

AEM P5.15

$$9.082 \ddot{y}(t) + b \dot{y}(t) + 890 y(t) = 0$$

```
clear
clf
y=dsolve('a*D2y+b*Dy+890*y=0','y(0)=0.15','Dy(0)=0','t');
y=subs(y,'a',9.082)
%
hold on
ezplot(subs(y,'b',200),[0,1])
gtext('b=200')          % Label the curve with mouse click
ezplot(subs(y,'b',179.8),[0,1])
gtext('b=179.8')
ezplot(subs(y,'b',100),[0,1])
gtext('b=100')
title('Solution to 9.082*D2y+b*Dy+890*y')
ylabel('y(t)')
hold off
```

$$y(0) = 0.15, \quad \dot{y}(0) = 0$$

Consider spherical dilatation where

$$\vec{v} = v(r, t) \hat{e}_r \quad \text{and} \quad v(r, t) = ra(t)$$

calculate $\nabla \cdot \vec{v}$ $(h_1, h_2, h_3) = (1, r, r \sin \theta)$

$$\nabla \cdot \vec{v} = \frac{1}{h_1 h_2 h_3} \left[\frac{\partial}{\partial r} (v_1 h_2 h_3) + \frac{\partial}{\partial \theta} (v_2 h_1 h_3) + \frac{\partial}{\partial \phi} (v_3 h_1 h_2) \right]$$

$$= \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (ra(t) (r) (r \sin \theta)) \right]$$

$$= \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (r^3 a \sin \theta) \right] = 3a \leftarrow$$

$$\nabla \cdot \vec{v} = \left(\hat{e}_r \frac{\partial}{\partial r} + \frac{\hat{e}_\theta}{r} \frac{\partial}{\partial \theta} + \frac{\hat{e}_\phi}{r \sin \theta} \frac{\partial}{\partial \phi} \right) \cdot v a(t) \hat{e}_r$$

$$\nabla \neq \hat{e}_r \frac{\partial}{\partial r} + \hat{e}_\theta \frac{\partial}{\partial \theta} + \hat{e}_\phi \frac{\partial}{\partial \phi}$$

$$\nabla \equiv \hat{e}_i \frac{\partial}{\partial x^i}$$

Hypothetical vector field

$$\vec{f}(r, \theta, \phi) = r^2 (\hat{e}_r + \sin \theta \hat{e}_\theta + \hat{e}_\phi)$$

a) compute $\nabla \times \vec{f}$

b) what is the value $(r, \theta, \phi) = (2, \pi, \frac{\pi}{2})$

$$\nabla \times \vec{f} = \frac{1}{h_1 h_2 h_3} \begin{vmatrix} h_1 \hat{e}_1 & h_2 \hat{e}_2 & h_3 \hat{e}_3 \\ \partial/\partial q^1 & \partial/\partial q^2 & \partial/\partial q^3 \\ h_1 f_1 & h_2 f_2 & h_3 f_3 \end{vmatrix}$$

$$(q^1, q^2, q^3) = (r, \theta, \phi)$$

$$(h_1, h_2, h_3) = (1, r, r \sin \theta)$$

$$\nabla \times \vec{f} = \frac{1}{r^2 \sin \theta} \begin{vmatrix} \hat{e}_r & r \hat{e}_\theta & r \sin \theta \hat{e}_\phi \\ \partial/\partial r & \partial/\partial \theta & \partial/\partial \phi \\ r^2 & r(r^2 \sin \theta) & r \sin \theta (r^2) \end{vmatrix}$$

$$= \hat{e}_r \left[\frac{\partial}{\partial \theta} (r^3 \sin \theta) - \frac{\partial}{\partial \phi} (r^3 \sin \theta) \right]$$

$$- r \hat{e}_\theta \left[\frac{\partial}{\partial r} (r^3 \sin \theta) - \frac{\partial}{\partial \phi} (r^2) \right]$$

$$+ r \sin \theta \hat{e}_\phi \left[\frac{\partial}{\partial r} (r^3 \sin \theta) - \frac{\partial}{\partial \theta} (r^2) \right]$$

$$= r \cot \theta \hat{e}_r - 3r \hat{e}_\theta + 3r \sin \theta \hat{e}_\phi$$

$$\nabla \times \vec{f}(2, \frac{\pi}{2}, \frac{\pi}{2}) = \frac{2}{\cancel{\tan(\frac{\pi}{2})}} \hat{e}_r - 6 \hat{e}_\theta + 6 \sin \frac{\pi}{2} \hat{e}_\phi$$

$$= -6 \hat{e}_\theta + 6 \hat{e}_\phi$$

\vec{c} Given $[T_{ik}] = \begin{bmatrix} 1 & 2 & 3 \\ 4 & 5 & 6 \\ 7 & 8 & 9 \end{bmatrix},$

$$\vec{c} = \hat{i}_1 + 2\hat{i}_2 + 3\hat{i}_3$$

a) write the symmetric and anti-symmetric parts

First, write each of the following in invariant (e.g. $T_{ik} = \overset{\leftrightarrow}{T}$, trace of tensor $\overset{\leftrightarrow}{T} = \text{tr}(\overset{\leftrightarrow}{T})$, etc., then compute using the given values

$$b) T_{ik} - \frac{1}{3} \delta_{ik} T_{ll}$$

$$c) \left(T_{ik} - \frac{1}{3} \delta_{ik} T_{ll} \right) c_i$$

$$d) \left(T_{ik} - \frac{1}{3} \delta_{ik} T_{ll} \right) c_i c_k$$

$$a) \sum_{ik} = \frac{1}{2} \left([T_{ik}] + [T_{ki}] \right)$$

$$= \begin{bmatrix} 1 & 3 & 5 \\ 3 & 5 & 7 \\ 5 & 7 & 9 \end{bmatrix}$$

$$A_{ik} = \frac{1}{2} \left([T_{ik}] - [T_{ki}] \right)$$

$$= \begin{bmatrix} 0 & -1 & -2 \\ 1 & 0 & -1 \\ 2 & 1 & 0 \end{bmatrix}$$

$$b) T_{ik} - \frac{1}{3} \delta_{ik} T_{\ell\ell} = \vec{T} - \frac{1}{3} \vec{I} \operatorname{tr}(\vec{T})$$

$$= \begin{bmatrix} 1 & 2 & 3 \\ 4 & 5 & 6 \\ 7 & 8 & 9 \end{bmatrix} - \frac{1}{3} (15) \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} -4 & 2 & 3 \\ 4 & 0 & 6 \\ 7 & 8 & 4 \end{bmatrix}$$



$$c) \left(T_{ik} - \frac{1}{3} \delta_{ik} T_{kk} \right) C_i = \begin{matrix} \vec{C} \cdot \vec{T} & \vec{C} \cdot \vec{T} \\ \vec{C}^T & \vec{C}^T \end{matrix} \begin{matrix} T_{ik} C_i \\ T_{ki} C_i \end{matrix}$$

$$C_1 T_{1k} + C_2 T_{2k} + C_3 T_{3k}$$

$$\approx C_1 T_{k1} + C_2 T_{k2} + C_3 T_{k3}$$

$$= \vec{C} \cdot \left[\vec{T} - \frac{1}{3} \text{tr}(\vec{T}) \mathbb{I} \right]$$

$$= \begin{bmatrix} 1 & 2 & 3 \end{bmatrix} \begin{bmatrix} -4 & 2 & 3 \\ 4 & 0 & 6 \\ 7 & 8 & 4 \end{bmatrix} = \begin{bmatrix} 25 & 26 & 27 \end{bmatrix}$$

$$d) (T_{ik} - \frac{1}{3} \delta_{ik} T_{ll}) c_i c_k$$

$$\vec{c} \cdot \vec{T} = \vec{T} \cdot \vec{c}$$

$$(\vec{A} \vec{B})^T = \vec{B}^T \vec{A}$$

$$= \vec{c} \cdot \vec{c} (\vec{T} - \frac{1}{3} \text{tr}(\vec{T}) \vec{I})$$

$$= \vec{c} \cdot \vec{c} ()$$

$$\underbrace{T_{ik} c_i c_k}$$

$$= \vec{c} \vec{c} : \left[\vec{T} - \frac{1}{3} \text{tr}(\vec{T}) \vec{I} \right]$$

$$\text{Set } M_{ik} = T_{ik} - \frac{1}{3} \delta_{ik} T_{ll}$$

$$= M_{ik} c_i c_k = M_{11} c_1 c_1 + M_{12} c_1 c_2 + \dots$$

$$\dots + M_{33} c_3 c_3$$

$$= 158$$

$$\vec{c} \vec{c} = \begin{bmatrix} c_1 c_1 & c_1 c_2 & c_1 c_3 \end{bmatrix}$$

~~→~~

$$\begin{bmatrix} c_1 \\ c_2 \\ c_3 \end{bmatrix}$$

$$\begin{bmatrix} c_1 & c_2 & c_3 \end{bmatrix}$$

~~→~~

$$\vec{c} \cdot \vec{T} \stackrel{?}{=} \vec{T}^T \cdot \vec{c}$$

$$\vec{T} \cdot \vec{c} \rightarrow T\vec{c}$$

$$\vec{c} \cdot \vec{T} \rightarrow \vec{c}T$$

$$\underbrace{c_i T_{ik}} \stackrel{?}{=} \underbrace{T_{ki} c_i}$$

$$\begin{aligned} & (c_1 T_{11} + c_2 T_{21} + c_3 T_{31}) \hat{e}_1 \\ & + (c_1 T_{12} + c_2 T_{22} + c_3 T_{32}) \hat{e}_2 + \dots \end{aligned}$$

$$\begin{aligned} & (T_{11} c_1 + T_{12} c_2 + T_{13} c_3) \hat{e}_1 \\ & + (T_{21} c_1 + T_{22} c_2 + T_{23} c_3) \hat{e}_2 + \dots \end{aligned}$$

$$\vec{T} = T_{ik} \hat{e}_i \hat{e}_k \quad \vec{T} = T \cdot \vec{e}$$

$$\vec{c} = c_j \hat{e}_j$$

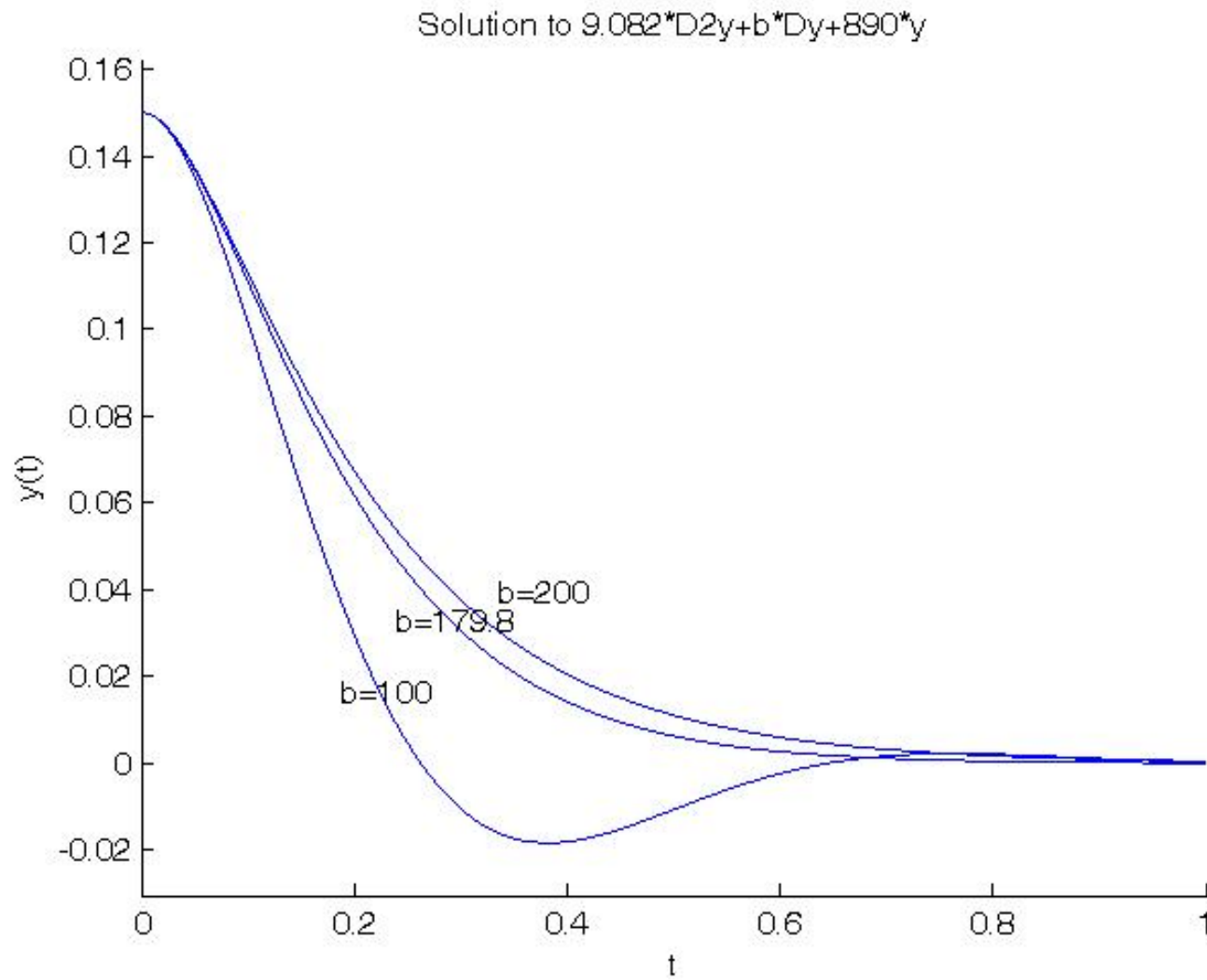
$$\hat{e}_j \hat{e}_j \cdot T_{ik} \hat{e}_i \hat{e}_k = c_i T_{ik}$$

$$T_{ik} \hat{e}_k \hat{e}_i \text{ or } T_{ki} \hat{e}_i \hat{e}_k$$

$$T_{ki} \hat{e}_i \hat{e}_k \cdot c \hat{e}_j = T_{ki} c_k$$

$$[AB]^T = B^T A^T$$

AEM P5.15



AEM P5.15

>> P5_15

y =

$$\begin{aligned} & 3/40 * (b + (b^2 - 808298/25)^{(1/2)}) / (b^2 - 808298/25)^{(1/2)} * \exp(- \\ & 250/4541 * (b - (b^2 - 808298/25)^{(1/2)}) * t) - 3/40 * (b - (b^2 - \\ & 808298/25)^{(1/2)}) / (b^2 - 808298/25)^{(1/2)} * \exp(-250/4541 * (b + (b^2 - \\ & 808298/25)^{(1/2)}) * t) \end{aligned}$$

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```
>> pretty(simplify(y))
```

$$\begin{aligned}
 & \frac{3}{40} \sqrt[5]{\exp\left(-\frac{50}{4541}(5b - \%1)t\right) b} \\
 & + \exp\left(-\frac{50}{4541}(5b - \%1)t\right) (25b^2 - 808298)^{1/2} \\
 & - 5 \exp\left(-\frac{50}{4541}(5b + \%1)t\right) b \\
 & + \exp\left(-\frac{50}{4541}(5b + \%1)t\right) (25b^2 - 808298)^{1/2} \sqrt{\quad} / \\
 & (25b^2 - 808298)^{1/2} \\
 & \%1 := (25b^2 - 808298)^{1/2}
 \end{aligned}$$

```
>>
```

Runge-Kutta Method: Newton's Law

AEM P5.33

Given: Newton's equation of motion for the 2-body problem

$$\ddot{x}(t) = -\frac{x(t)}{[x^2(t) + y^2(t)]^{3/2}}, \quad x(0) = 0.4, \quad \dot{x}(0) = 0,$$

$$\ddot{y}(t) = -\frac{y(t)}{[x^2(t) + y^2(t)]^{3/2}}, \quad y(0) = 0, \quad \dot{y}(0) = 2,$$

Runge-Kutta Method: Newton's Law

$$\left. \begin{array}{l} x_1 = x \\ x_2 = y \\ x_3 = \dot{x} \\ x_4 = \dot{y} \end{array} \right\} \rightarrow \left\{ \begin{array}{l} \dot{x}_1 = \dot{x} \\ \dot{x}_2 = \dot{y} \\ \dot{x}_3 = \ddot{x} = -\frac{x_1}{[x_1^2 + x_2^2]^{3/2}} \\ \dot{x}_4 = \ddot{y} = -\frac{x_2}{[x_1^2 + x_2^2]^{3/2}} \end{array} \right.$$

$$x_1(0) = 0.4$$

$$x_2(0) = 0$$

$$x_3(0) = 0$$

$$x_4(0) = 2$$

Runge-Kutta Method: Newton's Law

MATLAB ODE45:

%Script to generate orbits

%

```
[t,x] = ode45('orbitfun', [0 2*pi], [0.4 0 0 2]);
```

```
x1 = x(:,1);
```

```
x2 = x(:,2);
```

```
plot(x1,x2)
```

```
axis equal
```

```
grid
```

```
function dx = orbitfun(t,x)
```

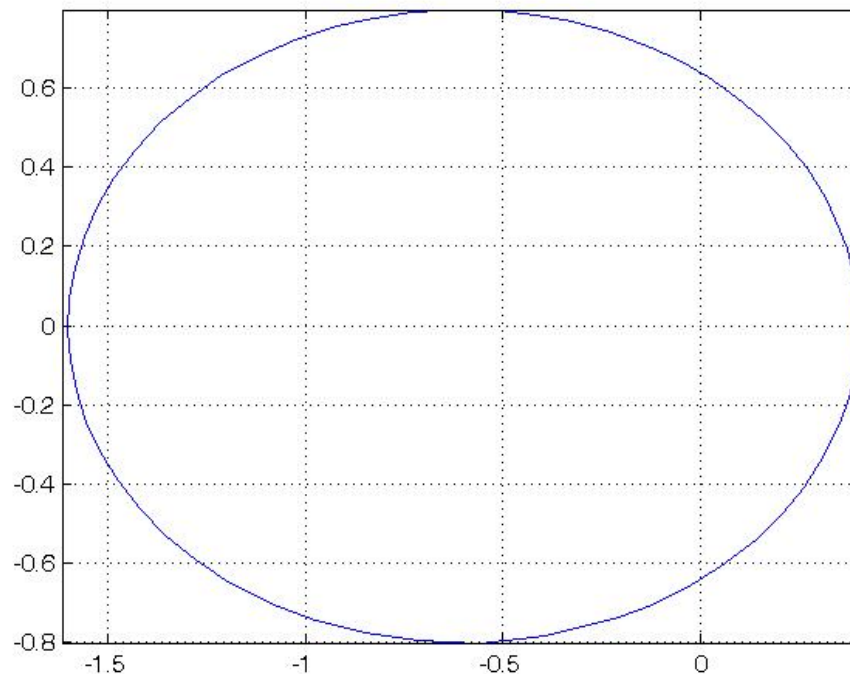
```
dx = zeros(4,1);
```

```
dx(1) = x(3);
```

```
dx(2) = x(4);
```

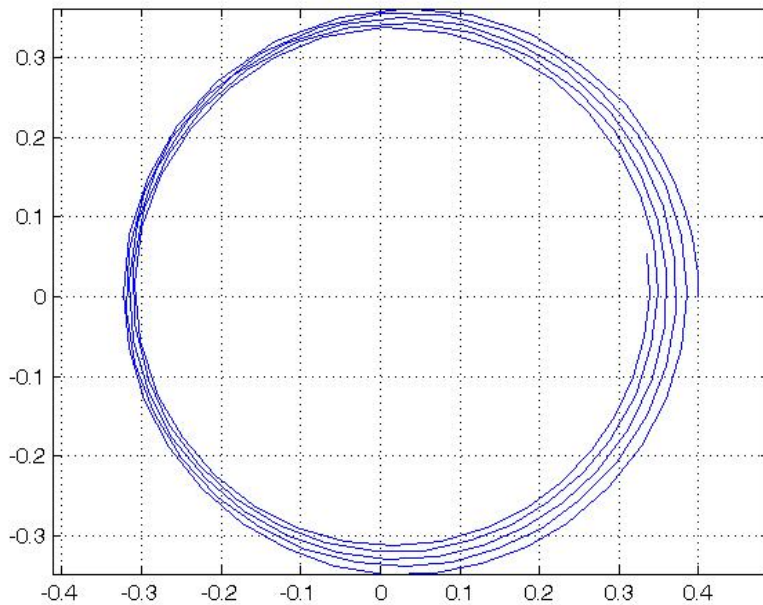
```
dx(3) = -x(1)/(x(1)^2 + x(2)^2)^1.5;
```

```
dx(4) = -x(2)/(x(1)^2 + x(2)^2)^1.5;
```

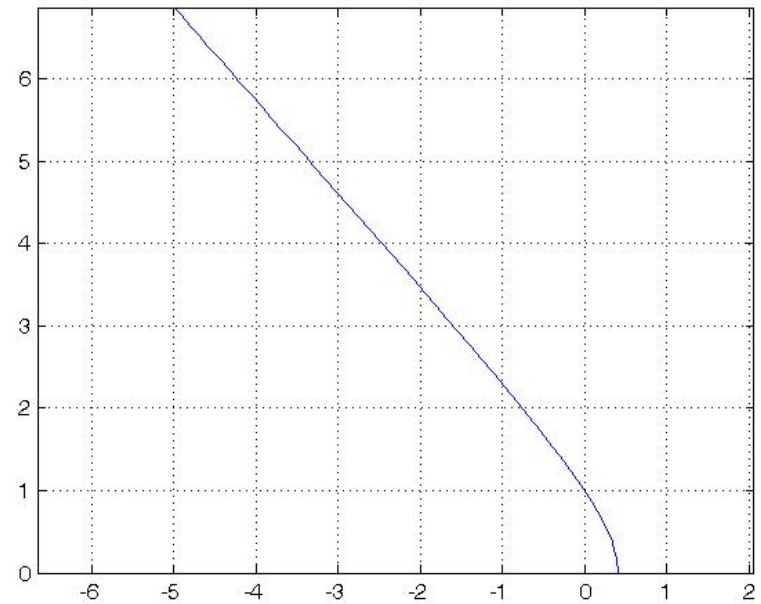


Runge-Kutta Method: Newton's Law

```
[t,x]=ode45('orbitfun',[0 2*pi],[0.4 0 0 1.5]);
```



```
[t,x]=ode45('orbitfun',[0 2*pi],[0.4 0 0 2.5]);
```



Runge-Kutta Method: Newton's Law

```
[t,x]=ode45('orbitfun',[0 5*pi],[0.4 0 0 2.1]);
```

