



# A.C.E.S.

*Air-breathing Cold Engine Start*



## Manufacturing Status Review



Alex Bertman, Jake Harrell, Tristan Isaacs, Alex Johnson, Matthew McKernan, T.R. Mitchell, Nicholas Moore, James Nguyen, Matthew Robak, Lucas Sorensen, Nicholas Taylor



A small graphic in the top left corner showing four aces of different suits (hearts, diamonds, clubs, spades) fanned out, with the ace of clubs featuring a turbine fan design.

# Outline

---

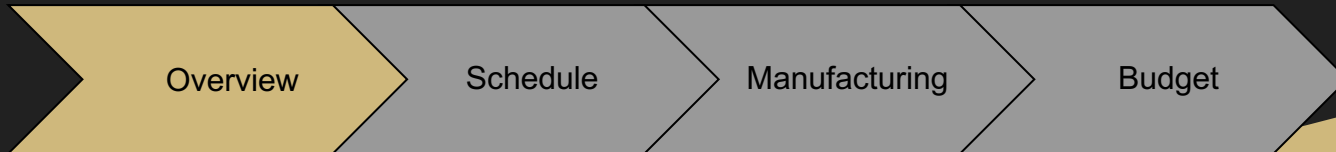
- Overview
- Schedule
- Manufacturing
- Budget





# Overview

3





# Project Description

---



- Design, build, and test a system to facilitate starting a JetCat P90-SXi jet engine at a temperature of  $-50^{\circ}\text{F}$  by:
  - Controlling the temperature and mass flow rate of the fuel into the engine
  - Ensuring that the engine electronics are within their operating temperature range
  - Ensuring that the heating system has sufficient power to heat the fuel delivery system and engine electronics
- Motivation
  - Air Force Research Lab (AFRL) conference
  - Proof of concept for high-altitude (cold-temperature) restart for jet-powered UAS







# Course Project Objectives

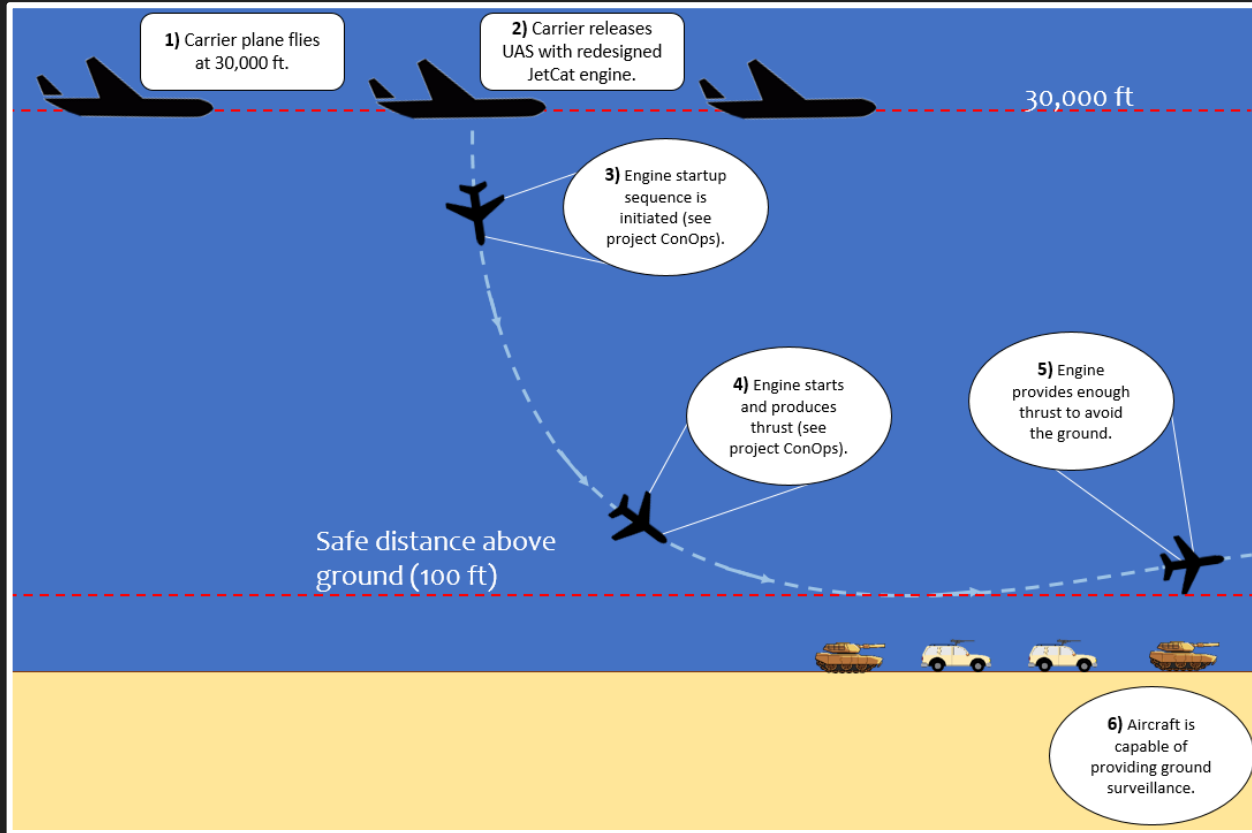


	<b>Fuel Delivery System (FDS)</b>	<b>Electronics Heating</b>	<b>Startup Time</b>	<b>AFRL Conference</b>
<b>Level 1</b>	System will control mass flow rate & temperature of fuel when placed in an environment cold-soaked to - 30°F.	The electronics will be heated to 60°F after being placed in an environment cold-soaked to - 30°F.	The fuel delivery and electronics heating systems objectives will be completed in less than 3 hours.	
<b>Level 2</b>	System will control mass flow rate & temperature of fuel when placed in an environment cold-soaked to - 40°F.	The electronics will be heated to 60°F after being placed in an environment cold-soaked to - 40°F.	The fuel delivery and electronics heating systems objectives will be completed in less than 1.5 hours.	
<b>Level 3</b>	System will control mass flow rate & temperature of fuel when placed in an environment cold-soaked to - 50°F.	The electronics will be heated to 60°F after being placed in an environment cold-soaked to - 50°F.	The fuel delivery and electronics heating systems objectives will be completed in less than 8 m 42 s	
<b>Level 4</b>				Entire system will be integrated with engine and successfully start within 3 hours.



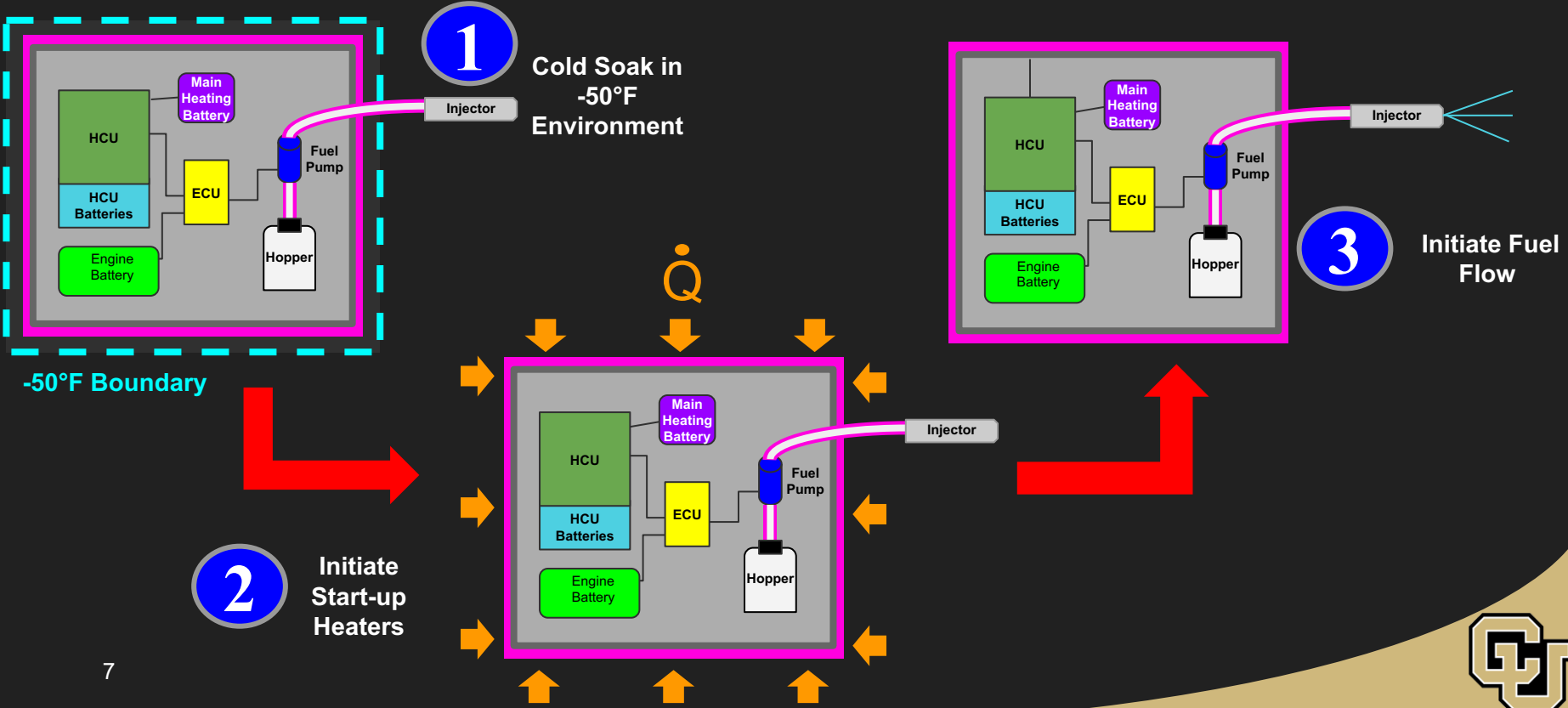


# Mission CONOPS





# Project Conops

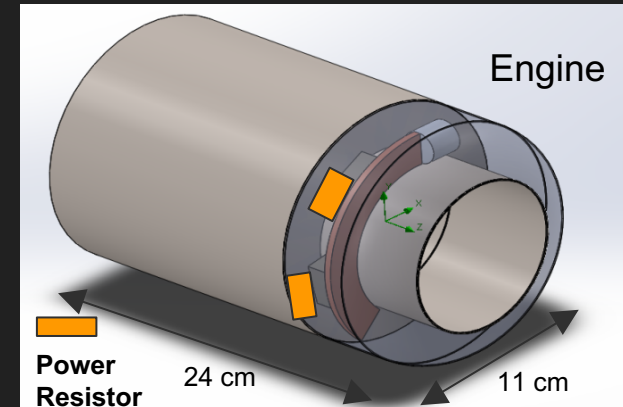
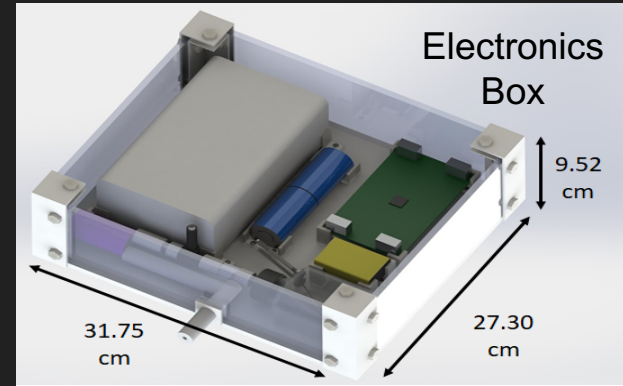




# Baseline Design

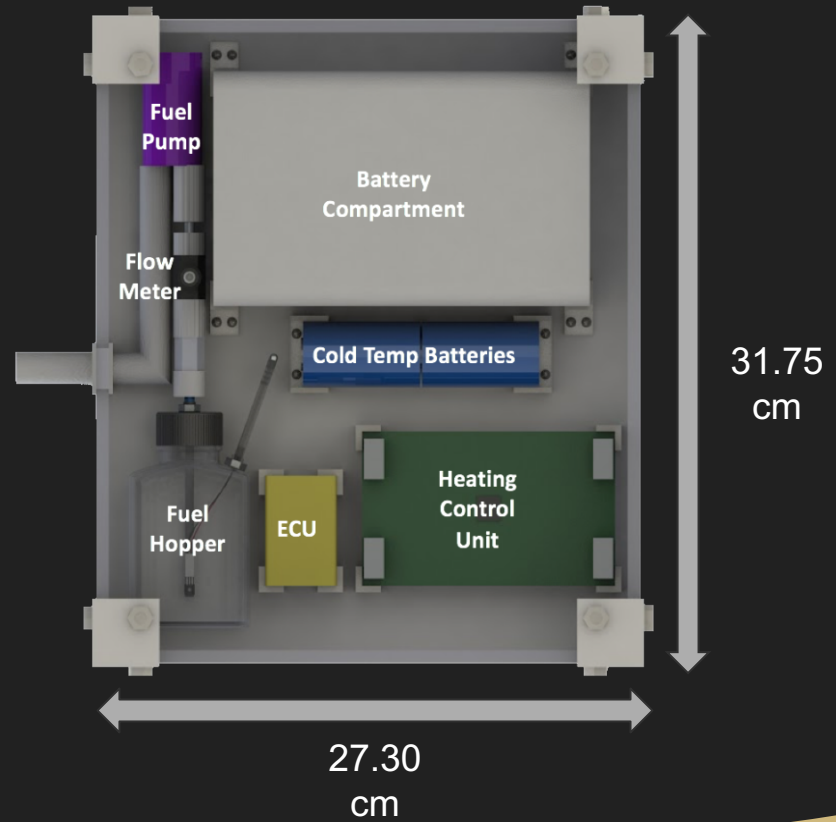


- **Initial Energy:** Cryogel insulated battery
- **Fuel Delivery System:** Resistive heating wire wrapped around fuel delivery components insulated in Cryogel
- **Electronics Heating:** Resistive heating within manufactured plastic box, ESB heated by power resistors in engine cowling
- **Heating Control Unit (HCU):** Microcontroller powered by cold temperature batteries, controls temperatures and fuel flow rate



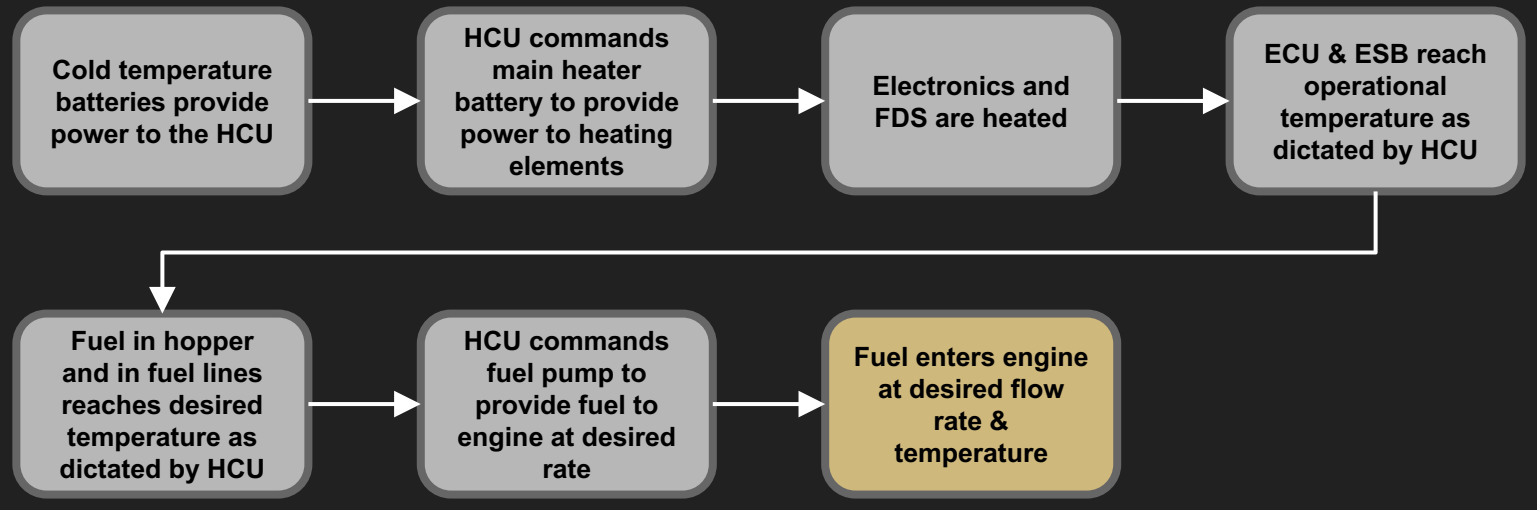


# Baseline Design



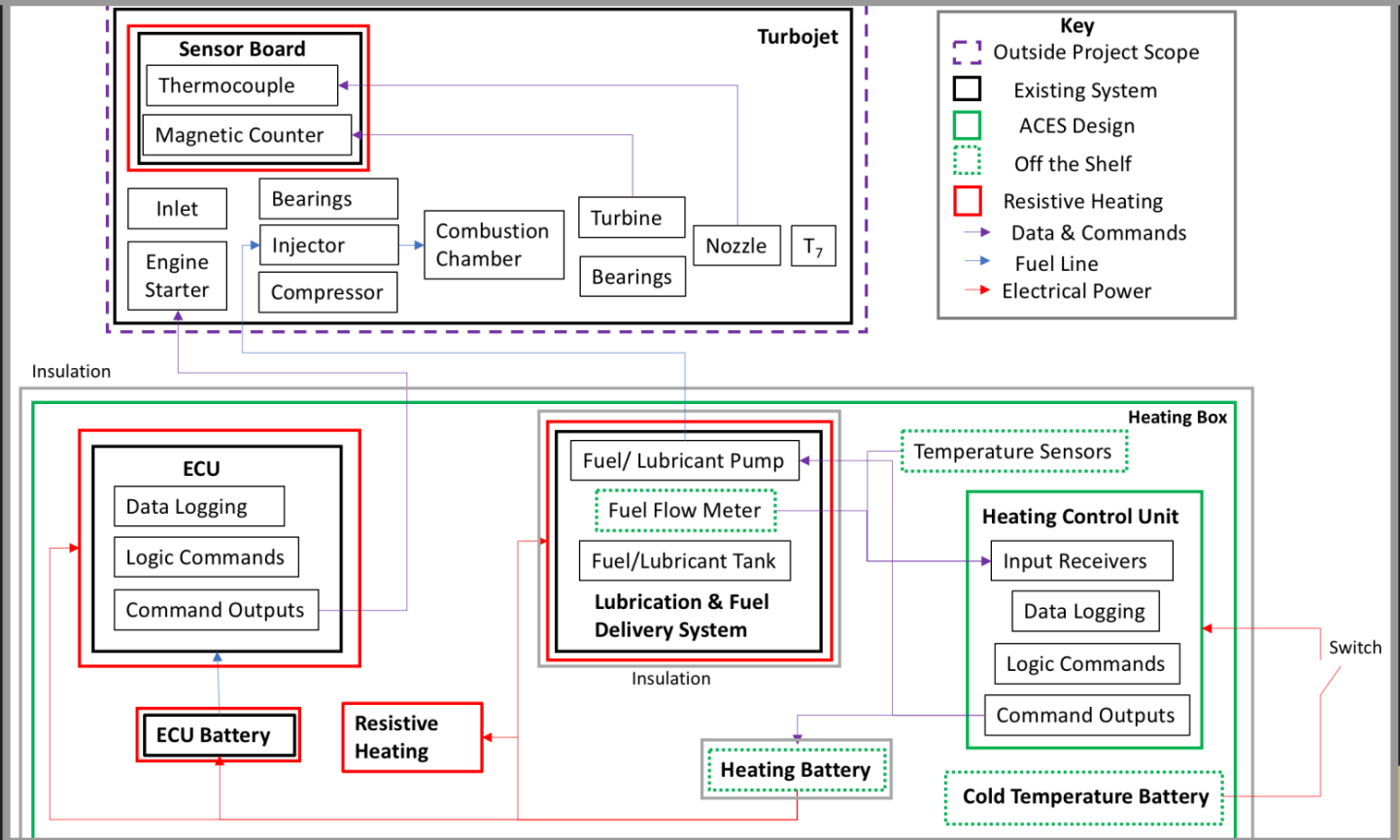


# Process Flow Diagram





# Functional Block Diagram





# Critical Project Elements

---



CPE 1: Temperature of main heater battery must be  $> 30^{\circ}\text{F}$  and  $< 122^{\circ}\text{F}$

CPE 2: ECU, ESB, and engine battery must be  $60^{\circ}\text{F}$  while not exceeding  $122^{\circ}\text{F}$  for the battery and  $150^{\circ}\text{F}$  for the ECU and ESB

CPE 3: Temperature of fuel in fuel delivery system must be  $> 60^{\circ}\text{F}$  and  $< 115^{\circ}\text{F}$

CPE 4: A Heating Control Unit (HCU) must control the mass flow rate of fuel and heating systems



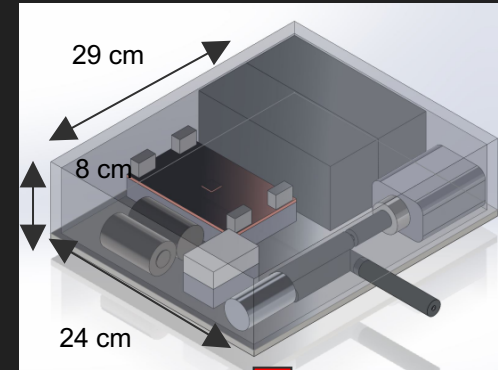




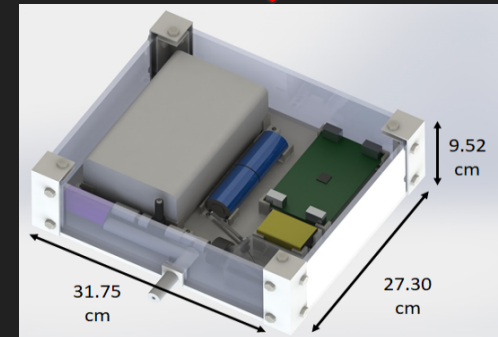
# Executive Summary



- Design Changes Since CDR
  - Multiple HCU PCB changes
    - Removed slave microcontroller
  - Refined electronics housing design
  - Changed resistive wire wrapping pattern
  - Added flow meter to CAD model
  - Refined fuel delivery system design
- Schedule
  - HCU PCB schedule has slipped
  - **With margin, still on schedule**
- Budget
  - **\$2037.71** spent to date
  - **On budget**



CDR Design

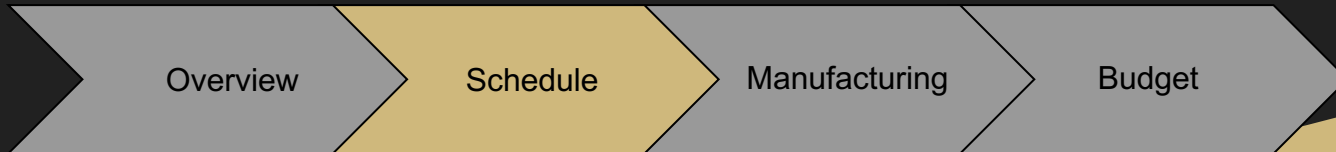


Current Design

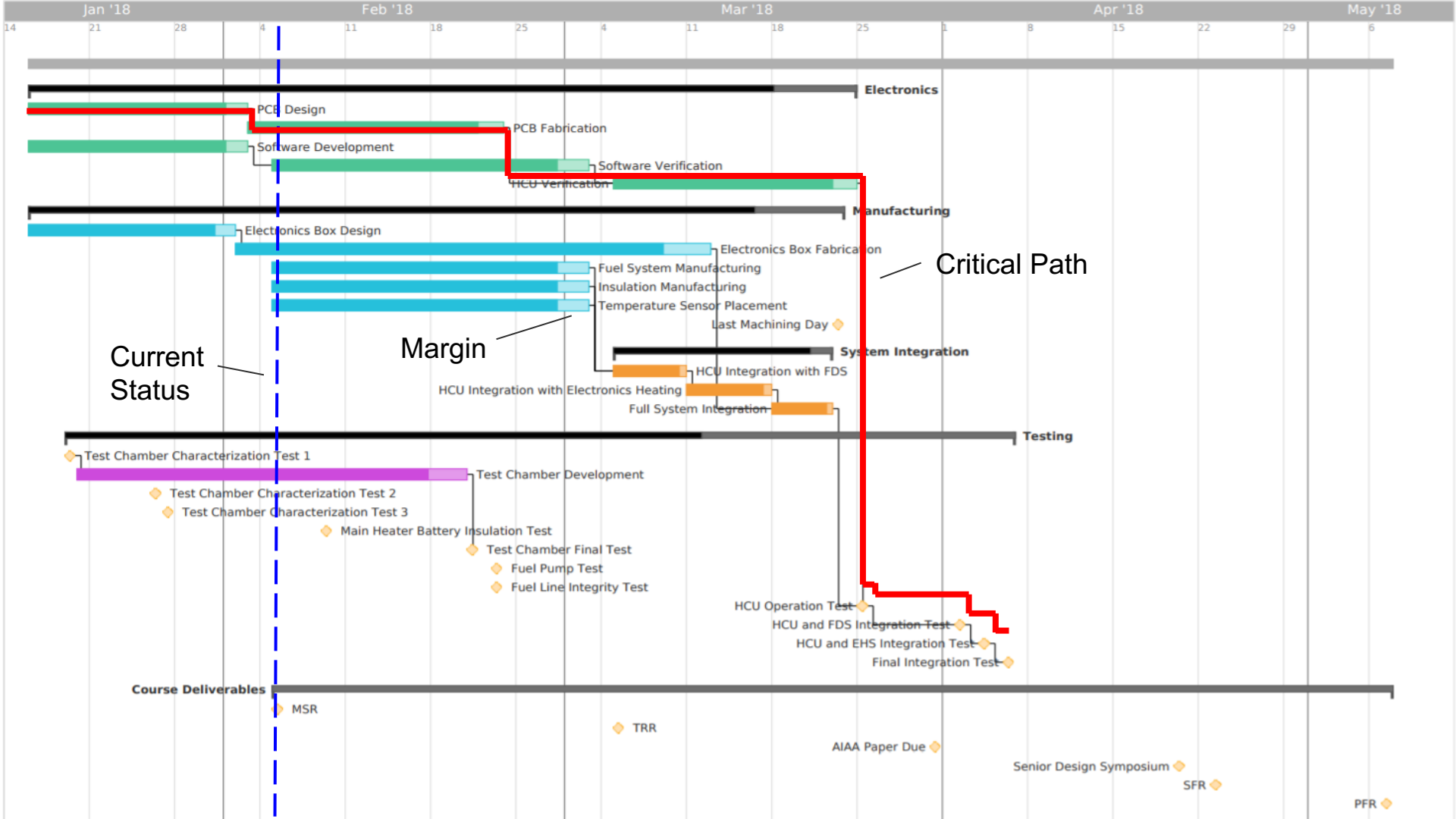




# Schedule





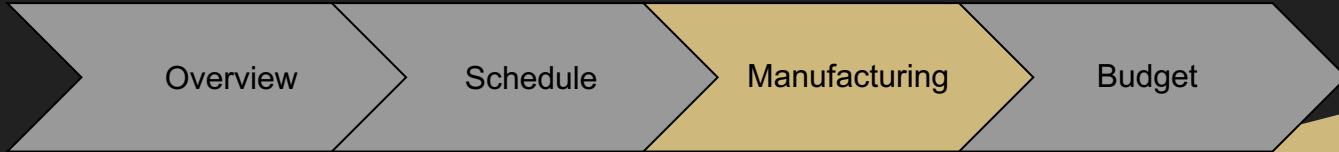
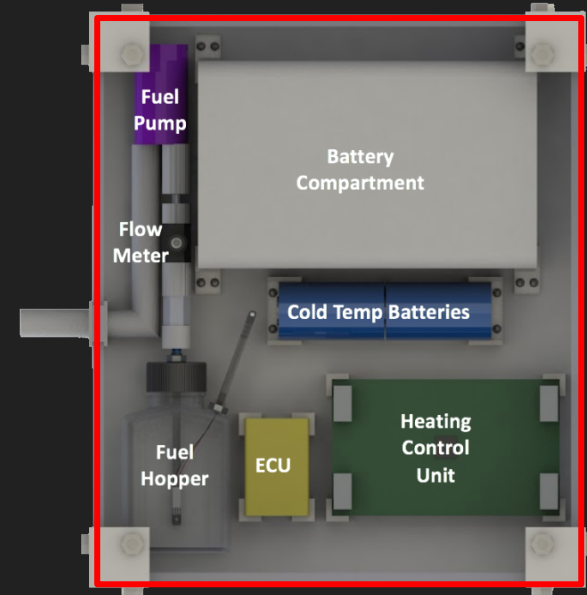




# Manufacturing

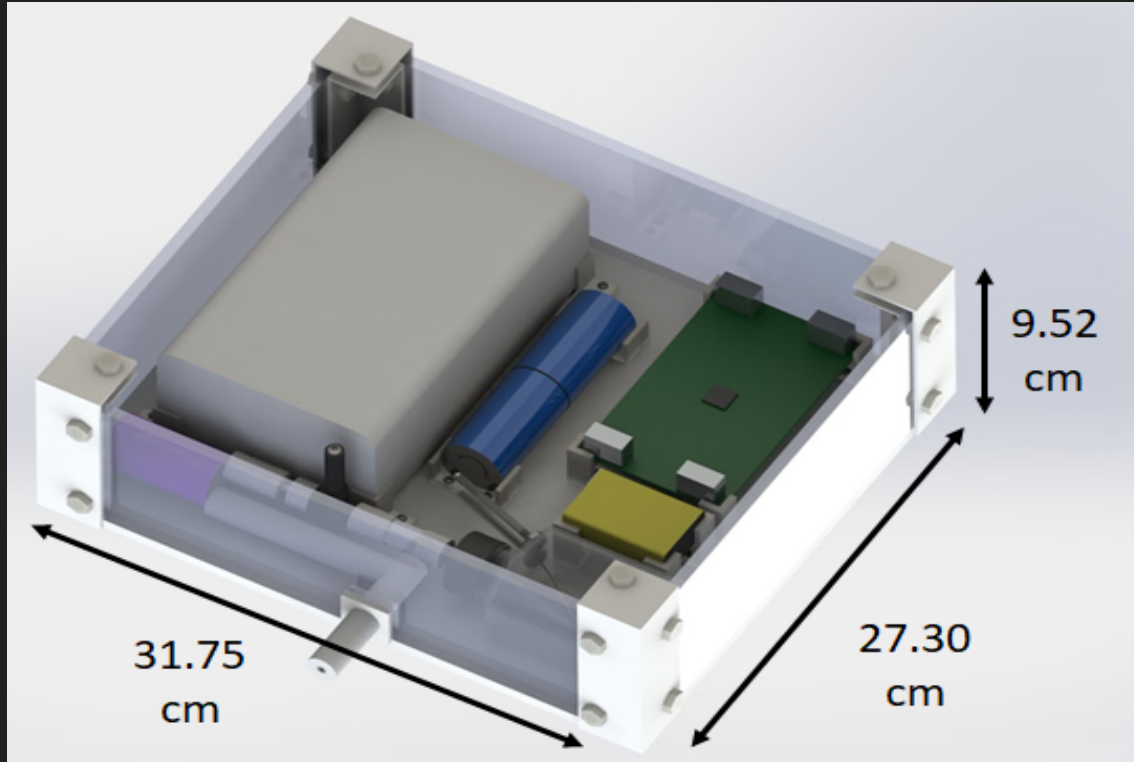


Electronics Box  
HCU Hardware  
HCU Software  
Testing Apparatus





# Electronics Box





# Electronics Box ProGrEss



## Status:

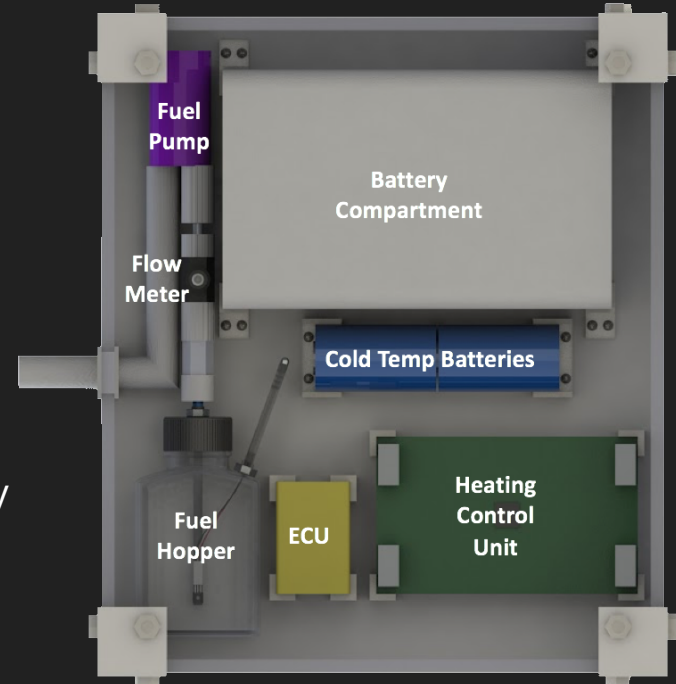
- Electronics box fully designed, including joints and mounting brackets
- Component layout has been finalized

## Tasks Still to be done:

- Updating of thermal simulations (for test comparisons)
- Layout design for wiring connections
- 3D Printing L-brackets, corner brackets, and mounting brackets
- Confirm resistor wire length after updated thermal simulations
- Construction of shell prototype and iterate as necessary

## Expected Lead Times:

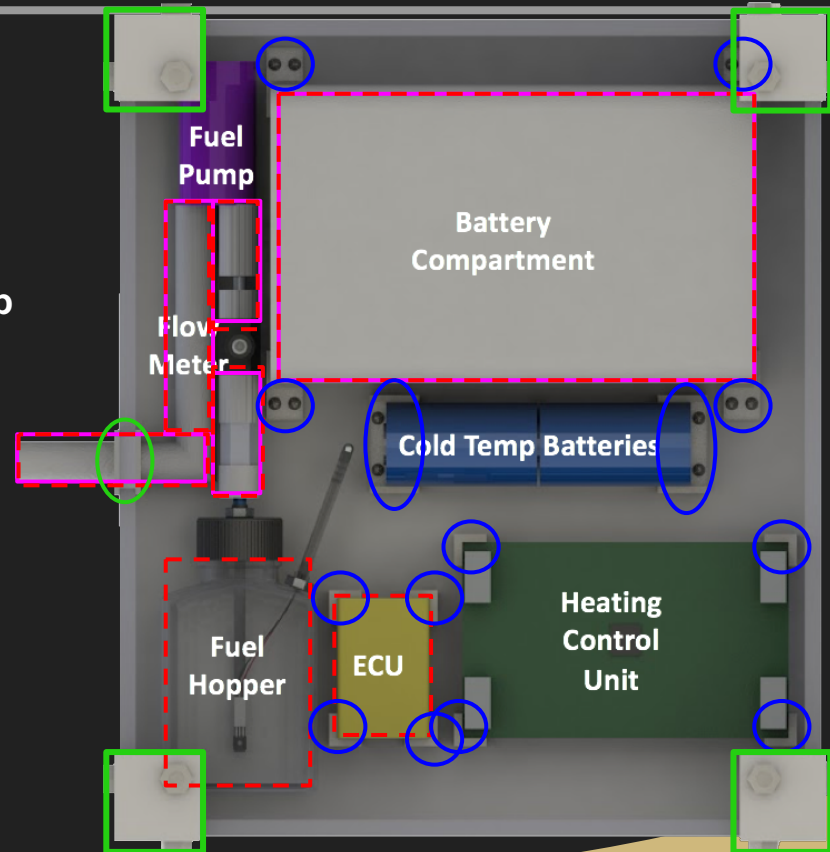
- 3D Printing Brackets: 1 week,
- Prototype: 2 weeks,
- Final Construction: 4 weeks





# Components

- Insulation
- Resistive heating elements
- Internal Component Mounts
- Electronics Box Corner Brackets & Clip







# Insulation Manufacturing

## Status:

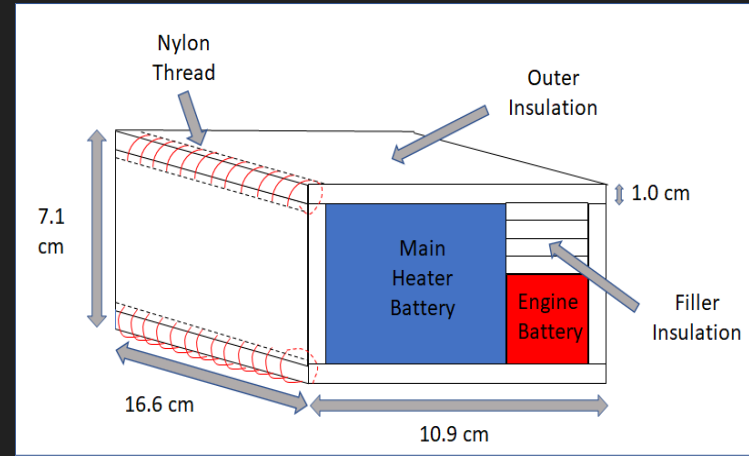
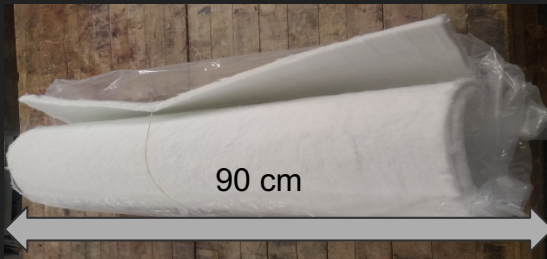
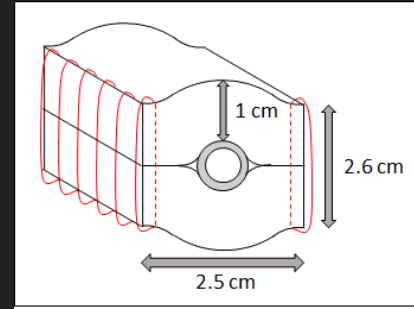
- Cryogel purchased and received
- Cryogel pieces will be sewn together; see diagrams

## Concerns:

- Cannot be bent easily at small scale
- Produces more dust than anticipated
- Airborne dust is an irritant but not hazardous

## Mitigation Strategies:

- Wear gloves, eye protection, long sleeve clothing, and respirator during manufacturing
- Wrap insulation in cellophane





# Resistive Elements Manufacturing



## Status:

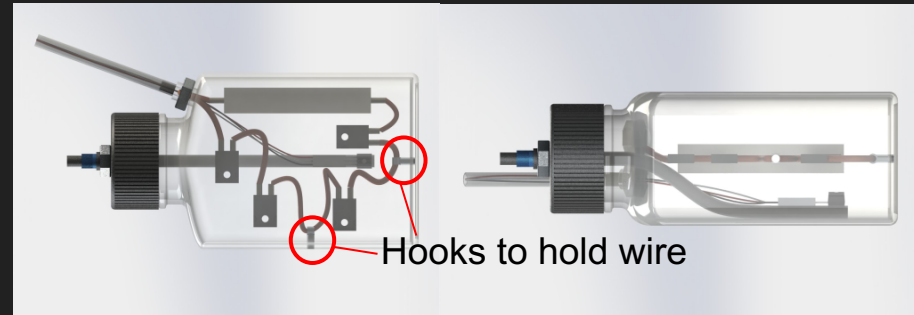
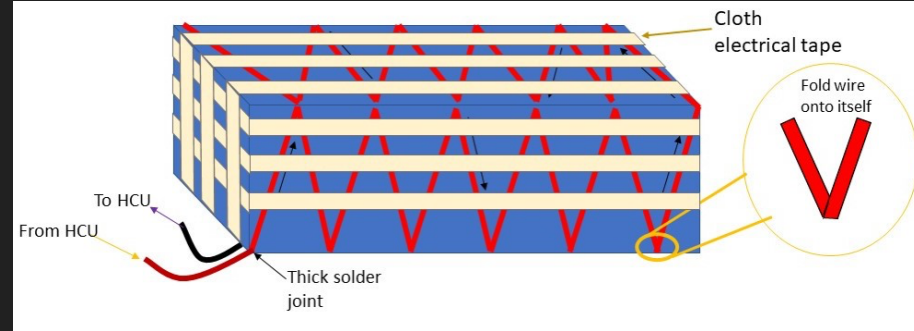
- Resistive wire ordered, ETA 2-3 wks
- Power resistors to be ordered, 1-2 wks lead time
- Wire will be wrapped around electronic components & batteries as shown on right
- Power resistor configuration within hopper shown on right

## Concerns:

- Induced magnetism from wire coils could affect electronics and batteries
- Little room inside fuel hopper to place resistors

## Mitigation Strategies:

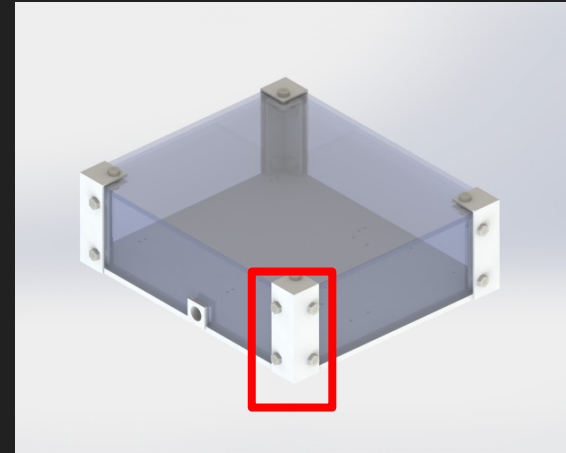
- V shaped wrap pattern eliminates wire coils around electronic components
- Bulky fuel filter within hopper removed, replaced with inlet at bottom of hopper





## Electronics Box Hardware - brackets

- 3D printed with ABS plus-P430 thermoplastic using uPrint SE Plus printer
  - *Alternative option: PolyLite ABS plastic with Lulzbot Taz5 printer*
- All dimensions in millimeters
- Tolerances for all dimensions are +/- 1 mm

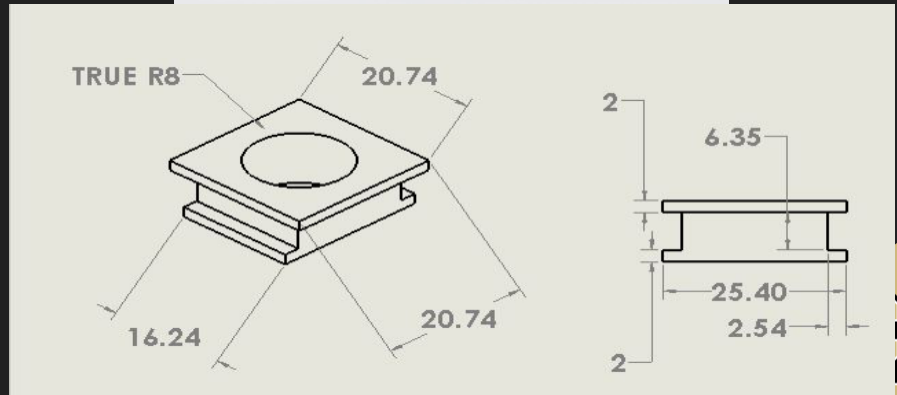
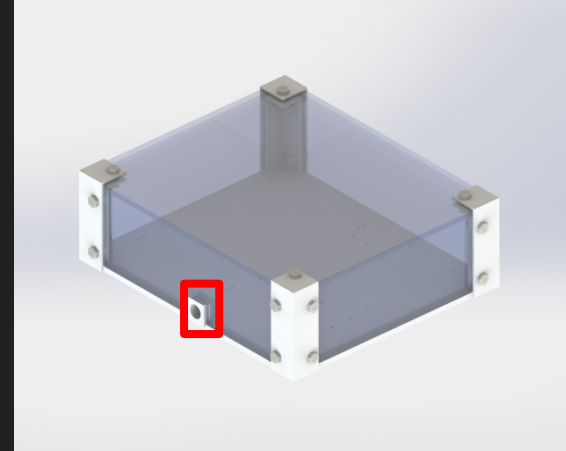




## Electronics Box Hardware - Clip



- 3D printed with ABS plus-P430 thermoplastic using uPrint SE Plus printer
  - *Alternative option: PolyLite ABS plastic with Lulzbot Taz5 printer*
- All dimensions in millimeters
- Tolerances for all dimensions are +/- 1 mm





# TEMPERATURE SENSORS

## Status:

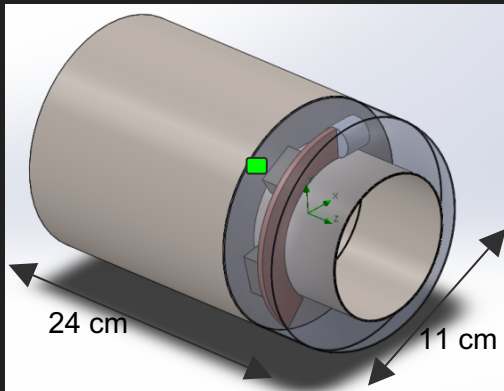
- **TMP-36 temperature sensors purchased and received**
- Sensors will be placed as shown in diagrams
- Sensors will be mounted using low temperature adhesives

## Concerns:

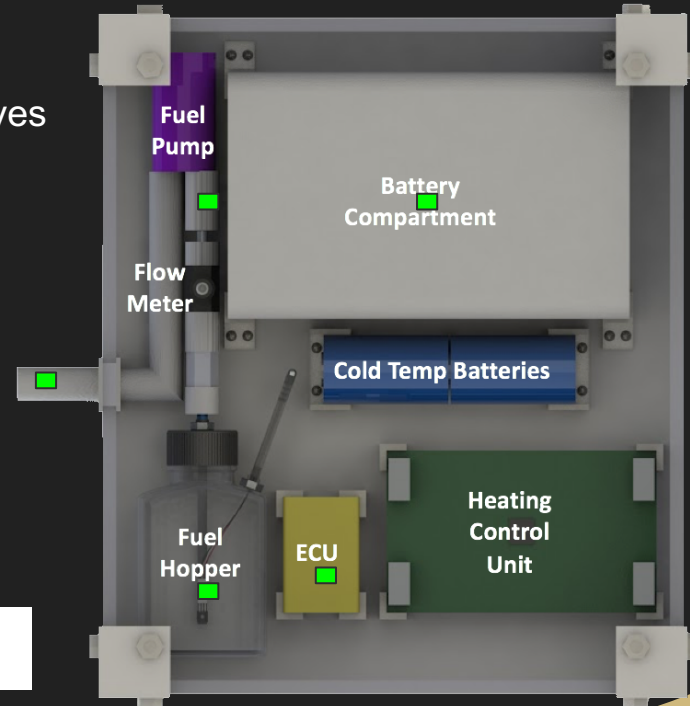
- Noise between sensors and HCU

## Mitigation Strategies:

- Digital sensor with onboard ADC an option to eliminate noise concerns



■ TMP36

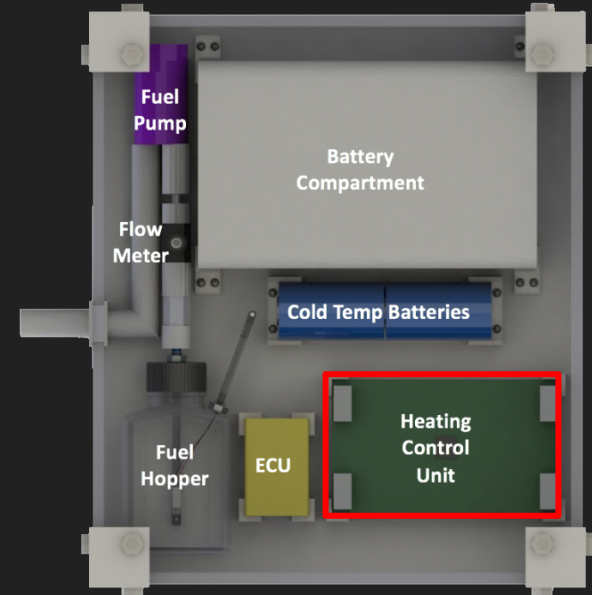




# Manufacturing



Electronics Box  
HCU Hardware  
HCU Software  
Testing Apparatus





# HCU PCB Fabrication



## Status:

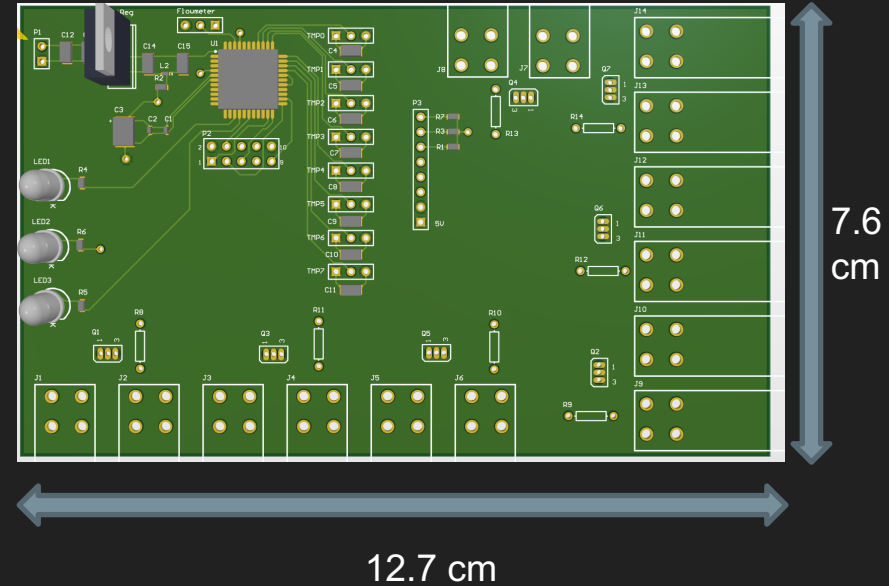
- Board completely designed in Altium
- Submitted to Advanced Electronics this week (lead time 1 week)

## Concerns:

- Population of the board will produce errors.
- Noise will produce inaccuracies with the ADC.

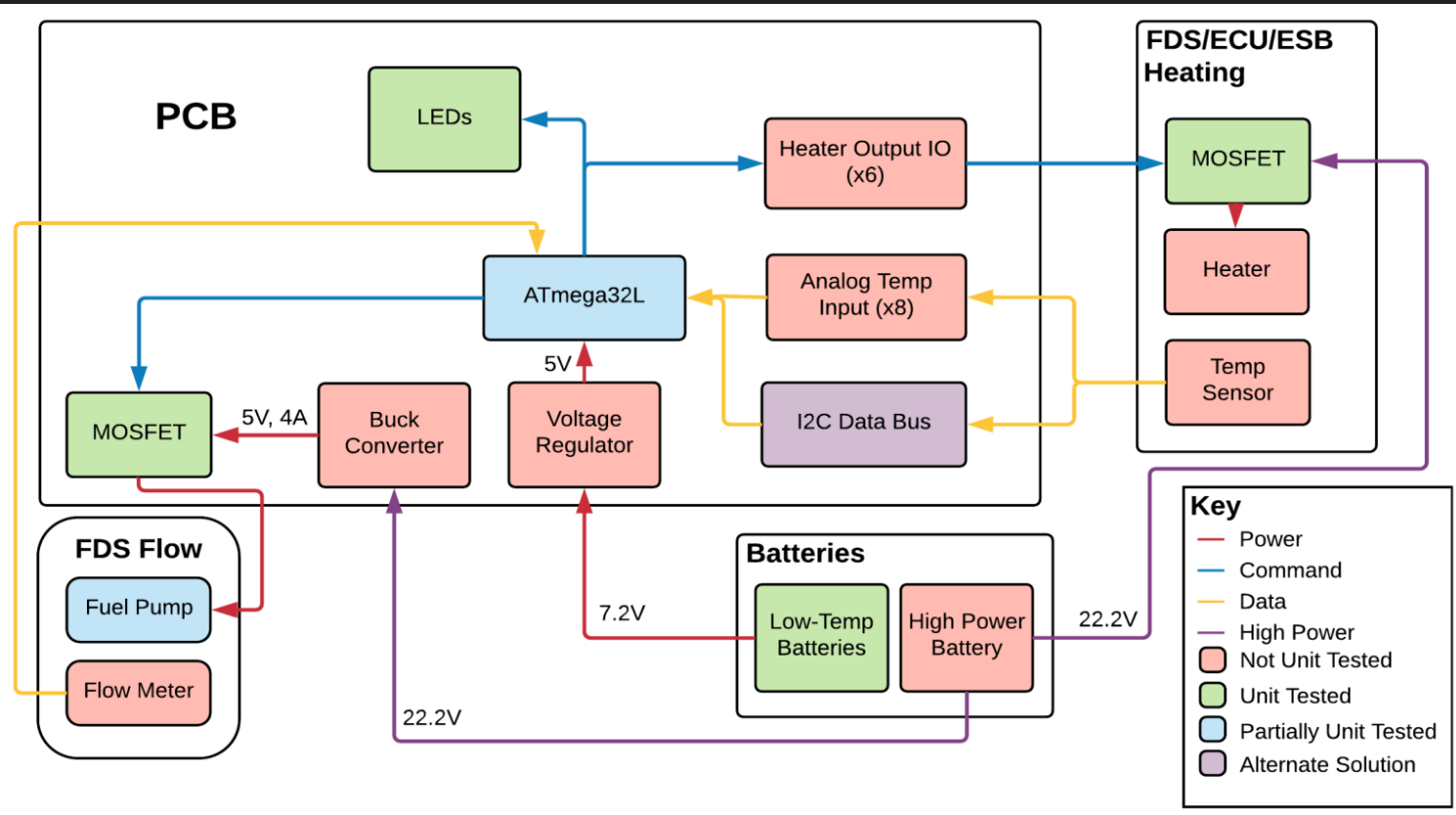
## Mitigation Strategies:

- Population of the board will occur in the ITLL or Bobby's lab (w/ supervision)
- $0\ \Omega$  resistors have been placed within the design to allow for debugging capabilities.





# Heating Control Unit: FBD



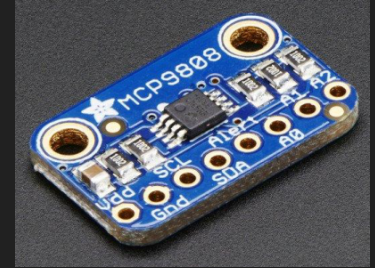




# HCU: Biggest Concerns



- **Noise in the analog temperature sensors.**
  - Long lead wires from sensor location to board (ESB->HCU) will act like antennas.
- **Solution**
  - Use temperature sensors which on board ADC which will then communicate with the HCU over I<sup>2</sup>C lines.
- **Magnetic field caused by resistor wire coils**
  - Faraday's Law states that a magnetic field will be induced by a coil of wire with a current flow through it.
- **Solution**
  - Surround component in Faraday Cage and electrically insulate from resistor wire.



From adafruit.com

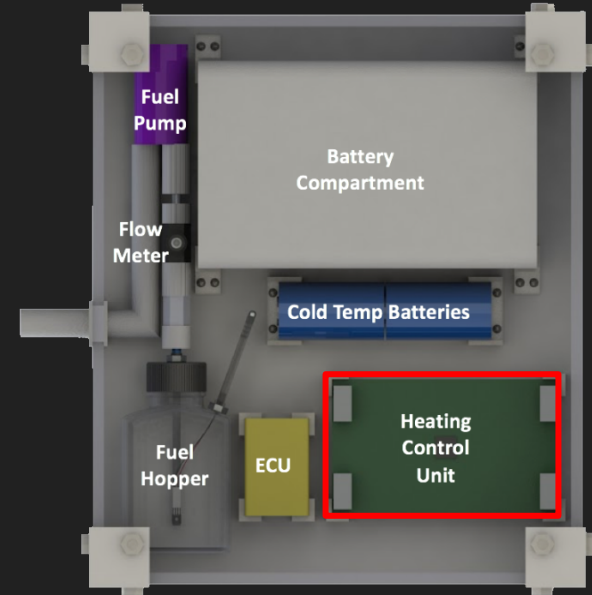




# Manufacturing



Electronics Box  
HCU Hardware  
HCU Software  
Testing Apparatus





# HCU Software Development



## Status:

- Code has been completely written
- Only Alive\_LED has been unit tested

## Concerns:

- Relying on one person for code writing and debugging could result in unresolved bugs

## Mitigation Strategies:

- Documentation for the code is being autogenerated by **Doxygen**
  - Will **greatly decrease the amount of time debugging** as other group member will be able to understand the code much faster
- Debugging wire harness is due to arrive by Mon Feb. 5 (JTAG)

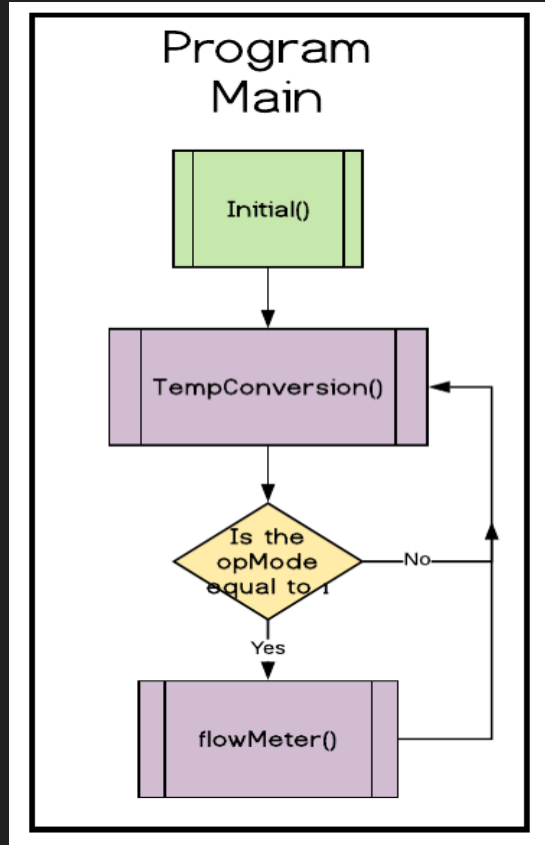




# HCU: SoftwARE



1. Performs TempConversion() on loop until all components reach desired temperature.
2. Performs TempConversion() and flowMeter() during the pumping phase.
3. TempConversion() only once fuel is exhausted.

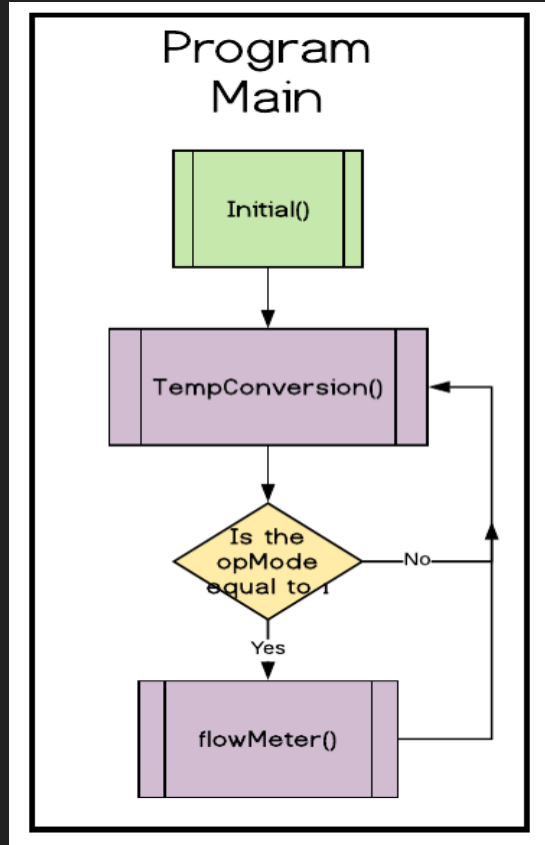




# HCU: SoftwARE



- TempConversion()
  - Measures temperature of all sensors
  - Compares temperature to desired and toggles MOSFETs for heating circuits
- flowMeter()
  - Counts pulse train from flow meter
  - Compares flow rate to desired and alters PWM duty cycle





# HCU: SoftwARE Timing



- Software timing is not an issue

- There are no exact timing constraints for code execution
- The only ISR's are very short and will be completed within ~10 clock cycles
  - ISR for Alive\_LED
  - Flow meter pulse train ISR
- All 6 temperature conversions should occur within **3 ms**
  - This was tested in the Atmel Simulator, built in feature of the IDE
- Flow meter code should take **0.262 sec**
  - This is because the sampling period is this amount of time.

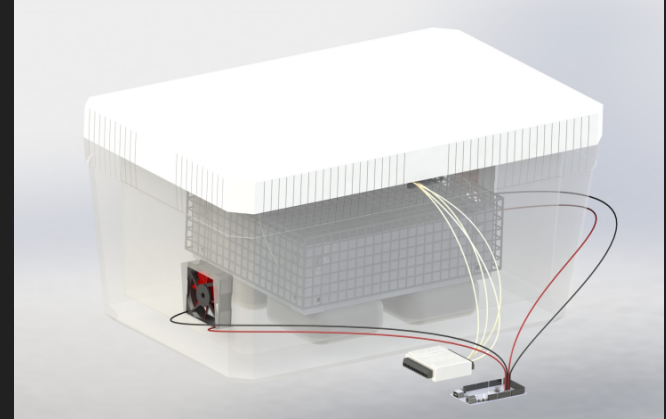




# Manufacturing



Electronics Box  
HCU Hardware  
HCU Software  
Testing Apparatus



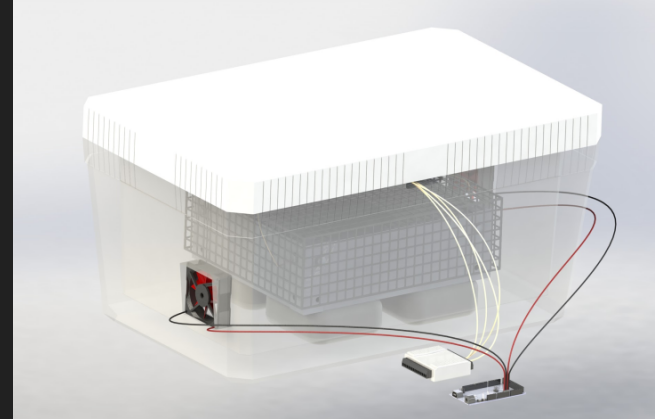
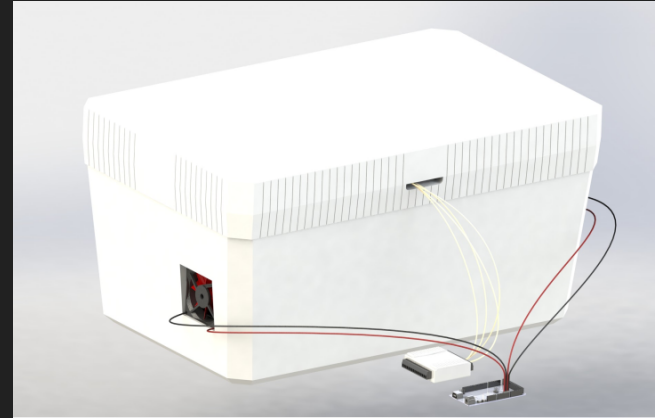


# Test Apparatus

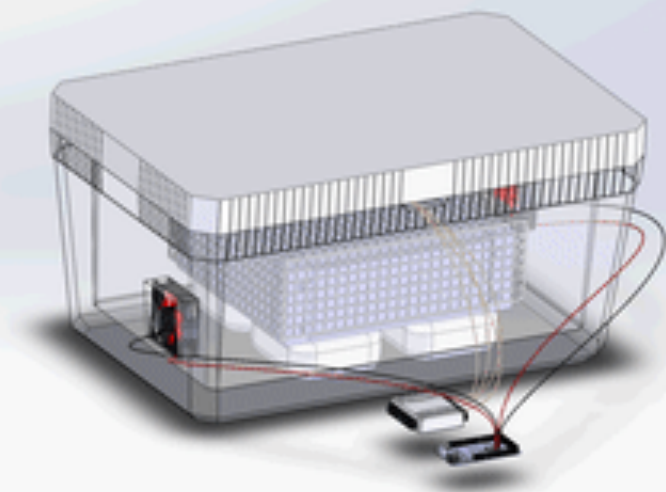


## Summary:

- Dry ice chamber with **active temperature control**
- Capable of sustaining  **$-50 \pm 5^\circ\text{F}$  for 1 hour**
- Ability to house **full electronics box** and engine
- Computer fans for temperature control
- Thermocouples for temperature validation and model verification
- Fully constructed









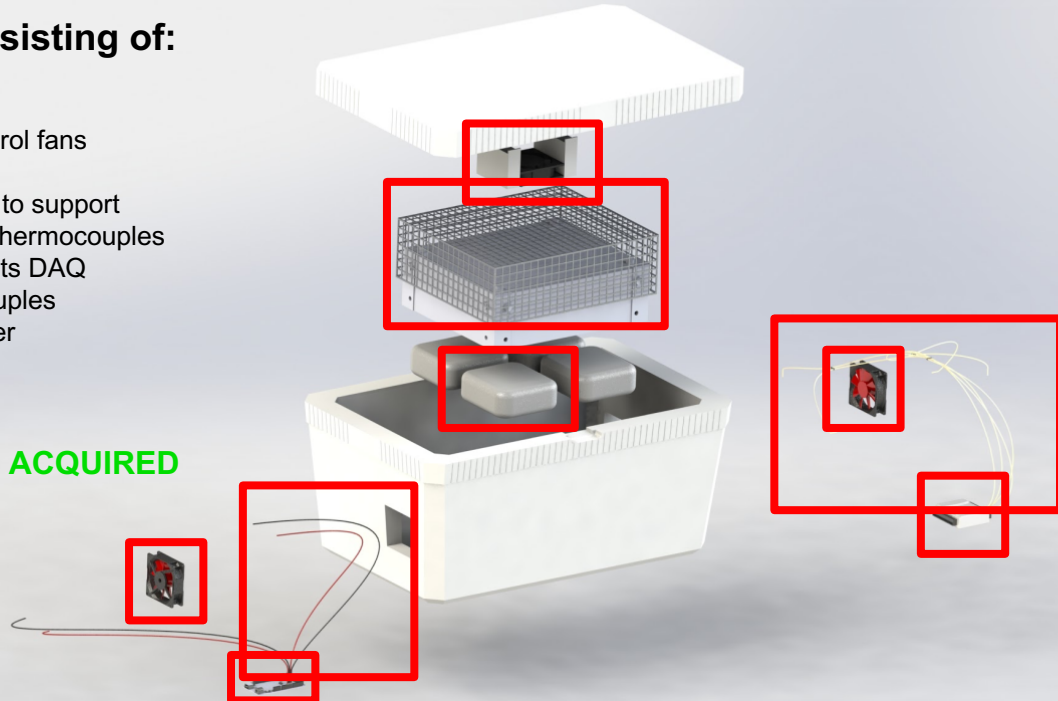
# Test Chamber Components



## Foam cooler consisting of:

1. Dry ice blocks
2. 2 temperature control fans
3. 1 circulation fan
4. Chicken wire cage to support additional ice and thermocouples
5. National Instruments DAQ
6. 4 type K thermocouples
7. Wiring for fan power
8. Arduino MEGA

**ALL COMPONENTS ACQUIRED**





# Test Apparatus Progress



## Status:

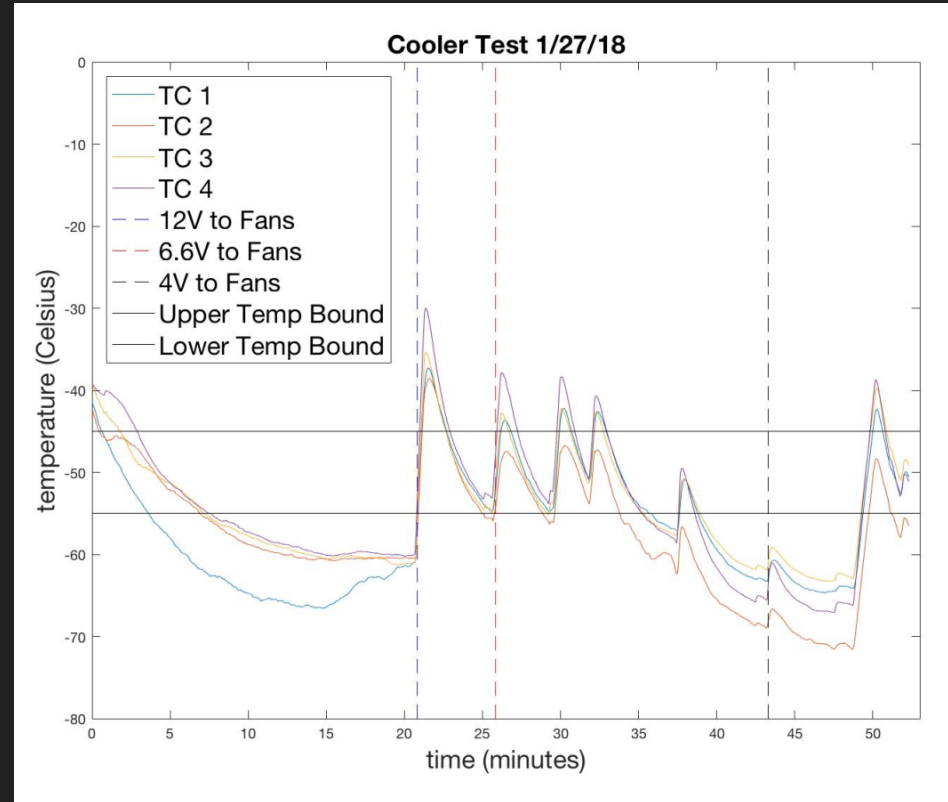
- Conducted initial trials with test chamber (see plot for results)
- **Test chamber feasibility demonstrated**
- **Test chamber fully constructed**

## Concerns:

- Test chamber warms too quickly to control when fans are operated
- Fans become very cold, could cease to function

## Mitigation Strategies:

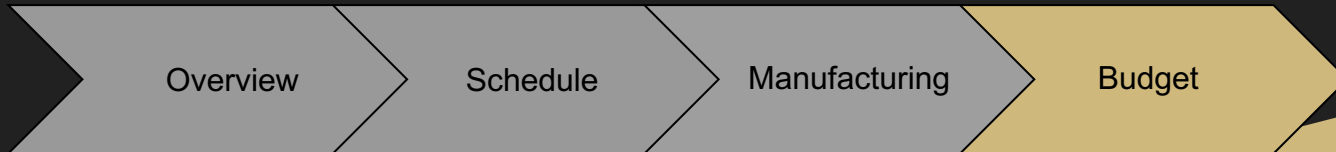
- Arduino fan control to be implemented
- Fans can be replaced between tests





# Budget

40

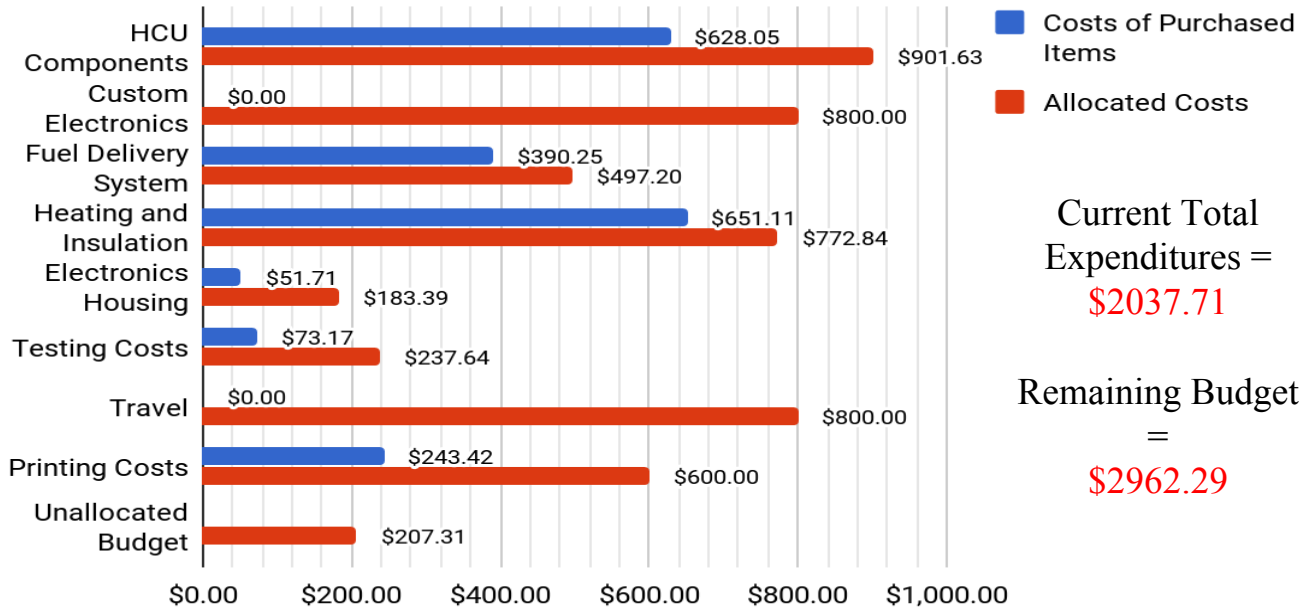




# Cost Plan



## ACES Purchases vs Allocated Budget



Current Total Expenditures =  
**\$2037.71**

Remaining Budget =  
**\$2962.29**





# Critical Parts Received



Part	Cost (Including Shipping)
Fuel Pump with Cable	\$383.90
ATMEGA32L, In circuit Debugger, Adapter Cable	\$290.58
Cryogel Insulation	\$319.84
Resistive Heaters	\$249.91
Altium License	\$124.01
Turnigy Heavy Duty Lip (Heater Battery)	\$81.36
HCU Batteries	\$92.91
<b>Total</b>	<b>\$1,542.51</b>





## Pending/Planned orders



Item	Price
PCB Printing	\$300
3D Printing L-Brackets	\$150 (U-print)
4 Thermocouple Breakout Boards	\$60
Miscellaneous Electronic Components	\$100
Custom Electronics	\$800
Travel Expenses	\$800
Printing	\$300
<b>Total</b>	<b>\$2,510</b>





## Lead Time for Ordered Components



Component	Lead Time
Resistive Wire	2-3 Weeks (Already Ordered)
PCB Printing	1 Week per iteration
3D Printing L-Brackets	1 Day per iteration
Thermocouple Breakout Board	1-2 Weeks







# Questions?





# BACKUP SLIDES



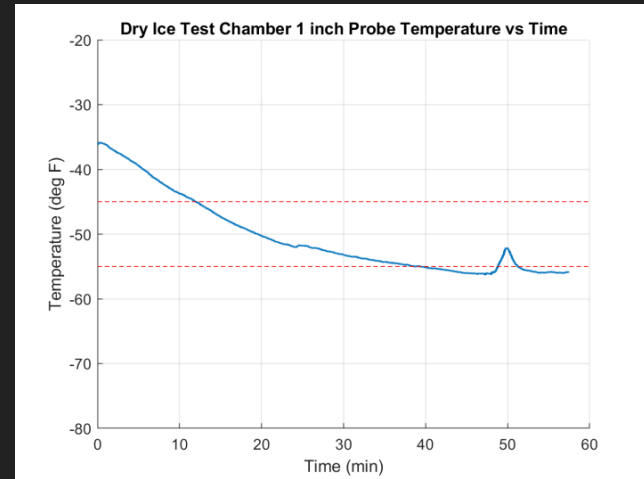


# Test Apparatus Progress



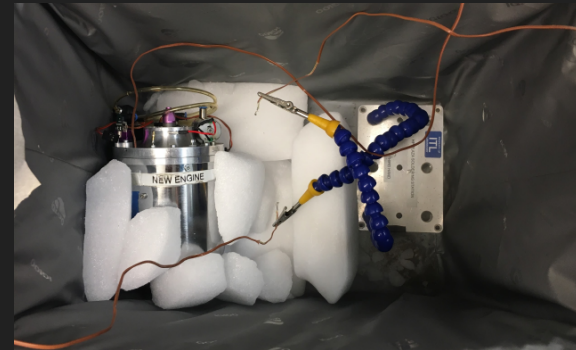
## First Run 1/19:

- Initial test conducted to establish feasibility of dry ice test chamber
- No active temperature control
- Two thermocouples placed with 5 inch vertical displacement
  - Placed at 1 inch and 6 inches above dry ice
  - Easy to maintain one, but proved test-bed was unhomogenized



## Takeaways and lessons learned:

- Issues with temperature control and uniform temperature distribution
- Surprisingly good temperature consistency
- **Sublimation a non issue**
- **Active temperature control necessary**





# Functional Requirements

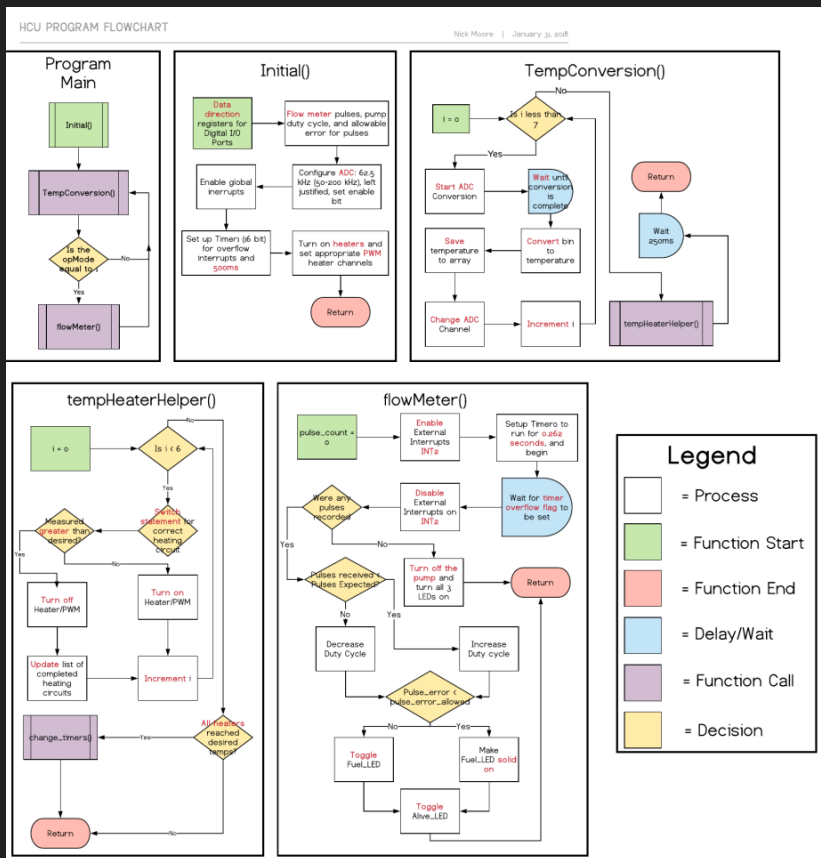


- **FR 1) ENERGY:** An initial energy source shall provide adequate power for the fuel delivery system heating and electronics heating.
- **FR 2.1) FDS:** The Fuel Delivery System shall provide a specified fuel flow rate from 0 to 4.8 g/s  $\pm$  0.13 g/s to the engine.
- **FR 2.2) FDS:** The Fuel Delivery System shall provide fuel at a specified temperature from 60 to 110°F  $\pm$  3.6°F to the engine.
- **FR 3) Electronics Heating:** The electronics (ECU, ESB, batteries) shall be heated to their operating temperature of 60°F.
- **FR 4.1) HCU:** The Heating Control Unit (HCU) shall monitor and regulate the temperature of the electronic components and fuel delivery heating systems.
- **FR 4.2) HCU:** The Heating Control Unit (HCU) shall monitor and regulate the mass flow rate of the fuel delivery heating system.



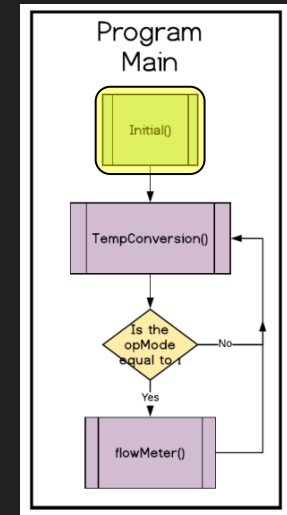
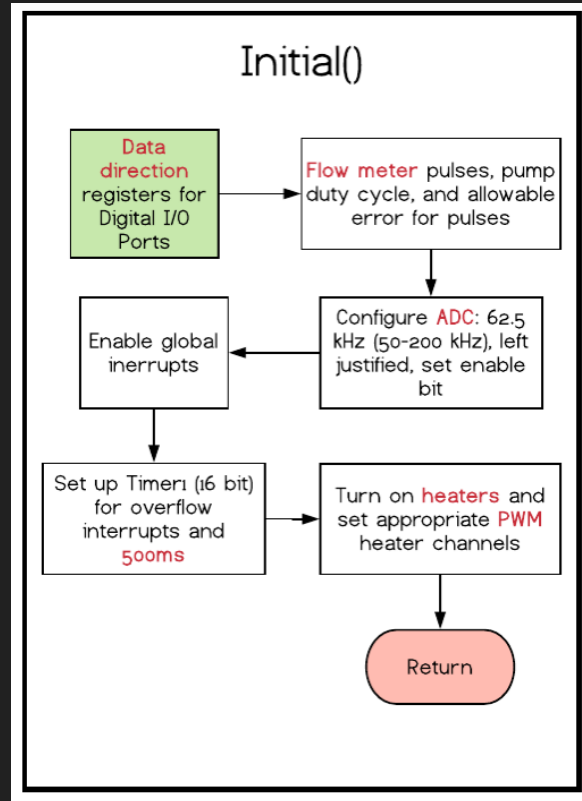


# HCU SOFTWARE OVERVIEW



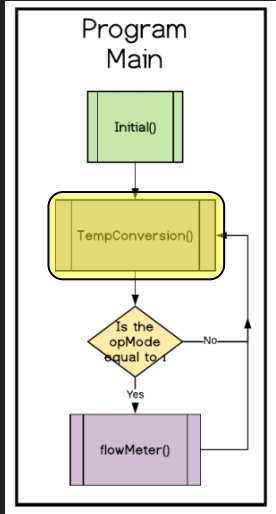
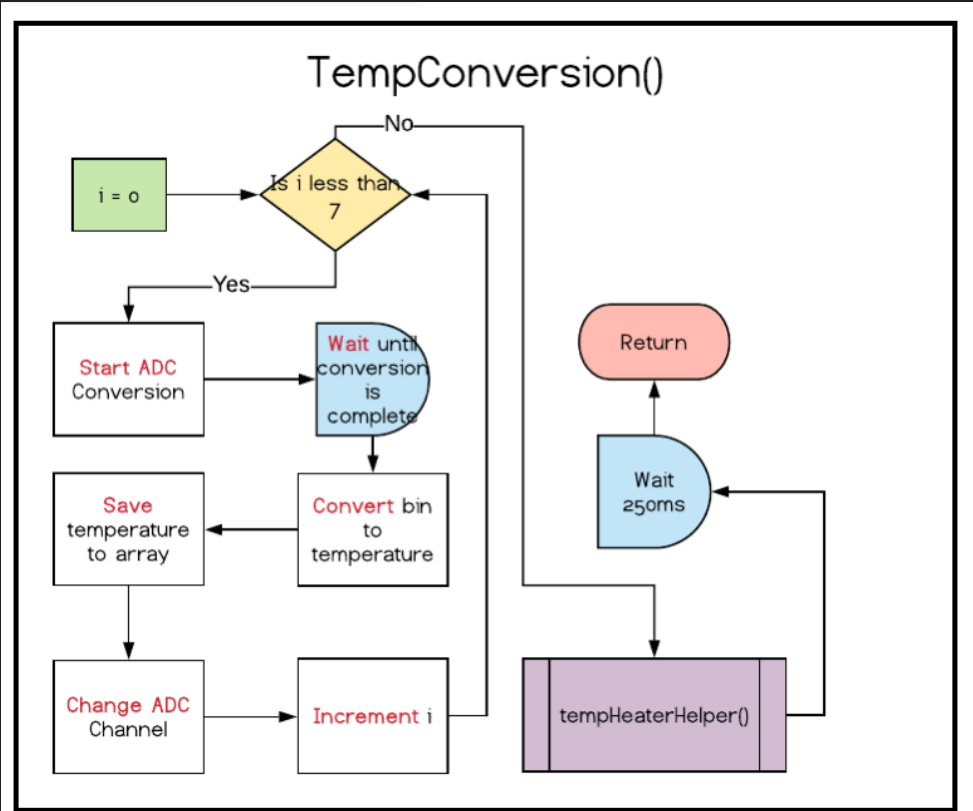


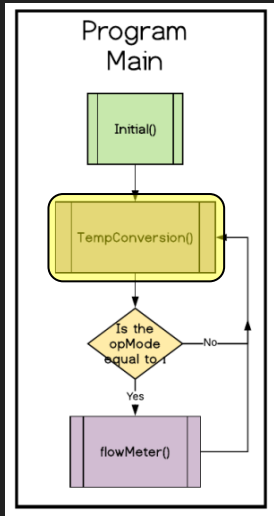
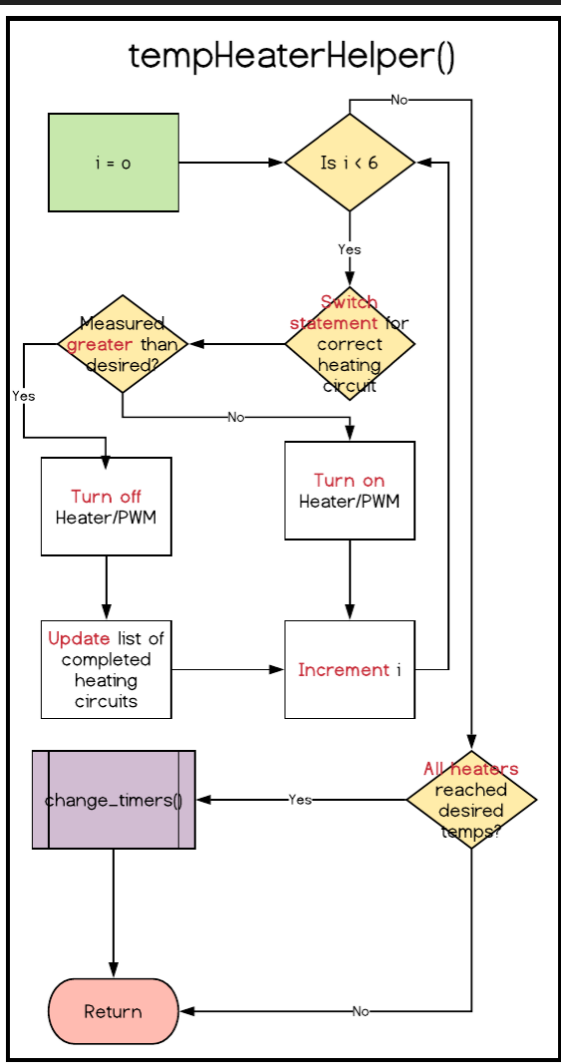
# HCU: SoftwARE





# HCU: SoftwARE

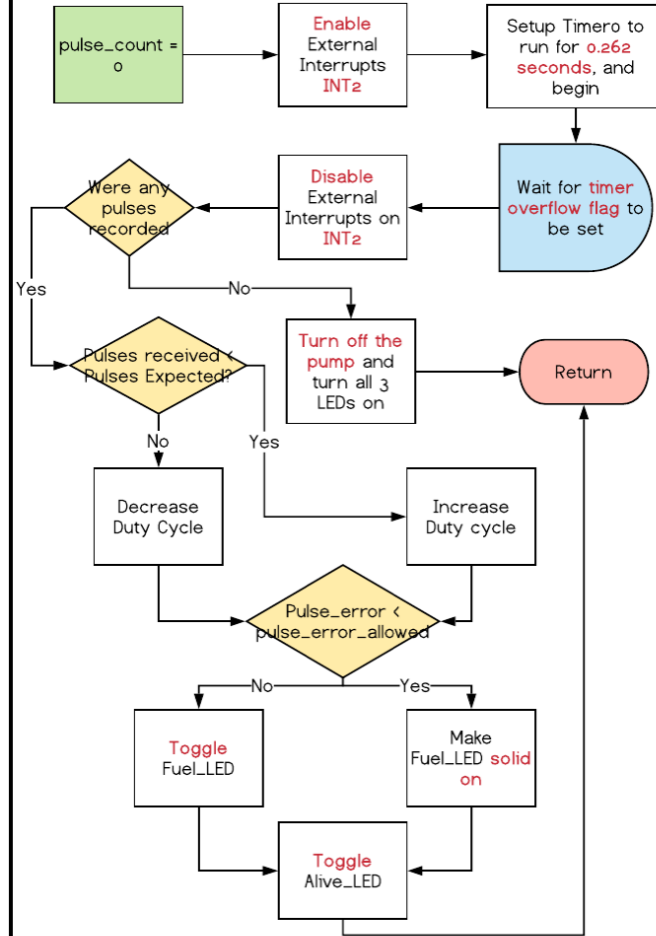




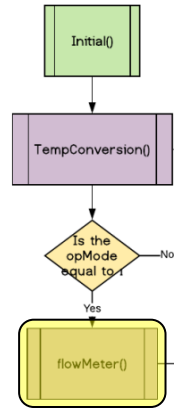


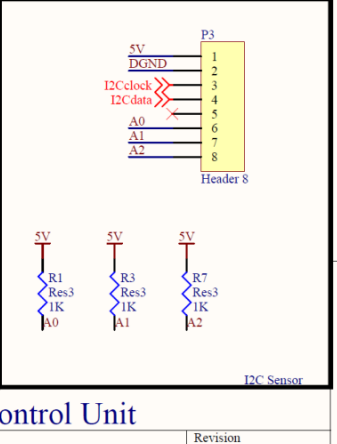
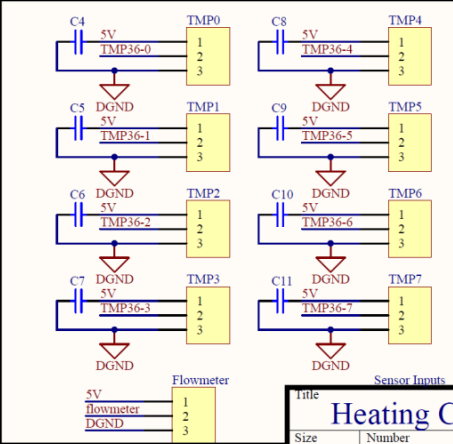
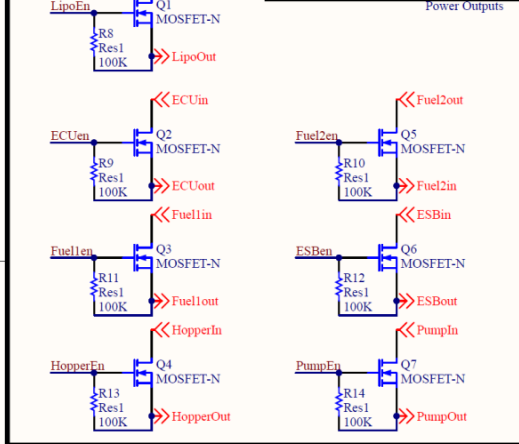
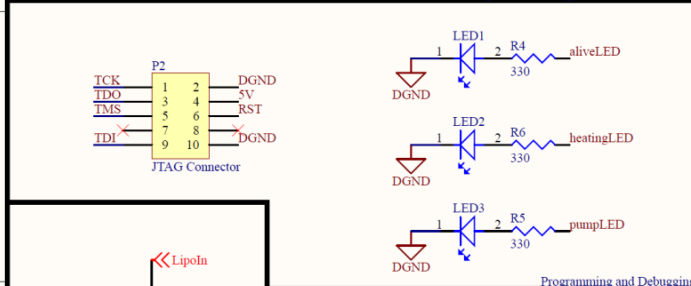
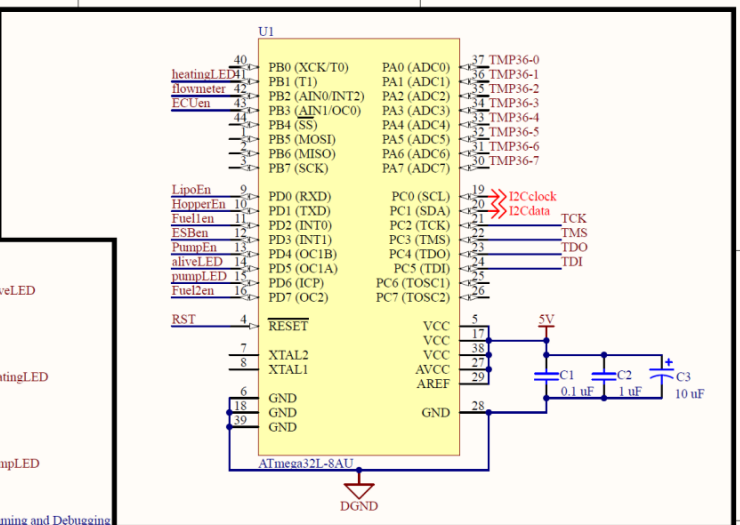
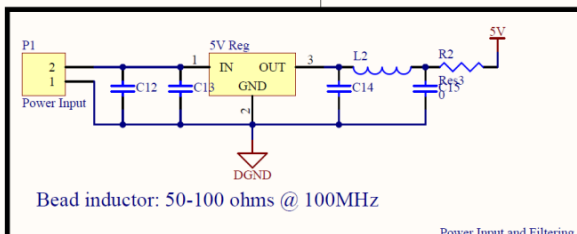


## flowMeter()



## Program Main



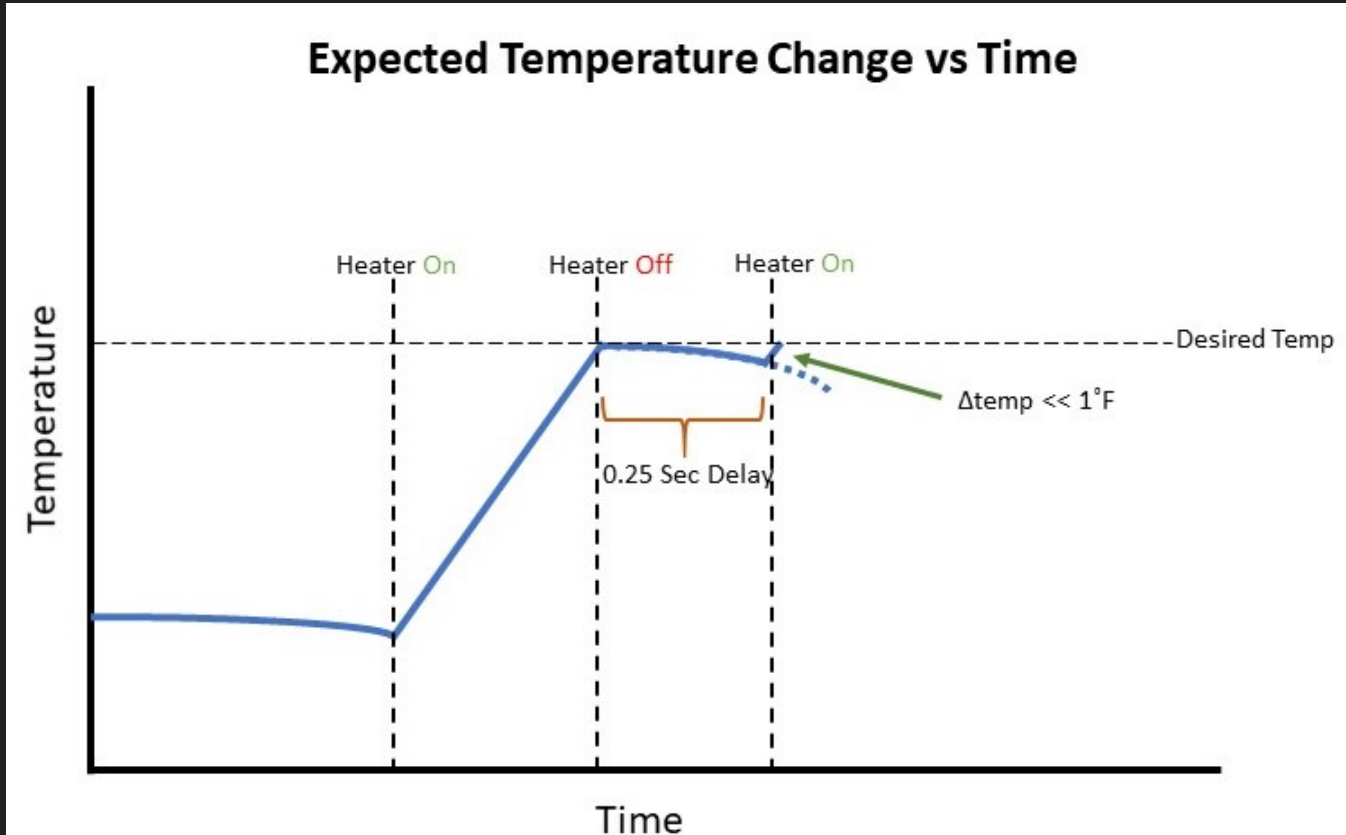


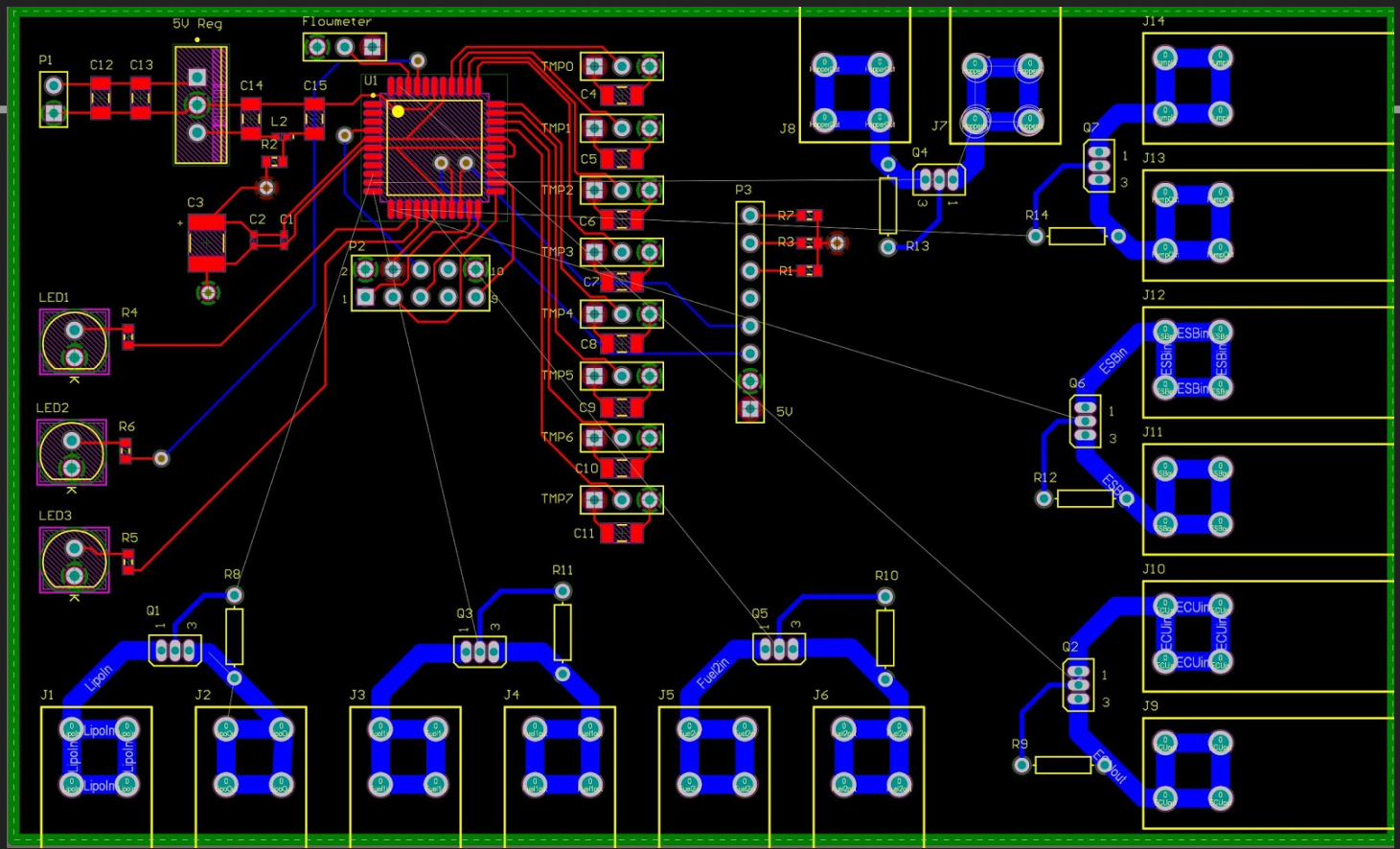
Title		
Size	Number	Revision
A	Main Schematic	1.3.2
Date:	2/3/2018	Sheet 1 of 2
File:	C:\Users\...Main.Sch.Doc	Drawn By: Alex Bertman





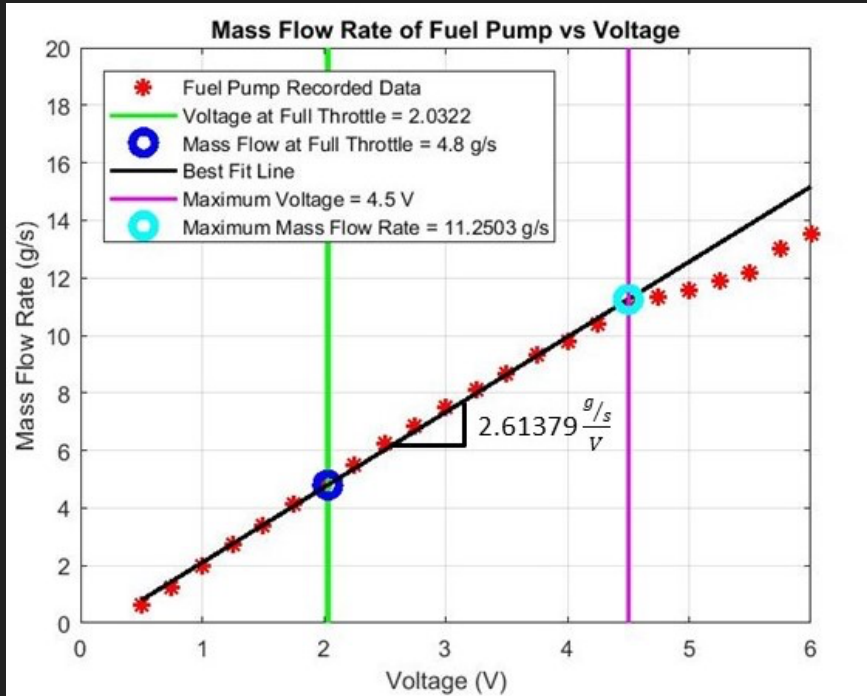
# HCU: SoftwARE Temp Control







# HCU: Software Mass flow



$$\frac{4.8 \text{ g}}{1 \text{ s}} \left| \frac{1 \text{ mL}}{0.81 \text{ g}} \right| \left| \frac{1 \text{ L}}{1000 \text{ mL}} \right| \left| \frac{110,000 \text{ pulses}}{1 \text{ L}} \right| \left| \frac{0.262144 \text{ s}}{1 \text{ period}} \right| = 171 \text{ pulses per period}$$

$$\frac{4.8 \text{ g}}{1 \text{ s}} \left| \frac{1 \text{ period}}{171 \text{ pulses}} \right| = 0.02807 \text{ g/s error per pulse}$$

$$\frac{1 \text{ pulse}}{1 \text{ pulse}} \left| \frac{0.02807 \text{ g/s}}{1 \text{ g/s}} \right| \left| \frac{0.382587 \text{ V}}{1 \text{ g/s}} \right| \left| \frac{20,000 \text{ counts}}{6 \text{ V}} \right| = 35.797 \Delta \text{OCR1B per pulse error}$$

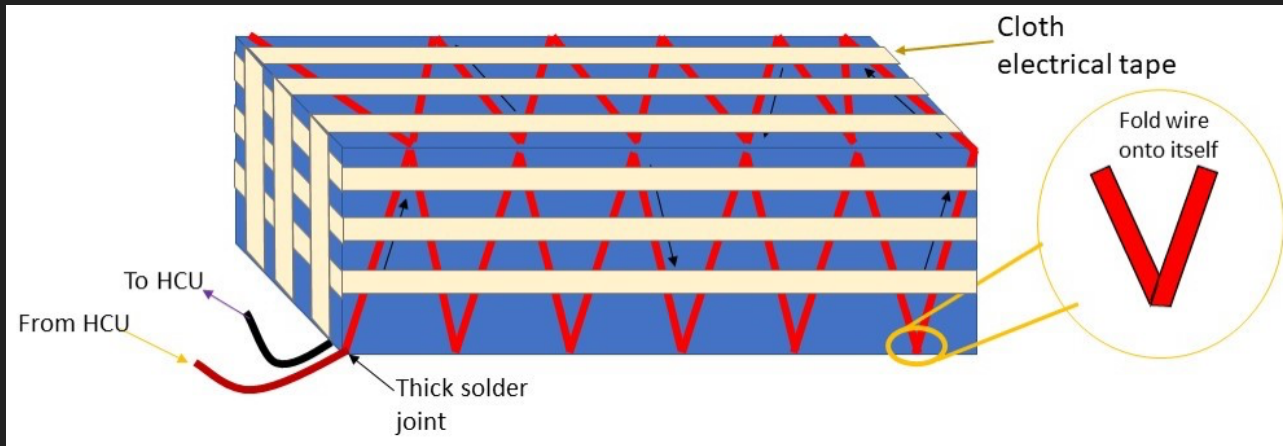




# Wire Wrapping



- Need to prevent wire wrapping from being an electromagnet.
  - Could/probably will disrupt the Lipo battery equalizing circuits and ECU
    - This configuration will produce destructive B fields.
  - Fuel lines do not have this concern so they will be wrapped in the more traditional sense.

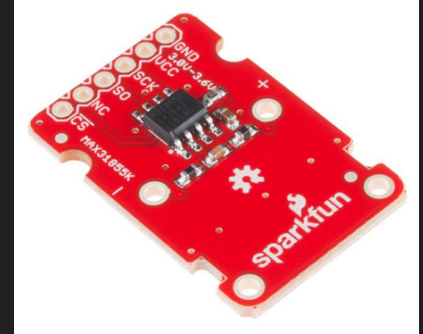




# Test Chamber Software



- Software is currently not written or tested
  - However, this is not a concern.
- Same conditional control law as for the heater circuits (controlled by the HCU)
  - Makes implementation very simple.
- Need to interface with the SparkFun Thermocouple Breakout Board - MAX31855K
  - This is done over the SPI lines for the Arduino.
  - Arduino has built in libraries which makes this incredibly easy.

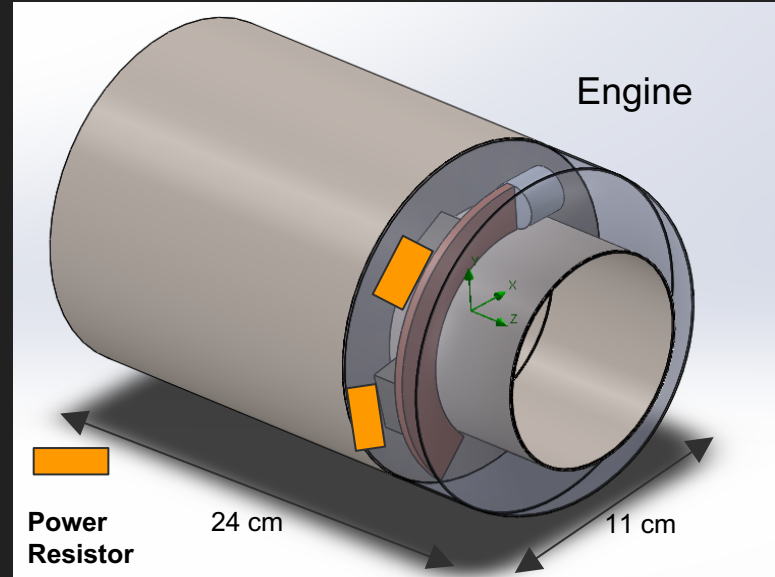




# Power Resistors for ESB



- **Vishay/Dale 12 Ohm 15W** power resistors to heat ESB (x2)
- These will be mounted to the underside of the ESB using a Cyanoacrylate
  - An example of this is super glue
  - These have temperature ranges of -67 to 482 F.







# Lengths of Resistor Wire



- 14 ft (0.88 $\Omega$ /ft) for both Lipo batteries
- 3.86 ft (3.17  $\Omega$ /ft) for the ECU
- 11.92 ft (3.17  $\Omega$ /ft) for outside the hopper
- 0.54 ft (3.17  $\Omega$ /ft) for Fuel line section 1
- 0.83 ft (3.17  $\Omega$ /ft) for Fuel line section 2
- 2.69 ft (3.17  $\Omega$ /ft) for Fuel line section 3

