

Laboratory for Atmospheric and Space Physics University of Colorado **Boulder**

Boulder Unmanned Sensor for Transport Events and Repositioner

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Project Motivation

- Dr. Wang's research at LASP suggests that charged particles could be lifted by Coulomb force
 - Dust transport events: micron-sized dust particles are charged by various sources in space and ejected from the surface of lowmass bodies
- Current instrument is too large for a space application in low-gravity
- Data could be collected with a smaller instrument in a CubeSat form factor, for a potential mission to an asteroid



Project Statement

- Dust BUSTER will <u>miniaturize</u>, <u>manufacture</u>, and <u>test</u> a **Technology Readiness Level (TRL) 4** dust instrument to characterize dust transport events similar to those that occur on asteroids
- To aid the instrument, the team will also <u>design</u> and <u>test</u> an <u>Autonomous Repositioning System (ARS)</u> to tilt a 6U CubeSat to a specified angle for dust collection

Overall Mission ConOps





Levels of Success

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	Level 1	Level 2	Level 3
Instrument	 - 2U TRL 4 dust instrument - Operates in vacuum chamber - Interfaces mechanically with CubeSat 	- Wire electrodes remain intact upon 10 m/s impact	-
CubeSat/ ARS	 Construct 6U CubeSat model Tilt CubeSat model up to 45 degrees on a flat surface Determine which side of the CubeSat has the least sun 	 Open loop autonomous tilt with 5° accuracy Operates on sandy surface 	 Closed loop tilt with 1° accuracy Instrument cover opens once under operator command
Software	 Detect dust via external trigger Send dust data over serial Post processing algorithm extracts mass, velocity, charge 	- Self-triggering dust detection algorithm	- Determine uncertainty in mass, velocity, and charge

FBD





Diagram

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Design Recap





Photodiodes

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Design Recap -Door Mechanism

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Design Recap -TIlting Mechanism

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Design Recap -Dust Instrument

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Critical Project Elements

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Critical Project Element	Relation to Testing
Sun determination	Full sky accuracy, closed-loop autonomous tilt
Tilting mechanism	Tilt actuation accuracy, closed-loop autonomous tilt
Surviving Impact	Instrument Dust Trajectory Sensor (DTS) and wire electrode impact testing
Real-time event detection	Analog electronics, DTS, electron shield, trigger algorithm, dust detection

Scheduling



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ARS Test Schedule



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Instrument Test Schedule

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Completed



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Instrument Test Readiness



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Instrument Testing Flow





Amplifier

Verify our implementation of customer's design for a charge sensitive amplifier, Req 5.1, 5.11, 5.12

Purpose: Measure each CSA's amplification of a simulated dust event.

Facility: Electronics Lab

- Power supply $(\pm 5V \& \pm 15V)$
- Waveform generator
- Oscilloscope & probe
- ESD mat/straps
- Assembled CSA PCB

Measurement: voltage of the amplified signal (expected test gain = 100)



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Analog Electronics. Charge Sensitive Amplifier

Verify our implementation of customer's design for a charge sensitive amplifier, Req 5.1, 5.11, 5.12

Purpose: Measure each CSA's amplification of a simulated dust event.

Facility: Electronics Lab

- Power supply $(\pm 5V \& \pm 15V)$
- Waveform generator
- Oscilloscope & probe
- ESD mat/straps
- Assembled CSA PCB

Measurement: voltage of the amplified signal (expected test gain = 100)



Impact Testing

Verify that Wire Electrodes can withstand 10 m/s impact (Req 2.5.1)

Purpose: Drop a DTS at successively larger impact velocities to characterize failure (when wire electrode becomes free to move) Facility: Idea Forge

- Lansmont 15D Shock Test Machine
- One DTS unit
- Accelerometer

Procedure:

- Mount DTS to drop test table
- Raise table up to desired height and drop
- Visually inspect wire electrodes for broken or freely moving wires
- Drop again at new height



Impact Testing

Verify that Wire Electrodes can withstand 10 m/s impact (Req 2.5.1)

Testing Set-up:

- Preliminary drops without DTS to determine drop height vs velocity relationship
- Wire electrodes installed as rigid bar (no slack) with no requirement for tension
- Analyzing for failure (wires free to move) at 3 locations after each test

Limitations:

- Only 1 DTS to test
- Material deformations are difficult to analyze



Intro to facilities and data

Location	IMPACT Lab (CU East Campus)		
Equipment (Customer Provided)	 Vacuum chamber (w/ pump) Vacuum wall cable interfaces Dust dropper Free electron emitter Power supplies (±5V, +3.4V, ±15V, ±5kV) Data acquisition 8-channel voltage DAQ Lab computer & software Translation table 		
Data Out	 Analog voltages DTS Stand-alone Electron Shield Voltage data file to calculate charge, mass, and velocity MCU/Trigger Software Full Instrument Test 		



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DTS Stand-alone

Verify wire electrode and CSA correctly respond to dust event

Purpose: Confirm the wire electrode connections and CSA conversion from charge to voltage, and signal amplification

Facility: IMPACT no vacuum

Measure: Live analog voltage output from CSA board (8 wires at a time)

Success: Signal roughly matches expected shape and voltage magnitude (~2 V)

Sample Shape:





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MCU/Trigger software

Verify trigger threshold and MCU data processing

Purpose: Test the ability of the trigger to correctly identify dust events and MCU's ability to process and send data over serial

Facility: IMPACT no vacuum Output: CSA digital voltage over serial





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Electron Shield

Verify that electrons are repelled in TRL4 environment

Purpose: Verify magnetic shield blocks electrons up to 100 eV of energy which would cause noise on the wire electrodes

Facility: IMPACT vacuum (for free electrons)

Procedure:

- Replace dust dropper with electron emitter
- Apply set voltage to emit electrons up to 100 eV of energy
- Measure response from CSA over test duration (1 min)

Measure: Digital Voltage Success: Null Voltage (random noise)



Full Instrument Test

Verify that instrument detects dust particles that enter the instrument. Req 2, 5, 6

Purpose: Detect a dust event and extract the charge, mass, and velocity of the particle.

Facility: IMPACT vacuum

Measure: Output digital voltage in a data file to postprocessing software, calculate charge, mass and velocity distribution in 6 positions (~90 events total)

Data Type	Expected Range	
Charge (Q)	1 - 160 fC	
Mass (<i>m</i>)	50-150 µg	
Velocity (<i>v</i>)	1 - 2 m/s	



ARS Test Readiness



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ARS Testing Flow

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Characterization of Sun-sensing

Verify the system can find the sun to within 1° over the full sky.

Requirements 3.21 and 3.22

Purpose: Characterize the accuracy of the photodiodes, covers, and algorithm across the sky.

Facility: Bobby's Lab with overhead lights off

Measurements:

- Measured light source position
 - 5ft distance to source, know position to 0.5" for 0.5°
- CubeSat calculated sun vector
 - Based on photodiode measurements

Full sky characterization:

• 32 locations that use all photodiode combinations



Open Loop Tilt Testing

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Verify the tilting mechanism can tilt the instrument up to 45° in 1° increments (+/- 0.5° accuracy) - Requirements 4.12,4.121



Tilt Testing

Verify the tilting mechanism can tilt the instrument up to 45° in 1° increments (+/- 0.5° accuracy) - Requirements 4.12,4.121

Angle measurements from test will be compared to our tilting model

- If within the allowable error, requirement is satisfied
- Accelerometer has an ¹/₄ degree resolution with ¹/₄ degree accuracy



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Tilt Testing

Verify the tilting mechanism can tilt the instrument up to 45° in 1° increments (+/- 0.5° accuracy) - Requirements 4.12,4.121

Angle measurements from test will be compared to our tilting model

- If within the allowable error, requirement is satisfied
- Accelerometer has an ¹/₄ degree resolution with ¹/₄ degree accuracy
- Expected Values from Accelerometer



Integrated Tilting and Sun-sensing Verify integration of tilting mechanism, door, and sun sensing for 1°

accuracy and closed-loop tilting - Requirements 3 and 4

Purpose: Measure the tilt angle of the Cubesat as it responds to light locations

Facility: Senior Project Depot

Measurements:

• Tilting angle of the cubesat using accelerometer - compare to calculated ideal tilt based on actual light source position



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Cost Plan

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Thank you!

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Feedback?

Slide Directory

<u>Title</u> <u>Project Overview</u>	Design Recap	Instrument Test	<u>ARS Test</u>	Budget	
Statement Motivation CONOPS Levels of Success FBD ARS FBD Instrument	Model & Photo Photodiode Cover Tilting Mechanism Dust Cover Instrument <u>CPE</u> Scheduling	Test flow CSA CSA Drop test Drop test Facilities DTS stand alone MCU/trigger Electron shield Full instrument	Test Flow Sun sensing Open loop tilt Tilt sensing Tilt sensing Tilt and Sun sensing Tilt and sun sensing	Updated Cost Plan Backup slides Photodiode cover Photodiode vector Locations Accelerometer Vacuum chamber Tilt testing Door testing Photodiode boards Instrument test Teensy Impact testing Impact testing	

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Photodiode Cover Calibration

Verify the photodiodes can measure the sun angle to within 0.5° over the 60° field of view. Requirements 3.21 and 3.22

Purpose: Provide a calibration for the
photodiode output to sun angleProcedure:
Set up

Facility: Bobby's Lab

- Cubesat
- QB50 Turntable
- Light Source (bike light)
- 5V Power Supply

Measurements: Output voltage to oscilloscope



• Turn 1° increments, measure voltage





Photodiode Cover Calibration

Verify the photodiodes can measure the sun angle to within 0.5° over the 60° field of view. Requirements 3.21 and 3.22

- Current comparison of model and true voltage as measured with the oscilloscope
 - Actual measurements will be done with microcontroller
- Random noise of ~0.1 V is higher than anticipated during design
- Mitigating with an active filter to remove noise
- All points will fall inside the allowable limits



Photodiode Vector Calibration

Verify the pointing of each individual photodiode. Requirements 3.21 and 3.22

Purpose: Provide a calibration for the pointing of each photodiode on the CubeSat

Facility: Bobby's Lab

- Cubesat
- QB50 Turntable
 - 1 increments, 0.25" to within 0.005"
- Light Source (bike light)
- 5V Power Supply

Measurements: Output voltage to microcontroller, resulting sun angles



Locations





Accelerometer Testing

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Verify that accelerometer can resolve less than 0.5 degree tilt angle

- Characterization of noise levels of digital output
- Machine Shop
 - ADXL345 Triple Axis Accelerometer
 - CNC
 - Accelerometer mount
- Procedure (this can also be a diagram)
 - Calibrate accelerometer
 - Take data at level (0 deg tilt) (10 s)
 - Move CNC known amount
 - Take data at tilt (10s)
 - Compare measure to computed
 - Reneat

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Vacuum Chamber Preparations

- Cleaning
 - Alcohol/flux remover cleaning for PCBs
 - Acetone and ethanol cleaning in ultrasonic bath for machined components

- Cleaned instrument stored in ESD bag for transport
- Proper Material Selection
 - Low outgassing materials: aluminum, PEEK, Delrin
 - Vented Bolts



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- Procedure
 - Suspend pulley from roof support

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- Attach Rope to Pulley
- Attach one end of rope to cubesat fixtures
- Attach other end to counterweight
- Begin Tilting
- Measure Tilting
- Repeat last 2 steps



Door Testing

Purpose: Ensure the Door Opens Correctly Facility: Senior Project Depot Measurements:

Does the door protect this instrument

Procedure:

- Attach door to power supply and Teensy
- Activate door mechanism at level and tilting platform



Photodiodes Boards

Verify functionality of PCB and overall design

- Voltage relative to intensity of light
- Electronics lab
 - Power supply (5V)
 - Oscilloscope/multimeter

- Procedure
 - Connect PCB to power supply and measuring device

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- Turn on and check readout in ambient light
- In a dark room, position light source 5ft away and check readout at different angles

Instrument Test

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Test will must be done 90 times:

- 6 different translation table positions
- 3 different sized dust particles
- 5 recorded events for each particle size in each position

= 90 total events

These numbers are customer specified.

Teensy Shield Test (backup)

Verify functionality of PCB and filter design

- Fit check, verify all components are powered correctly and outputting information
- Electronics lab
 - Power supply $(5V)^{\dagger}$
 - Oscilloscope
 - Multimeter

- Procedure
 - Fit check all COTS boards (do not solder on yet)

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- Connect to power supply and check proper power distribution
- Connect photodiode and check output after filter
- Solder on COTS boards one at a time and check functionality of each

Impact Testing Models

- Rigid Bar Statics Model
 - Wire's fail at 4 N impact force
 - Unable to correlate impact velocity to impact force without impact time
- Solidworks Model
 - Proper Material Selection
 - 10 m/s impact results show stresses not exceeding failure stress

Impact Testing Model

Assumptions

- Entire DTS is bonded
- Landing on rigid surface
- Perfectly inelastic collision
- **Stainless steel 304 wire electrodes**
- Ultimate Tensile Strength: 505 MPa

Solder Stoppers

- Length (~ 2.2 mm) designed to shear at 520 MPa normal stress
- Wire will fracture before solder joint shears off



