

University of <u>CO</u>lorado <u>M</u>odel <u>P</u>ositioning - St<u>A</u>tic - <u>SyS</u>tem

Test Readiness Review 02 March 2016

Nicholas Gilland, Brandon Harris, Kristian Kates, Ryan Matheson, Amanda Olguin, Kyle Skjerven, Anna Waltemath, Alex Wood

University of Colorado at Boulder





Agenda

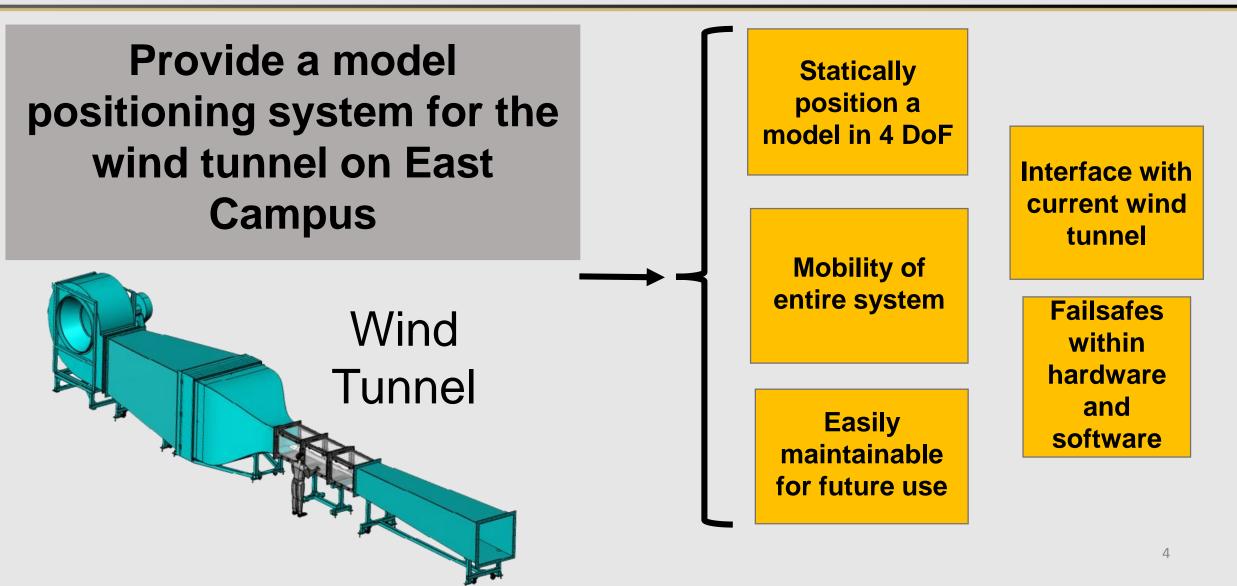
Section	Presenter
Overview	Alex
Schedule	Mandy
Test Readiness	Kristian & Nick
Budget	Mandy



Overview



Project Purpose and Objectives

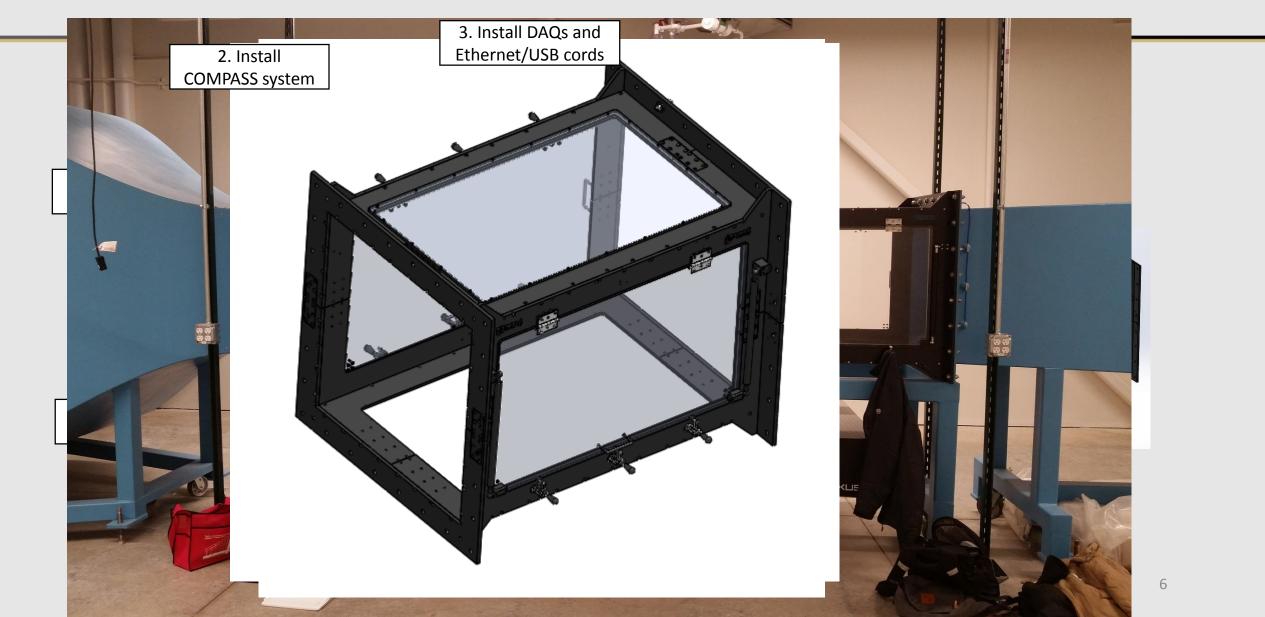


COMPASS

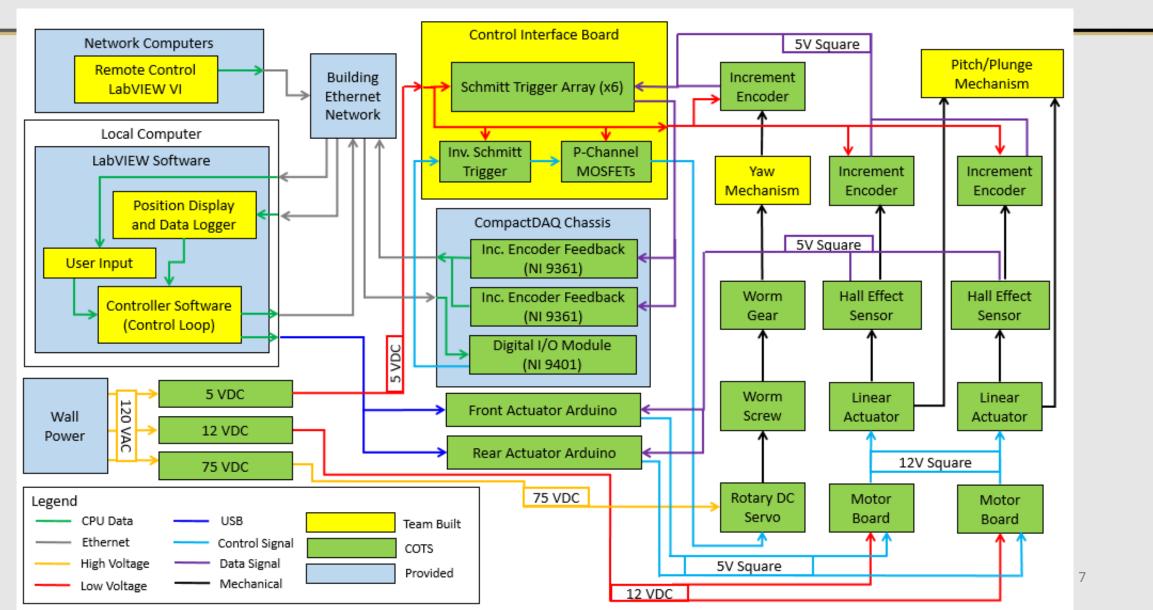
Levels of Success

	DoF	Range	Position/Ang ular Accuracy	Testing Expectations	Levels of Communication	Highest Level of Success
Level 1	Pitch Yaw	Pitch = $\pm 30^{\circ}$ Yaw = $\pm 30^{\circ}$	Pitch = $\pm 0.1^{\circ}$ Yaw = $\pm 0.1^{\circ}$	Basic verification of movement	Command through local computer	Reached by:
Level 2	"" Roll	" " Roll = ± 45°	" " Roll = ± 0.5°	VICON w/o static load	и и	
Level 3	""	" " Plunge = ± 10 cm	""" Plunge = ± 5 mm	VICON w/ static load	Remote command through local area network	← Complete by Design Symposium
Level 4	. "	и и	"	In tunnel w/ aerodynamic load	"	Complete by End of Semester

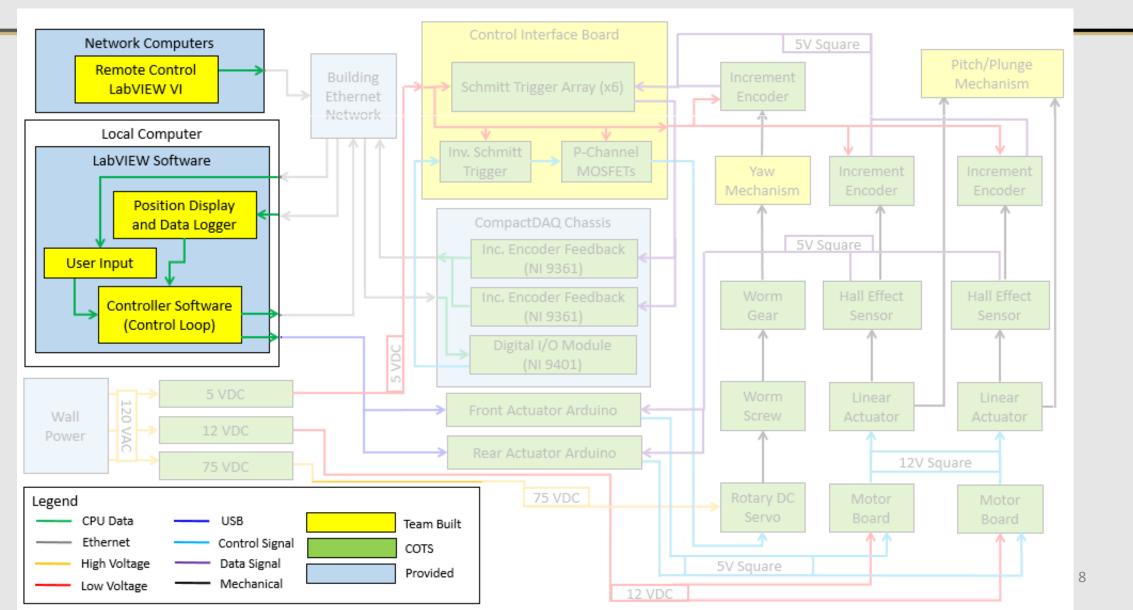
Operational CONOPS



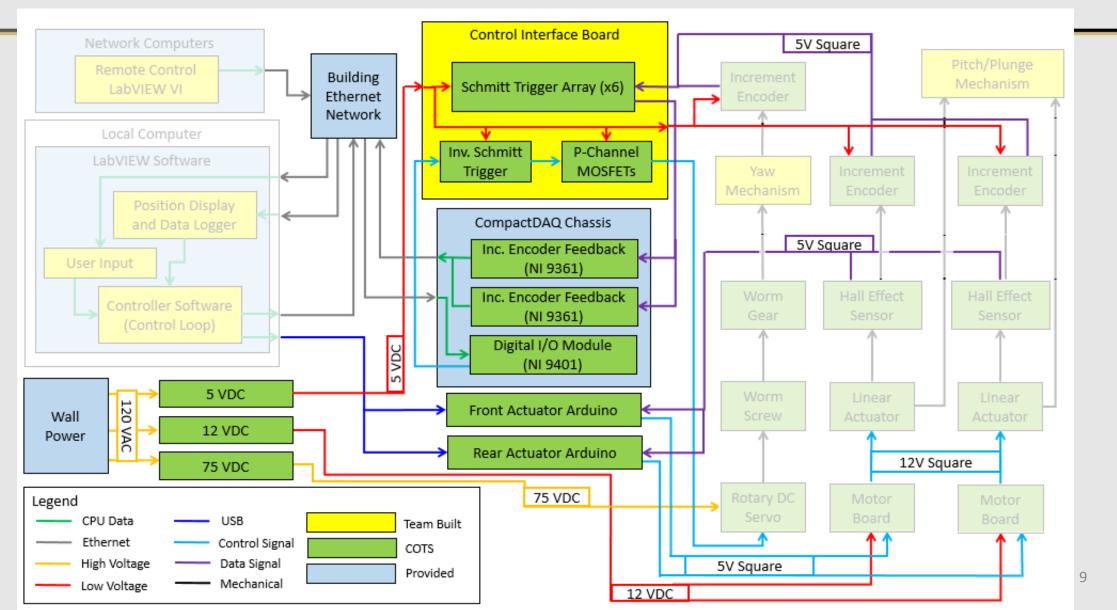
Functional Block Diagram



FBD - Computer/Software

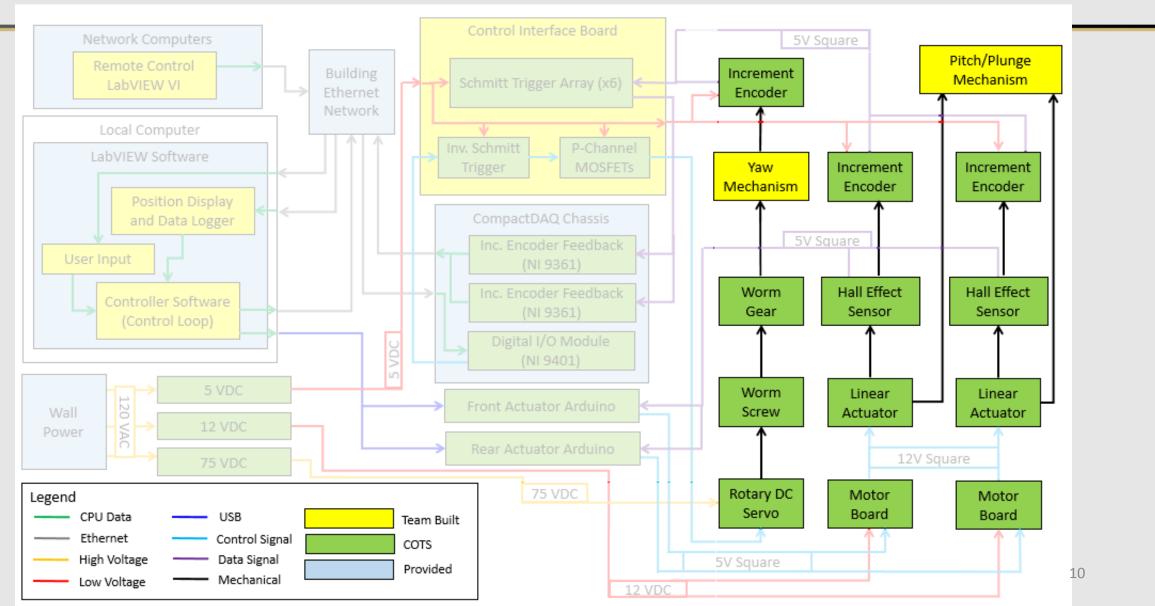


FBD - Electrical Interface

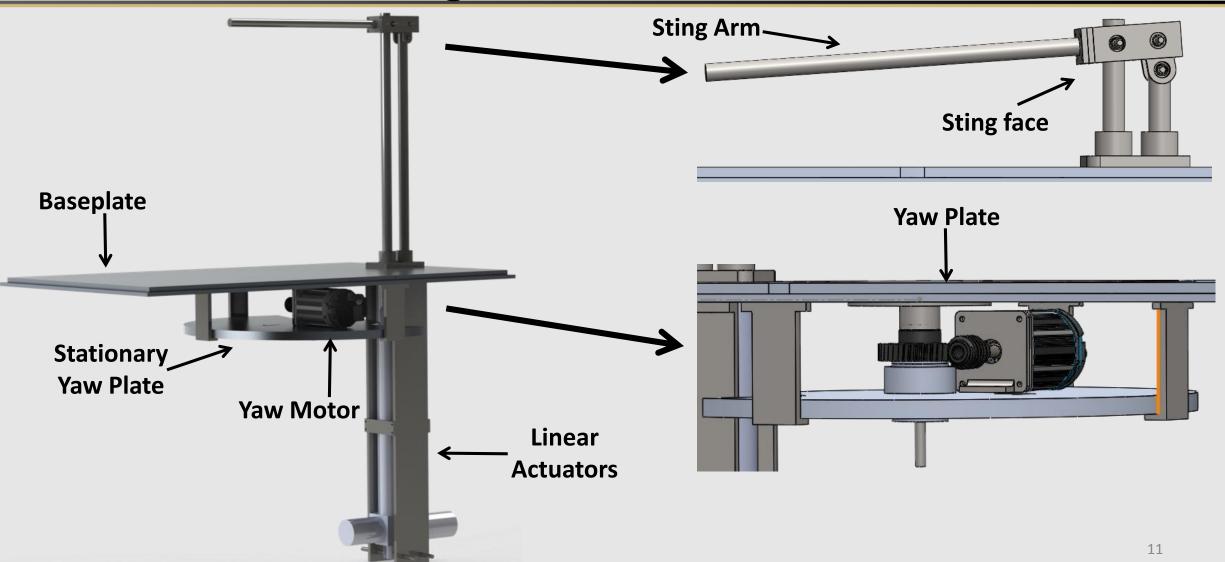


COMPASS

FBD - Positioning System

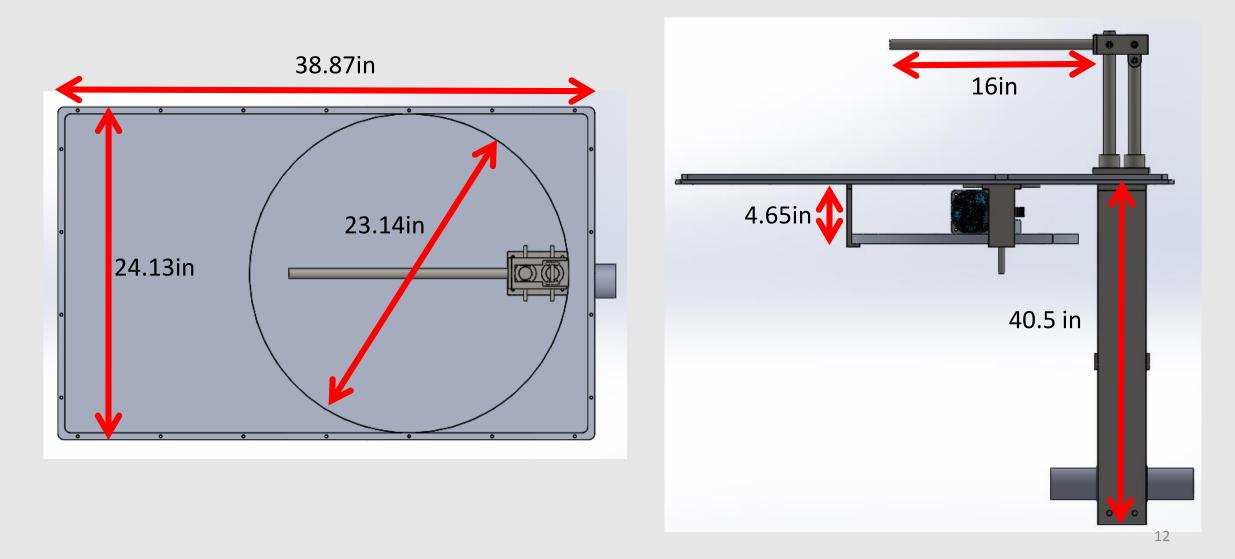


Baseline Design





Baseline Design: Major Dimensions





Critical Project Elements

CPE.1: Manufacturing base plate

- Issues with installation in the wind tunnel if manufactured incorrectly
- Lack of funds to buy material for a second plate

CPE.2: LabVIEW communication

CPE.2.1: DAQ communication for yaw system

- Signal generation not behaving as expected
- CPE.2.2: Arduino communication for linear system
 - Establishing consistent communication

CPE.3: VICON testing system

• Learning curve and usage

CPE.4: System verification within the wind tunnel while under aerodynamic load

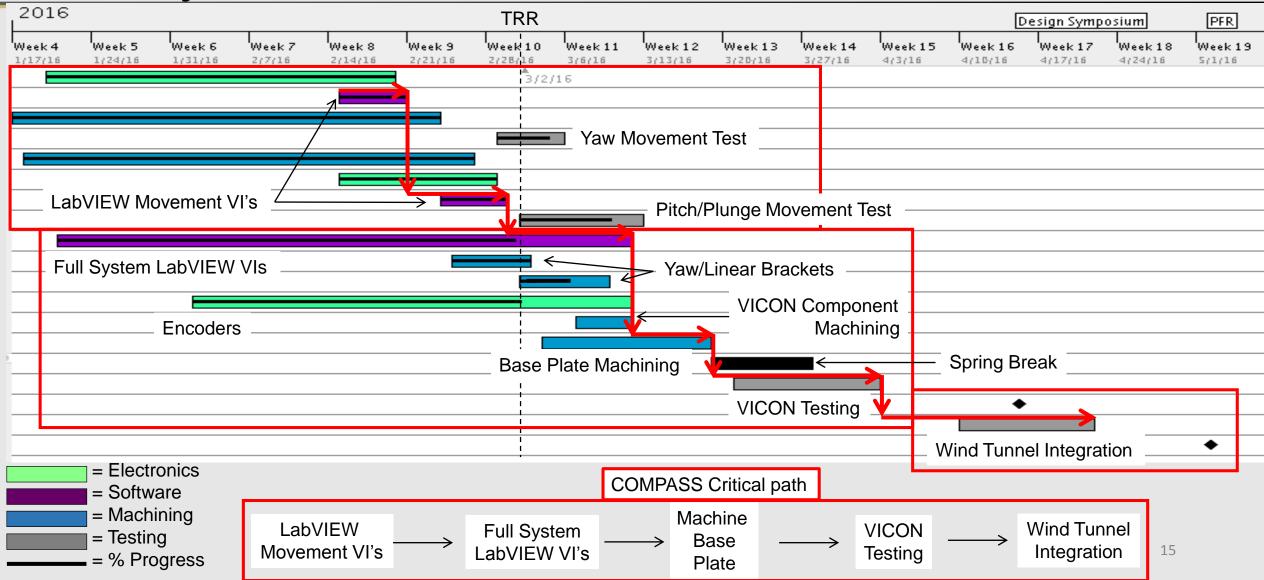
- High number of man hours to complete test
- Availability of resources



Schedule



Project Schedule

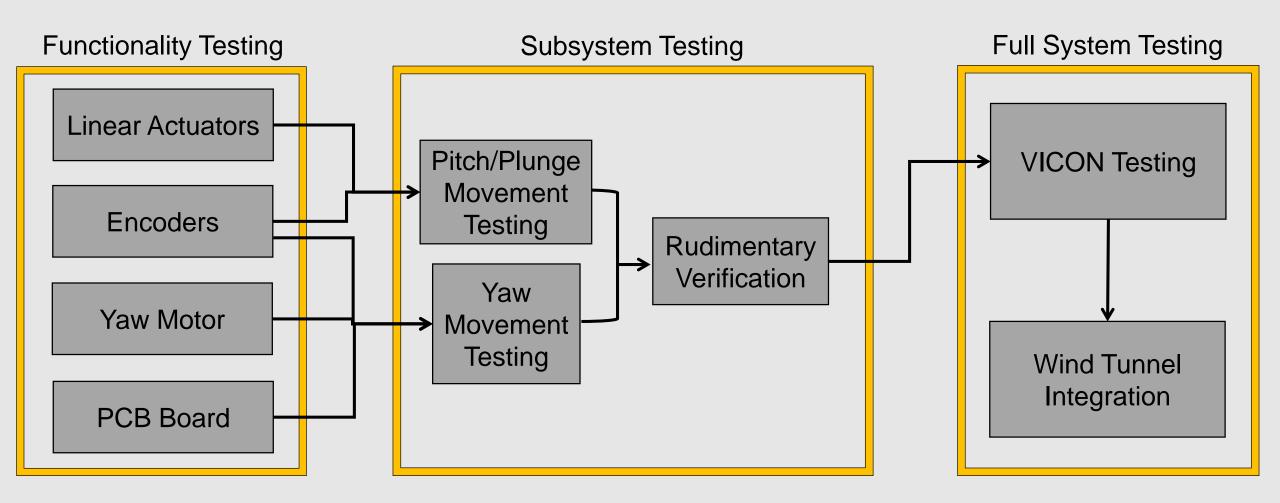




Test Readiness



Testing Scope





In Progress Testing



Component Functionality Tests

Testing For: General functionality of procured components

Test Fixtures: N/A

Test Equipment: Multimeter, Oscilloscope, Power Supplies etc.

Test Procedure: Verify that the procured component meets specs provided by manufacturer.



Component Functionality Testing

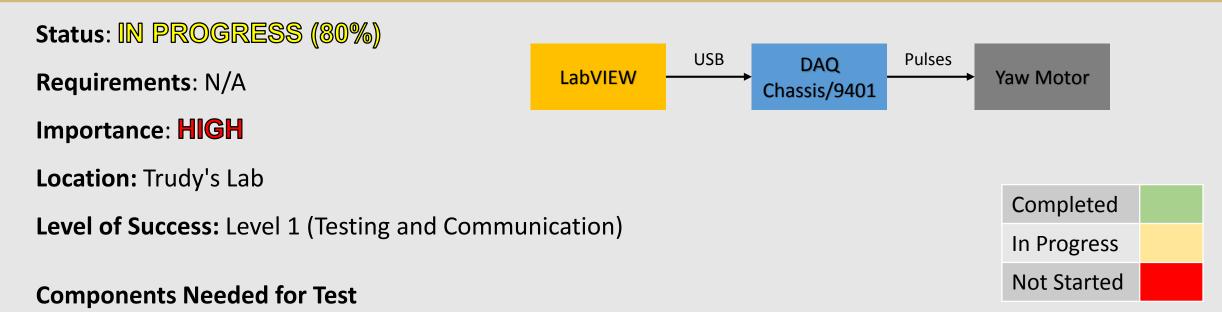
Status: 80%

Requirements: N/A (Tests do not satisfy any Design Requirements) **Importance**: LOW

Component	Test	Pass/Fail
Teknic Servo Motor	Programming and basic movement (Control from Arduino)	Pass
PCB Board Rev. B	Test of Schmitt Trigger and MOSFET circuit (Square Wave Input)	Pass
NI 9401 Module	Digital I/O output to trigger MOSFET gate	Pass
Linear Actuator	Actuation using provided Arduino and code	Pass
Incremental Encoder	Observe square wave output on lines	Pass
NI 9401 Module	Square wave pulse train output at 800 Hz	Fail
NI 9361 Module	Measure counts from incremental encoders	Fail



Yaw Subsystem Movement



Mechanical	Yaw Plate	Yaw Base Plate	Motor Bracket	Yaw Shaft Flange	Spacer	Worm Gear	Worm Screw	Bearings
Electrical	Teknic Motor	NI 9401	PCB Board	Power Supply				
Software	Yaw Control VI							



Yaw Subsystem Movement

- **Testing For:** Ability to command and control entire yaw subsystem via LabVIEW interface.
- **Test Fixtures:** Hydraulic lift cart, board supports
- Test Equipment: Yaw Subsystem, Station w/ LabVIEW, Connecting cables
- Test Procedure: Command angle via LabVIEW, monitor yaw subsystem and record movement visually
- Data Collected: Visually validate movement

Challenges:

- Cannot hit appropriate frequency of pulse train for angle change input
- Ensuring minimal slop in yaw gear



Pitch/Plunge Subsystem Movement



Components Needed for Test

Mechanical	Linear Bracket x2	Linear Collar x2	Linear Bearing Mount	Completed	
Electrical	Linear Actuator x2	Arduino x2	Power Supply	In Progress	
Software	Linear Control VI	NI Visa		Not Started	

Actuator

Actuator

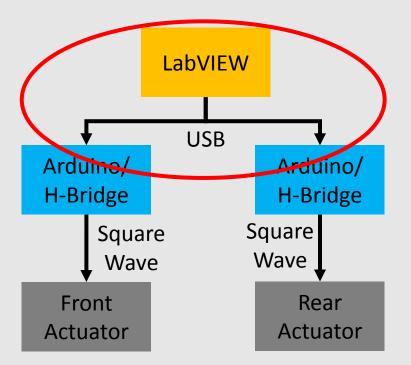
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Pitch/Plunge Subsystem Movement

- Testing For: Ability to command and control entire pitch/plunge subsystem via LabVIEW interface.
- Test Fixtures: Clamps, sawhorses
- Test Equipment: Pitch/Plunge Subsystem, Station w/ LabVIEW, Connecting cables.
- Test Procedure: Command pitch/plunge via LabVIEW, monitor and record motion of system.
- Data Collected: Visually validate movement

Challenges:

 Consistent and reliable communication between linear actuator Arduinos and LabVIEW





Future Tests



Rudimentary Verification

Status: NOT STARTED

Importance: MEDIUM

Location: Trudy's Lab

Level of Success: Level 1

Design Requirements Verified

Range: \pm 30° yaw, \pm 30° pitch, \pm 45° roll, \pm 10 cm plunge

Accuracy: $\pm 0.1^{\circ}$ yaw, $\pm 0.1^{\circ}$ pitch, $\pm 0.5^{\circ}$ roll, ± 0.5 mm plunge

Completed	
In Progress	
Not Started	

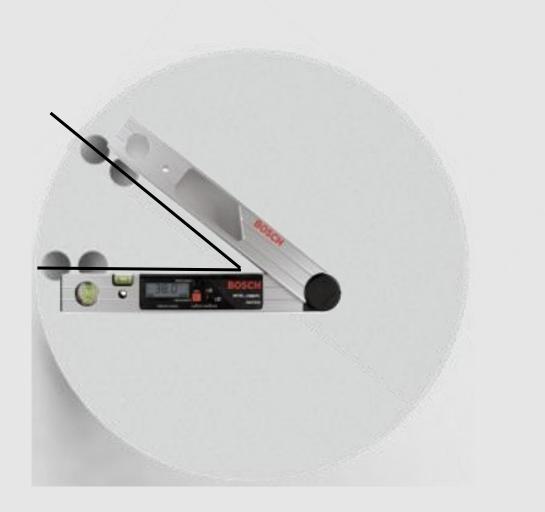
Components Needed for Test

Mechanical	Base Plate			
Electrical	Yaw Subsystem	Linear Subsystem	DAQ Chassis	Encoder x3
Software	Yaw Control VI	Linear Control VI		



Rudimentary Verification

- **Testing For:** Rudimentary verification of accuracy and range in pitch, roll, yaw and plunge in addition to code debugging.
- Test Fixtures: Sawhorses to support entire COMPASS system
- Test Equipment: Pitch/Plunge Subsystem, Yaw Subsystem, station w/ LabVIEW, digital protractor, digital angle finder
- **Test Procedure**: Command pitch, yaw, plunge to COMPASS, wait for system to finish movement and then record new position with test equipment.
- Data Collected: Angle measurements with rudimentary tools are accurate to within 0.1°



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VICON

Status: NOT STARTED

Importance: **HIGH**

Location: Idea Forge

Level of Success: Level 2 (Without Load)

Level 3 (With Load)

Design Requirements Verified

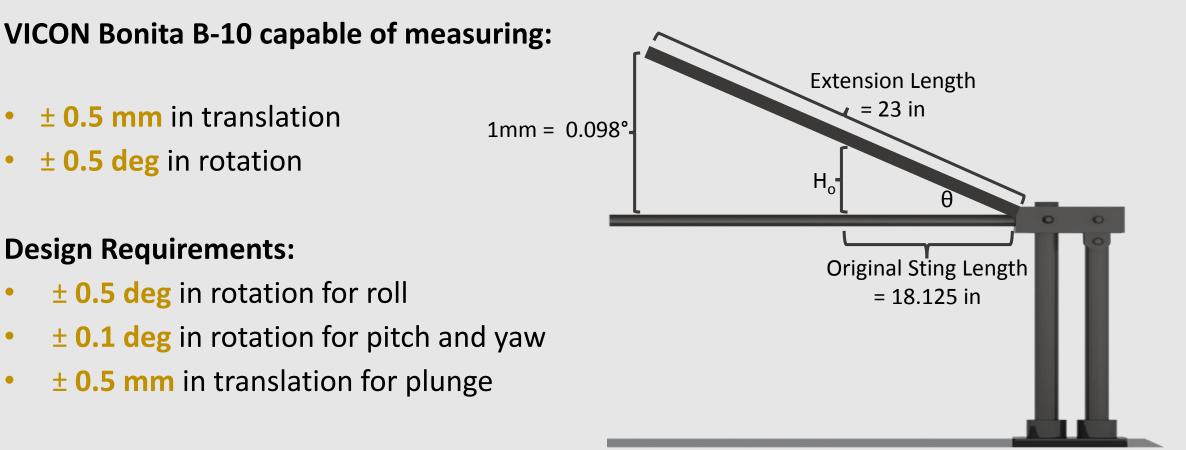
Range: \pm 30° yaw, \pm 30° pitch, \pm 45° roll, \pm 10 cm plunge

Accuracy: $\pm 0.1^{\circ}$ yaw, $\pm 0.1^{\circ}$ pitch, $\pm 0.5^{\circ}$ roll, ± 0.5 mm plunge

Components Needed for Test:

Mechanical	Extension Arm	Universal Mounting Hub
Software	Initialization VI's	Functionality VI's

VICON



To verify COMPASS entirely using VICON, an extension arm must be implemented to reach needed displacements for pitch and yaw. Angle of movement can be solved for using displacement of nodes and the law of cosines.



VICON Without Load

- **Testing For:** Precision verification of accuracy and Range in pitch, roll, yaw and plunge
- Test Fixtures: Extension Arm, Sawhorses
- Test Equipment: Full COMPASS system, VICON system, pearl markers, station w/ LabVIEW, station w/ VICON
- Test Procedure: Record initial location of pearl markers from VICON. Command new position to COMPASS and record new coordinates of markers. Repeat as needed.
- Data Collected: x, y, z location of pearl markers. Using location and extension arm length, can back out angle traveled through.





VICON With Load

- **Connect** universal mounting hub and pearl markers to sting shaft. Record x, y, z location of pearl markers from VICON
- Connect simulated load of airfoil with maximum lift to mounting hub: 16 kg
- Command COMPASS to maximum deflection of 30° in pitch
- Collect x, y, z location data of pearl markers from VICON
- Analyze x, y, z location data to calculate and confirm pitch angle (law of cosines)







Wind Tunnel: Aerodynamic Load

Status: NOT COMPLETE

Importance: **HIGH**

Location: East Campus Wind Tunnel

Level of Success: Level 4

Design Requirements Verified

Range: \pm 30° yaw, \pm 30° pitch, \pm 45° roll, \pm 10 cm plunge

Accuracy: $\pm 0.1^{\circ}$ yaw, $\pm 0.1^{\circ}$ pitch, $\pm 0.5^{\circ}$ roll, ± 0.5 mm plunge

Highest Level of Success

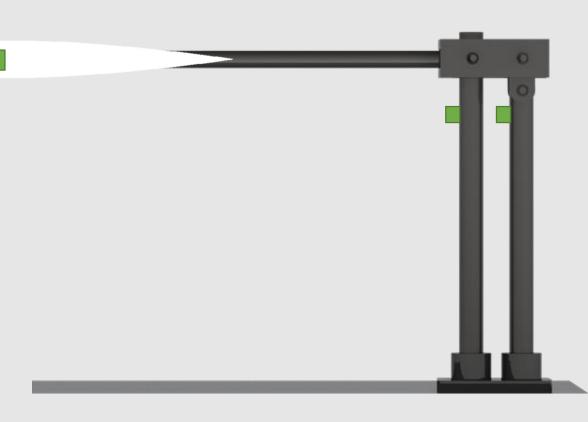
Components Needed For Test

Electronics	ADXL 326 Accel.	EVAL-ADXL326Z	NI 9209	NI 9234	PCB 325C65 x2
Software	Accelerometer VI				



Wind Tunnel: Aerodynamic Load

- Testing For: Accuracy and Range in pitch under aerodynamic load as well as vibration characterization of linear actuators
- Test Fixtures: Wind Tunnel
- Test Equipment: Full COMPASS system, station w/ LabVIEW, accelerometer, test airfoil
- Test Procedure: Place full COMPASS system under aerodynamic load to measure deflection and characterize vibrations.
- Data Collected: Acceleration in 2 axis to calculate pitch angle and vibration measurements of linear actuators





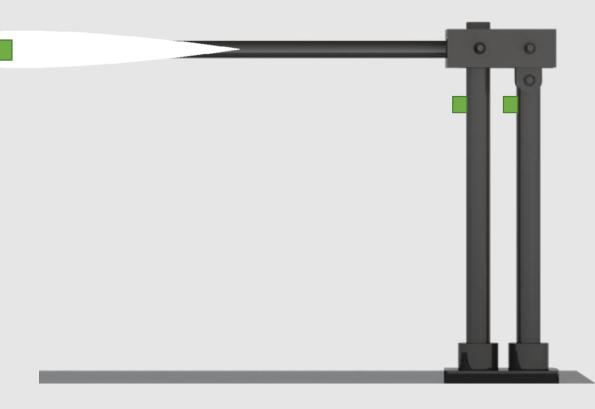
Wind Tunnel Testing

Pitch Verification

- Mount ADXL 326 accelerometer inside hollow airfoil: Sensitivity of 57 mv/g
- A 0.1° inclination corresponds to 1.7 mv
- Read signal using 24 bit NI 9209 DAQ module. Resolution of 1.2 mv/bin
- 1.2 mv/bin < 1.7 mv: SUFFICIENT FOR VERIFICATION
- Challenges: Shielding for noise

Vibration Characterization

- Mount PCB 325C65 accelerometers to linear actuators to characterize vibrations: sensitivity of 100 mv/g
- Read signal using 24 bit NI 9234
 DAQ module

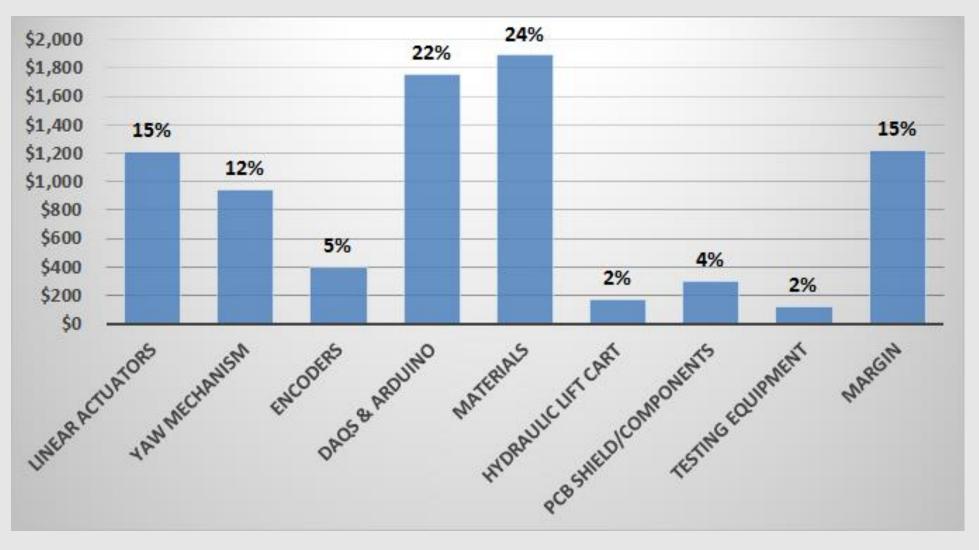




Budget

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Budget



Course Budget: \$5000 EEF Addition: \$3000 Total Budget: \$8000 Margin: \$1212, 15%

Parts Price Change From MSR

Parts	Expected Price at MSR	Actual Price at TRR	
Baseplate/Yaw plate Aluminum	\$1196	\$1196	
Encoders	\$1644	\$395	
NI (2 9361/9401) + Arduino	\$1778	\$1756	
Yaw Motor	\$781	\$781	
Linear Actuators	\$1205	\$1205	
Hydraulic Lift Cart	\$172	\$172	
Bearings	\$388	\$388	
Worm Gear/Screw	\$166	\$166	
Sting Rods	\$25	\$30	
Materials + Screws/Nuts	\$90	\$82	
PCB Shield/Components	\$119	\$297	
Testing Equipment	Not Assessed	\$120	
Printing	\$200	\$200	
Margin	\$236, 3%	\$1212, 15%	37

COMPASS



Additional Funding Allocation

• 12% Margin Allocation

- Additional Cabling
- Linear Actuator Rod Fairing
- Linear Actuator Bracket Aluminum
- PCB Components



Questions?



Back-up Slides

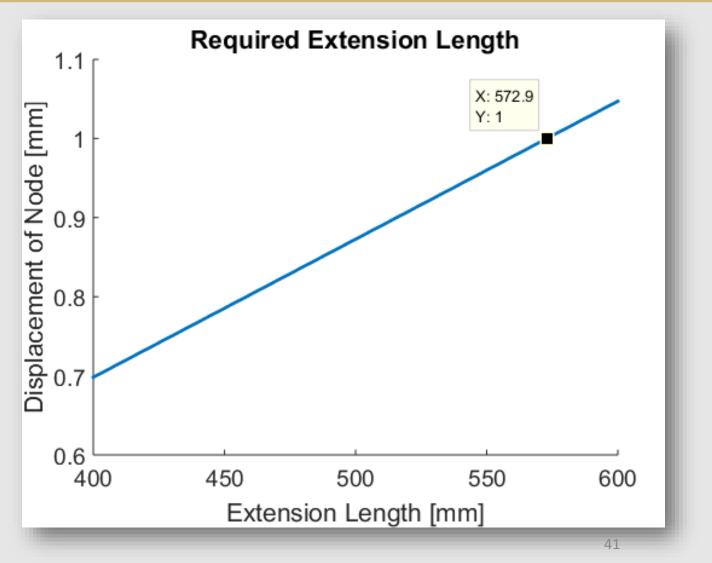
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Extension Calculation

Result:

1 mm displacement = **573 mm** arm

573 mm ≈ **23 inch** arm





Calculations for Accelerometer

 $\Delta A_{OUT}[g] \cong 1\,g \times \sin(P)$

For 0.1 degrees, A = 1.7 mv

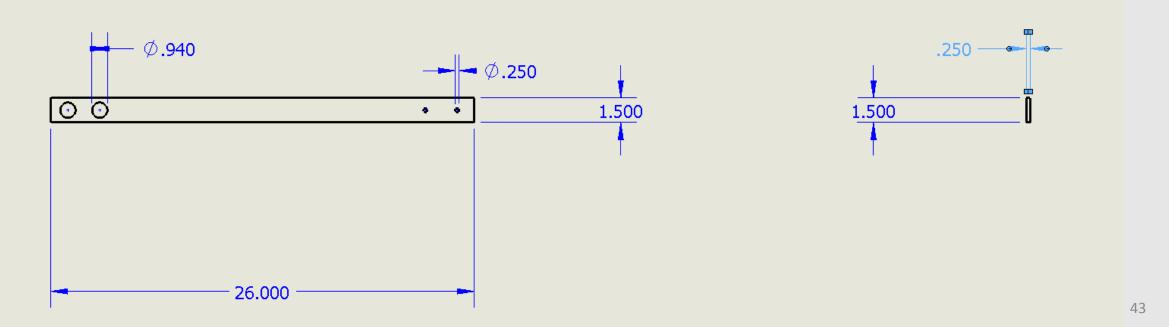
$$\frac{A_{X,OUT}}{A_{Y,OUT}} = \frac{1 g \times \sin(\theta)}{1 g \times \cos(\theta)} = \tan(\theta)$$
$$\theta = \tan^{-1} \left(\frac{A_{X,OUT}}{A_{Y,OUT}}\right)$$

Use to calculate angles from measurements



Extension Arm

- Length: 26 inches
- Width: 1.5 inches
- Large Hole D: 0.94 inches
- Small Hole D: 0.25 inches

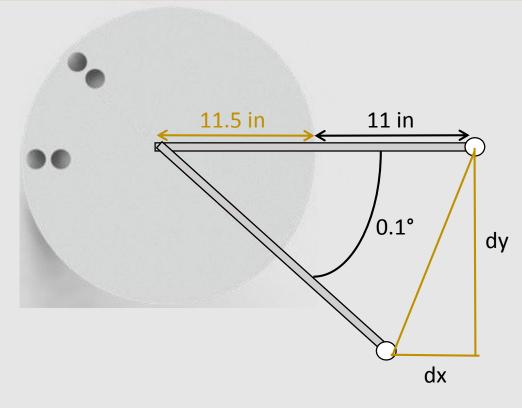


Yaw Verification Test

Main Objective: Validate range and accuracy requirements of yaw mechanism.

Requirements: ± 30° range, ± 0.1° accuracy.

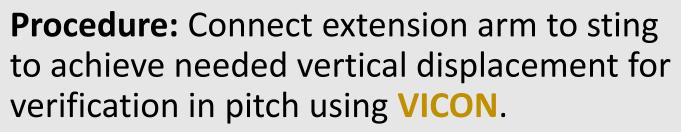
Procedure: Connect extension arm to yaw plate to achieve needed horizontal displacement for verification in yaw using VICON.



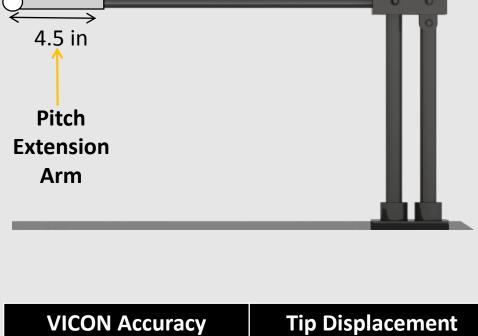
VICON Accuracy	Tip Displacement
0.5 mm	1 mm

Pitch Verification Test

```
Main Objective: Validate range and
accuracy requirements of pitch mechanism.
Requirements: ± 30° range, ± 0.1° accuracy.
Location: Idea Forge
```



Provided Displacement: 1 mm per 0.1°



0.5 mm

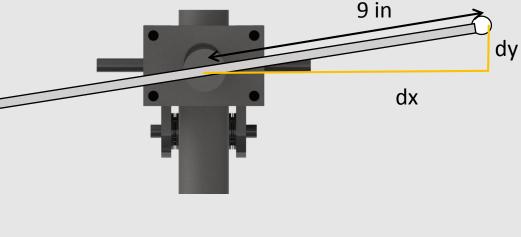
1 mm

Roll Verification Test

Main Objective: Validate range and accuracy requirements of roll mechanism. Requirements: ± 45° range, ± 0.5° accuracy. Location: Idea Forge

Procedure: Connect extension arm to sting to achieve needed vertical displacement for verification in roll using VICON. (or use VICON's 0.5° rotation accuracy)

Provided Displacement: 1 mm per 0.5°



VICON Accuracy	Tip Displacement
0.5 mm	1 mm

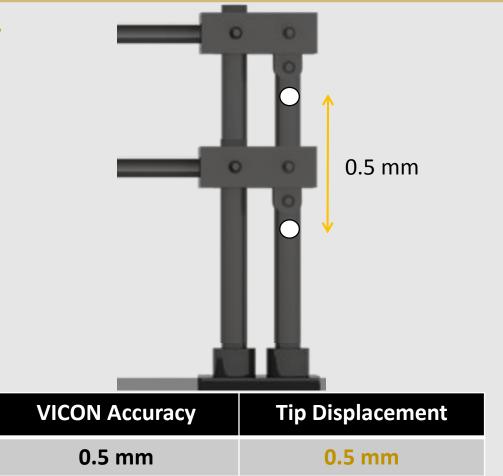
Plunge Verification Test

Main Objective: Validate range and accuracy requirements of plunge mechanism.

Requirements: ± 10 cm range, ± 0.5 mm accuracy.

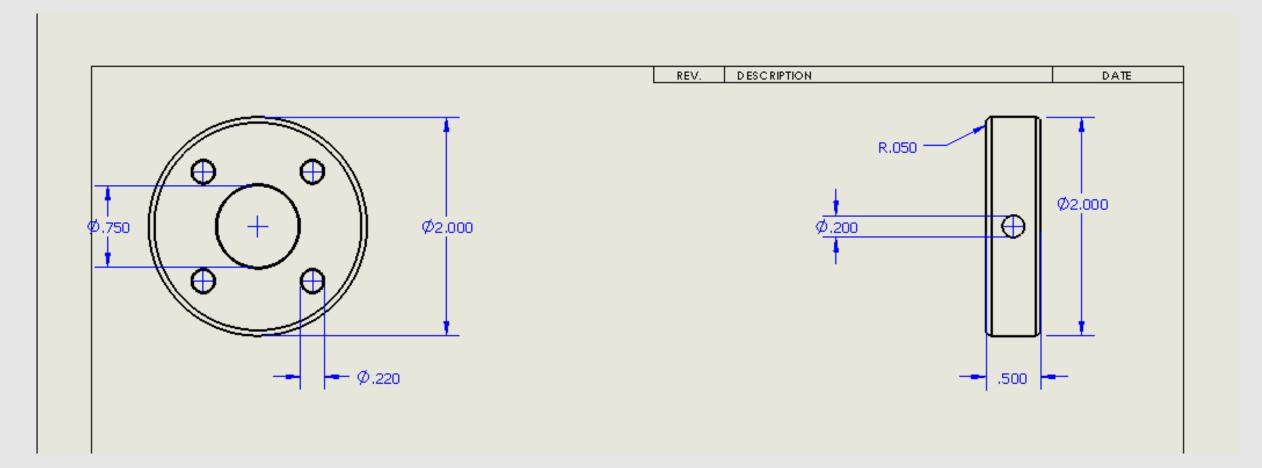
Location: Idea Forge

Procedure: Connect nodes to COMPASS, command plunge to and verify plunge requirements using VICON.





Universal Mounting Hub





Accuracy Characterization

- Yaw can be characterized directly by incremental encoder
 - Assembly has revealed adjustments need to be made for slop
- Pitch and Plunge elements require mechanical modifications
 - Assembly of future system will allow characterization of accuracies
- Challenges:
 - Linear actuator hall effect sensor setup not expected
 - Linear actuators not to requested specifications
 - Yaw accuracy directly characterized but slop introduces control issues
 - Slop in any part of system will introduce control issues
- Goal: Characterize controllable accuracy of the system



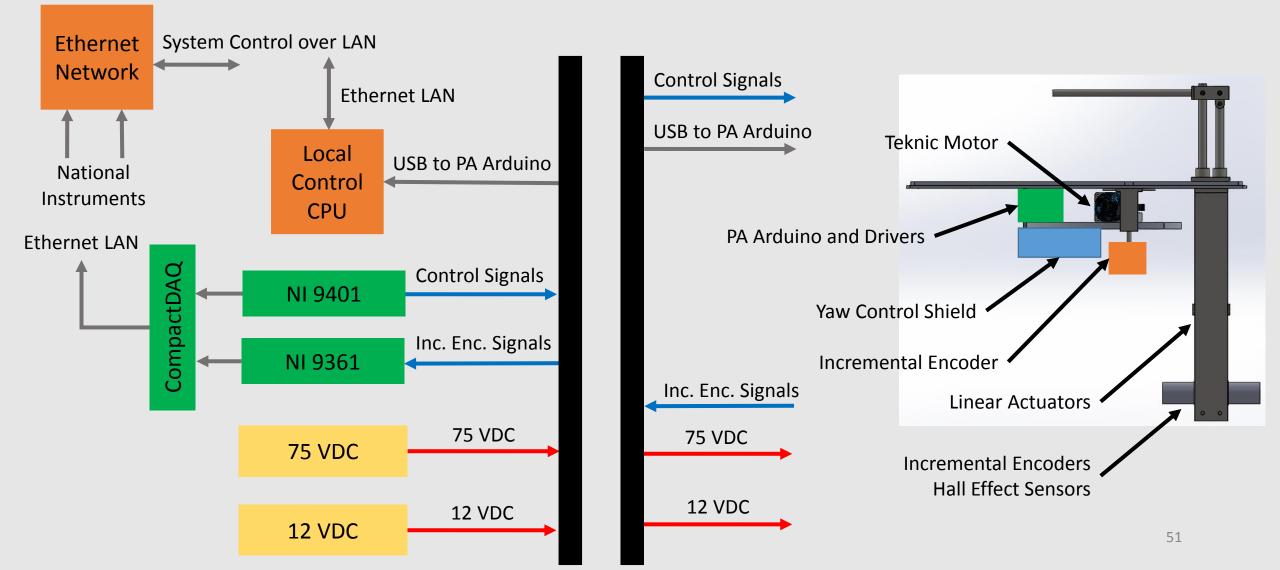
Accuracy Characterization

- Steps for characterizing system performance:
 - 1. Record error of incremental encoders
 - 2. Measure slop of assembled yaw and pitch/plunge subsystems
 - 3. Characterize total error through the system
 - 4. Characterize controllable error

Component (Yaw DOF)	Error (degrees)	Component (P/P DOF)	Error (mm/degrees)
Absolute Encoder	0.1°	Linear Actuator	0.0406 mm 0.045°
Radial Bearing Slop	TBD with Testing	Incremental Encoder	0.00857 mm 0.0095°
Total	0.1°	Sting Assembly Slop	TBD with Testing
		Total	0.0406 mm 0.045°

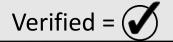


Overall System Setup



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Software Scope and Status



Initialization and Safety

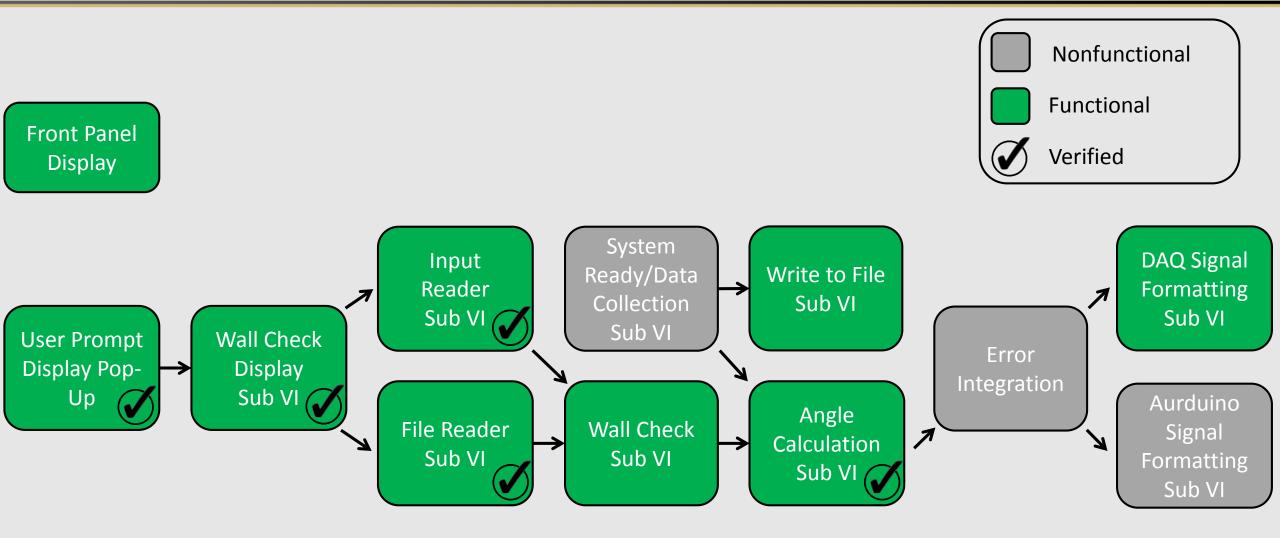
Functionality	COMPASS VI	Status
Allow for a variable sting length	User Prompt	
Read a User file	File Reader VI	
Allow for manual inputs	Input Reader VI	
Prevent damage to the system, the wind tunnel and fillets	Wall Check VI Wall Display VI	

Positioning

Function	COMPASS VI	Status
Format signals to the motor	Format to DAQ VI Format to Aurduino VI	
Read position from encoders	Collect Time + Position VI	
Calculate where to position the system based on attitude	Angle Calculation VI	
Write position to data file and allow user to save file	Write to File VI	

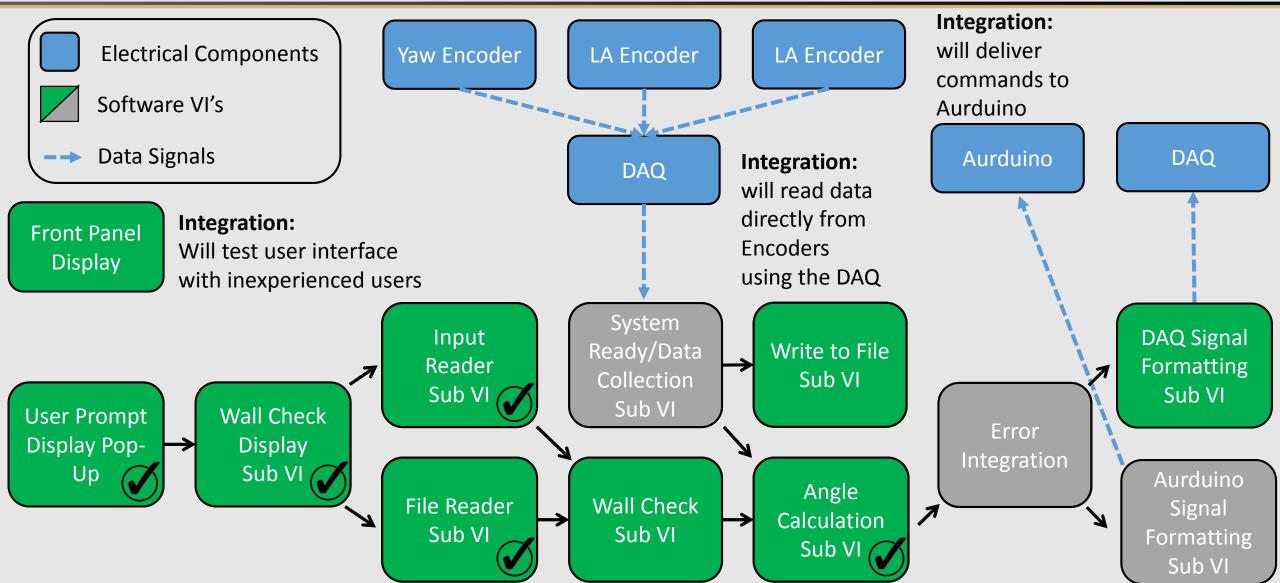


Software Completion and Flow



COMPASS

Software Integration Plan





Electronics Scope and Status

Verified =

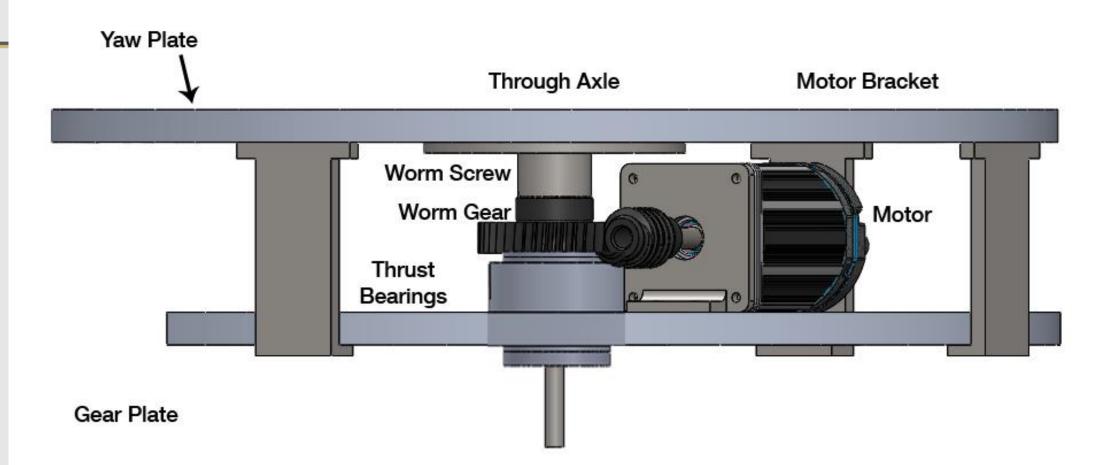
Pitch and Plunge

Functionality	Component(s)	Status
Powering the system	Transformers	
Position the system	Actuators/Hall Effect	
Feedback to LabVIEW	NI9361/Encoders	
Commanding a position	Arduinos/Motor Boards	

Positioning

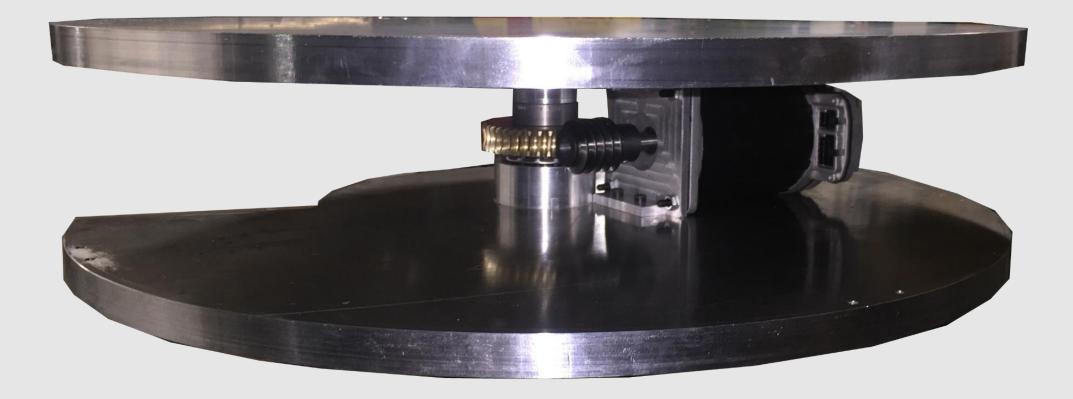
Functionality	Component(s)	Status
Powering the system	Transformer	
Position the system	Motor/Internal Encoder	
Feedback to LabVIEW	NI9361/Encoder	
Commanding a position	NI9401/PCB	

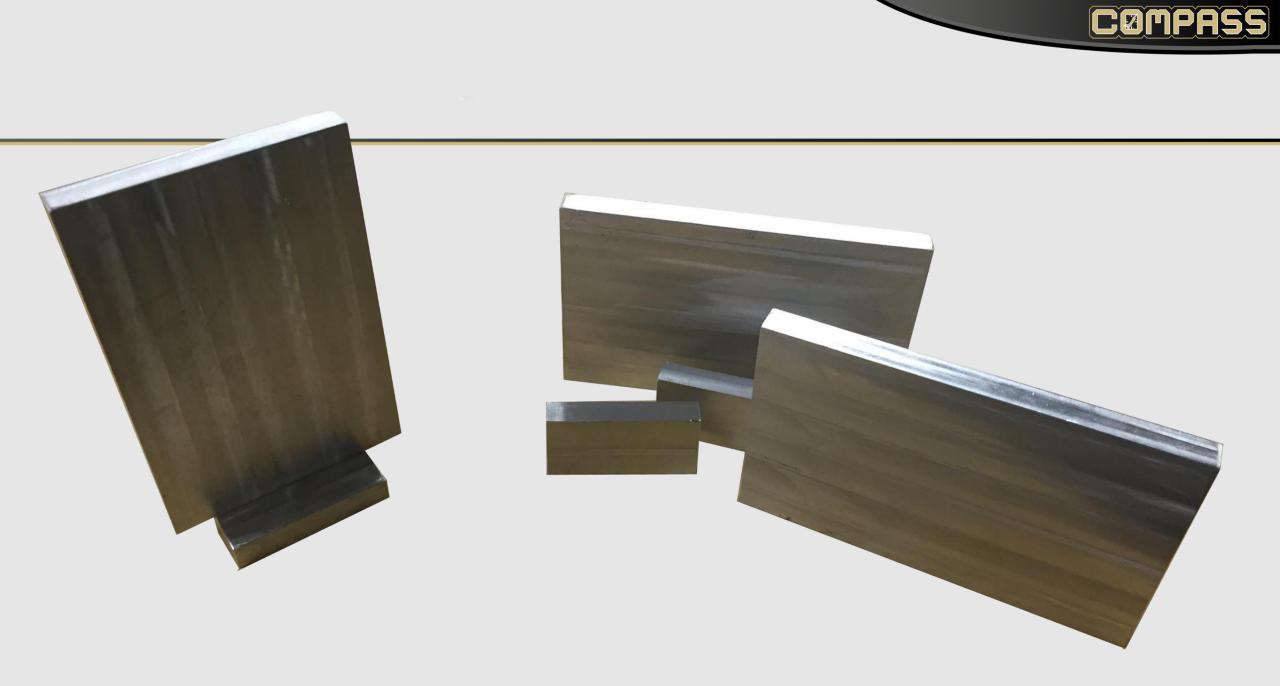


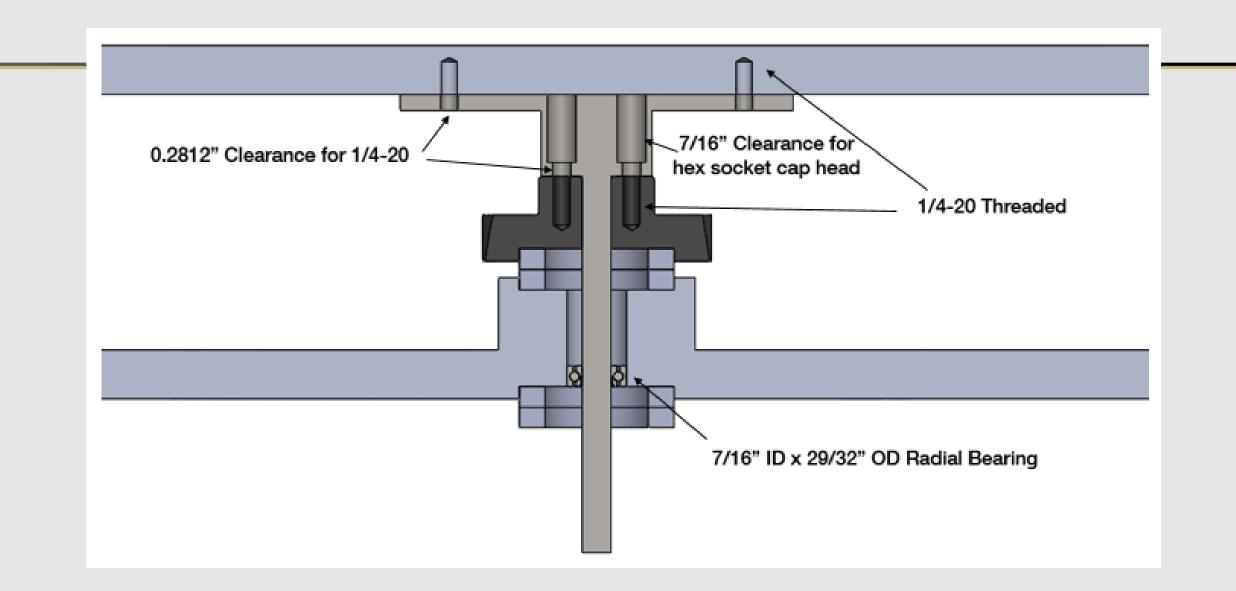


Gear Plate Brackets

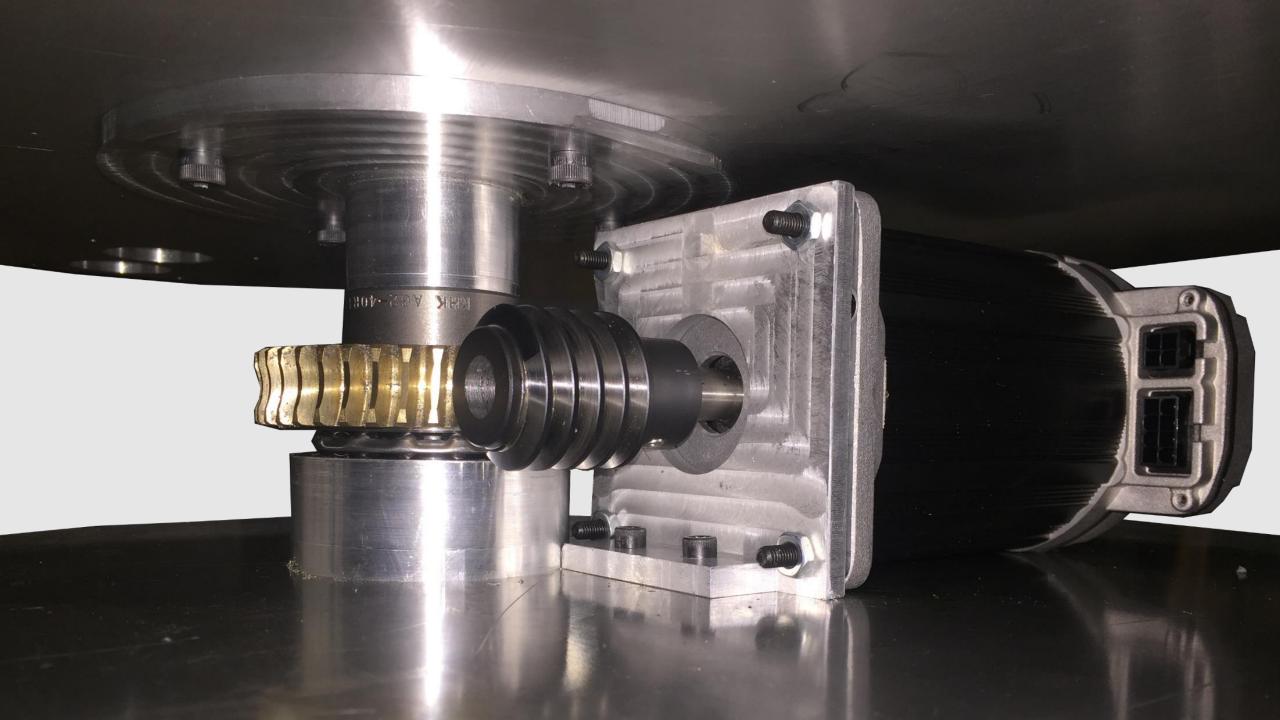






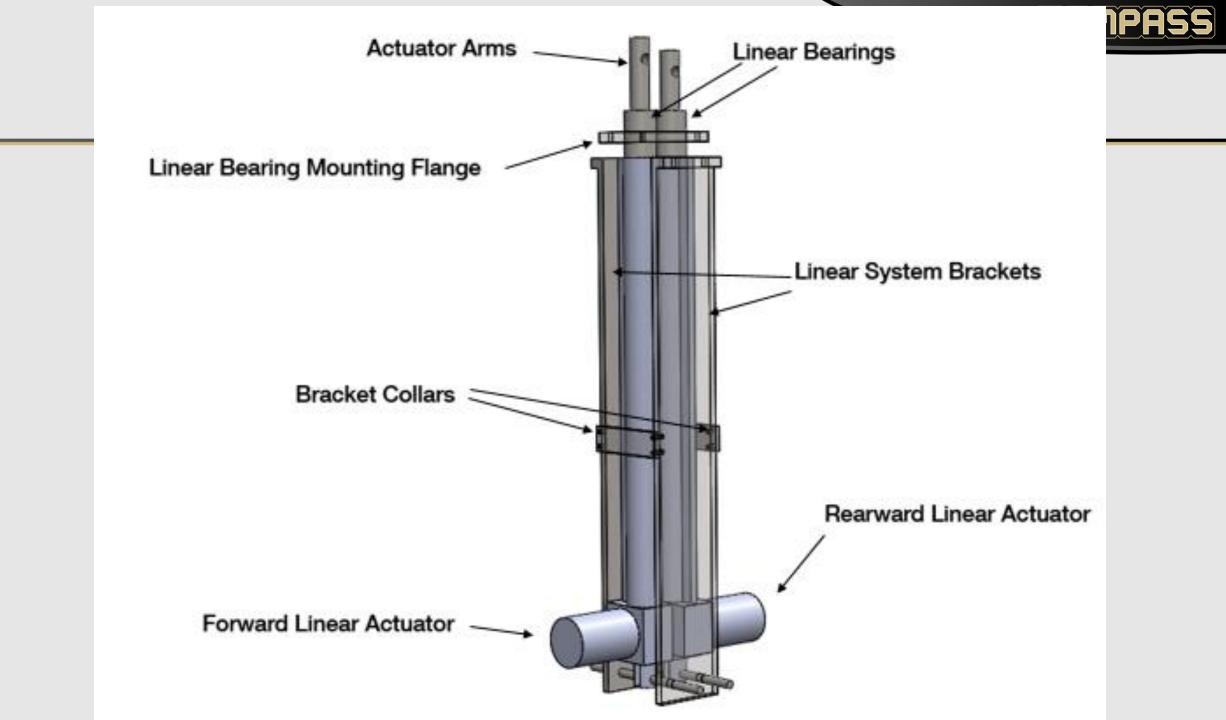


COMPASS



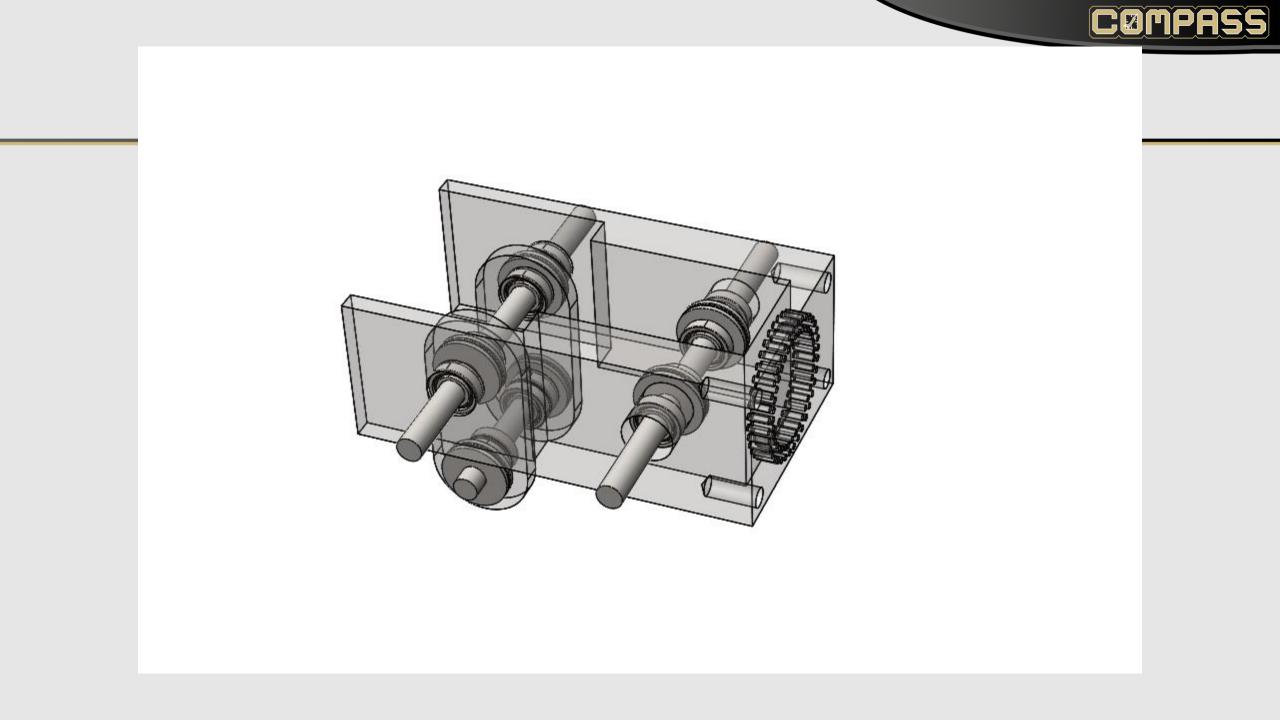














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Subsystem Failsafes

- Yaw DOF Subsystem
 - Software: Angle Check VI and User Input VI
 - Hardware: Motor Torque Limit with mechanical hard stop
- Pitch and Plunge DOF Subsystem
 - Software: Angle Check VI, User Input VI, no max length command input
 - Hardware: N/A
- Electrical: Power Cutoff Switch