	pr1_pru1_pru_r31_1	gpio2[/]



## Expansion Header P9 Pinout



PIN	PROC	NAME	MODE0	MODE1	MODE2	MODE3	MODE4	MODE5	MODE6	MODE7
1,2						GND				
3,4						DC_3.3V VDD_5V				
5,6 7,8						SYS_5V				
9						PWR_BUT				
10	A10					SYS_RESETn				
11	T17	UART4 RXD	gpmc_wait0	mii2 crs	gpmc_csn4	rmii2_crs_dv	mmc1_sdcd		uart4_rxd_mux2	gpio0[30]
12	U18	GPIO1_28	gpmc_be1n	mii2_col	gpmc_csn6	mmc2_dat3	gpmc_dir		mcasp0_aclkr_mux3	gpio1[28]
13	U17	UART4_TXD	gpmc_wpn	mii2_rxerr	gpmc_csn5	rmii2_rxerr	mmc2_sdcd		uart4_txd_mux2	gpio0[31]
14	U14	EHRPWM1A	gpmc_a2	mii2_txd3	rgmii2_td3	mmc2_dat1	gpmc_a18		ehrpwm1A_mux1	gpio1[18]
15	R13	GPIO1_16	gpmc_a0	gmii2_txen	rmii2_tctl	mii2_txen	gpmc_a16		ehrpwm1_tripzone_input	gpio1[16]
16	T14	EHRPWM1B	gpmc_a3	mii2_txd2	rgmii2_td2	mmc2_dat2	gpmc_a19		ehrpwm1B_mux1	gpio1[19]
17	A16	I2C1_SCL	spi0_cs0	mmc2_sdwp	I2C1_SCL	ehrpwm0_synci	pr1_uart0_txd			gpio0[5]
18	B16	I2C1_SDA	spi0_d1	mmc1_sdwp	I2C1_SDA	ehrpwm0_tripzone	pr1_uart0_rxd			gpio0[4]
19	D17	I2C2_SCL	uart1_rtsn	timer5	dcan0_rx	I2C2_SCL	spi1_cs1	pr1_uart0_rts_n		gpio0[13]
20	D18	I2C2_SDA	uart1_ctsn	timer6	dcan0_tx	I2C2_SDA	spi1_cs0	pr1_uart0_cts_n		gpio0[12]
21	B17	UART2_TXD	spi0_d0	uart2_txd	I2C2_SCL	ehrpwm0B	pr1_uart0_rts_n		EMU3_mux1	gpio0[3]
22	A17	UART2_RXD	spi0_sclk	uart2_rxd	I2C2_SDA	ehrpwm0A	pr1_uart0_cts_n		EMU2_mux1	gpio0[2]
23	V14	GPI01_17	gpmc_a1	gmii2_rxdv	rgmii2_rxdv	mmc2_dat0	gpmc_a17		ehrpwm0_synco	gpio1[17]
24	D15	UART1_TXD	uart1_txd	mmc2_sdwp	dcan1_rx	I2C1_SCL		pr1_uart0_txd	pr1_pru0_pru_r31_16	gpio0[15]
25	A14	GPI03_21*	mcasp0_ahclkx	eQEP0_strobe	mcasp0_axr3	mcasp1_axr1	EMU4_mux2	pr1_pru0_pru_r30_7	pr1_pru0_pru_r31_7	gpio3[21]
26	D16	UART1_RXD	uart1_rxd	mmc1_sdwp	dcan1_tx	I2C1_SDA		pr1_uart0_rxd	pr1_pru1_pru_r31_16	gpio0[14]
27	C13	GPIO3_19	mcasp0_fsr	eQEP0B_in	mcasp0_axr3	mcasp1_fsx	EMU2_mux2	pr1_pru0_pru_r30_5	pr1_pru0_pru_r31_5	gpio3[19]
28	C12	SPI1_CS0	mcasp0_ahclkr	ehrpwm0_synci	mcasp0_axr2	spi1_cs0	eCAP2_in_PWM2_out	pr1_pru0_pru_r30_3	pr1_pru0_pru_r31_3	gpio3[17]
29	B13	SPI1_D0	mcasp0_fsx	ehrpwm0B		spi1_d0	mmc1_sdcd_mux1	pr1_pru0_pru_r30_1	pr1_pru0_pru_r31_1	gpio3[15]
30	D12	SPI1_D1	mcasp0_axr0	ehrpwm0_tripzone		spi1_d1	mmc2_sdcd_mux1	pr1_pru0_pru_r30_2	pr1_pru0_pru_r31_2	gpio3[16]
31	A13	SPI1_SCLK	mcasp0_aclkx	ehrpwm0A		spi1_sclk	mmc0_sdcd_mux1	pr1_pru0_pru_r30_0	pr1_pru0_pru_r31_0	gpio3[14]
32						VADC				
33	C8					AIN4				
34						AGND				
35	A8					AIN6				
36	B8					AIN5				
37	B7					AIN2				
38	A7					AIN3				
39	B6					AIN0				
40	C7					AIN1				
41#	D14	CLKOUT2	xdma_event_intr1		tclkin	clkout2	timer7_mux1	pr1_pru0_pru_r31_16	EMU3_mux0	gpio0[20]
	D13	GPIO3_20	mcasp0_axr1	eQEP0_index	., ,	Mcasp1_axr0	emu3	pr1_pru0_pru_r30_6	pr1_pru0_pru_r31_6	gpio3[20]
42@	C18	GPI00_7	eCAP0_in_PWM0_out	uart3_txd	spi1_cs1	pr1_ecap0_ecap_capin_apwm_o	spi1_sclk	mmc0_sdwp	xdma_event_intr2	gpio0[7]
	B12	GPIO3_18	Mcasp0_aclkr	eQEP0A_in	Mcaspo_axr2	Mcasp1_aclkx		pr1_pru0_pru_r30_4	pr1_pru0_pru_r31_4	gpio3[18]
43-46						GND				





### BACKUP: MEASUREMENTS

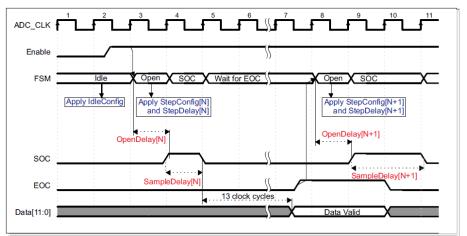


### Measurement Timing



## Lidar range and beam attitude measurements shall be taken within one microsecond.

- 15 ADC clocks per sample
   625 ns full conversion
- Reading the quadrature decoder registers may be accomplished in this time (~4 ns)



TimeGe

#### Full measurement time is limited by the ADC

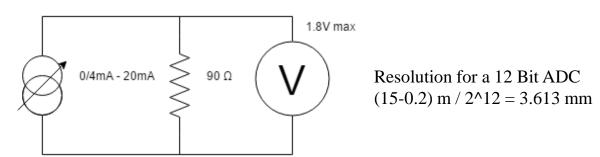


#### Lidar Measurement



The lidar shall send range data to the on-board processor or DAQ

- The lidar produces a 0/4 mA to 20 mA current loop based off of the range between two set points A and B.
- A 90 ohm resistor is used to turn this into a 1.8 V max signal which is read on the microcontroller's ADC





## Quadrature Decoding



The beam attitude measurement shall be sent to the on-board processor or DAQ.

- The BeagleBone Black has quadrature decoders that interface directly with the OPS optical encoders
- Taking measurements is as simple as reading each counter



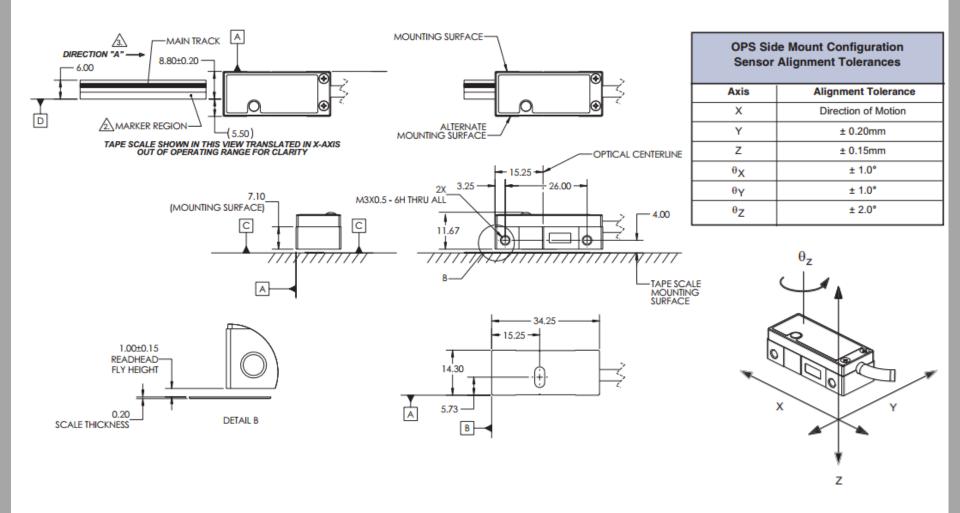


## BACKUP: ENCODER INTEGRATION



## OPS Encoder Mounting Side

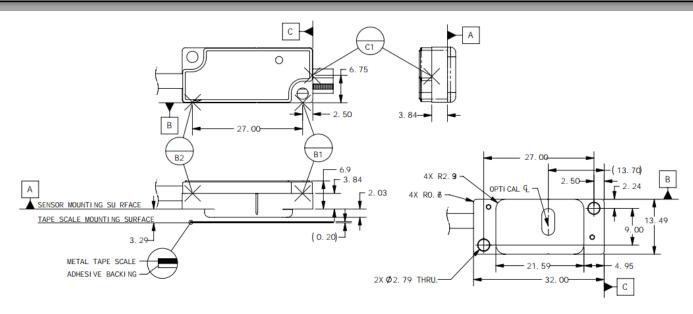






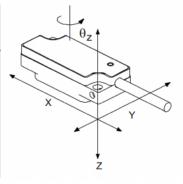
#### OPS Encoder Mounting Top





#### **Wide Alignment Tolerances**

OPS Top Mount Configuration Sensor Alignment Tolerances					
Axis	Alignment Tolerance				
X	Direction of Motion				
Y	± 0.20mm				
Z	± 0.15mm				
θX	± 1.0°				
θY	± 1.0°				
θZ	± 2.0°				



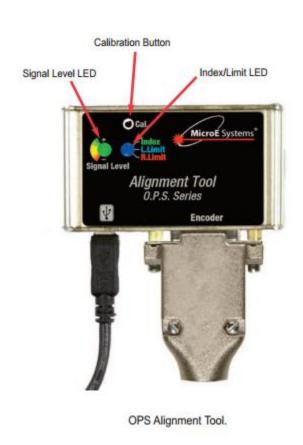
#### Sensor Size & Weight (top mount sensor)

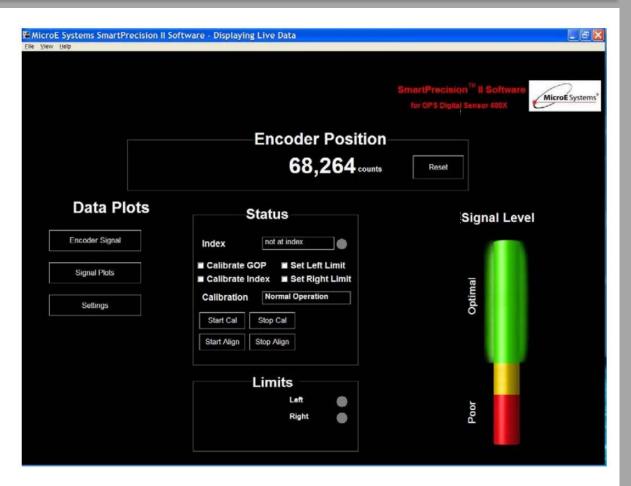
Height	Width	Length
0.35[8.93mm]	0.53 [13.49mm]	1.26 [32.00mm]
Weight	6g (without cable)	



## OPS Encoder Alignment











# BACKUP: COMMUNICATION

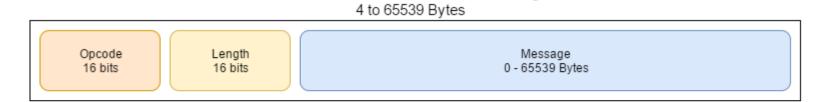


# Communication Between Microcontroller and PC



- UART 115200 bits / sec
- USB 2.0 480 Mbits / sec (high speed)
- Ethernet/IP 10/100/1000 Mbits /sec

Controller-PC communication layer agnostic to protocol



Communication Packet Between PC and BeagleBone

Sample Message Data

Encoder 0
32 Bits

Encoder 1
32 Bits

ADC from VDM28
12 bits extended to
32

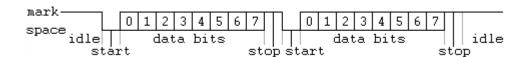


## **UART**



- Will require an FTDI
- 115200 bits/s
- 8 data bits per packet1 start and 1 stop
- 11520 bytes/s







## Ethernet data rate feasibility



#### IPv4

Max Ethernet packet 1518 bytes

68 bytes of UDP overhead (with IP and Ethernet frames)

1472 bytes left for data → 60 measurements per packet

1512 byte total packet size

100 Mb/s: 8127 frames/sec \* 1512 bytes/frame =

12.288 Mbytes/s

1000 Mb/s: 81274 frames/sec \* 1512 bytes/frame =

122.8 Mbytes/s



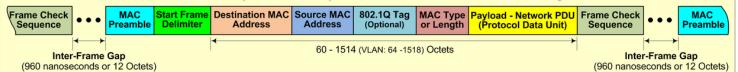
## Ethernet UDP Overhead



#### Fast Ethernet (IEEE 802.3u) - UDP

Maximum Ethernet frames and data throughput rate calculations.

#### Fast Ethernet (IEEE 802.3u) Frame Structure with UDP Datagram



## Fast Ethernet Frame Component Size With UDP Datagram

With ODP Datagram						
Frame Component	Component Size					
MAC Preamble	7 Octets of: <b>10</b> °	7 Octets of: 10101010				
Start Frame Delimiter	1 Octet of: 10	1 Octet of: 10101011				
Destination MAC Address	6 Octets	6 Octets				
Source MAC Address	6 Octets	6 Octets				
802.1Q VLAN TAG ID (Optional)	4 Octets (Optional)					
MAC Type or Length	2 Octets	2 Octets				
MTU	IP Header	20 Octets				
(Maximum Transmission Unit)	UDP Header	8 Octets				
Payload Network PDU	Data/Padding	18 - 1472 Octets				
Protocol Data Unit:	Total:	<b>46 - 1500 Octets</b> (Max: 1504 – VLAN)				
Frame Check Sequence (CRC)	4 Octets					
Inter-Frame Gap  ● ● ●	12 Octets (960 nanoseconds)					
Total Physical Frame Size:	84 - 1538 Octets (Max: 1544 -VLAN)					

#### Fast Ethernet Maximum Frame and Data Throughput Rate Calculation with UDP Datagram

Rate Term	Value					
Fast Ethernet Bit Rate	100 Mbit/sec -or- 100Mb/sec					
Fast Ethernet Bit Time	10 nanoseconds (.00000001 seconds)					
1 Octet (Byte)	8 Bits					
Max Octet Rate	(100Mb/sec)/((8 Bits) = 12,500,000 Octets/sec					
Max Frame Rate (84 Octet Frames) Min Packet (60 Bytes + 4 Bytes CRC)	(100Mb/sec)/((8 Bits)*(84 Octets/Frame)) = 148,810 Frames/sec (FPS)					
Max UDP Data Rate (84 Octet Frames) Min UDP Packet (60 Bytes + 4 Bytes CRC)	(148,810 Frames/sec)*(18 Bytes/Frame) = 2,678,571 Bytes/sec					
Max Frame Rate (1538 Octet Frames) Max Packet (1514 Bytes + 4 Bytes CRC)	(100Mb/sec)/((8 Bits)*(1538 Octets/Frame)) = 8,127 Frames/sec (FPS)					
Max UDP Data Rate (1538 Octet Frames) Max UDP Packet (1514 Bytes + 4 Bytes CRC)	(8,127 Frames/sec)*(1472 Bytes/Frame) = 11,963,589 Bytes/sec					
Max Fast Ethernet Frame Bandwidth Max Packet (60 Bytes + 4 Bytes CRC) Max Packet (60 Bytes)	(148,810 Frames/sec)*(64 Bytes/Frame) = 9,523,840 Bytes/sec ( 9.082641 MiB/s) (148,810 Frames/sec)*(60 Bytes/Frame) = 8,928,600 Bytes/sec ( 8.514977 MiB/s)					
Max Fast Ethernet Frame Bandwidth Max Packet (1514 Bytes + 4 Bytes CRC) Max Packet (1514 Bytes)	(8,127 Frames/sec)*(1518 Bytes/Frame) = 12,336,786 Bytes/sec (11.765276 MiB/s) (8,127 Frames/sec)*(1514 Bytes/Frame) = 12,304,278 Bytes/sec (11.734274 MiB/s)					

\*\*\* Note 1: Units - M: 1,000,000 Mi: 1,048,576



## Ethernet UDP Overhead



#### Gigabit Ethernet (IEEE 802.3ab) - UDP

Maximum Ethernet frames and data throughput rate calculations.

## Gigabit Ethernet (IEEE 802.3ab) Frame Structure with UDP Datagram MAC Start Frame Destination MAC Source MAC Address Address (Optional) or Length (Protocol Data Unit) Frame Check Sequence

Inter-Frame Gap (96 nanoseconds or 12 Octets)

Frame Check

Sequence

60 - 1514 (VLAN: 64 -1518) Octets

|←────| Inter-Frame Gar

123,048,836 Bytes/sec (117.348515 MiB/s)

Inter-Frame Gap (96 nanoseconds or 12 Octets)

MAC

#### Gigabit Ethernet Frame Component Size With UDP Datagram

With ODF Datagram						
Frame Component	Component Size					
MAC Preamble	7 Octets of: 10101010					
Start Frame Delimiter	1 Octet of: 10	1 Octet of: 10101011				
Destination MAC Address	6 Octets	6 Octets				
Source MAC Address	6 Octets					
802.1Q VLAN TAG ID (Optional)	4 Octets (Optional)					
MAC Type or Length	2 Octets					
MTU	IP Header	20 Octets				
(Maximum Transmission Unit)	UDP Header	8 Octets				
Payload Network PDU	Data/Padding	18 - 1472 Octets				
Protocol Data Unit:	***Total:	<b>46 - 1500 Octets</b> (Max: 1504 – VLAN)				
Frame Check Sequence (CRC)	4 Octets					
Inter-Frame Gap • • •	12 Octets (96 nanoseconds)					
Total Physical Frame Size:	84 - 1538 Octets (Max: 1544 -VLAN)					

#### Gigabit Ethernet Maximum Frame and Data Throughput Rate Calculation with UDP Datagram

Rate Term	Value
Gigabit Ethernet Bit Rate	1000 Mbit/sec -or- 1000Mb/sec
Gigabit Ethernet Bit Time	1 nanosecond (.000000001 seconds)
1 Octet (Byte)	8 Bits
Max Octet Rate	(1000Mb/sec)/((8 Bits) = 125,000,000 Octets/sec
Max Frame Rate (84 Octet Frames) Min Packet (60 Bytes + 4 Bytes CRC)	(1000Mb/sec)/((8 Bits)*(84 Octets/Frame)) = 1,488,095 Frames/sec (FPS)
Max UDP Data Rate (84 Octet Frames) Min UDP Packet (60 Bytes + 4 Bytes CRC)	(1,488,095 Frames/sec)*(18 Bytes/Frame) = 26,785,714 Bytes/sec
Max Frame Rate (1538 Octet Frames) Max Packet (1514 Bytes + 4 Bytes CRC)	(1000Mb/sec)/((8 Bits)*(1538 Octets/Frame)) = 81,274 Frames/sec (FPS)
Max UDP Data Rate (1538 Octet Frames) Max UDP Packet (1514 Bytes + 4 Bytes CRC)	(81,274 Frames/sec)*(1472 Bytes/Frame) = 119,635,891 Bytes/sec
Max Gigabit Ethernet Frame Bandwidth Max Packet (60 Bytes + 4 Bytes CRC) Max Packet (60 Bytes)	(1,488,095 Frames/sec)*(64 Bytes/Frame) = 95,238,080 Bytes/sec ( 90.876031 MiB/s) (1,488,095 Frames/sec)*(60 Bytes/Frame) = 89,285,700 Bytes/sec ( 85.149477 MiB/s)
Max Gigabit Ethernet Frame Bandwidth Max Packet (1514 Bytes + 4 Bytes CRC)	(81,274 Frames/sec)*(1518 Bytes/Frame) = 123,373,932 Bytes/sec (117.658550 MiB/s) (81,274 Frames/sec)*(1514 Bytes/Frame) =

<sup>\*\*\*</sup> Note 1: IEEE 802.3ab - Gigabit Ethernet over copper twisted-pair cabling.

<sup>\*\*\*</sup> Note 2: Gigabit Ethernet allows for larger MTUs (Jumbo or Super Jumbo Frames).

<sup>\*\*\*</sup> Note 3: Units – M: 1,000,000 Mi: 1,048,576



## USB Data Rate Feasibility



#### Universal Serial Bus Specification Revision 2.0

Table 5-10. High-speed Bulk Transaction Limits

Protocol	Overhead (55 bytes)	(3x4 SYNC bytes, 3 PID bytes, 2 EP/ADDR+CRC bytes, 2 CRC16, and a 3x(1+11) byte interpacket delay (EOP, etc.))				
Data Payload	Max Bandwidth (bytes/second)	Microframe Bandwidth per Transfer	Bandwidth Transfers Rema		Bytes/ Microframe Useful Data	
1	1064000	1%	133	52	133	
2	2096000	1%	131	33	262	
4	4064000	1%	127	7	508	
8	7616000	1%	119	3	952	
16	13440000	1%	105	45	1680	
32	22016000	1%	86	18	2752	
64	32256000	2%	63	3	4032	
128	40960000	2%	40	180	5120	
256	49152000	4%	24	36	6144	
512	53248000	8%	13	129	6656	
	60000000				7500	

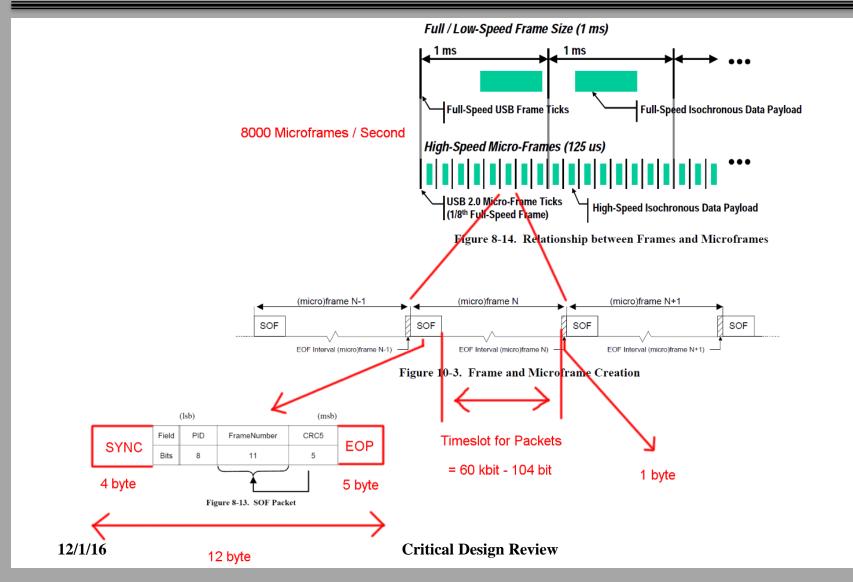
21 measurements for the maximum data payload produces a 508-byte data payload Speeds should be over 50 million bytes a second

Max



## **USB 2.0**

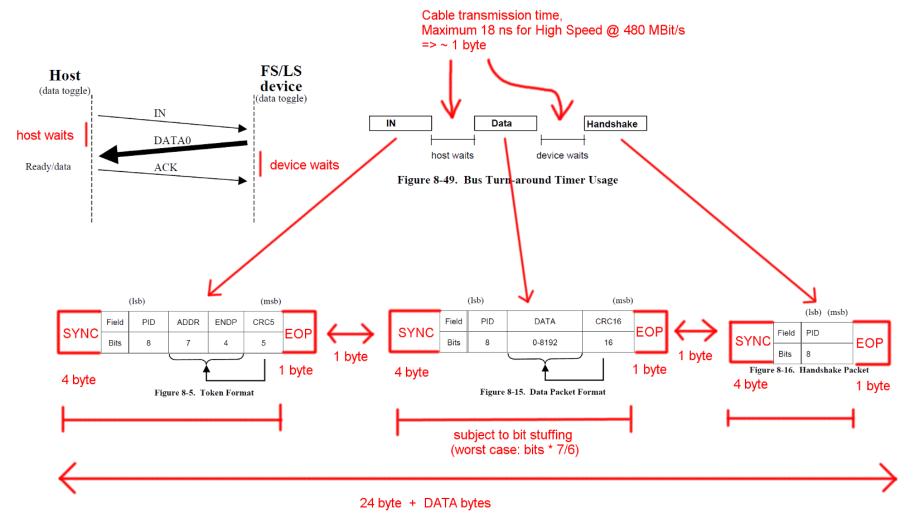






## **USB 2.0**

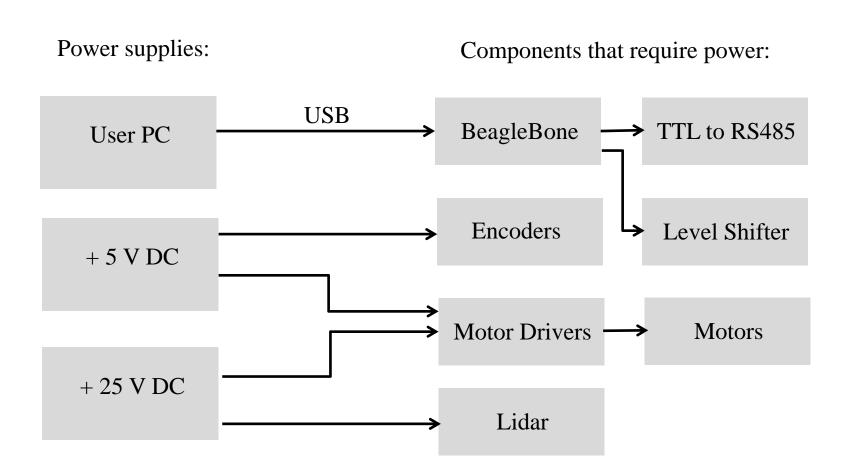






## Power

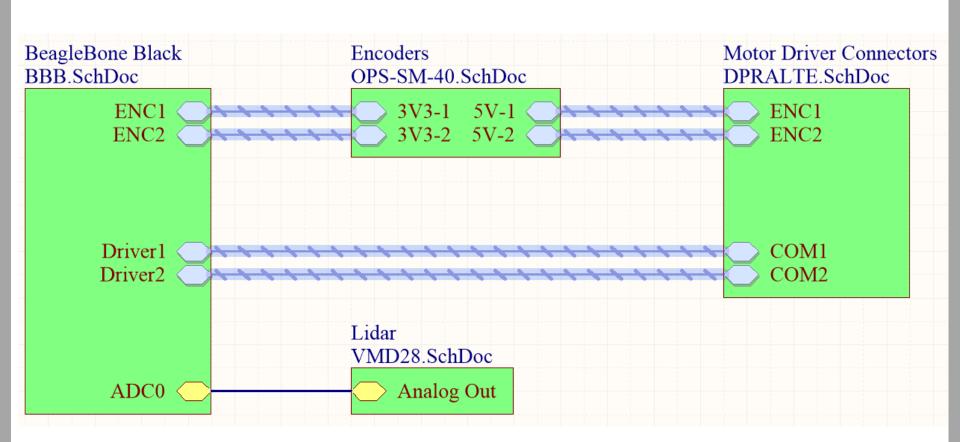






## Top Level Connections

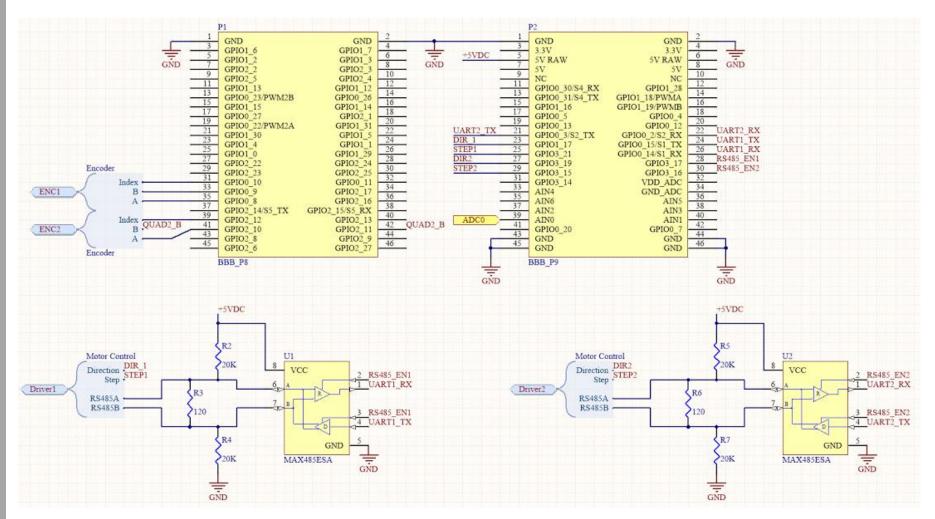






# BeagleBone Connections

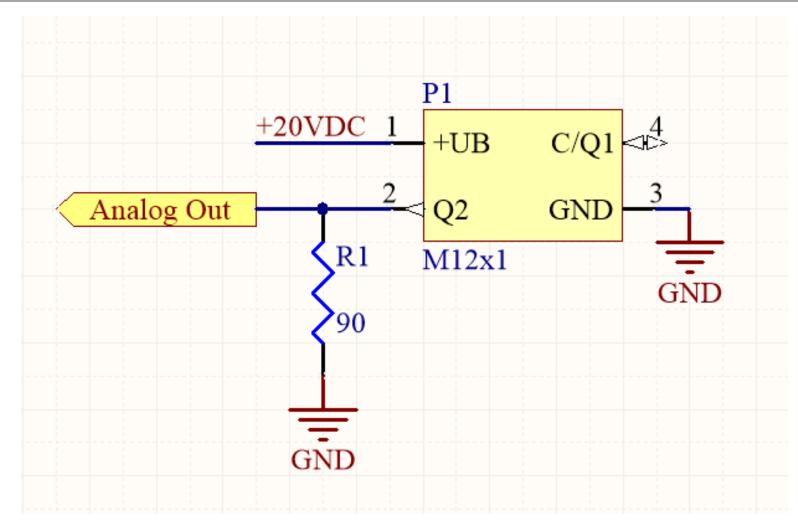






## Lidar Connections









# BACKUP: VERIFICATION AND VALIDATION



## Artec Eva Lite 3D Scanner



### **Specifications**

- 3D resolution: 0.5 mm
- 3D point accuracy: 0.1 mm

## **Output**

 Creates a SOLIDWORKS file of the scanned object/surface





# Scan Pattern Feasibility



#### **Solution**: Perform two system-level tests:

- 1. Verify that the sensor package can obtain measurements with the required resolution in a longer period of time
- 2. Verify the ability of the system to perform a 60-second scan/analysis, even though the required resolution cannot be met

#### Resolution requirement test:

- Lidar frequency: 100 Hz
- Point spacing (exterior): 2.5 cm
- Exterior spiral arc length: 32.32 m
- Time to complete scan: ~12 min
- Maximum prism angular acceleration
- Required angular velocity:

#### 4.6536 rpm

• Maximum prism angular acceleration: 4.8e-6 rad/s<sup>2</sup>

#### <u>Time requirement test:</u>

- Spiral spacing of 8.66 cm gives 59 total spirals
- Time: 50 seconds (leaving margin for analysis)
- Required lidar frequency: 382 Hz
- Required angular velocity:71.0763 rpm
- Maximum prism angular acceleration:
   3.76e-6 rad/s<sup>2</sup>



# Test Setup (Lidar)



- Receive return through glass
  - Shoot lidar through panes of glass and use oscilloscope to determine if the lidar is receiving a return
- Range, error and precision
  - Over a timespan on 60 sec, consistent measurements with accuracy of +/- 5 cm must be taken
- Reflective tape
  - Measure signal return accuracy, consistency, and strength from surface with and without retro-reflective tape
- Sample Frequency
  - Using an oscilloscope, determine time (in milliseconds) between range measurements



# Test Setup (Prisms)



- Must be able to turn beam 20°
  - Mount prisms parallel to lidar, manually rotate prisms and verify 20° beam divergence
- Returns through glass
  - If the lidar does not receive returns through glass,
     replace the glass with coated prisms and repeat
     trials



# Test Setup (Encoders)



- Determine functionality of hardware
  - Connect encoders to microprocessor and motors
  - Manually move motor to verify functionality of encoders
    - This is just to test connectivity and verify that communication is working properly



# Test Setup (Motors)



- Motor functionality
  - Connect motors to motor drivers and provide any arbitrary commands, verify response happens
- All required motor rates must be achievable
  - Once motor, encoder, driver, and microprocessor system is fully integrated
  - Command to maximum rate of 71 rpm and hold for 50 sec
  - Command to minimum rate of 4 rpm and hold for 13 min
- Time to accelerate to desired motor rates
  - Given motor rate commands, verify motor accelerations are within desired bounds from encoder output analysis



## Test Setup (Motor Drivers)



- Given any input the drivers must change the position of the motors
  - This can be visually verified
- Verify that command accuracy of 0.1° can be met
  - Can verify commanded vs actual by manually comparing commanded angle and actual angle
  - Can more accurately verify by comparing computational models of prism rotation, given a single motor angle displacement, against encoder positions
    - Measuring initial and final laser position physically and predictively in software



# Test Setup (Software)



- Unit tests
  - x, y, z coordinate rotation
  - Read in IMU data, prism positions, and range
    - Combine to produce range measurement
  - Actual hazard output should match expected
    - Expected generated by software mockup
  - Generate health and status reports
  - Output results



# Full Budget



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Item	Quantity	Cost per (USD)	Total Discounts (USD)	Total cost (USD)	Source/Notes
Lidar:					
Pepperl-Fuchs VDM28-15-L-IO/73c/110/122	1	470.17	470.17	0	https://www.carltonbates.com/Photoelectric-Sensors-Reflex-Reflective-Block-Style-/PEPPI
Motors:					
ULT-165-A-12-A-x-00x	1	1033		1033	Requested quote from "http://www.celeramotion.com/"
Encoders:					
OPS read head (OD 3.937" ID 2.756")	2	560		1120	Requested quote from "http://www.celeramotion.com/"
Bearings:					
VA030CP0 Thin Section Bearing 3"x3 1/2"x1/4" inch Open	4	81.77		327.08	http://www.vxb.com/VA030CP0-Thin-Section-3-x3-1-2-x1-4-inch-Open-p/kit8779.htm
Microprocessor:					
BeagleBone Black Rev C.1	1	55		55	https://www.adafruit.com/products/1996
Reflective Tape					
Oralite (Reflexite) R99 Rail Microprismatic Retroreflective					
Conspicuity Tape: 4 in. x 15 ft. (White)	9	36.22		325.98	https://www.amazon.com/gp/product/B00420CDOK/ref=s9_simh_gw_g469_i4_r?ie=UTFE
Motor Drivers:					
DPRALTE - 020B080	2	605	363	847	https://www.servo2go.com/product.php?ID=101890#details
Risley prism:					
P-WRC059 coated with BBAR 400-700 nm	2	108		216	Requested quote from "http://www.rossoptical.com/"
Materials					
Mat: misc electronics				300	
Mat: Fasteners				166	
Mat: Stock				849.77	
Mat: Tooling				812	can increase to 922 if budget provides, buying additional tools for arseni
Shipping				330	Assuming \$15 per item
Total (Cost)				6381.83	Tax exempt
Budget				5000	Provided by the CU Aerospace Engineering department
					UROP Funding
				2300	Thayer
Total (Budget)				8300	·
Percent Margin				23.11048193	



# Full Budget: Materials



Part	Mat	Shape	Н	L / diameter	W	Stock Dims	Cost	Quantity	Total	All from mcmaster
Base Plate	Al 6061	Plate	0.625	10	8	.625x12x12	74.9	1	149.8	
Lidar Stand	Al 6061	Block	2.992	3.75	2.5	3x3.5x6	53.5	1	53.5	
Side Bracket	Al 6061	Plate	0.5	3.32	2.25	.5x3x6	68.1	2	136.2	
Top Bracket	Al 6061	Plate	0.125	3.32	3	.125x4x6	60.19	1	60.19	
Outside Bracket	Al 6061	Plate	0.402	7	7	.438x8x8	32.86	2	65.72	
Main Housing	Al 6061	Plate	1.3	7	7	1.5x8x8	71.12	2	142.24	
Bearing Clamp	Al 6061	Cyl	0.163	4.25		4.75x0.5	11.32	2	22.64	
Prism Enclosure	Al 6061	Cyl	0.625	3.125		3.25x1	9.34	2	18.68	
Rotor	Al 6061	Cyl	0.523	5		6x.75	77.2	2	154.4	
Encoder Hub	304 SS	Cyl	0.188	3.272		3.5x.5	23.2	2	46.4	
									849.77	Total (USD)
Screws	Quantity	Cost Per 25	Total			Dowel Pin screw equivil	Quantity	Cost Per Pack	Total	
#2-56	50	7.2	14.4	All from mcmaster		#2-26	10	8.23	8.23	All from mcmaster
#4-40	50	7.5	15			#4-40	6	9.79	9.79	
#8-32	75	7.63	22.89			#8-32	6	11.67	11.67	
1/4 20	20	7.84	7.84			1/4 20	6	13.13	13.13	
			60.13	Total (USD)					42.82	Total (USD)
Machining Tools	Quantity	Cost Per	Total			Cabling	100			
End Mills	1	186.72	186.72	from mscdirect						
Drill Kit	1	100	100							
Edge Finder	1	63.39	63.39							GRAND TOTAL (USD)
			350.11	Total (USD)						1402.