

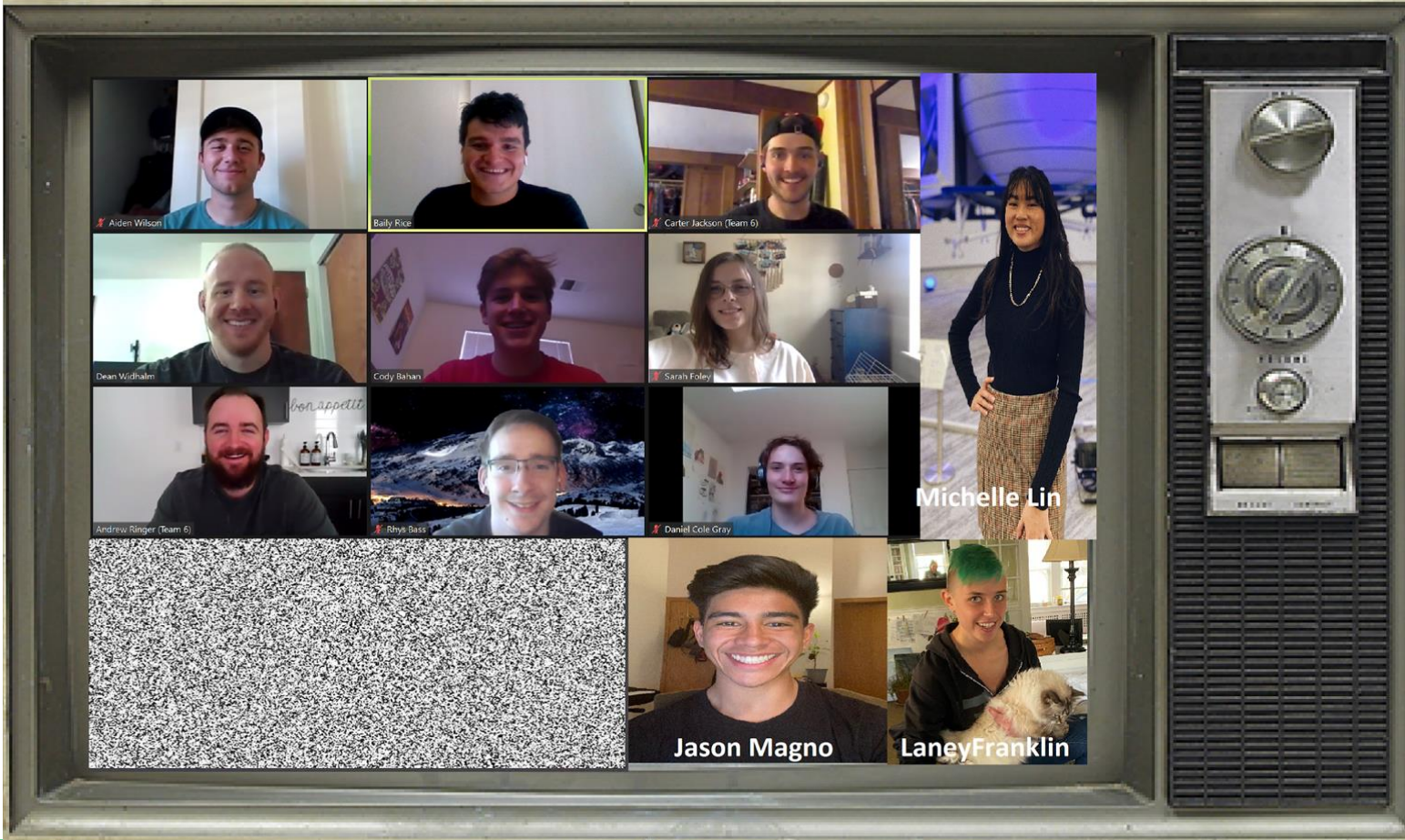
# CHAIR



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Cueing via Haptics And Inner-ear Responses

# CHAIR Team Members



# Project Motivation

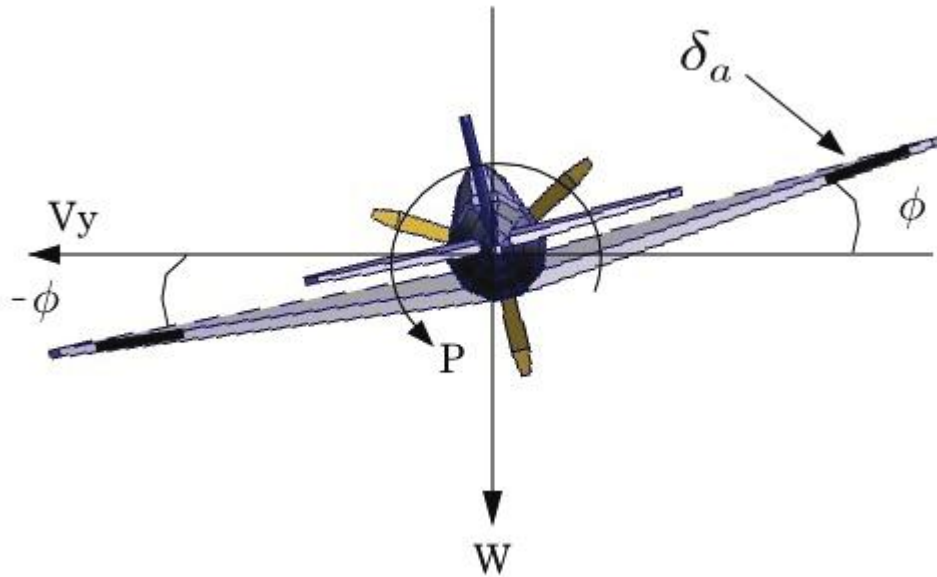
- Create a cueing device capable of providing additional sensory information to remote pilots
- Provide our customer a research tool to validate this proof of concept before application to industry



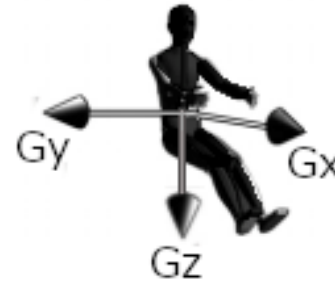
NBC News: <https://www.nbcnews.com/technolog/virtual-cockpit-what-it-takes-fly-drone-1C9319684>

# General Overview

- The CHAIR project is focused on cueing a pure roll response



- Take advantage of two separate sensory pathways
- Accurate cueing of pure roll still provides useful information



# Terms and Definitions

Test Subject: The person who is receiving cues from the CHAIR system

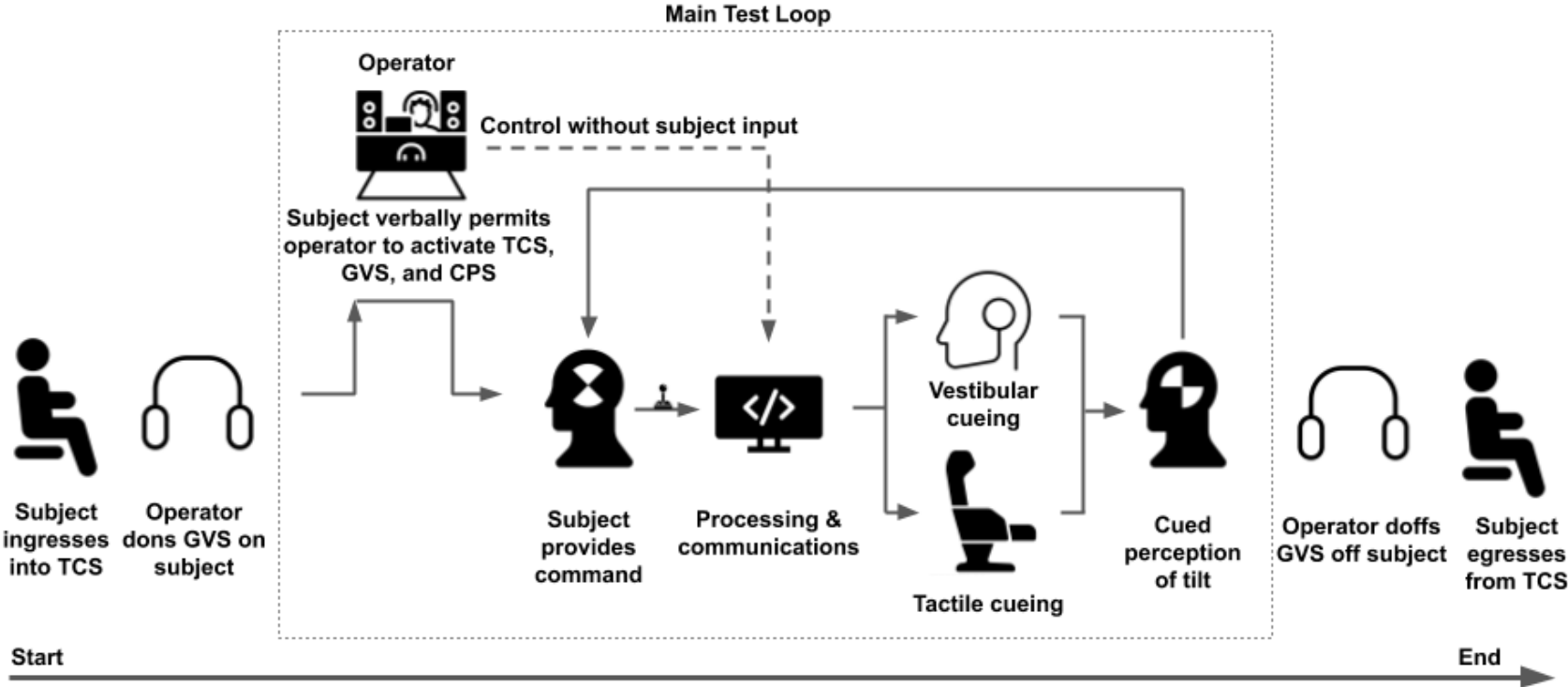
Test Operator: The person who is controlling input cues to the test subject

TCS: Tactile cueing system, applies pressure stimulation to subject

GVS: Galvanic vestibular stimulator, electrically stimulates the inner ear to elicit a feeling of tilt or motion in the subject

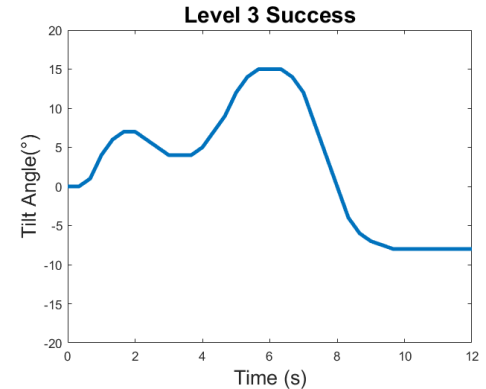
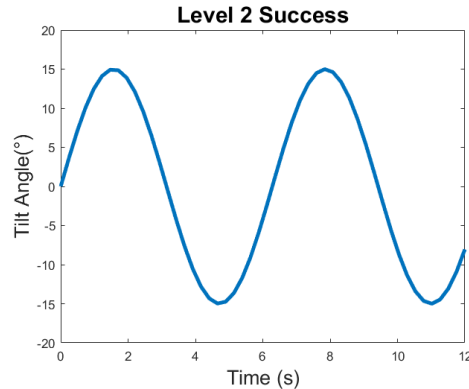
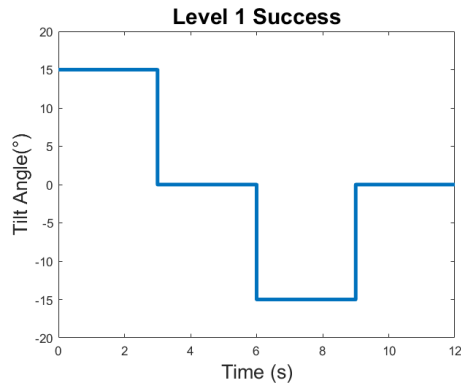
CPS: Central processing system, performs relevant computations, receives inputs from and sends outputs to subsystems

# ConOps

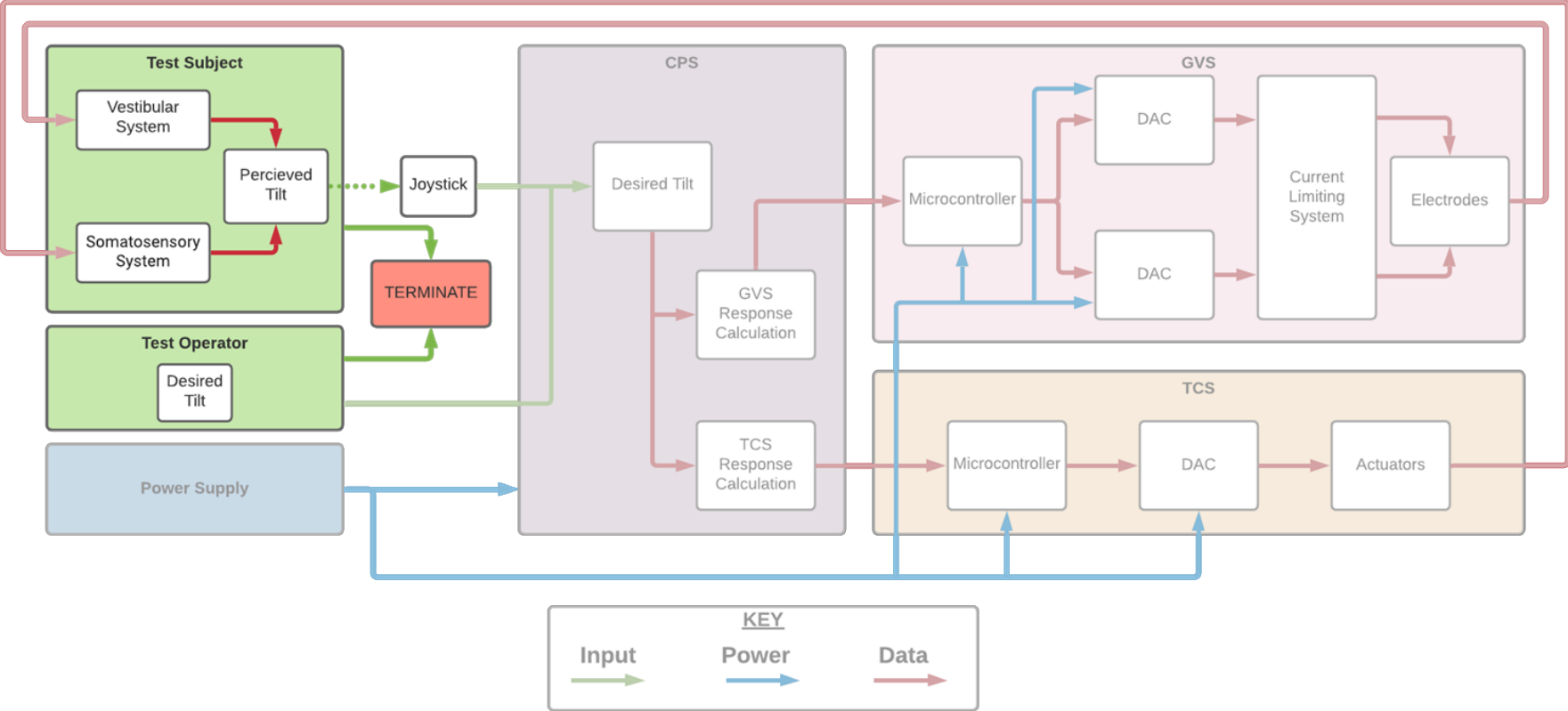


# Objectives

- Level 1
  - Cue static tilt angles of up to  $15^\circ$  in either direction
  - Implementation of joystick control from first level
- Level 2
  - Cue sinusoidal tilt profiles with maximum amplitude of  $15^\circ$
- Level 3
  - Cue any tilt profile as commanded by the test subject up to  $15^\circ$



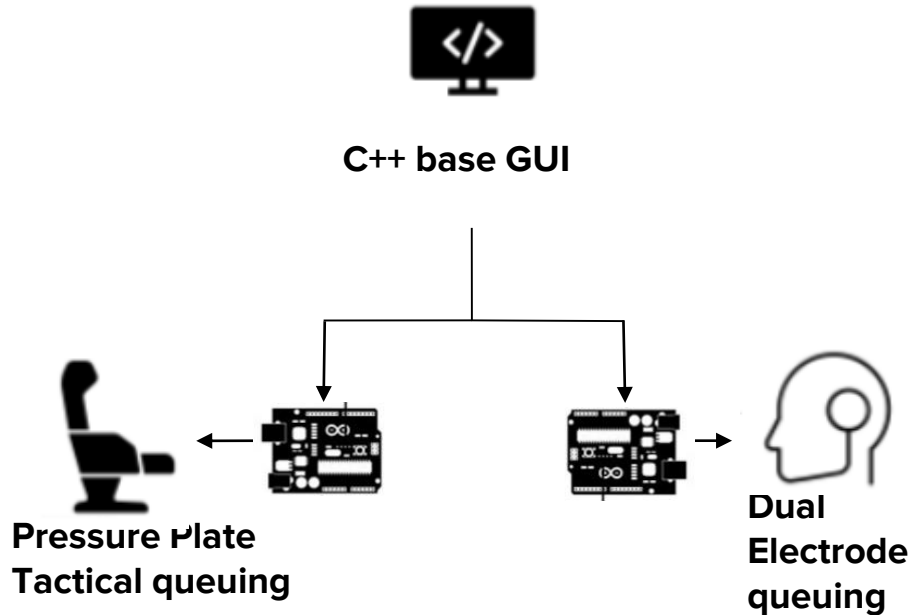
# Functional Block Diagram





# Baseline Design

- **TCS:** Physical pressure cueing performed by side and bottom actuators (8 total)
- **GVS:** Custom dual electrode system
- **CPS:** One processing unit (running code written in C++) that commands two microcontrollers
- **Microcontrollers:** Two microcontrollers, one for the TCS and one for the GVS

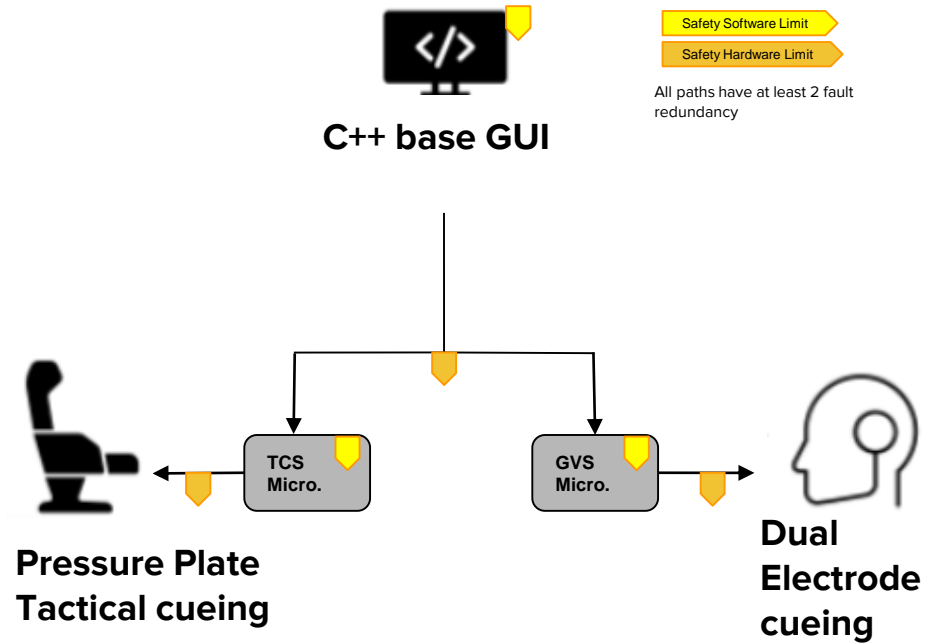


# Applicable NASA Safety Standards

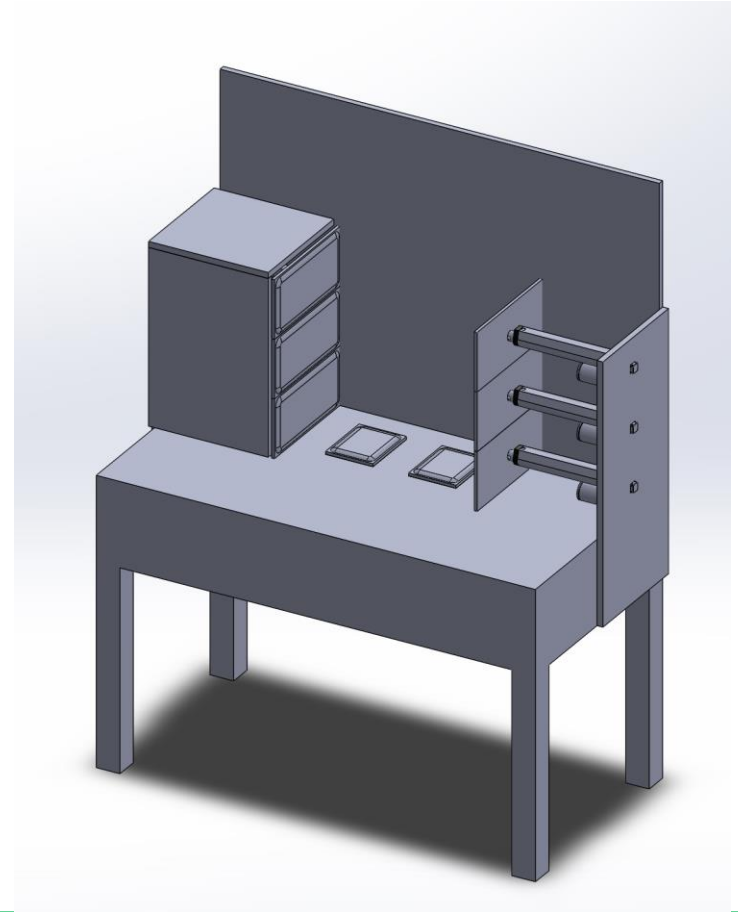
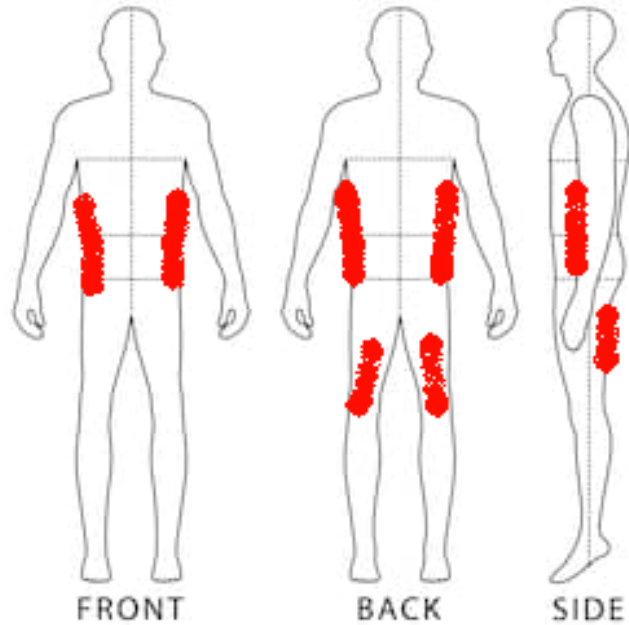
- Safety standards will be implemented with respect to NASA safety standard publications as our framework.
- For example we will be referencing the NASA standards for models and simulations (NASA-STD-7009A) along with NASA software safety guidebook (NASA-GB-8719.13).
- 7 total documents referenced in appendix.

# Safety Overview

- Hardware and software safety limits
- Both the GVS and TCS queuing 'paths' contain redundant (4 total) safety limits
- Test subject controls a system 'shutdown switch' that will stop all cueing

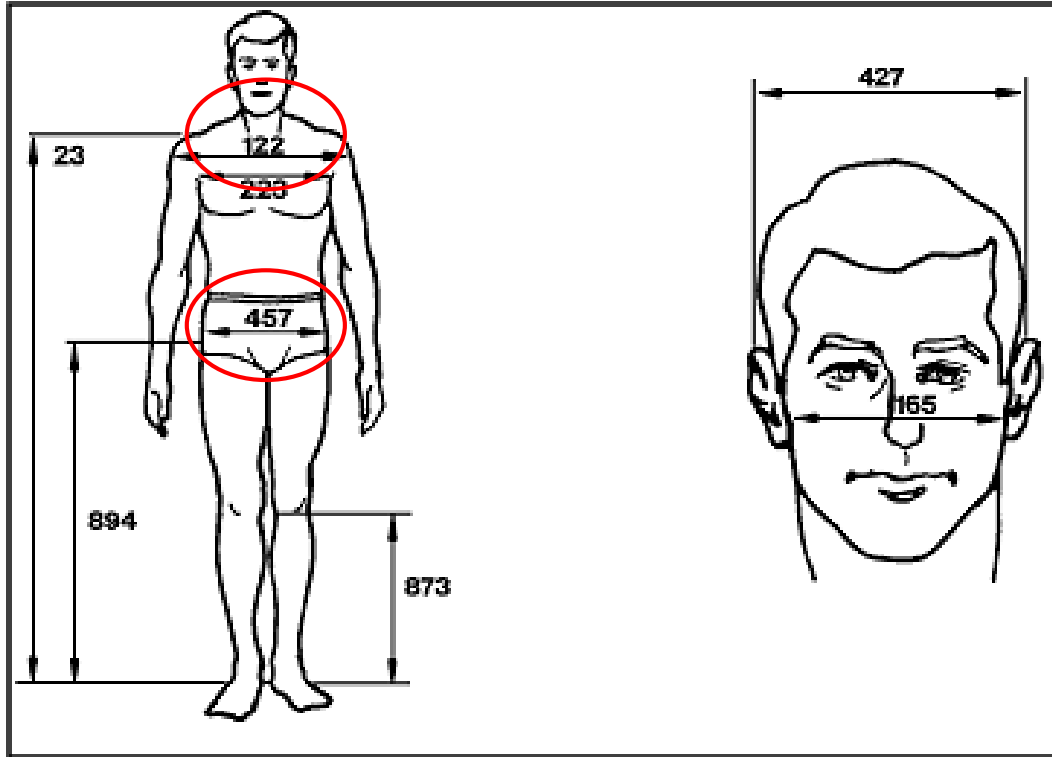


# Tactile Cueing System (TCS) Feasibility



Note: Actuators not drawn to scale

# Anthropometric considerations



Must accommodate customer (Dr. Clark)  
Aim to accommodate 50th percentile male dimensions as well if these are larger

Shoulder breadth (19.3 in) [Dr. Clark's: 18.3 in]

Hip breadth (14.1 in) [Dr. Clark's: 14.4 in]

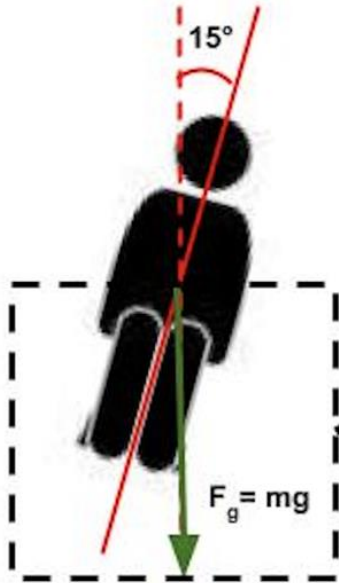
Additional anthropometric measurements must be considered for the GVS and joystick

# TCS Feasibility - Required Force Derivation

Primary focus is accommodating the customer

Anthropometric dimensions are independent

Increase capabilities and provide margin without impact to hardware selection (manufacturing)



\*representative

# TCS Feasibility - Required Force Derivation

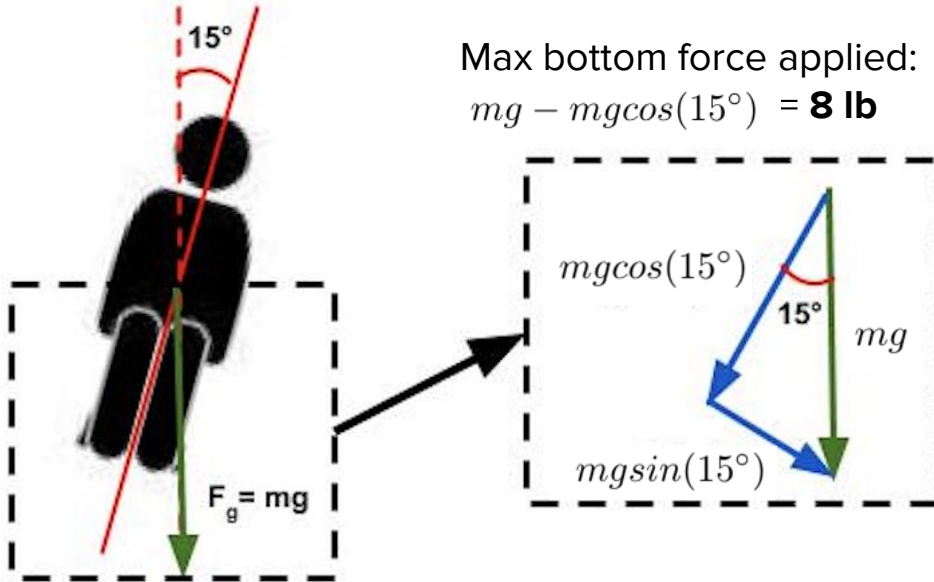
Max test subject weight: **215 lb**

Max side force applied:

$$mg\sin(15^\circ) = \mathbf{55\ lb}$$

Max bottom force applied:

$$mg - mg\cos(15^\circ) = \mathbf{8\ lb}$$



\*representative

# TCS Feasibility - Required Force Derivation

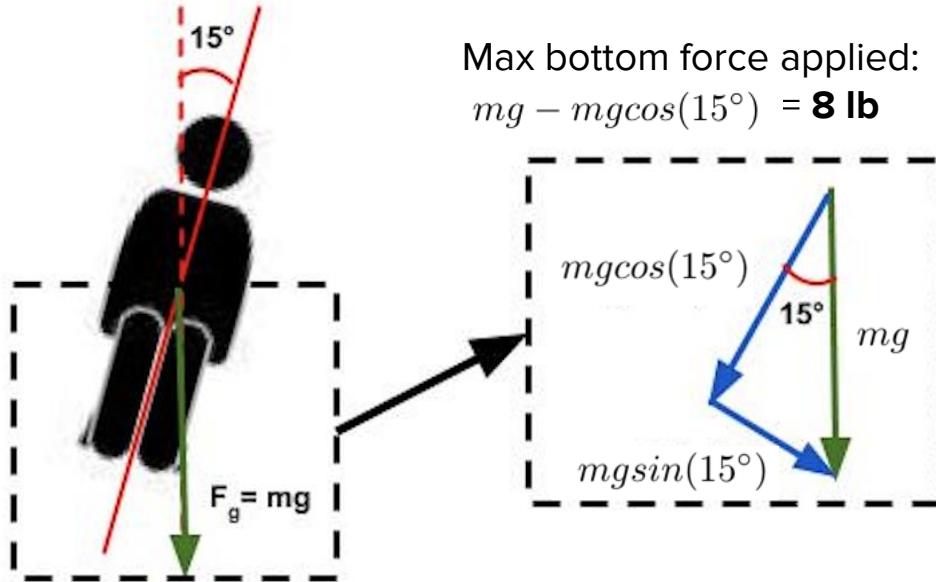
Max test subject weight: **215 lb**

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$$mg\sin(15^\circ) = \mathbf{55\ lb}$$

Max bottom force applied:

$$mg - mg\cos(15^\circ) = \mathbf{8\ lb}$$



\*representative

## DR 1.1.3

Side actuators must apply a total of **55 lb** to mimic horizontal gravitational force during 15° roll angle

Bottom actuators must apply a total of **8 lb** to mimic vertical gravitational force during 15° roll angle

Weight is supported by the primary structure



# TCS Feasibility - Stimulation Area + Force Per Actuator



## Side Actuators per side:

50 lbf Progressive Automations Linear Actuator

Required min pressure	Force from 15° roll angle	Maximum stimulated area	# of modules	Force per actuator	Force margin
0.38 psi	55 lbf	145 sq in	3	19 lbf	31 lbf

Feasible

## Bottom Actuators per side:

Required min pressure	Force from 15° roll angle	Maximum stimulated area	# of modules	Force per actuator	Force margin
0.38 psi	8 lbf	20 sq in	1	8 lbf	42 lbf

Feasible

# TCS Feasibility - Stimulation Area + Force Per Actuator



## Side Actuators per side: DR 1.1.1

50 lbf Progressive Automations Linear Actuator

Required min pressure	Force from 15° roll angle	Maximum stimulated area	# of modules	Force per actuator	Force margin
0.38 psi	55 lbf	145 sq in	3	19 lbf	31 lbf

Feasible

## Bottom Actuators per side:

Required min pressure	Force from 15° roll angle	Maximum stimulated area	# of modules	Force per actuator	Force margin
0.38 psi	8 lbf	20 sq in	1	8 lbf	42 lbf

Feasible

## Safety Limit: DR 2.1

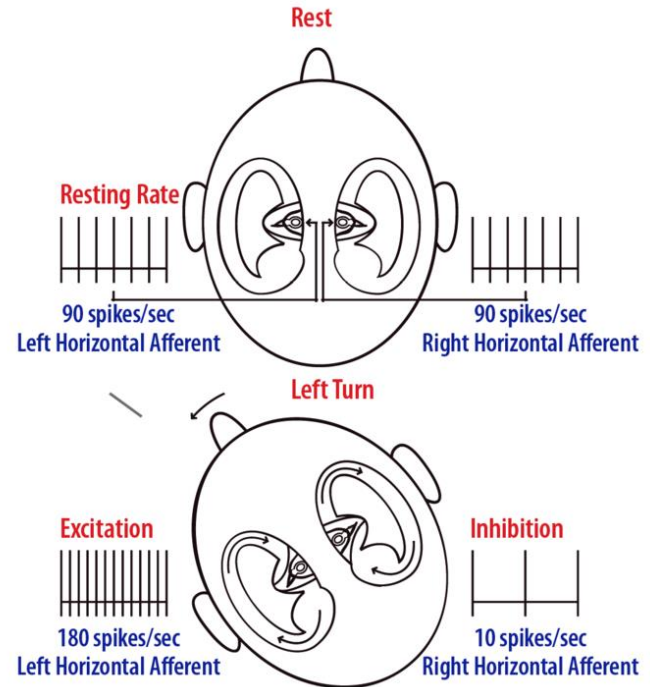
Actual force limit of actuator	# of modules	Maximum pressure capability	Pressure limit (requirement)	Factor of safety
50 lbf	8	1.03 psi	3.63 psi	3.5

Safe

# Galvanic Vestibular Stimulator (GVS) Feasibility

- Vestibular nerves “spike” in voltage at regular intervals when at rest.
- Changes in this spike rate results in the perception of motion
- GVS alters the spike rate by applying currents across the vestibular nerves.

$$\ominus_{\text{perceived}} = f(A)$$



From nobaproject.com

# GVS Feasibility

*Design Requirement 1.2.1:* The GVS shall be able to cue roll angles about the body x axis up to 15° from the nominal upright position.

- Customer identification that 15° can be cued within the 4mA provided.
- Customer also requested linear model of current vs angle with a TBD slope
- Current to angle transfer function will be worked on by customer as part of his research

**Feasible**, Customer (A GVS Expert) has provided a maximum current which will achieve his research goals

# GVS Feasibility

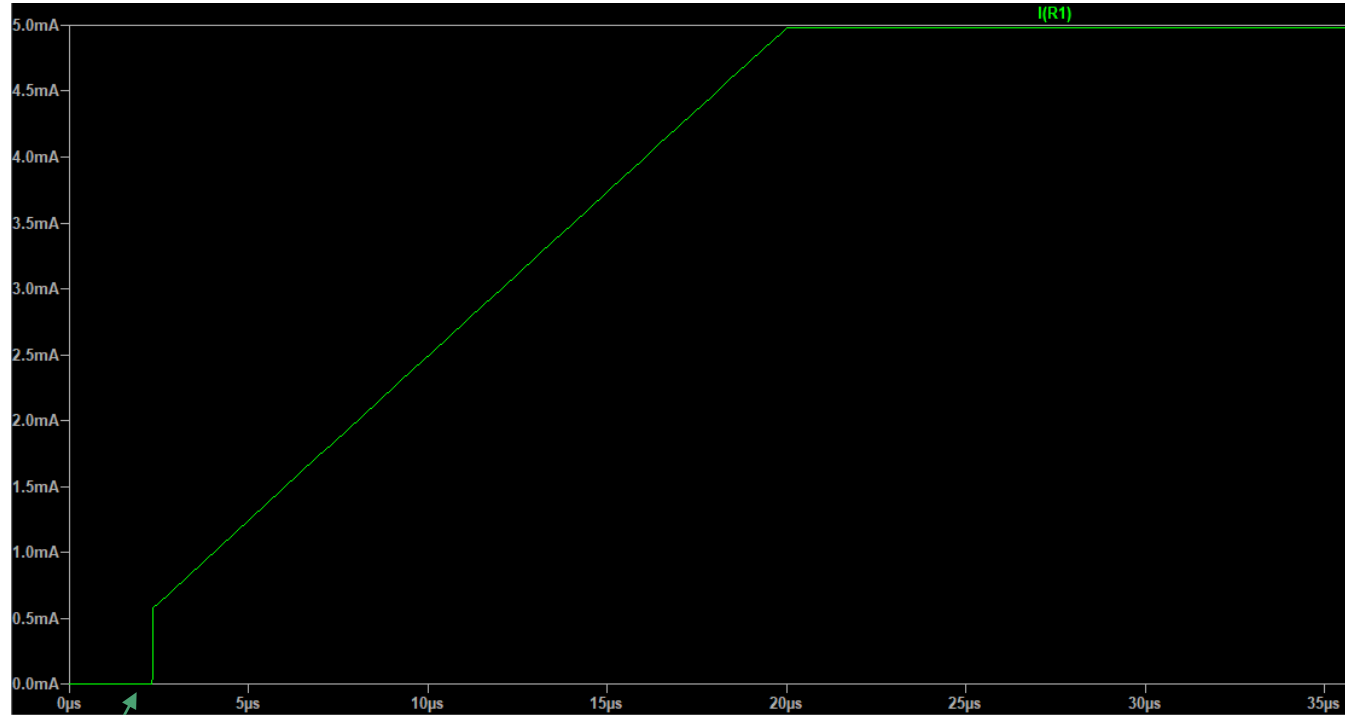
*Design Requirement 2.3:* The GVS will have a maximum output that will not exceed 4 mA of current

- The GVS will be designed as a two fault tolerant system
  - Software:
    - Software limitations prevent commands that draw more than 4 mA (DR 2.4).
    - DACs with built-in ammeters will be used to verify circuit conditions
  - Hardware:
    - Hardware current limiting circuitry will be installed for each electrode wire
    - The DACs maximum current can be set with resistors.

**Feasible,** Both Software and Hardware have two fault tolerance

# GVS Models

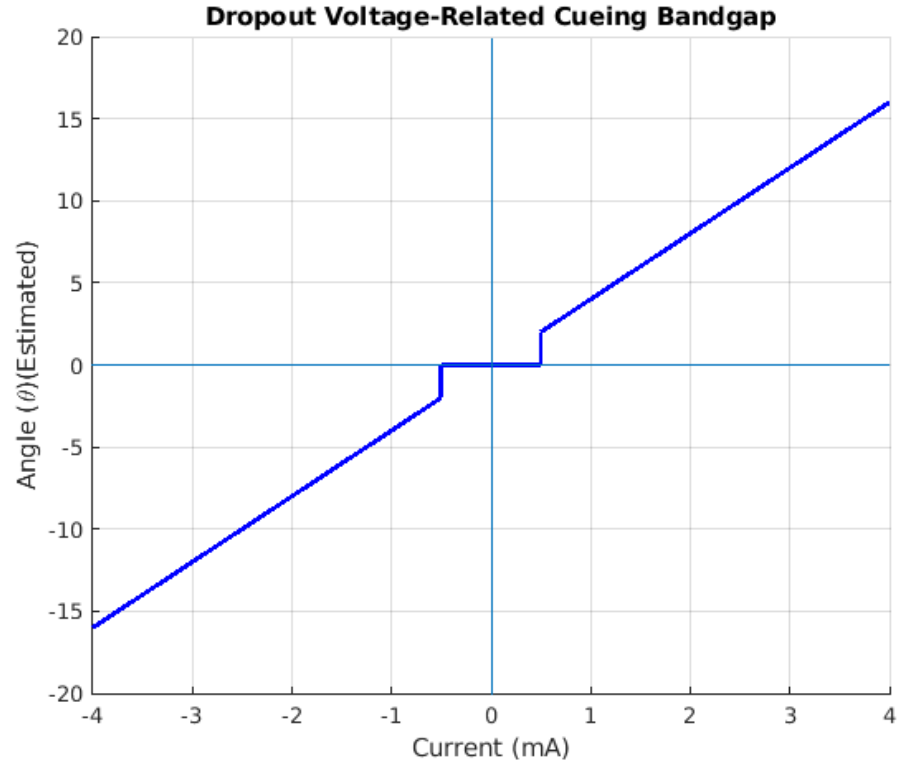
- LTspice Model of AD5770R DAC shows time response for 5mA command across a 1k $\Omega$  load



Result of Dropout Voltage

# GVS Models

- A fixed dropout voltage means our system has a minimum current it can generate ( $V_{\min} = I_{\min}R$ )
- For a  $1k\Omega$  load, this would be around  $0.5\text{mA}$
- Navy Medical Research Institute study found that the minimum perception threshold for current is  $1\text{mA}$

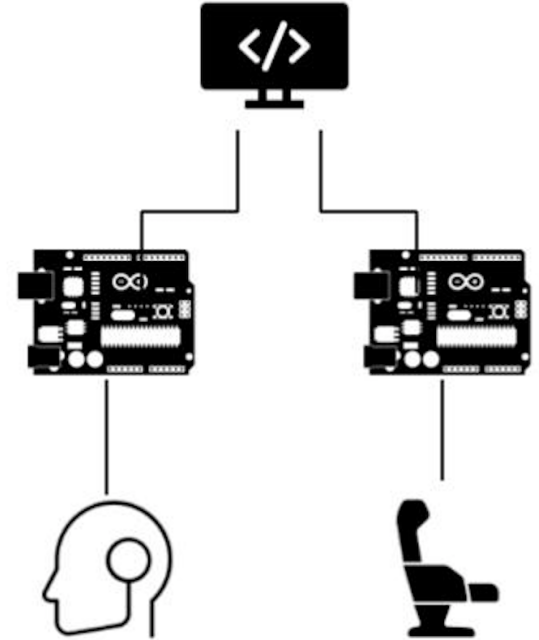


# Central Processing System (CPS) Feasibility

- Takes inputs from controller or subject
- Command Force/Current via TCS and GVS

## **DR 3.2 :**

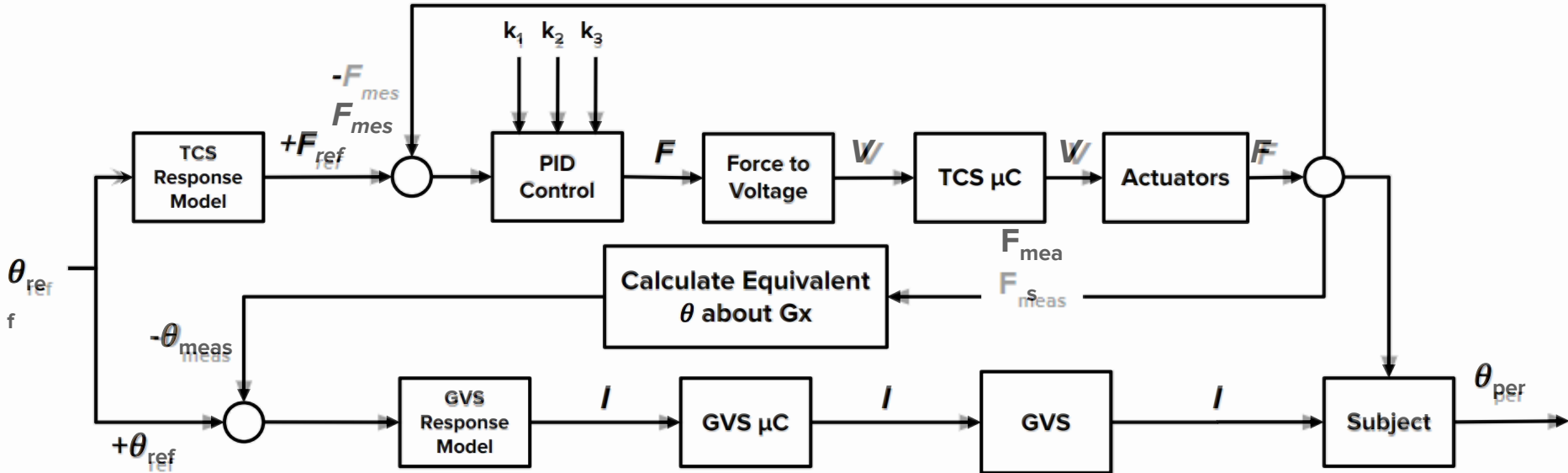
Time delay between the TCS and GVS cues as experienced by the test subject is less than 100 ms.



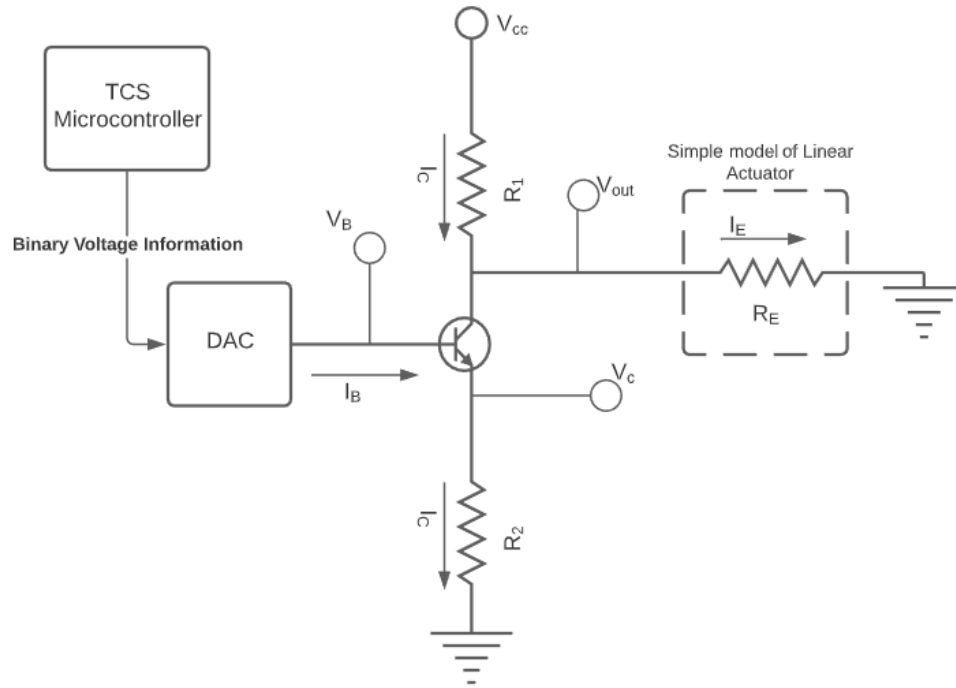


# CPS Feasibility- TCS and GVS Control Logic

$\theta_{ref}$ : commanded angle about Gx  
 $\theta_{per}$ : angle about Gx subject perceives  
 $\theta_{meas}$ : equivalent cued angle about Gx by TCS



# CPS Feasibility- Commanding Voltage to the TCS

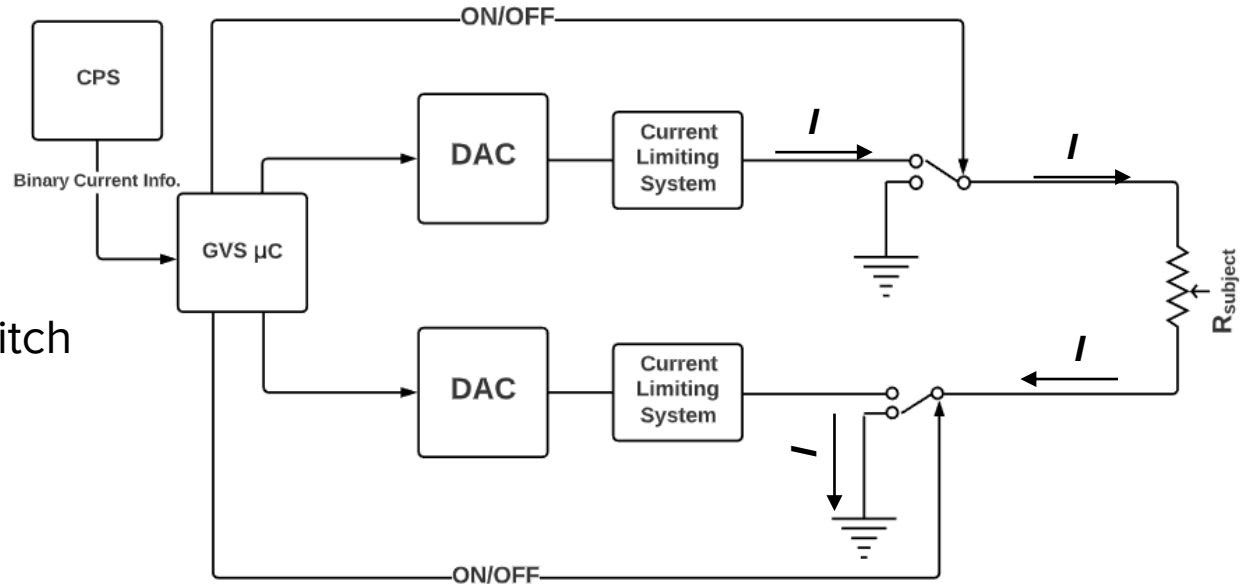


$$V_{out} = V_{cc} - \left(1 - \frac{1}{1 + \beta}\right) \frac{V_B - 0.6V}{R_2} R_1$$

- CPS must be able to command any voltage to actuators
- Looking at a single actuator, voltage output can be controlled through *npn* BJT circuit
  - Equation derived in Appendix
- Scalable to many actuators

**Feasible,** Power supplied to actuators ( $V_{out}$ ) can be commanded at any value (limited by power source, circuit design)

# CPS Feasibility- Commanding Current to GVS



- CPS must be able to switch which electrode is the cathode/anode

**Feasible,** a signal can be sent from CPS to GVS through a USB or similar connector to command the desired current. An inexpensive current source DAC will allow for this.

# CPS Feasibility - Future Considerations

- Development of transfer functions for control logic
- Sensor placement and reliability
  - Forces felt by subject will change with their position in the seat and as they move
- Time response model development of Tactile Cueing System
- Validation of control system outputs concerning human perception

# CHAIR Status Summary

## Potential Problem Areas

- Acquiring Access to Facilities.
  - Provide detailed schematics to PILOT's staff for fabrication.
- Safely Conduct Meetings.
  - Follow CDC guidelines & utilize GitHub
- Acquisition of Components.
  - Start searching for parts early.

	CPS	TCS	GVS
Feasible	<ul style="list-style-type: none"><li>- Control Both MCU's</li><li>- Software Logic</li></ul>	<ul style="list-style-type: none"><li>- Pressure Limit Safety</li><li>- TCS Circuit Design</li><li>- Required Actuator Force</li></ul>	<ul style="list-style-type: none"><li>- Current Limited to 4mA</li><li>- GVS Current Control Model</li><li>- GVS Circuit Design</li></ul>
Some Work Still Needed	<ul style="list-style-type: none"><li>- Cost</li><li>- Component Selection</li></ul>	<ul style="list-style-type: none"><li>- Cost</li></ul>	<ul style="list-style-type: none"><li>- Cost</li></ul>
Excess Work Still Needed	<ul style="list-style-type: none"><li>- System Verification</li><li>- UI Design</li></ul>	<ul style="list-style-type: none"><li>- Fabrication</li></ul>	<ul style="list-style-type: none"><li>- Fabrication</li></ul>
Status Summary	The CPS is yet to be fully proven feasible.	The TCS is mostly proven feasible.	The GVS is mostly proven feasible.

# Thank you

Questions?

# BACKUP SLIDES

# Table of Appendices

[Anthropometric Considerations](#)

[TCS Spatial Requirements](#)

[Software Flowchart](#)

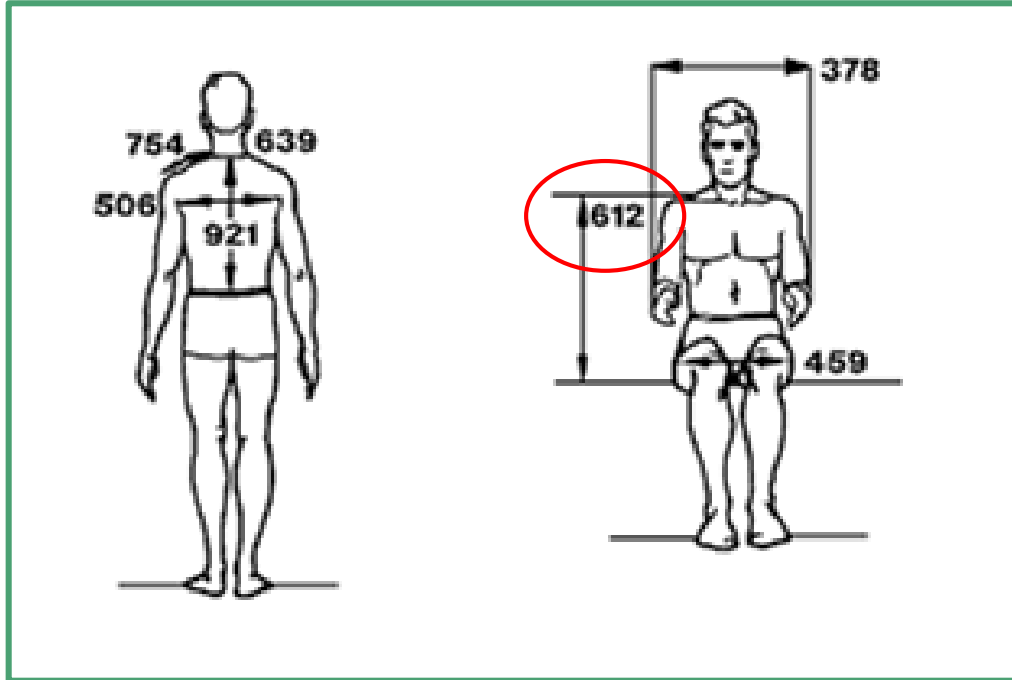
[Derivation of CPS to TCS Actuator Output](#)

[Applicable NASA Safety Standards](#)

[Actuator Spec Plots](#)



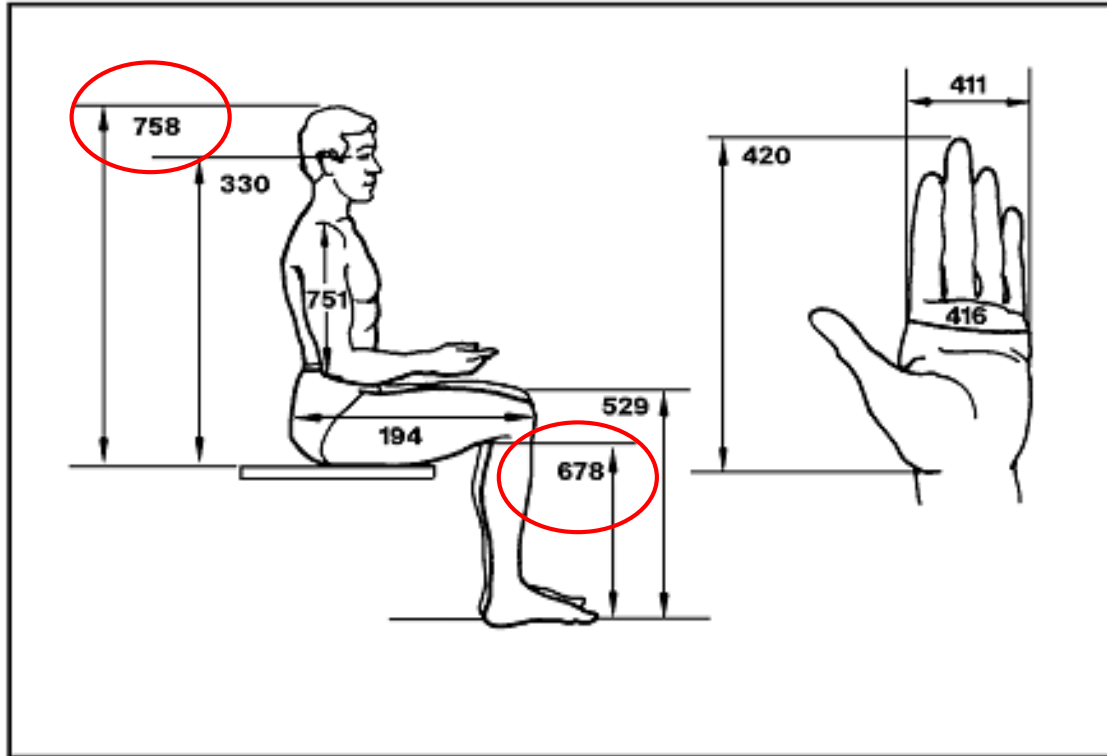
# Anthropometric considerations



Must accommodate customer (Dr. Clark)  
Aim to accommodate 50th percentile male  
dimensions as well if these are larger

Mid-shoulder height, sitting (25.7 in):  
effectiveness

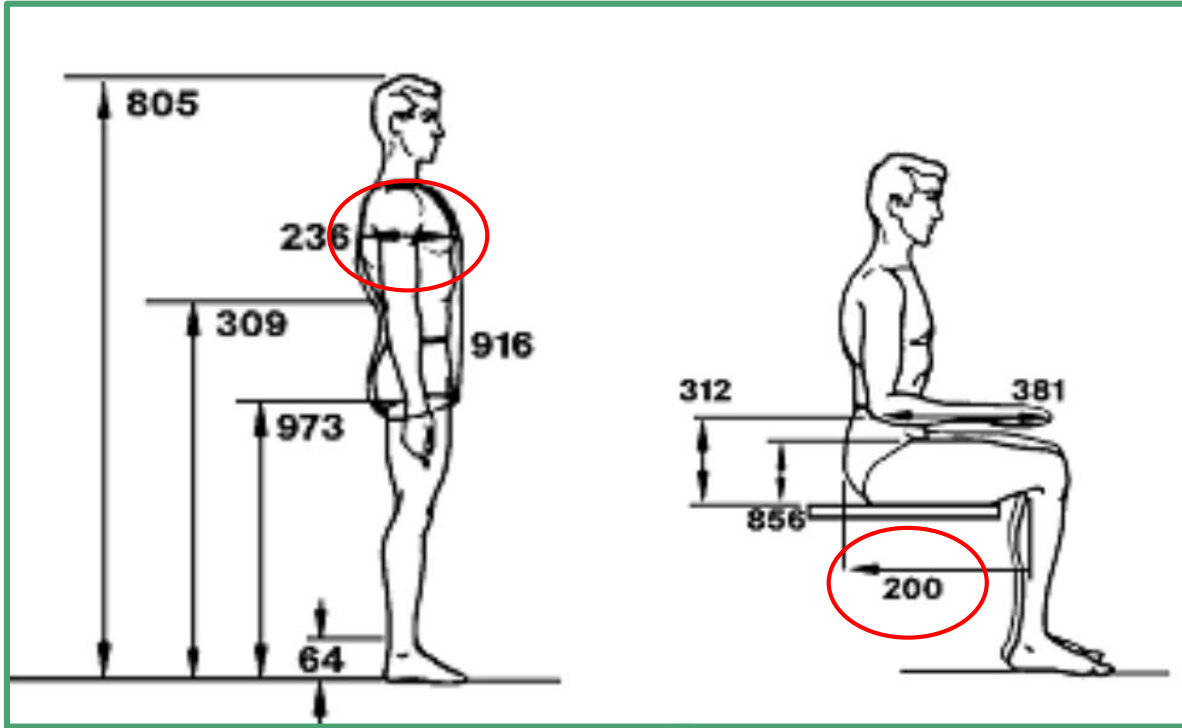
# Anthropometric considerations



Sitting height (73.1 in) - effectiveness

Popliteal height (17.5) - comfort

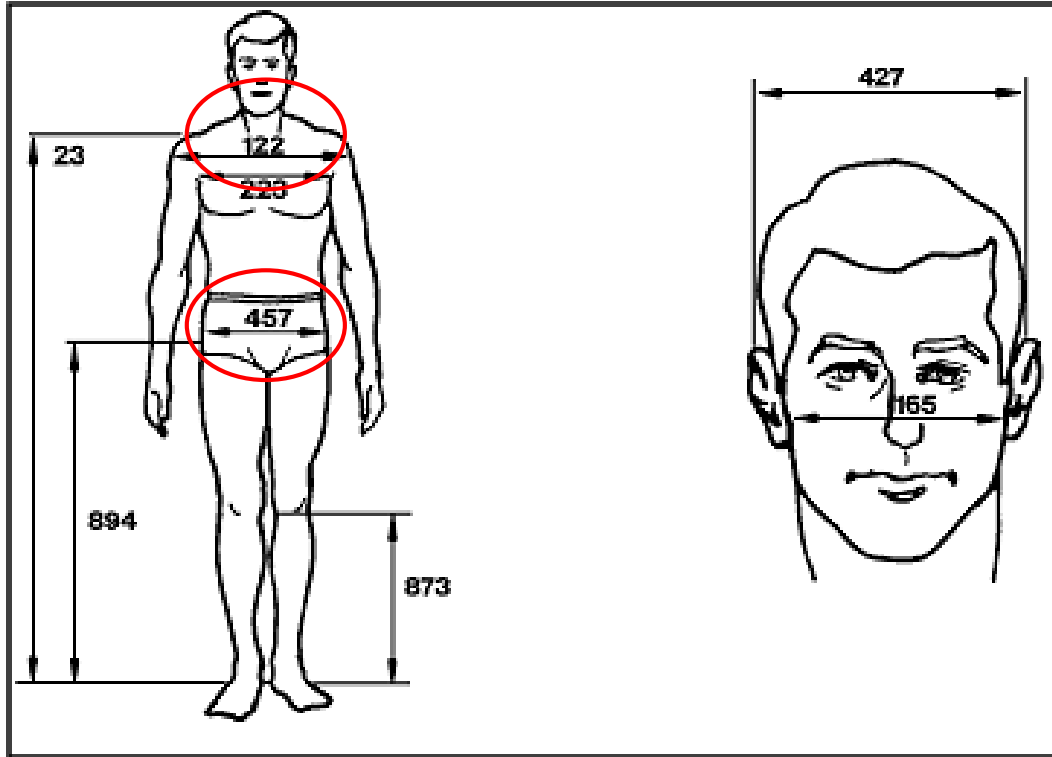
# Anthropometric considerations



Bust depth (9.8 in): effectiveness

Buttock-popliteal length (20.2 in):  
comfort

# Anthropometric considerations



Shoulder breadth (19.3 in): effectiveness  
[Dr. Clark's: 18.3 in]

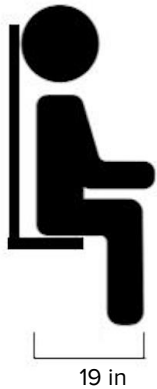
Hip breadth (14.1 in): comfort  
[Dr. Clark's: 14.4 in]

Addition anthropometric measurements must  
be considered for the GVS and joystick

# TCS Feasibility - Spatial Requirements

*Design Requirement 4.1:* “The TCS shall operate within a space no larger than 6’ x 6’.”

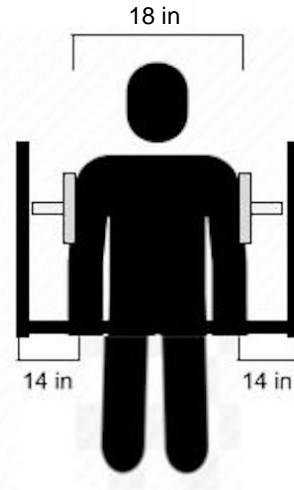
Back to Front:



- Customer femur length is 19”
- Total back to front length around **22”**

**Feasible** with 50” of margin

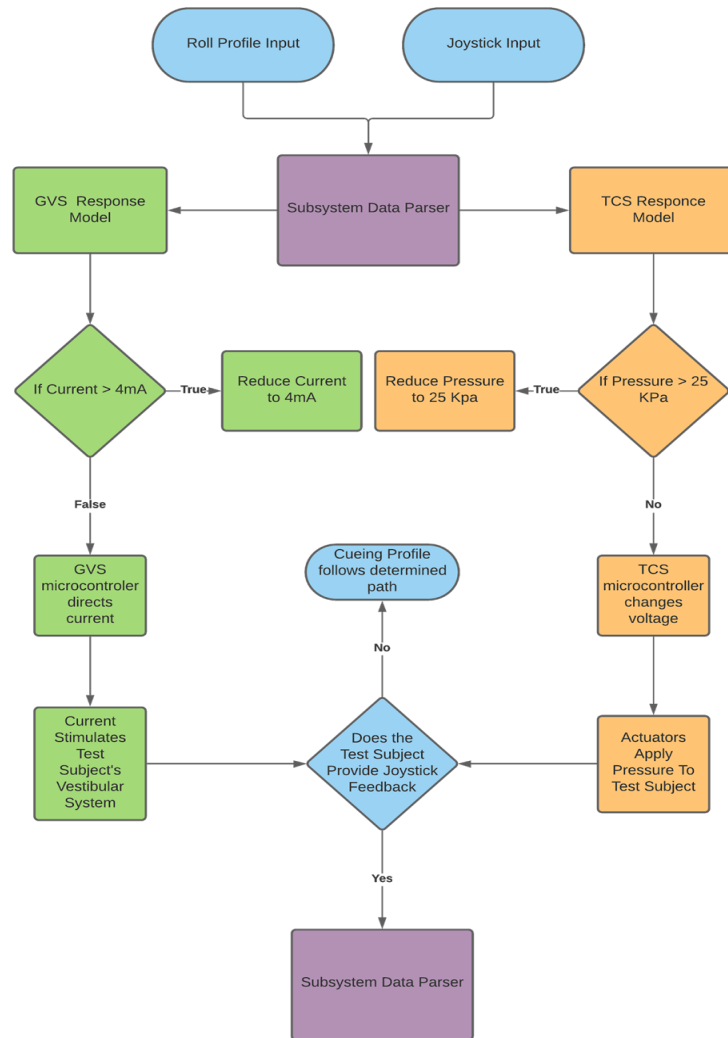
Side to Side:



- Customer shoulder width is 18”
- 150 lb actuators range from 10”- 14” extended
- PA actuator is 10”, plus about 4” for pressure plate and chair
- Total side to side length around **44”**

**Feasible** with 28” of margin

# General Software Flowchart



## Derivation of CPS to TCS Actuator Output

$$\begin{aligned}I_C &= \beta I_B \\V_E &= V_B - 0.6\mathbf{V} \\I_E &= \frac{V_E}{R_2} \\I_C &= I_E - I_B \\V_{out} &= V_{cc} - I_c R_1\end{aligned}$$

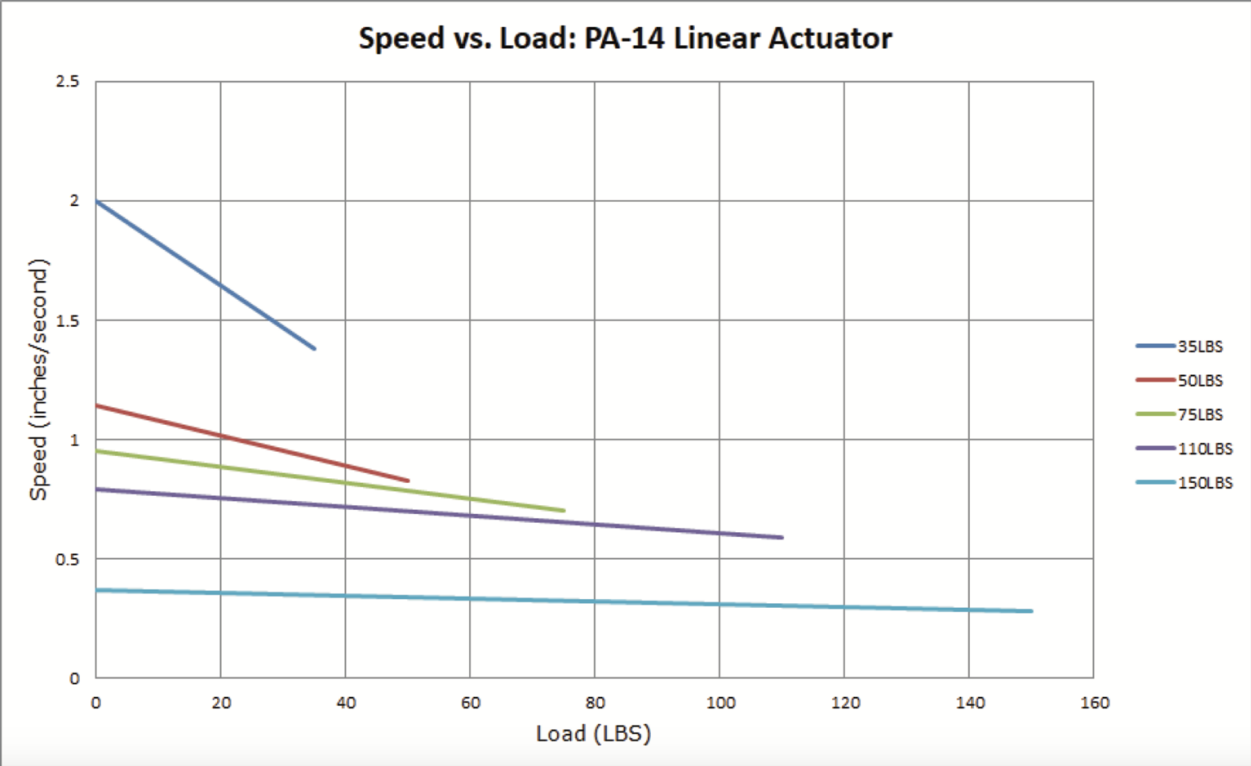
$$\begin{aligned}V_{out} &= V_{cc} - I_c R_1 \\V_{out} &= V_{cc} - (I_E - I_B) R_1 \\V_{out} &= V_{cc} - \left(1 - \frac{1}{1 + \beta}\right) I_E R_1 \\V_{out} &= V_{cc} - \left(1 - \frac{1}{1 + \beta}\right) \frac{V_B - 0.6\mathbf{V}}{R_2} R_1\end{aligned}$$

# Applicable NASA Safety Standards

<b>NASA Safety Standard</b>	<b>Doc.</b>
NASA SAFETY CULTURE HANDBOOK	NASA-HDBK-8709.24
NASA RELIABILITY AND MAINTAINABILITY (R&M) STANDARD FOR SPACEFLIGHT AND SUPPORT SYSTEMS	NASA-STD-8729.1A
STANDARD FOR MODELS AND SIMULATIONS	NASA-STD-7009A
ELECTRICAL, ELECTRONIC, AND ELECTROMECHANICAL (EEE) PARTS ASSURANCE STANDARD	NASA-STD-8739.10
NASA SOFTWARE SAFETY GUIDEBOOK	NASA-GB-8719.13
SOFTWARE FORMAL INSPECTIONS STANDARD	NASA-STD-8739.9
SOFTWARE ASSURANCE AND SOFTWARE SAFETY STANDARD	NASA-STD-8739.8A



# 50 lb Actuator Speed vs. Load



# 50 lb Actuator Current vs. Load

