



Actuated Electromagnetic System for Ice Removal

Manufacturing Status Review February 4, 2016

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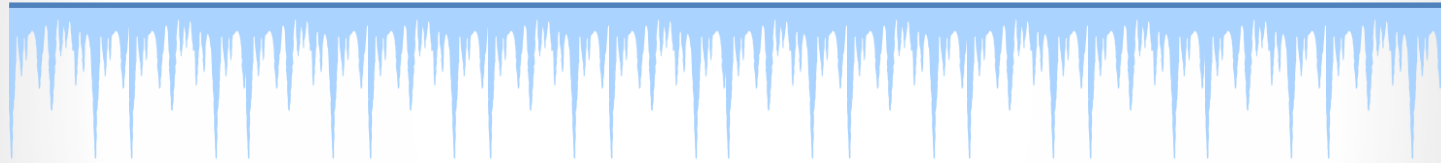
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Overview



Overview

Schedule

Ballistic
Pendulum

Wing Section

Test Setup

Budget

Problem Statement & Objectives

Design, build, and test a small-scale prototype of a deicing system for the **Orion UAV**.



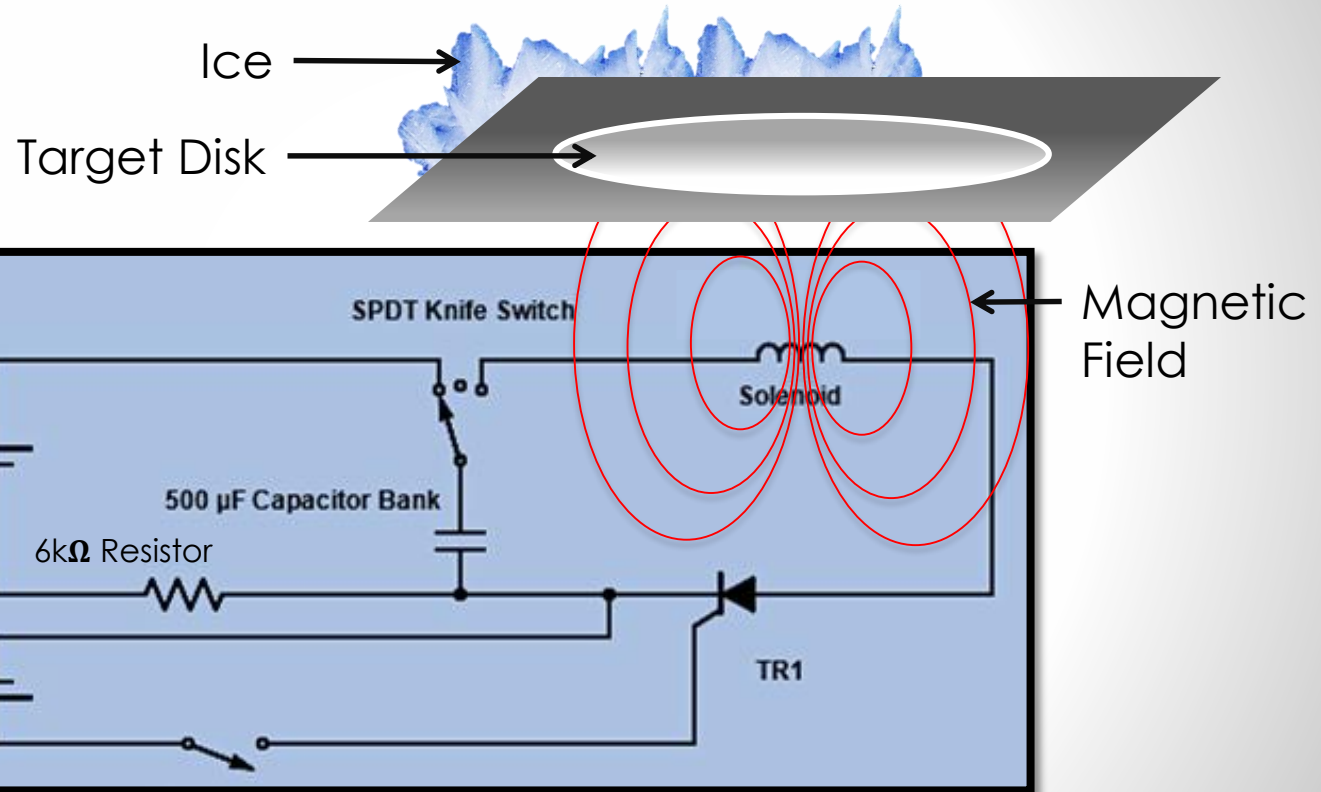
Orion UAV

Functional Requirements

- FR.1 - The full-scale system shall be integrable with the Orion UAV.
- FR.2 - The prototype shall remove ice.
- FR.3 - The full-scale system shall use less than 4kW-hr to deice the wing section.

Design Overview

Electromagnetic Deicing Mechanism



Deicing Mechanism = Baseline design used for all levels of success

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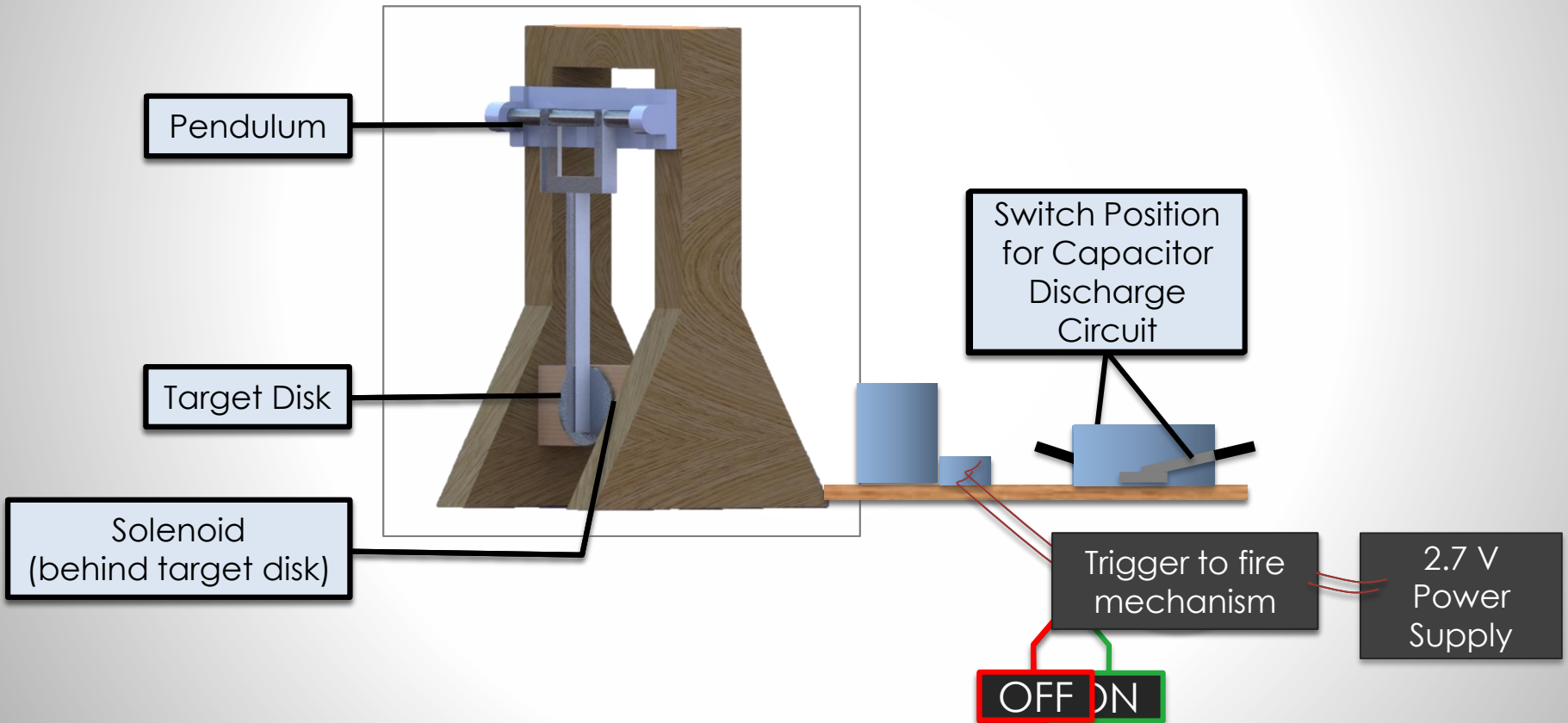
Test Setup

Budget

Level 1 Success – Design & Conops

Purpose of Level 1:

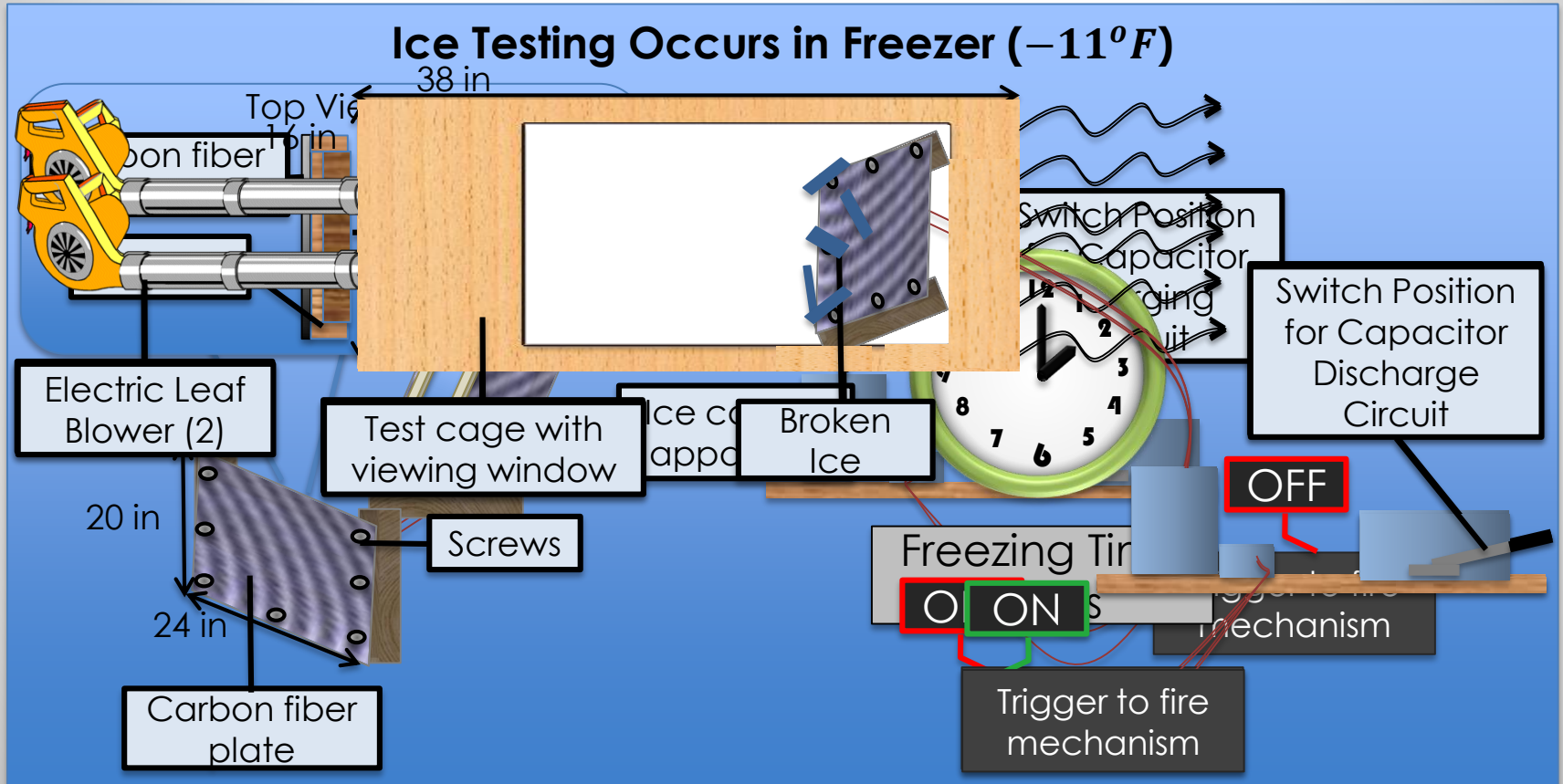
- Proof of Concept of circuit and electromagnetic theory
- Solenoid Force Model verification



Level 2 Success - Design & Conops

Purpose of Level 2:

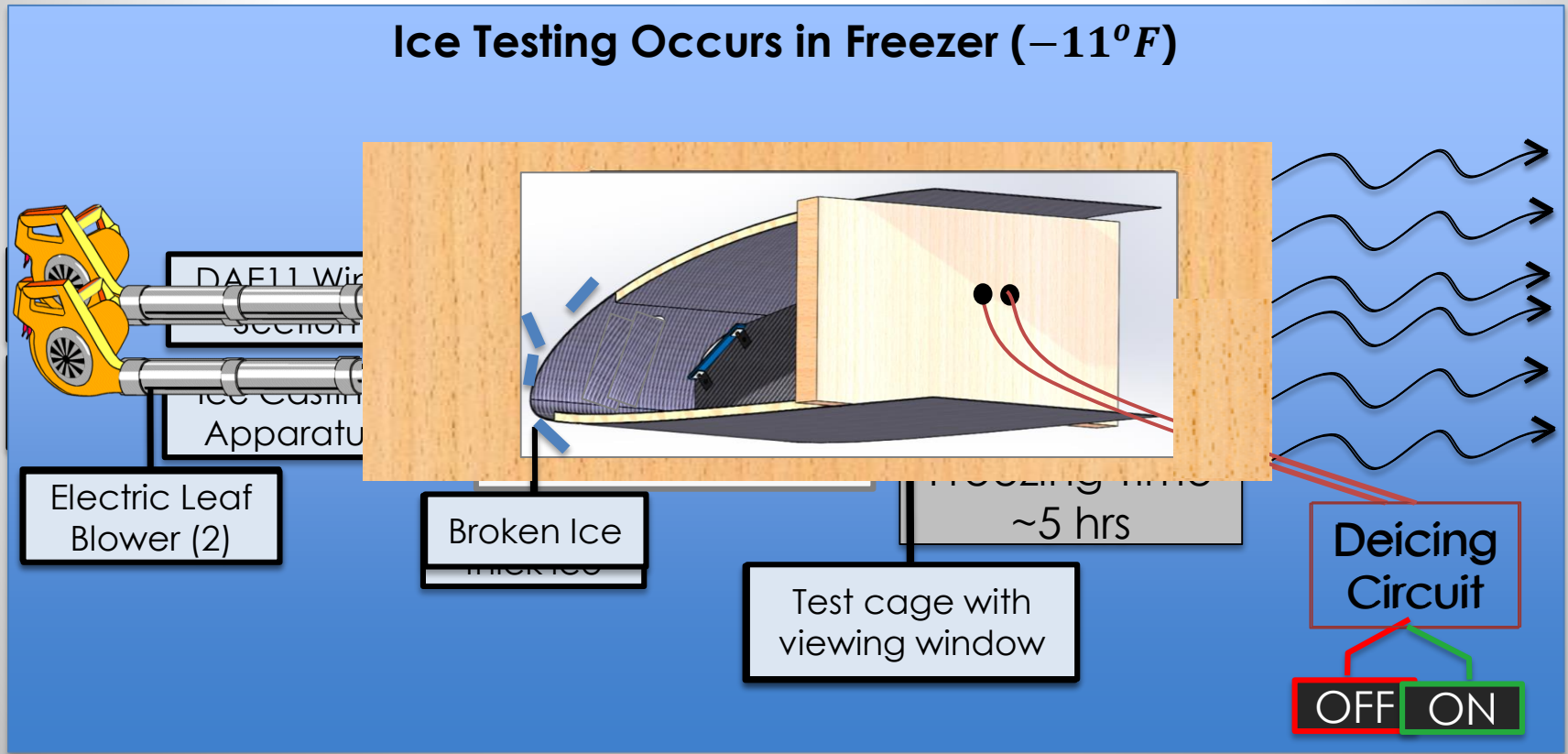
- Testing for ice breaking capabilities on a carbon fiber test article



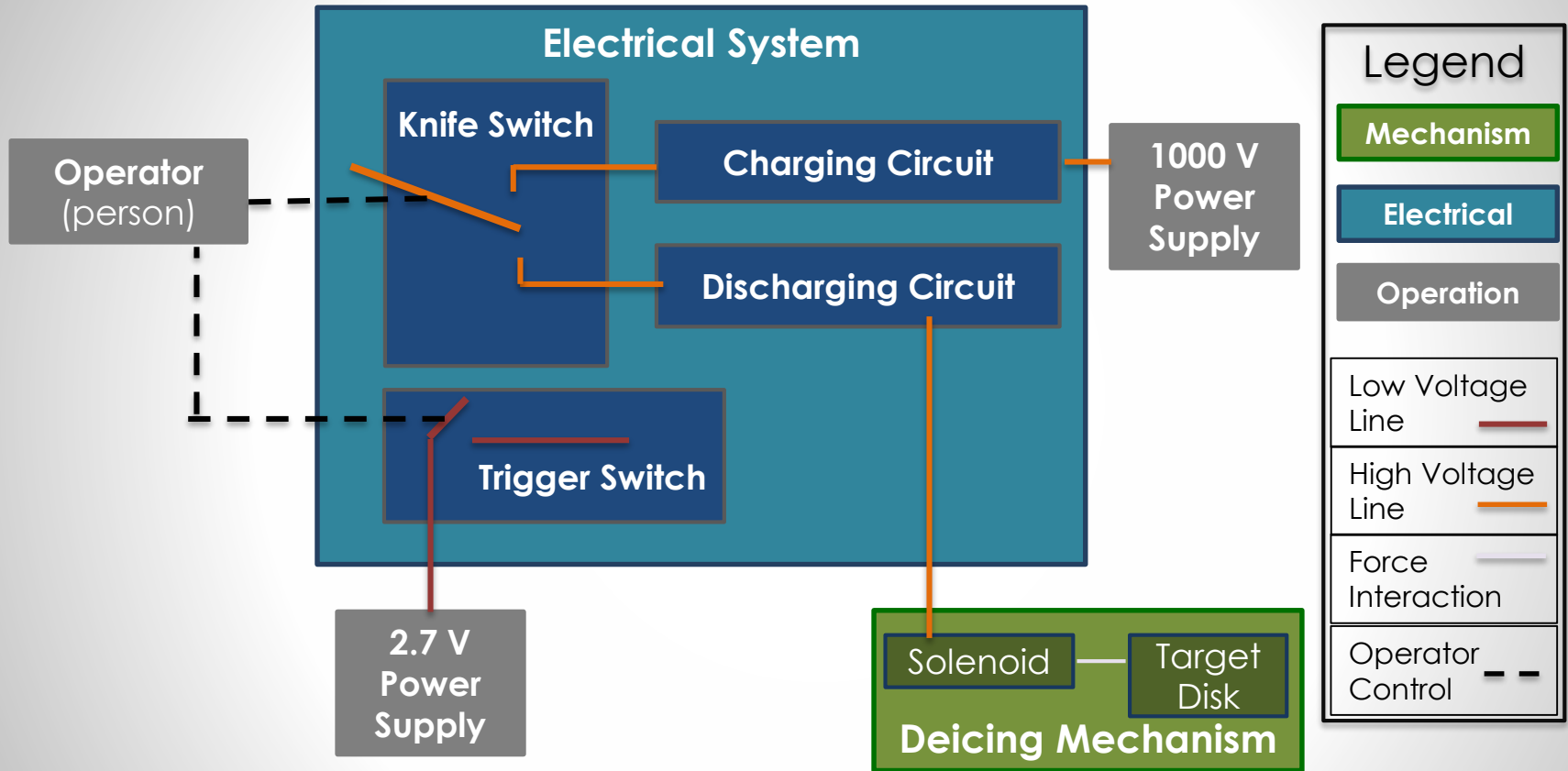
Level 3 Success – Design & Conops

Purpose of Level 3:

- Integration into wing structure like Orion UAV
- Testing in flight like wing section and conditions

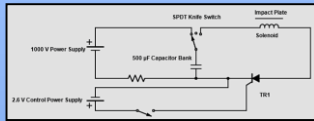


Functional Block Diagram



Critical Project Elements

Ballistic Pendulum

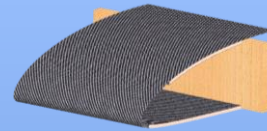


Deicing Mechanism

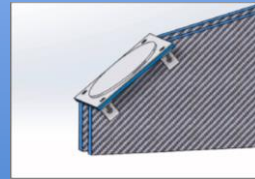


Pendulum Assembly

Wing Section



Test Section

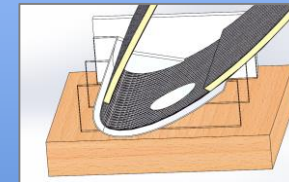


Housing Unit & Support Structure

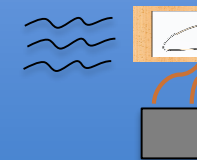
Test Setup



Sensors



Ice Casting



Wind Speed & Test Cage

Overview

Schedule

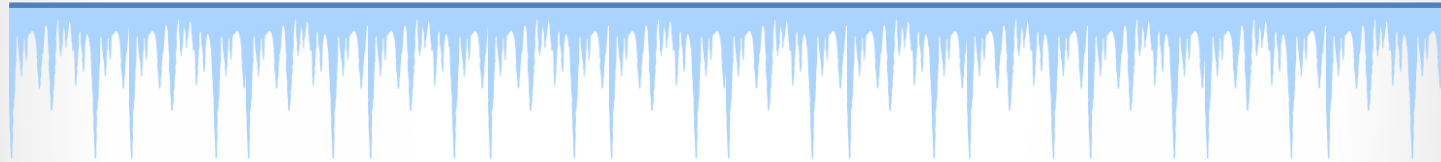
Ballistic
Pendulum

Wing Section

Test Setup

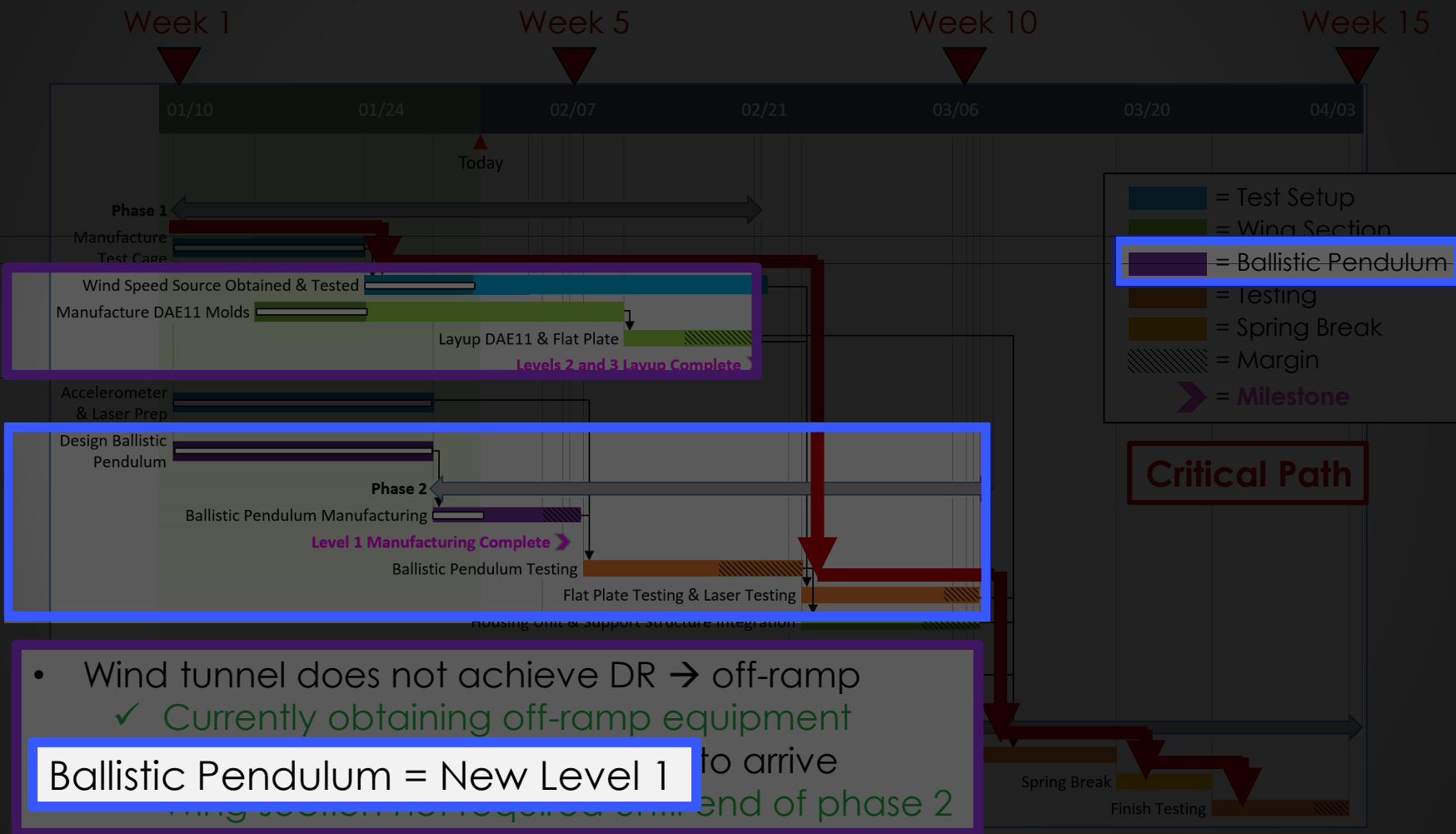
Budget

Schedule



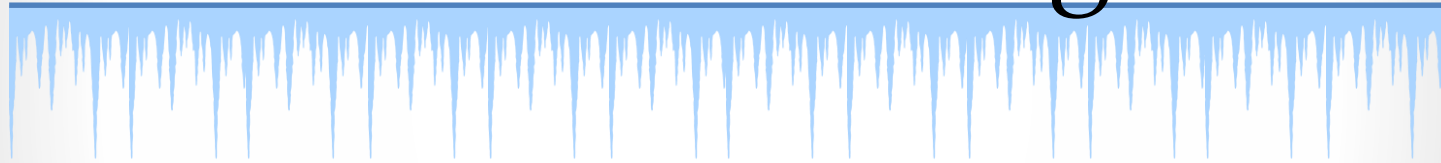
Current Schedule

Spring 2016



- Wind tunnel does not achieve DR → off-ramp
- ✓ Currently obtaining off-ramp equipment to arrive at end of phase 2
- Ballistic Pendulum = New Level 1

Manufacturing



Ballistic Pendulum → Wing Section → Test Setup



Ballistic Pendulum - Description

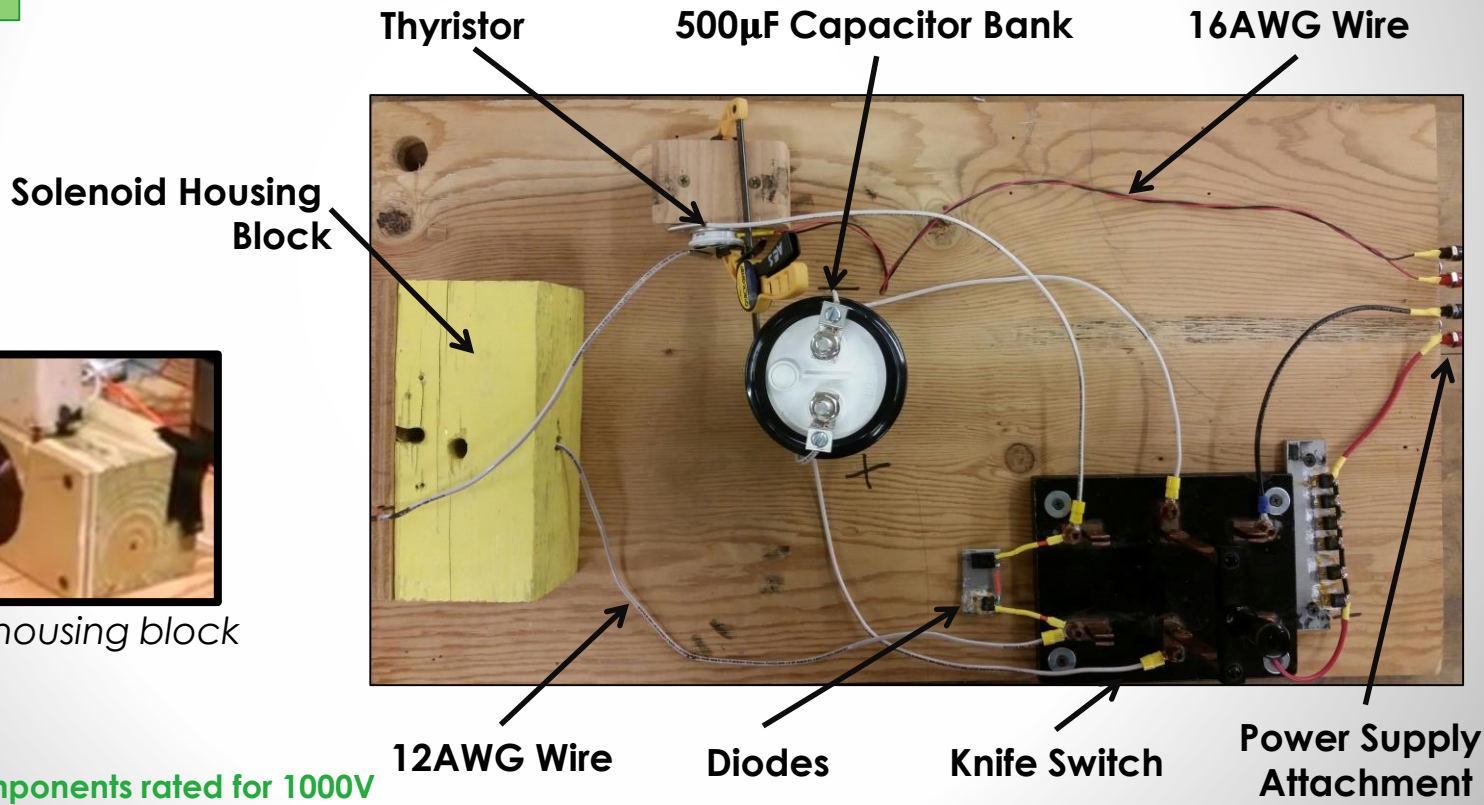
Ballistic Pendulum =

Deicing Mechanism

+

Pendulum Assembly

Completed



* All circuit components rated for 1000V

Ballistic Pendulum – Proof of Concept

Ballistic Pendulum =

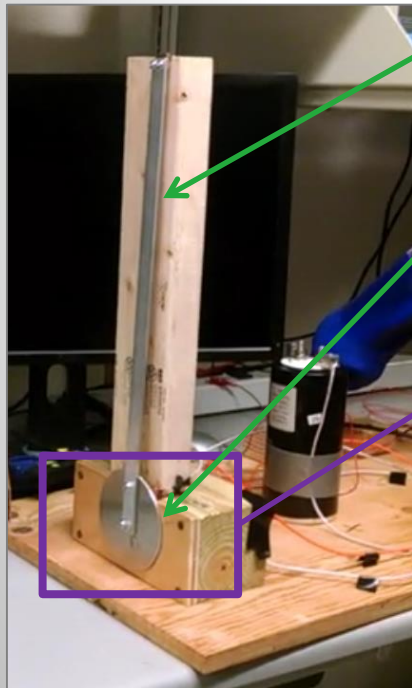
Deicing Mechanism

+

Pendulum Assembly

Completed

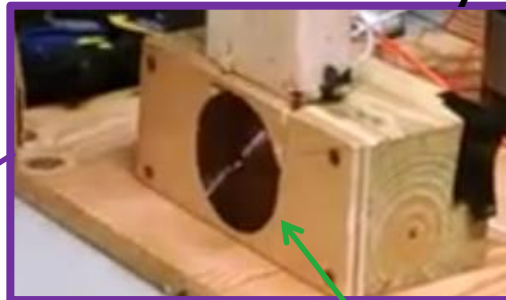
Pendulum Assembly*



Pendulum Arm

Target Disk

Solenoid Assembly



Solenoid

* Deicing mechanism not shown

Proof-of-Concept Test* (50V)



Ballistic Pendulum - Overview

Ballistic Pendulum = **Deicing Mechanism** + **Pendulum Assembly**

In Progress

* All components built in-house

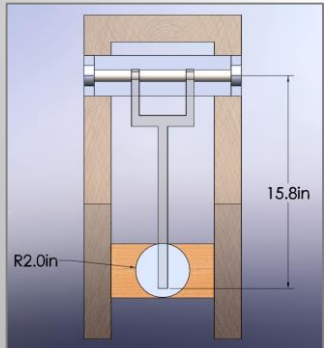
- Aluminum Brace
- Aluminum Pillow Bearing
- Aluminum Pivot Rod
- Acrylic Pendulum Arm

- Aluminum Target Disk
- Accelerometer

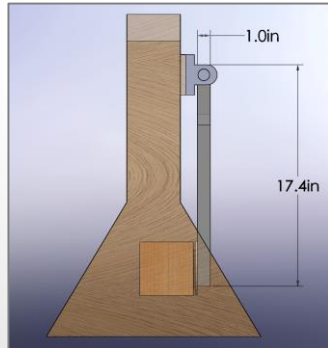
*All materials received except acrylic - 2/2



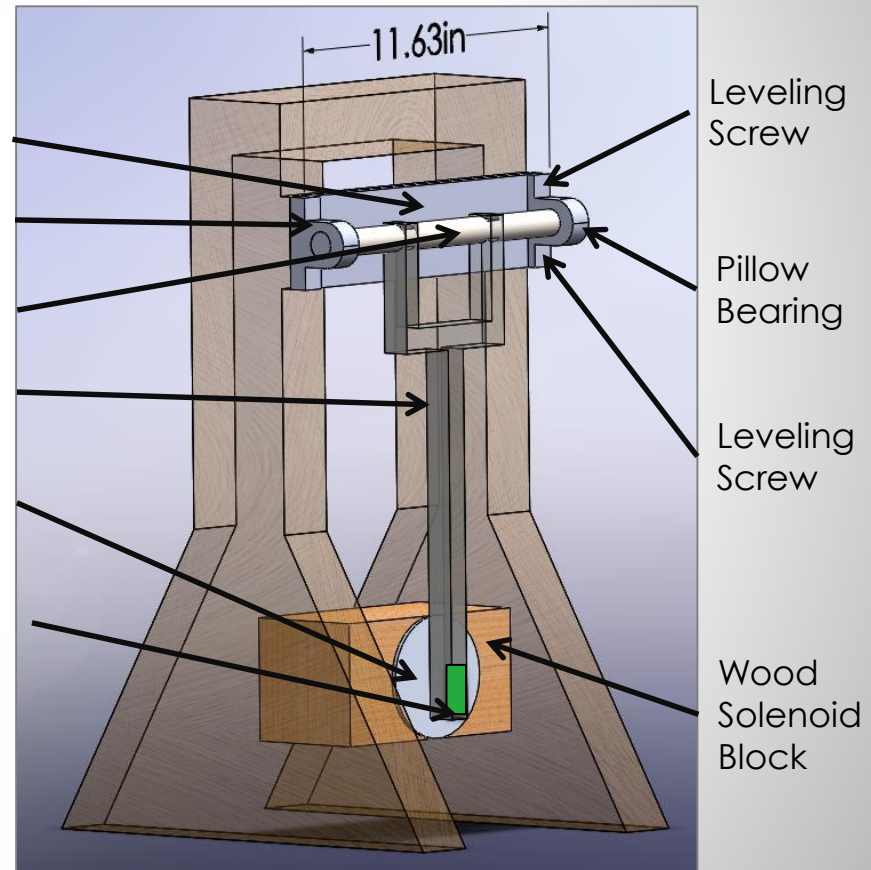
Pendulum Assembly



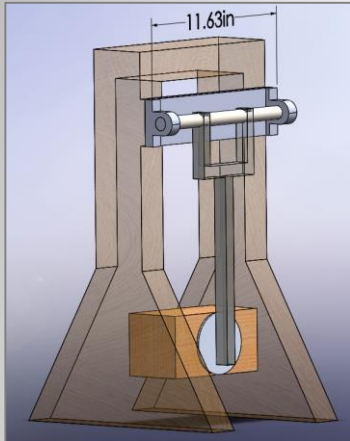
Front View



Side View



Ballistic Pendulum – Manufacturing Status



Ballistic Pendulum

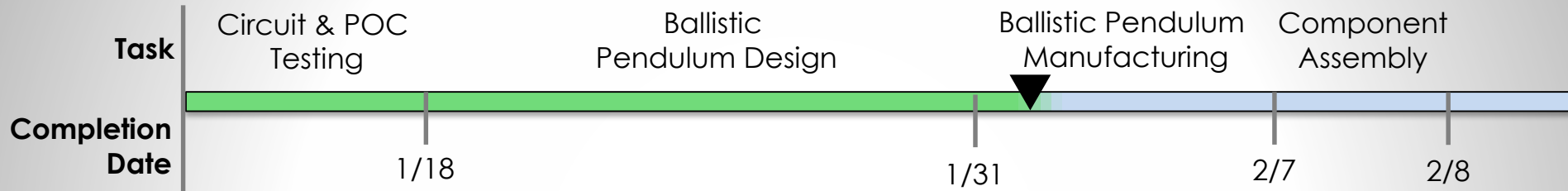
**70% Complete
(Man-Hours)**



Task	Status	Estimated Man-Hours Remaining	Completion Date
Build Deicing Circuit & Test Proof of Concept	Complete	---	1/18
Build Circuit Safety Cage	Complete	---	1/31
Obtain Ballistic Pendulum Materials	Complete	---	2/1
Machine Pillow Bearings	In Progress	4	2/8
Machine Acrylic Pendulum Arm	In Progress	3	2/8
Machine Aluminum Pivot Rod	In Progress	2	2/8
Machine Aluminum Brace	In Progress	1	2/8

Total Man-Hours Remaining for Manufacturing Ballistic Pendulum: **10 hrs**



Ballistic Pendulum – Overall Status



 = Up-to-Date Progress
 = On-Schedule Marker

STATUS: On Schedule

Manufacturing for level 1 success completed by: **2/8**

25% margin in days

Wing Section – DAE 11 Layup Overview

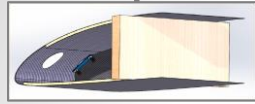
Wing Section

=

DAE11 Wing Layup

+

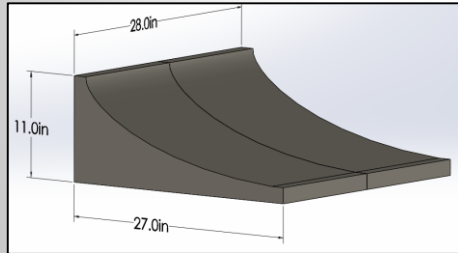
Housing Unit & Support Structure



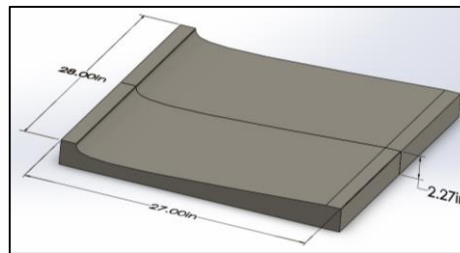
Machine Foam Molds → Prepreg Carbon Fiber Layup → Vacuum Bag → Cure (3 hrs, 250°F)

In Progress

Upper Surface Mold



Lower Surface Mold



Machine Foam Molds (CNC) – Upper & Lower Surfaces

- Allow for 14% margin in span during layup
- Negative layup with upper and lower surfaces attached

Future Work



DAE11 Layup with Wooden Spar

- Includes rib on one end (dragon plate with epoxy)

*** All materials are received with excess foam and prepreg**

Overview

Schedule

Ballistic
Pendulum

Wing Section

Test Setup

Budget

Wing Section - Housing Unit & Support Structure Overview

Wing Section =

DAE11 Wing Layout

+

Housing Unit & Support Structure

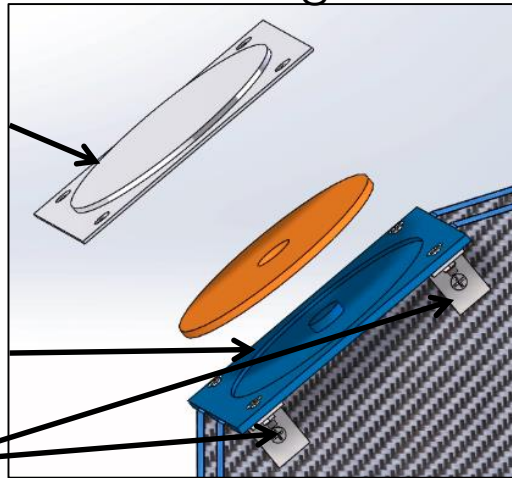
Future Work

Housing Unit

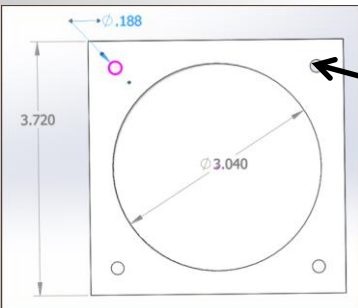
Top Plate (Polyethylene)

Base Plate (Polyethylene)

L-Brackets & Through Holes



Solenoid Housing Unit



Top Plate with Dimensions [in]

Countersink screw

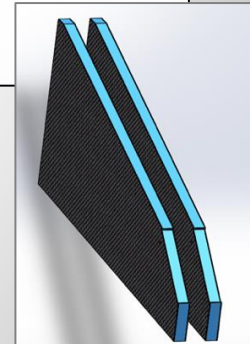
Support Structure

Epoxy + Dado Joint

9.00in

17.20in

Carbon fiber (foam core) support structures (Dragon Plate) – laser cut
(1 support structure = 2 laser cut dragon plates)

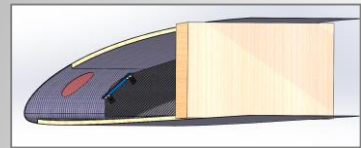


Front View of 1 Support Structure

***All materials are received**



Wing Section – Manufacturing Status



Wing Section

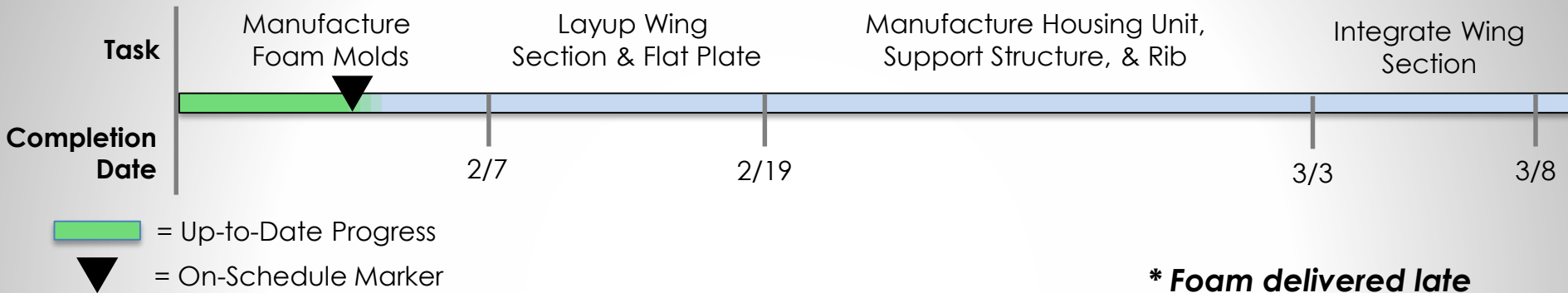
Task	Status	Estimated Man-Hours Remaining	Completion Date
Obtain All Wing Section Materials	Complete	---	1/24
Machine Foam Molds	In Progress	10	2/8
Layup DAE11 & Flat Plate	Future Work	6	2/19
Machine Housing Unit	Future Work	5	3/3
Cut Support Structure & Rib	Future Work	6	3/3
Wing Section Integration	Future Work	4	3/8

**10% Complete
(Man-Hours)**

Total Man-Hours Remaining for Manufacturing Wing Section: **31 hrs**



Wing Section - Overall Status



STATUS: On Schedule *

*** Foam delivered late**
 → Schedule allowed for adjustment in layup completion date

Manufacturing for level **2** wing section completed by: 2/19

20% margin in days

Manufacturing for level **3** wing section completed by: 3/8

30% margin in days

Test Setup - Sensors Description



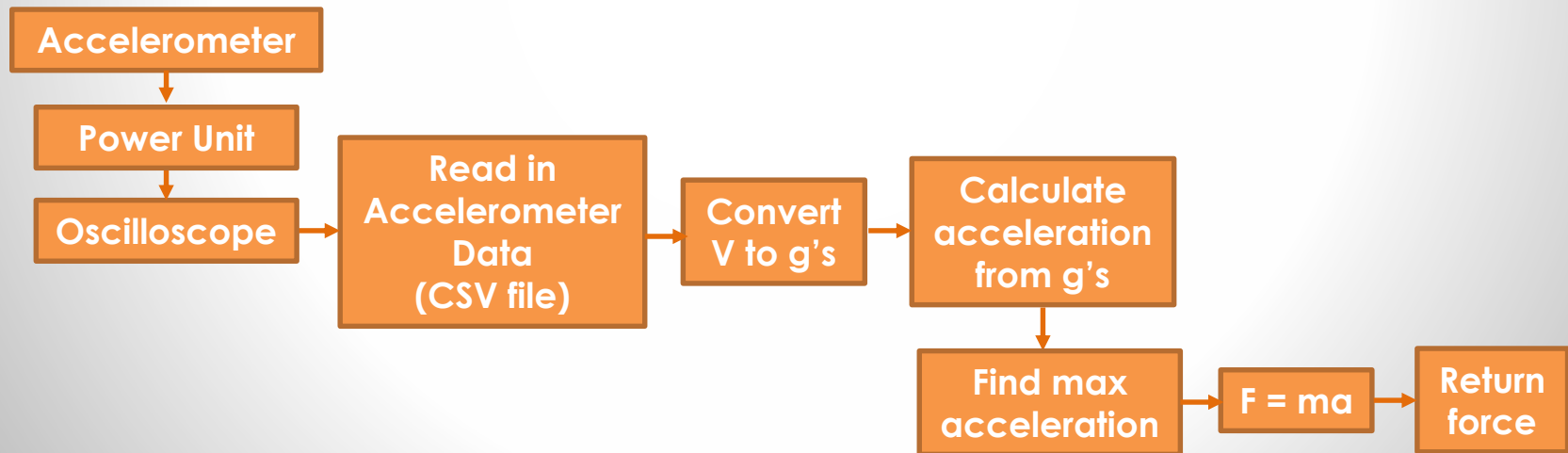
Sensor Goal → Verify force model and ANSYS model with force calculation

Methods to Calculate Force

- Accelerometer (acceleration) → **Ballistic Pendulum (Level 1)**
- High Speed Camera (height of pendulum arm) → **Carbon Fiber (Levels 2 & 3)**
- Laser (deflection) → **Carbon Fiber (Levels 2 & 3)**

Completed

Accelerometer Software Development: Force Calculation Function



Test Setup – Wind Source Description

Test Setup = **Sensors** + **Wind Source** + **Ice Casting**

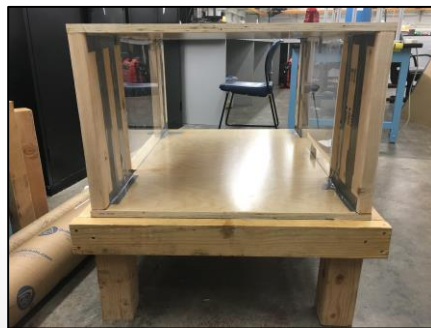
Test Cage

Completed



Side View of Test Cage

Front View of Test Cage



Cross-Section
25 in x 16 in

Wind Source

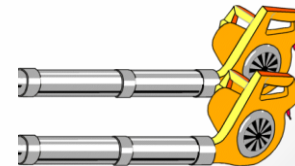
In Progress

Original Source: Wind Tunnel
→ Cannot achieve wind speed requirement

- Off Ramp -

2 Leaf Blowers

Currently achieving 25% lower than required average wind speed through simulation



* Currently in possession of 1 leaf blower, in process of obtaining 2nd

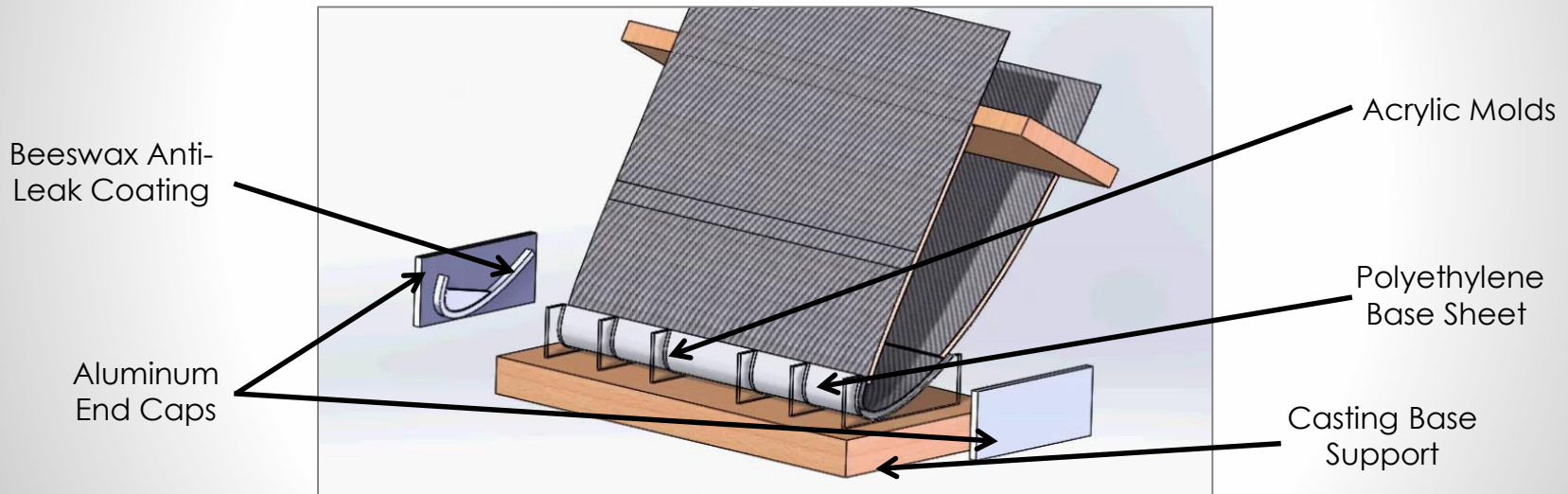


Test Setup – Ice Casting Apparatus Description

Test Setup = **Sensors** + **Wind Source** + **Ice Casting**

Future Work

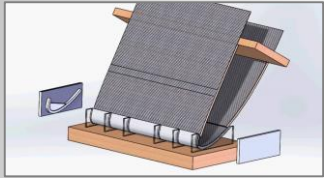
Ice Casting Apparatus



Note: end caps held in place with clamp

* Currently in possession of all materials for ice casting apparatus

Test Setup - Manufacturing Status



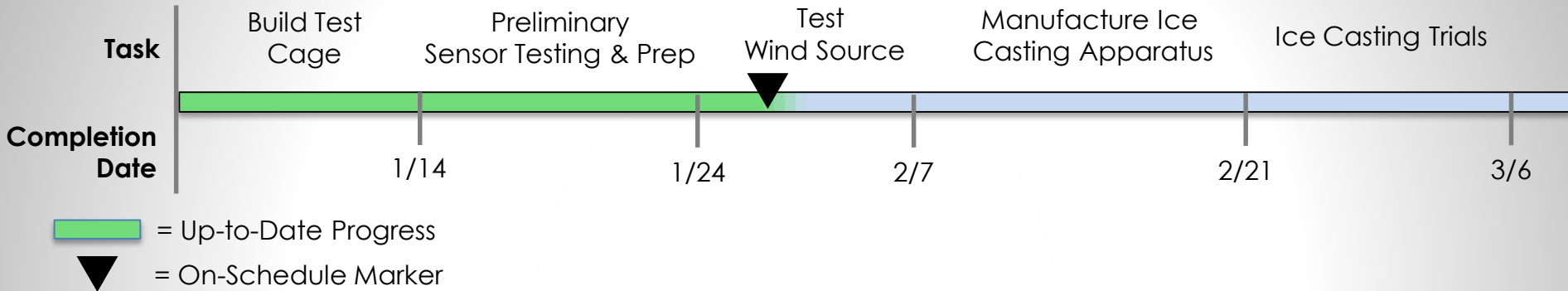
**20% Complete
(Man-Hours)**

Task	Status	Estimated Man-Hours Remaining	Completion Date
Obtain all materials for test cage & ice casting, build test cage	Complete	---	1/18
Preliminary Laser Test	Complete	---	1/22
Accelerometer Code & Preliminary Test	Complete	---	1/23
Obtain Second Leaf Blower	In Progress	1	2/19
Machine Aluminum End Caps	Future Work	7	3/3
Cut Acrylic Molds	Future Work	5	3/3
Ice Casting Integration	Future Work	3	3/8

Total Man-Hours Remaining for Manufacturing Wing Section: **16 hrs**



Test Section - Overall Status

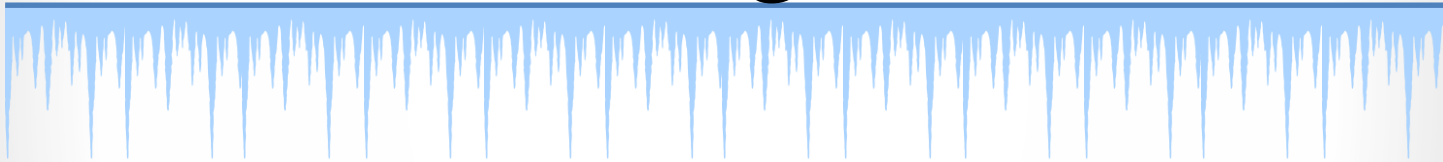


STATUS: On Schedule

Manufacturing for level **3** test section complete by: 3/6

33% margin in days

Budget



Overview

Schedule

Ballistic
Pendulum

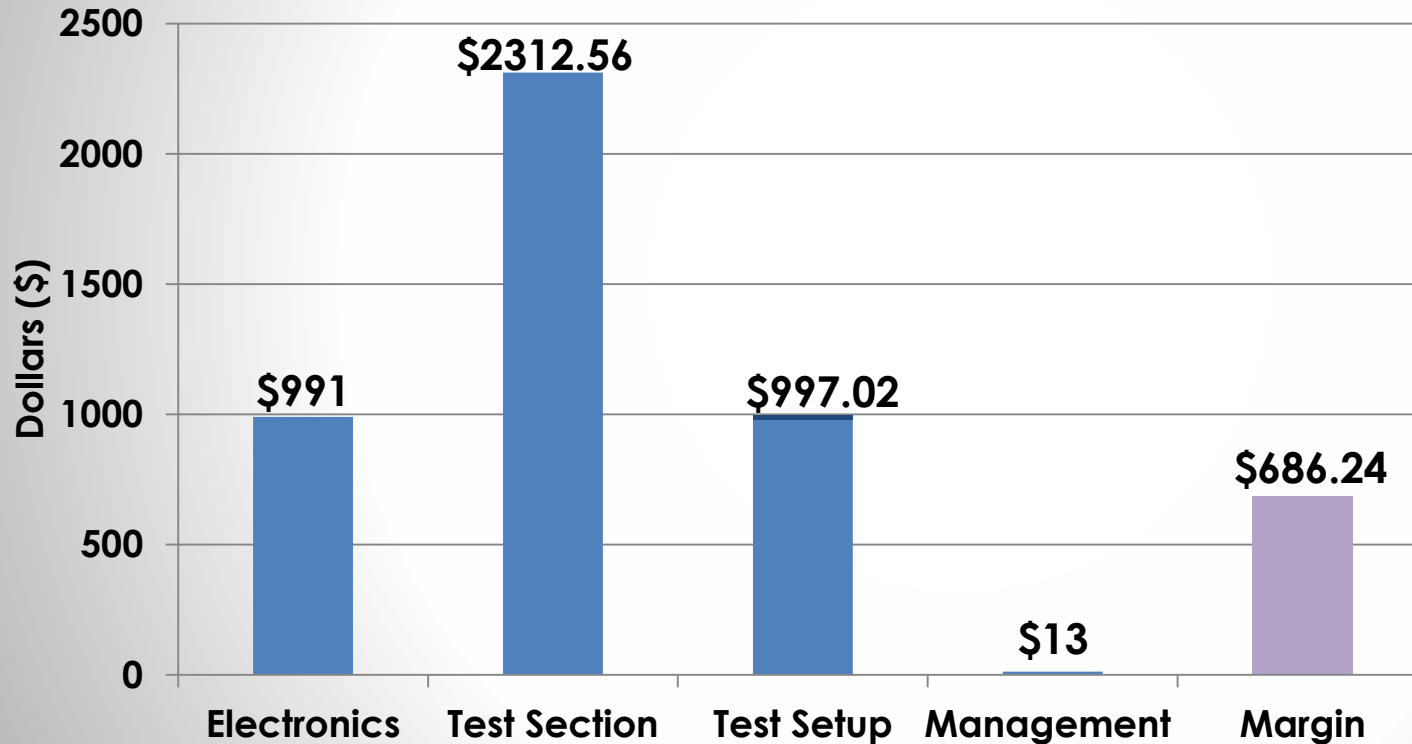
Wing Section

Test Setup

Budget

Budget Status

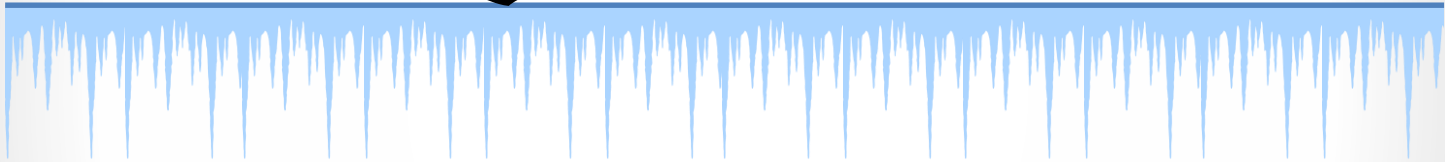
Budget Distribution



Potential Uses of Margin

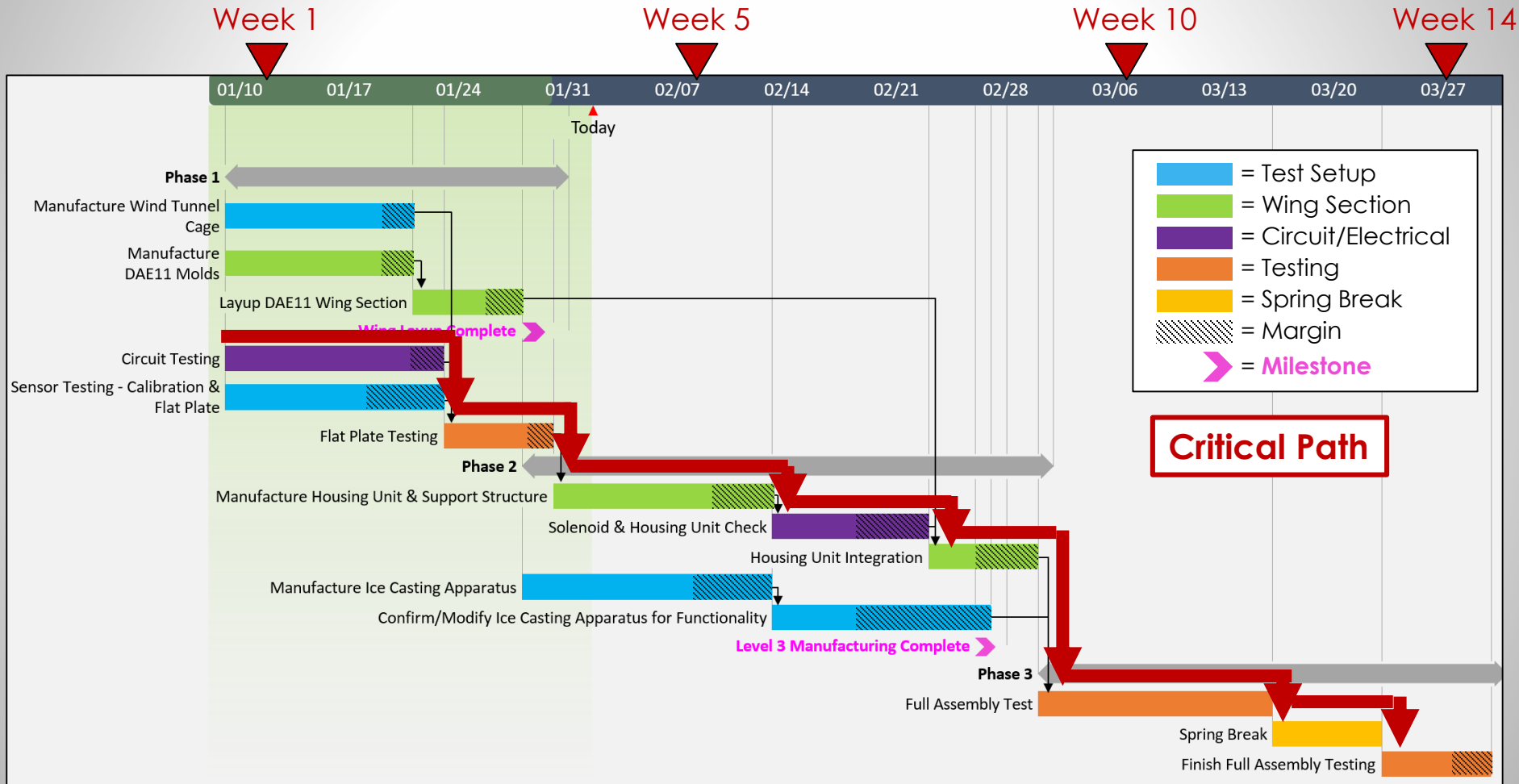
- Additional...
 - Thyristors
 - Capacitor
- Leaf blower(s)
- Improvement of electronic set up

Questions?



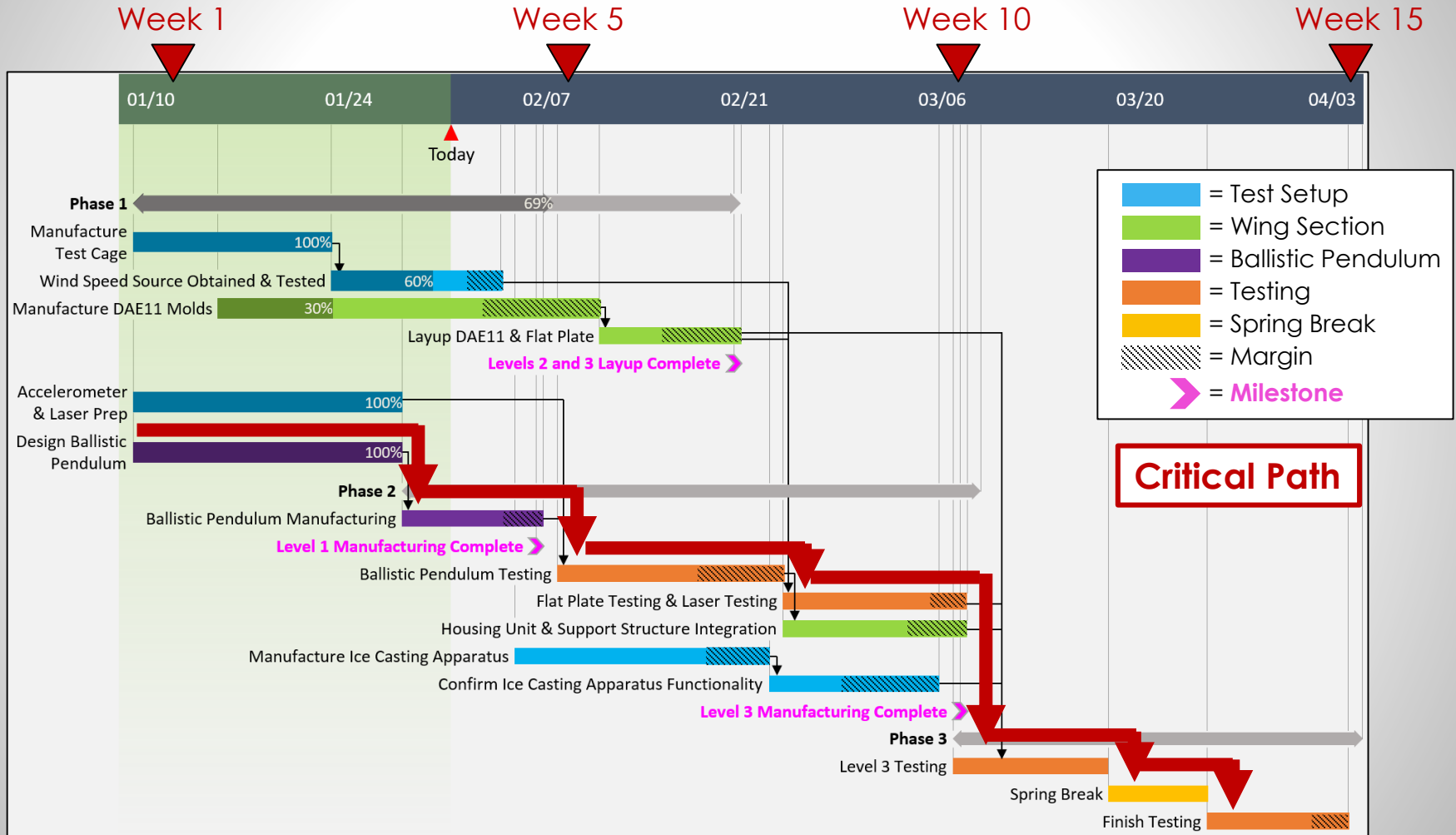
Previous Work Plan (CDR)

Spring 2016

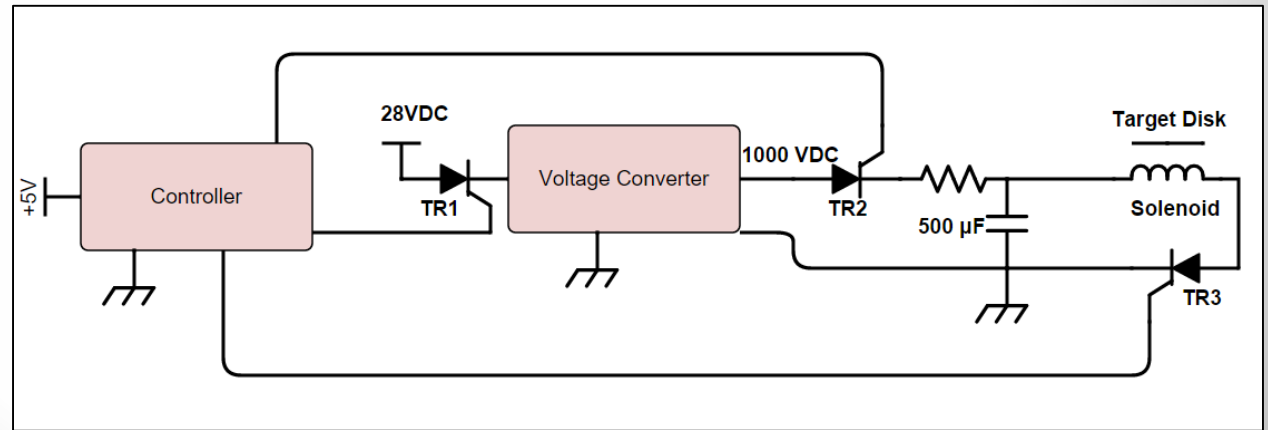
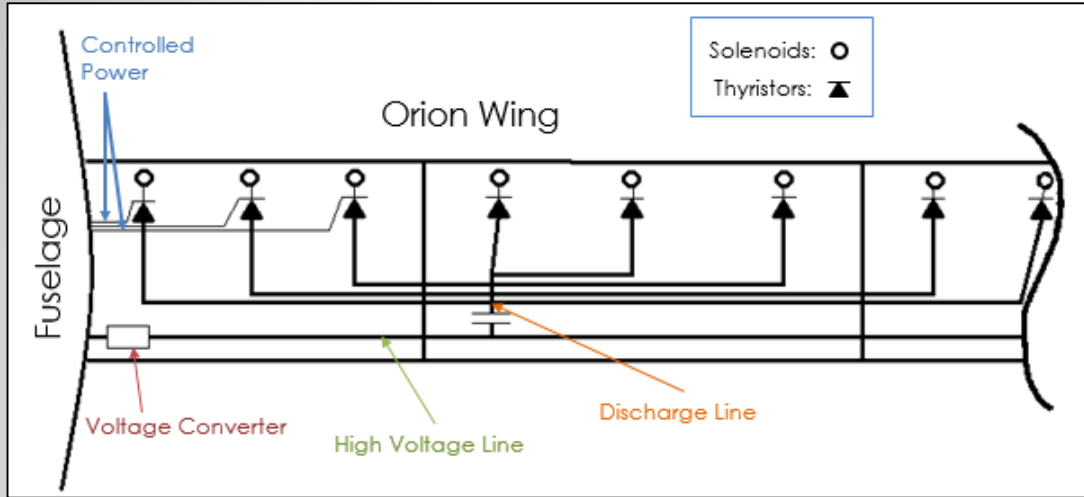


Current Work Plan Backup

Spring 2016



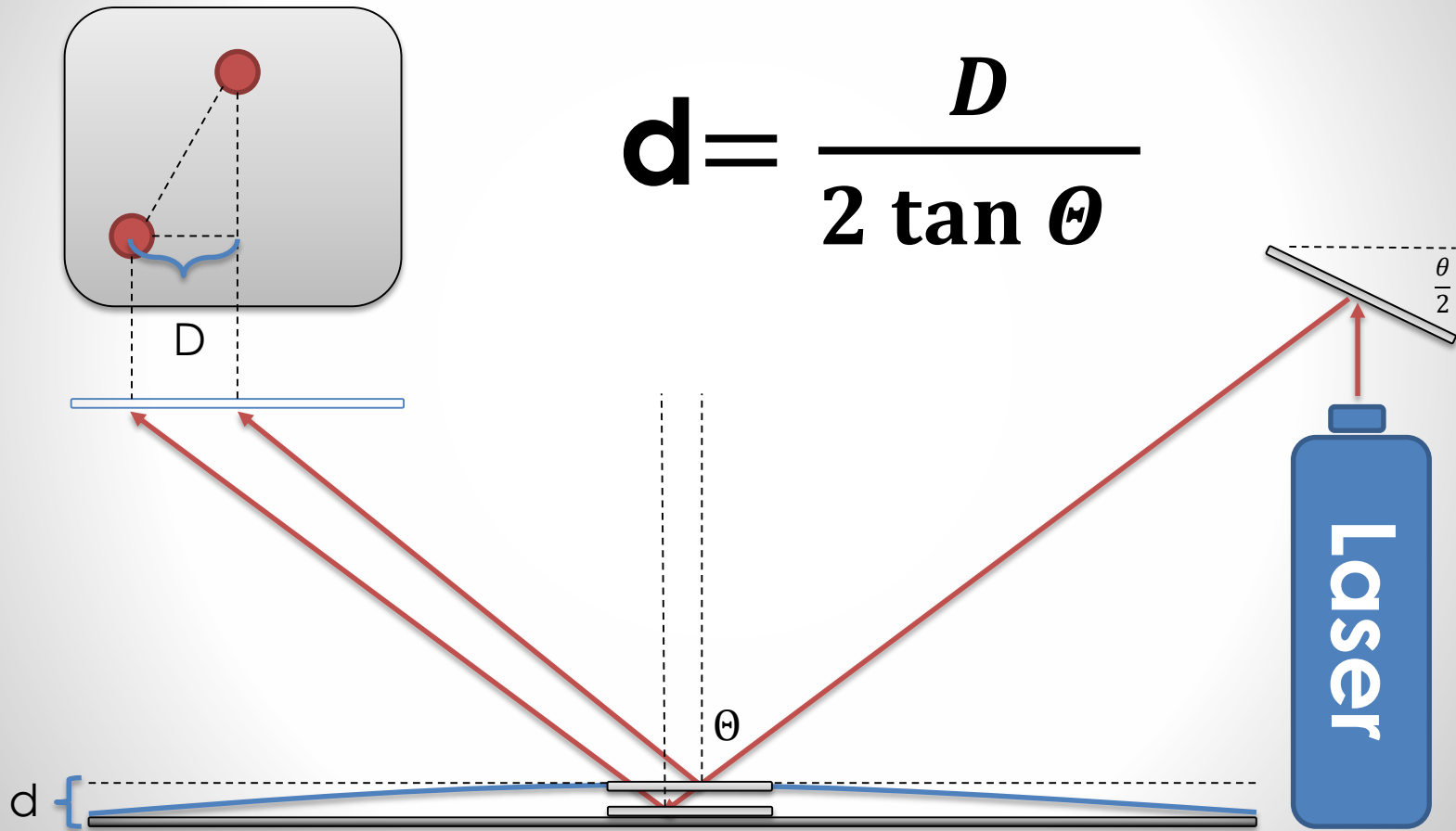
Full-span Backup



Full-span Weight Budget

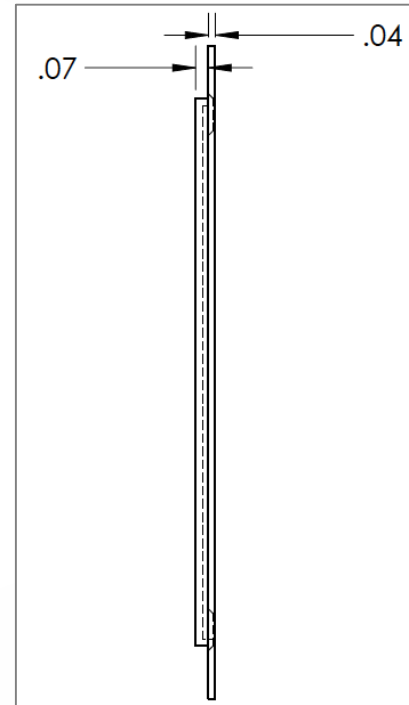
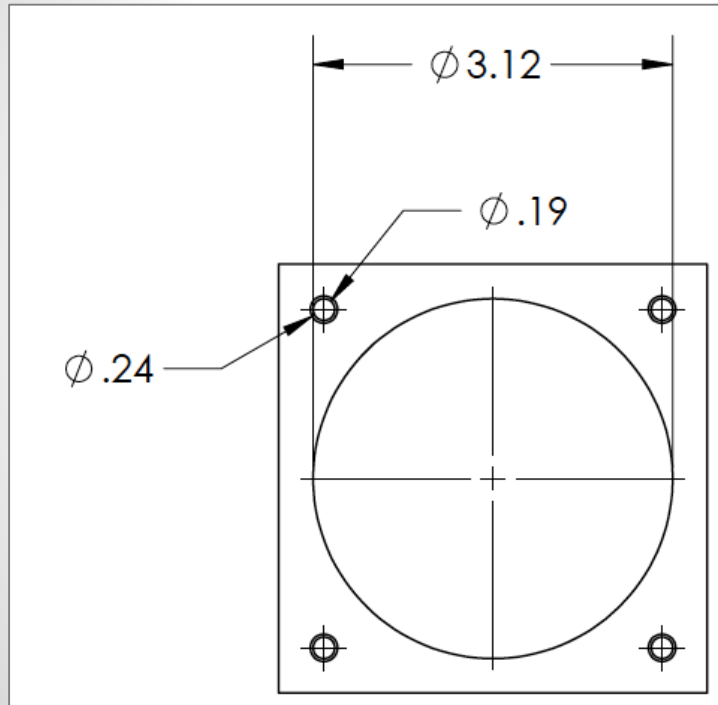
Item	Weight (lb)
Solenoids + Target Plates (76)	38.3
Housings (76)	69.3
Capacitors + Mounting (10)	27.2
Wire + Mounting	30.7
Voltage Converters	55.0
Total	221 lb

Sensors Backup



Wing Section Backup - Housing Unit

Housing Unit SolidWorks Designs

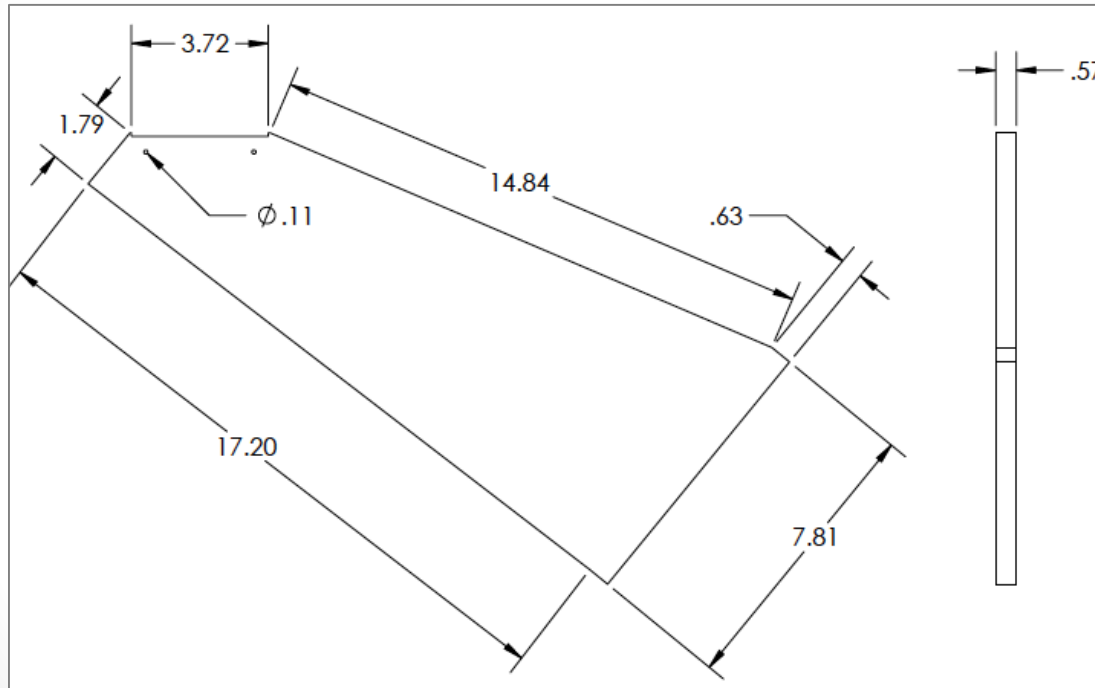


* All units in inches

Tolerance = 0.01 inches

Wing Section Backup - Support Structure

Support Structure SolidWorks Designs

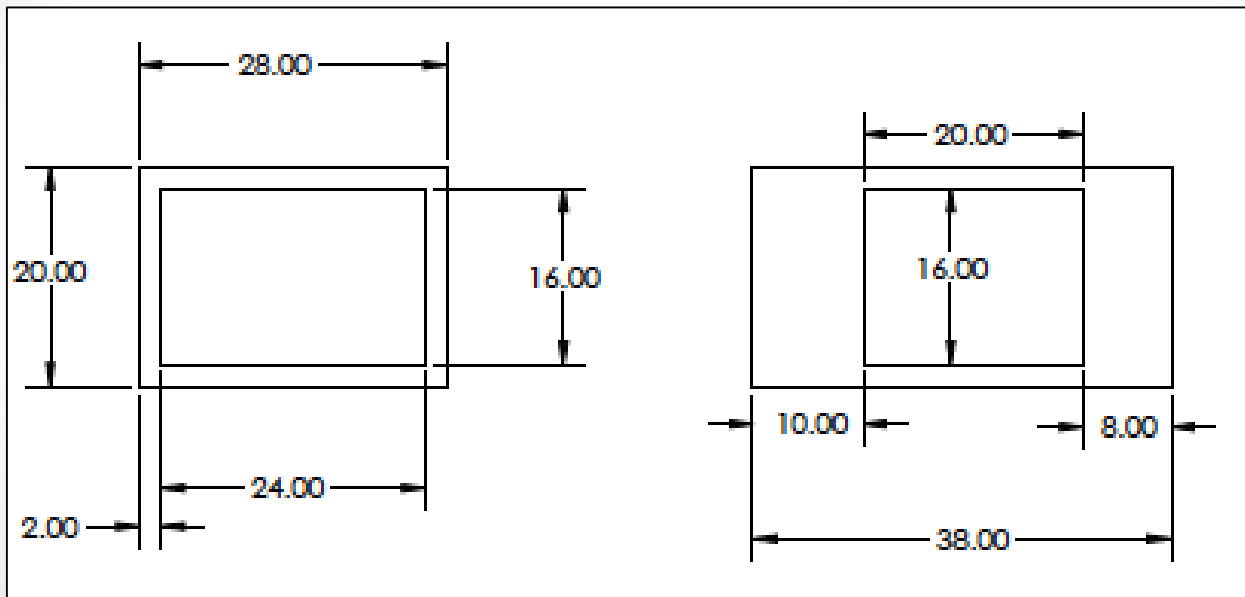


* All units in inches

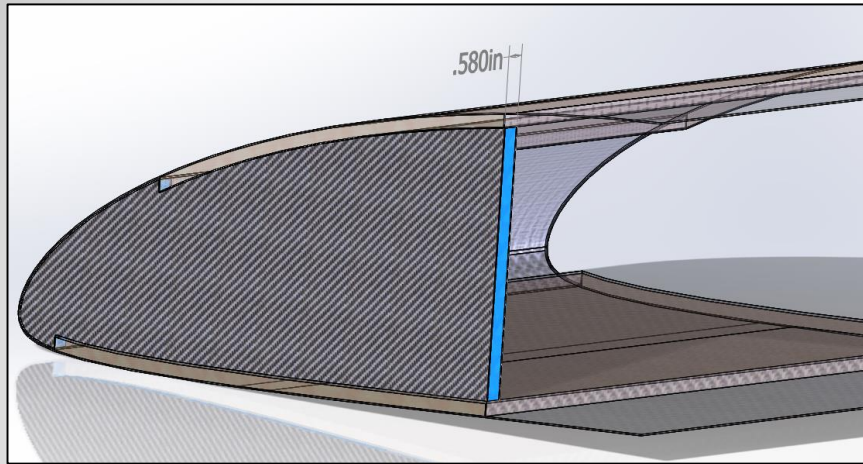
Tolerance = 0.01 inches

Test Cage Backup

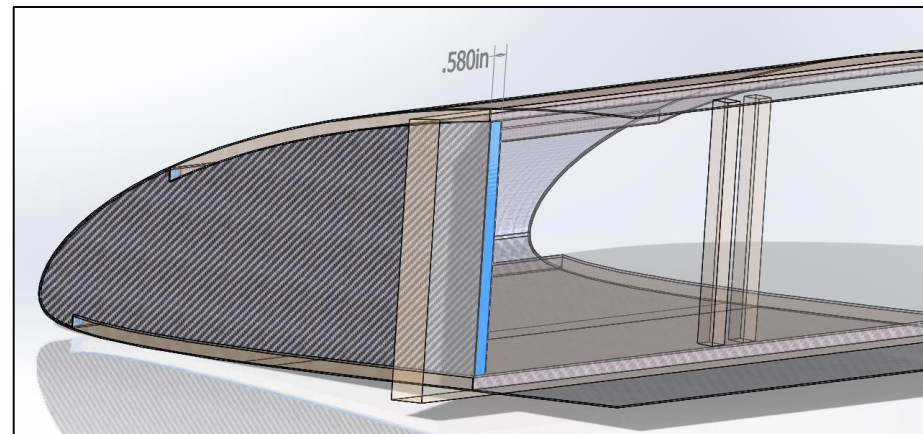
Test Cage (all units in inches)



Wing Section Rib Backup



Dragon Plate Rib
(Carbon Fiber plates with foam core)



Test Setup Backup

Leaf Blower Simulation

→ Two Leaf blowers simulated in test cage with exit velocity = 250 knots

